Lift coefficient against angle of attack
Airfoil: NACA 63-418, Re = 3.2E6
Solver: OpenFOAM V6, RASModel: RANS kOmegaSSTLM, Ti: 0.8 %

Drag coefficient against angle of attack
Airfoil: NACA 63-418, Re = 3.2E6
Solver: OpenFOAM V6, RASModel: RANS kOmegaSSTLM, Ti: 0.8 %

Coordinates (Airfoil: NACA 63-418)

Suction side transition location
Airfoil: NACA 63-418, Re = 3.2E6
Solver: OpenFOAM V6, RASModel: RANS kOmegaSSTLM, Ti: 0.8 %