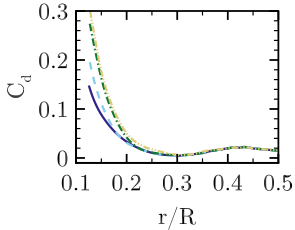
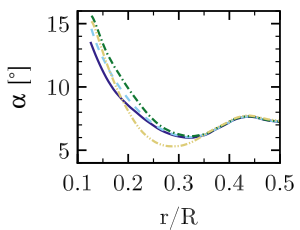


(a) Lift coefficient  $C_l$



(b) Lift coefficient  $C_d$



(c) Angle of attack  $\alpha$