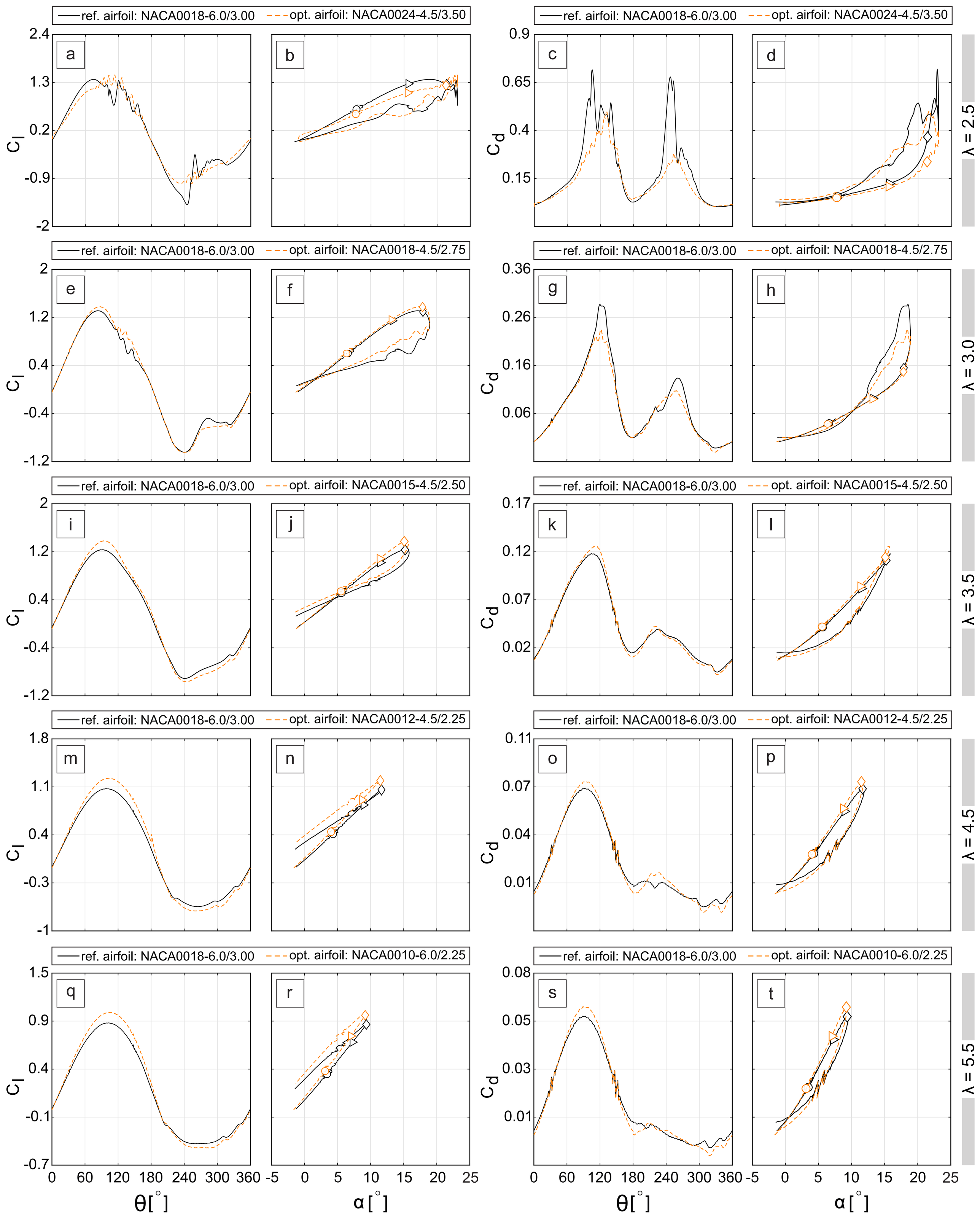
ref. airfoil: NACA0018-6.0/3.00
opt. airfoil: NACA0024-4.5/3.50
ref. airfoil: NACA0018-6.0/3.00
opt. airfoil: NACA0018-4.5/2.75
ref. airfoil: NACA0018-6.0/3.00
opt. airfoil: NACA0015-4.5/2.50
ref. airfoil: NACA0018-6.0/3.00
opt. airfoil: NACA0012-4.5/2.25
ref. airfoil: NACA0018-6.0/3.00
opt. airfoil: NACA0010-6.0/2.25
a
b
c
d
e
f
g
h
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l
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r
s
t
C_l
C_d
$\lambda = 2.5$
$\lambda = 3.0$
$\lambda = 3.5$
$\lambda = 4.5$
$\lambda = 5.5$
$\theta[^\circ]$
$\alpha[^\circ]$