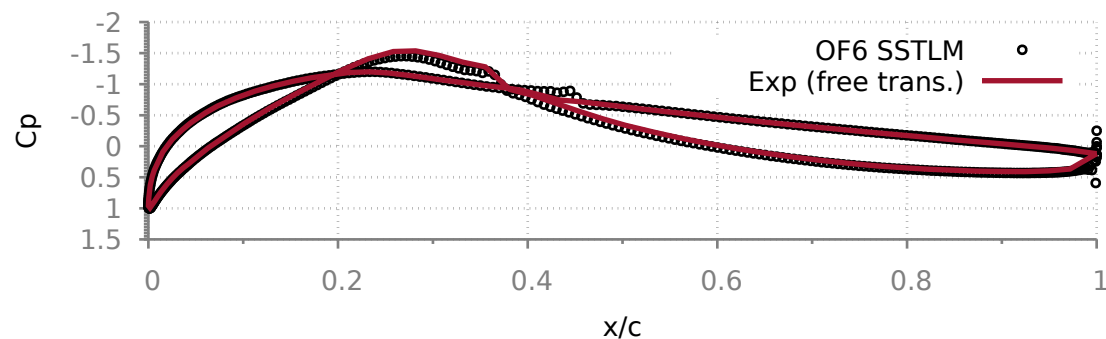
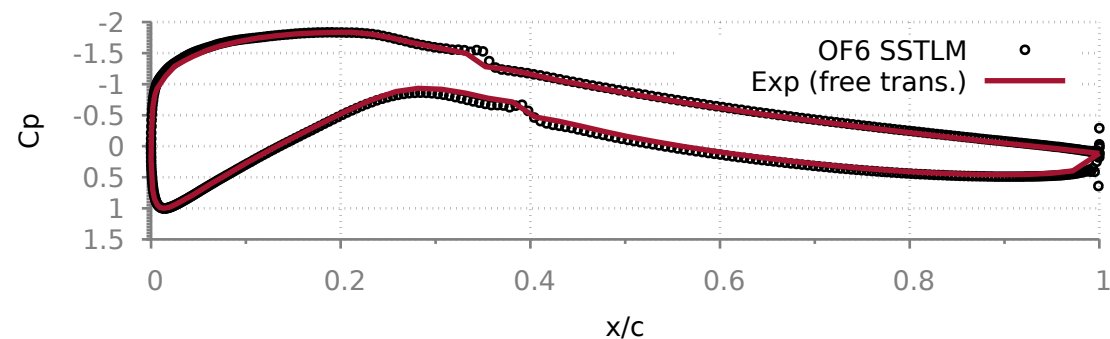


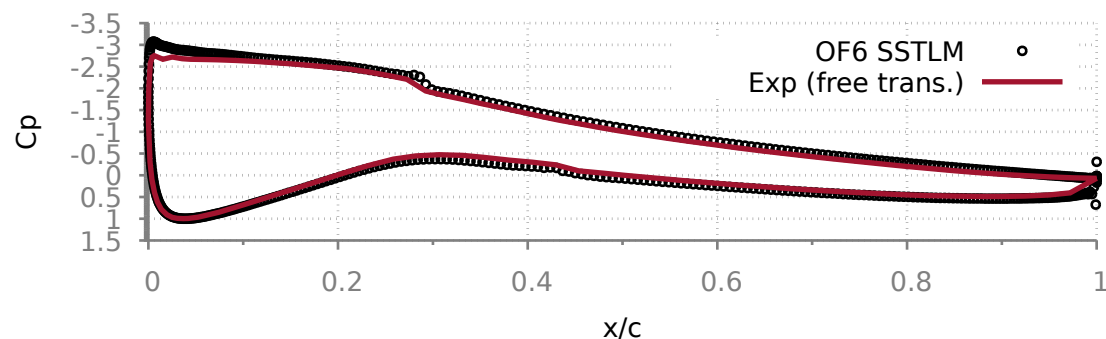
Chordwise pressure distribution
Airfoil: DU97-W-300, $Re = 2.0E6$
Solver: OpenFOAM V6, RASModel: RANS kOmegaSSTLM, $Ti: 0.1\%$
 $AoA=0deg$



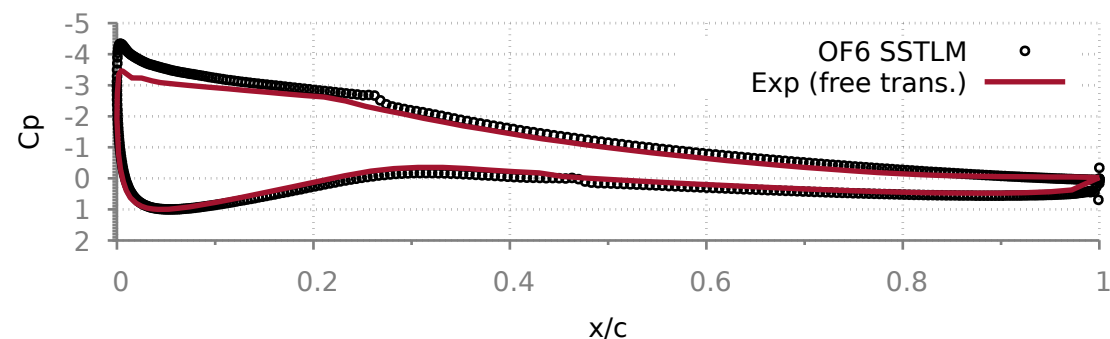
Chordwise pressure distribution
Airfoil: DU97-W-300, $Re = 2.0E6$
Solver: OpenFOAM V6, RASModel: RANS kOmegaSSTLM, $Ti: 0.1\%$
 $AoA=5deg$



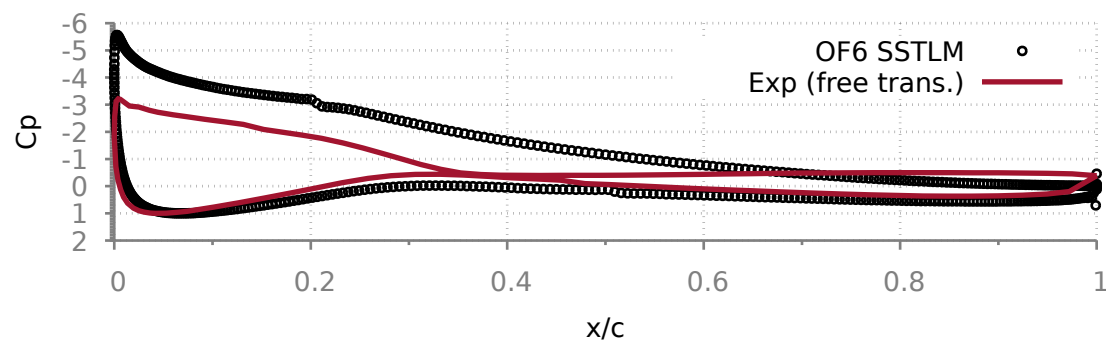
Chordwise pressure distribution
Airfoil: DU97-W-300, $Re = 2.0E6$
Solver: OpenFOAM V6, RASModel: RANS kOmegaSSTLM, $Ti: 0.1\%$
 $AoA=10deg$



Chordwise pressure distribution
Airfoil: DU97-W-300, $Re = 2.0E6$
Solver: OpenFOAM V6, RASModel: RANS kOmegaSSTLM, $Ti: 0.1\%$
 $AoA=12.5deg$



Chordwise pressure distribution
Airfoil: DU97-W-300, $Re = 2.0E6$
Solver: OpenFOAM V6, RASModel: RANS kOmegaSSTLM, $Ti: 0.1\%$
 $AoA=15deg$



Chordwise pressure distribution
Airfoil: DU97-W-300, $Re = 2.0E6$
Solver: OpenFOAM V6, RASModel: RANS kOmegaSSTLM, $Ti: 0.1\%$
 $AoA=20deg$

