Effect of tip spacing, thrust coefficient and turbine spacing in multi-rotor wind turbines and farms

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Abstract. Large eddy simulations (LES) are performed to study the wakes of a multi-rotor wind turbine configuration comprising of four identical rotors mounted on a single tower. The multi-rotor turbine wakes are compared to the wake of a conventional turbine comprising of a single rotor per tower with the same frontal area, hub height and thrust coefficient. The multi-rotor turbine wakes are found to recover faster, while the turbulence intensity in the wake is smaller, compared to the
5 wake of the conventional turbine. The differences with the wake of a conventional turbine increase as the spacing between the tips of the rotors in the multi-rotor configuration increases. The differences are also sensitive to the thrust coefficients used for all rotors, with more pronounced differences for larger thrust coefficients. The interaction between multiple multi-rotor turbines is contrasted with that between multiple single-rotor turbines by considering wind farms with five turbine units aligned perfectly with each other and with the wind direction. Similar to the isolated turbine results, multi-rotor wind farms show smaller
10 wake losses and smaller turbulence intensity compared to wind farms comprised of conventional single-rotor turbines. The benefits of multi-rotor wind farms over single-rotor wind farms increase with increasing tip spacing, irrespective of the axial spacing and thrust coefficient. The mean velocity profiles and relative powers of turbines obtained from the LES results are predicted reasonably accurately by an analytical model assuming Gaussian radial profiles of the velocity deficits and a hybrid linear-quadratic model for merging of wakes. These results show that a larger planform energy flux can be achieved without
15 significantly increased fatigue loads by using multi-rotor turbines instead of conventional, single-rotor turbines.

1 Introduction

Wind energy is among the fastest growing sources of renewable energy worldwide. Understanding and mitigating the deleterious effects of interactions between wakes of multiple turbines is critical for efficient utilization of the wind resource. In
20 large wind farms, the wake interactions can limit the planform energy flux. The turbulent wake interactions also determine fatigue loads on downstream turbines, which has a direct bearing on the levelized cost of energy. Previous work has shown that wake losses are closely tied to wind farm layout parameters such as inter-turbine spacing (Meyers and Meneveau, 2012; Yang

et al., 2012), alignment between columns and the wind direction (Stevens et al., 2014a; Ghaisas and Archer, 2016), horizontal
staggering between adjacent rows (Archer et al., 2013) and vertical staggering of similar or dissimilar turbines (Vasel-Be-Hagh
25 and Archer, 2017; Xie et al., 2017; Zhang et al., 2019).

The idea of mounting multiple rotors per tower has been explored in recent years (Jamieson and Branney, 2012, 2014;
Chasapogiannis et al., 2014; Ghaisas et al., 2018; van der Laan et al., 2019). For example, Jamieson and Branney (2012) showed
that the power generated by a turbine rotor of diameter D scales with the rotor area ($\sim D^2$), while the cost of the turbine scales
with the weights of the blades and of the supporting tower structure, which themselves follow a cubic scaling ($\sim D^3$). This
30 poses a challenge to upscaling the design of current single-rotor turbines to very large systems, but makes multi-rotor turbines
an attractive alternative (Jamieson and Branney, 2012). Structural considerations with designing a 20 MW multi-rotor system
were investigated in Jamieson and Branney (2014). Their results suggested that for a 45-rotor 20 MW system, the need for a
more complicated tower structure would be overwhelmed by benefits due to reduced rotor and drive train costs. Chasapogiannis
et al. (2014) studied the aerodynamics of a 7-rotor system, with the tips of the blades of adjacent rotors spaced 0.05 diameters
35 apart. Interference due to adjacent rotors was found to lead to approximately 3 % increase in power, while about 2 % increase
in the blade loading amplitude was observed.

Analysis of the wake of a 4-rotor turbine was carried out in our previous work (Ghaisas et al., 2018) using large eddy
simulation (LES). It was shown that the multi-rotor turbine wakes recover faster compared to wakes of an equivalent single-
rotor turbine. The turbulent kinetic energy added due to multi-rotor turbines was also lesser than that due to an equivalent
40 single-rotor turbine. Wind farms comprising of five aligned turbines spaced four diameters apart were also considered in this
study. The potential for reduced wake losses as well as reduced fatigue loads was clearly pointed out.

The results for the wake of an isolated turbine were confirmed recently in van der Laan et al. (2019) using a combination
of field observations and numerical simulations. van der Laan et al. (2019) also studied the aerodynamics of individual and
combined rotors. It was found that rotor interaction can lead to an increase of up to 2 % in the power generation, similar to
45 that reported in Chasapogiannis et al. (2014). Only isolated multi-rotor turbines were studied in this paper; interaction between
multiple multi-rotor turbines was not considered.

In this paper, we extend our previous work (Ghaisas et al., 2018) by considering a larger number and range of multi-rotor
wind turbine and farm design parameters. A schematic of the multi-rotor turbine considered here is shown in Fig. 1(b). Four
rotors with identical diameters, d, are mounted on a tower with height H_T (Fig. 1b). The tips of the rotors are separated by s_h
50 and s_v in the horizontal and vertical, respectively. As a result, the rotors are centered at $H_T \pm (s_v + d)/2$, and the mean hub-
height is H_T. The multi-rotor configuration (henceforth referred to as 4-rotor turbine) is compared to a conventional turbine
with a single rotor (referred to as 1-rotor turbine) with diameter $D = 2d$ per tower with height H_T (Fig. 1a). The total frontal
rotor area is $\pi D^2/4$ in each case.

The primary aim of this paper is to quantify the benefits associated with the wakes of multi-rotor turbines for a wide range
55 of tip spacings, thrust coefficients and inter-turbine spacings using LES. A second aim is to develop an analytical modeling
framework, combining elements from previously published studies, and to evaluate its ability to predict the mean velocity
profiles in the wakes of multi-rotor wind farms. This study differs from that of van der Laan et al. (2019) in several respects.
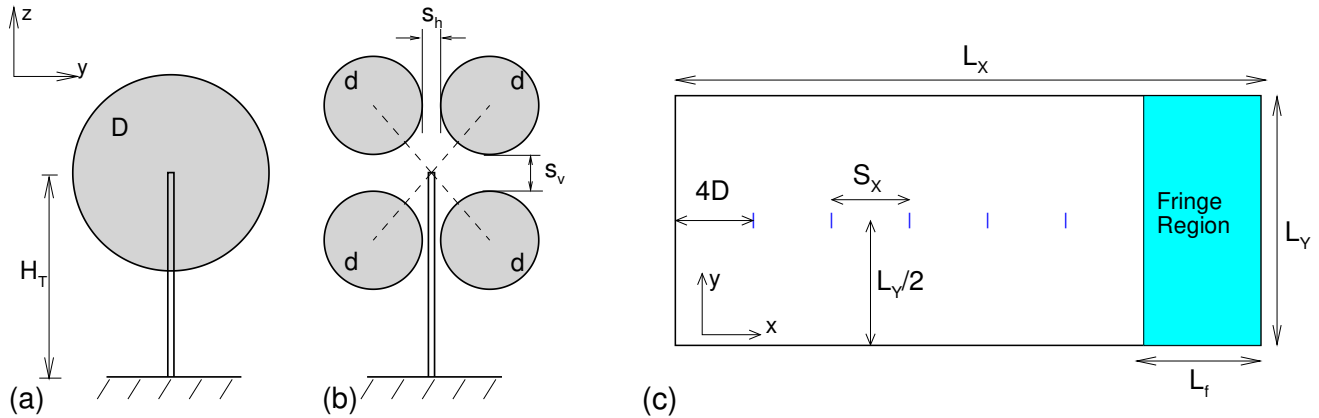
Figure 1. Schematic of (a) conventional 1-rotor turbine and (b) 4-rotor turbines. Tower height H_T is identical for both turbines. Diameters are related by $D = 2d$. Spacing between tips is s_h in horizontal and s_v in vertical. (c) Schematic of the computational domain in plan view, not to scale. Blue lines denote turbine locations.

First, all rotors of the turbine are assumed to operate in the 'Region II' of the power curve. Thus, a constant and uniform thrust coefficient is applied to all four rotors of the 4-rotor turbine. The controller adopted in van der Laan et al. (2019), on the other hand, allowed for small variations of up to 5 % in the thrust coefficients of the lower and upper rotors. Second, the dimensions of the turbines are different in the two studies. Assuming a typical value of 1000 m for the boundary layer height, the rotors of the 4-rotor turbines in the present study have a diameter of 50 m, which is approximately 1.7 times that of the 4R-V29 turbine considered in van der Laan et al. (2019). Finally, only isolated 4-rotor turbines are considered in van der Laan et al. (2019), while interactions between multiple 4-rotor turbines is studied in this paper.

This paper is organized as follows. The LES methodology, details of the simulations and the analytical framework are described in Sect. 2. Results of isolated 4-rotor turbines are described in Sect. 3, while results of wind farms comprised of 4-rotor turbines are described in Sect. 4. In each case, LES results are presented followed by predictions of the analytical modeling framework. Sect. 5 presents a brief summary and the conclusions.

2 Numerical Methodology

2.1 Simulation Framework

The standard LES-filtered incompressible Navier-Stokes equations are solved on a structured uniform Cartesian mesh using Fourier-collocation in x and y directions, sixth-order staggered compact finite-differences in the z direction and a total variation diminishing (TVD) fourth-order Runge-Kutta time-stepping scheme. Non-periodicity is imposed in the x direction using a fringe region technique (Nordström et al., 1999). Partial dealiasing is achieved by applying the standard 2/3 rule in x, y and the use of skew-symmetric form for the convective terms in the z direction. The effect of sub-filter scales is modeled using the

Anisotropic Minimum Dissipation (AMD) model (Rozema et al., 2015). Wind turbine forces are modeled as momentum sinks using the actuator drag-disk model (Calaf et al., 2010). The turbine forces in the LES are defined in terms of the disk-averaged velocity and a 'local thrust coefficient', C_T'. The local thrust coefficient (assuming validity of the inviscid actuator-disk theory) is related to the nominal thrust coefficient, C_T, by $C_T = 16C_T'/(C_T'+4)^2$. Algebraic wall models based on the Monin-Obukhov

80   similarity theory are used to specify the shear stresses at the bottom wall. Subgrid-scale stresses in the rest of the domain are smaller than the sub-filter scale stresses by around 8-10 orders of magnitude and, hence, are neglected in these simulations. The code has been validated over several previously published studies (Ghate and Lele, 2017; Ghaisas et al., 2017; Ghate et al., 2018).

2.2   Cases Simulated

85   Half-channel (HC) simulations are carried out using the concurrent precursor-simulation methodology (Stevens et al., 2014b) on domains of length L_x, L_y, L_z in the three coordinate directions. A schematic of the simulation domain is shown in Fig. 1(c). All simulations use $(L_y, L_z) = (\pi/2, 1)H$, while $L_x = \pi H$ or $1.25\pi H$, depending on the case. Here H is the height of the half-channel. The flow in the 'precursor' simulation is driven by a constant imposed pressure gradient, $-u_*^2/H$, where u_* is the friction velocity at the bottom wall. The HC configuration is used as a model for the neutrally-stratified atmospheric boundary

90   layer (ABL) with the Coriolis forces neglected (Stevens et al., 2014a; Calaf et al., 2010), and we use the terms HC and ABL interchangeably. The surface roughness height at the bottom wall is $z_0 = 10^{-4}H$. This corresponds to rough land, and has been used in previous wind turbine studies (Calaf et al., 2010). The turbulence intensity at a typical hub height of $0.1H$ is approximately 8%. All results are normalized using scales H and u_*, with typical values $H = 1000$ m, $u_* = 0.45$ m/s.

Precursor simulations (without turbines and with streamwise periodicity) are carried out first for 50 time units (1 time

95   unit = H/u_*), so as to achieve a fully-developed statistically stationary state. These velocity fields are then used to initialize the 'precursor' and 'main' simulation domains. Turbines are introduced in the 'main' domain, and a portion of this domain, of length $L_f = 0.15L_x$, is forced with the velocity field from the 'precursor' domain at each time step. Simulations in this concurrent precursor-simulation mode are carried out for a further 20 time units, with time-averaging performed using samples stored every 10 time steps over the last 12 time units.

100   The suite of simulations carried out is listed in Table 1. In the first set of simulations (IT1, IT2, IT3), isolated turbines are simulated with a baseline 1-rotor configuration with $D = 0.1H$, and a baseline 4-rotor configuration with $d = 0.05H$ and $s_h = s_v = s = 0.05d$. Six additional (set IT2) isolated 4-rotor turbine simulations are carried out with varying s to study the effect of tip spacing in the 4-rotor configuration. The thrust coefficient is fixed for this first set of simulations. In the second set (IT2-C_T'), four isolated turbine simulations are carried out to study the effect of varying thrust coefficient. In the third set

105   of simulations (sets WF*), a line of five 1-rotor turbines separated by a distance S_X in the streamwise direction is compared to a similar configuration with a line of five 4-rotor turbines separated by S_X in the streamwise direction. A total of 20 wind farms are simulated, considering different combinations of S_X, C_T' and s. The same thrust coefficient is used for all rotors in one simulation. All isolated turbines, and the most upstream turbine in the five-turbine cases, are located at $x = 0$, where the domain inlet is at $x = -4D$. The turbine towers are located at $y = L_y/2$ in the spanwise direction and the tower height is

Table 1. Suite of isolated turbine (sets IT*) and wind farm (sets WF*) simulations. Domain lengths are non-dimensionalized by height H, with label D1 denoting ($\pi \times \pi/2 \times 1$) and DA denoting ($1.25\pi \times \pi/2 \times 1$). Grid sizes shown are for 'main' domain. Equal number of grid points are additionally required for the 'precursor' domain in each case. Label G1 denotes $192 \times 96 \times 128$; G2 denotes $256 \times 128 \times 160$; G3 denotes $320 \times 160 \times 200$; and G2A denotes $320 \times 128 \times 160$ grid points. Axial spacing is undefined for isolated turbine simulations. Local thrust coefficients are $C'_T = 1, 4/3, 2$, corresponding to nominal $C_T = 0.64, 0.75, 8/9$, respectively.

Set	Domain	Grid	Tip Spacing, $s_h/d = s_v/d = s/d$	Thrust Coefficient, C'_T	Axial Spacing, S_X
IT1-s	D1	G1	1-Rot, 0.05	4/3	-
IT2-s	D1	G2	1-Rot, 0.0, 0.05, 0.1, 0.2, 0.25, 0.5, 1.0	4/3	-
IT3-s	D1	G3	1-Rot, 0.05	4/3	-
IT2-C'_T	D1	G2	1-Rot, 0.1	1.0, 2.0	-
WF2-C'_T	D1	G2	1-Rot, 0.1, 0.25, 0.5	1.0, 4/3, 2.0	$4D$
WF2-SX	DA	G2A	1-Rot, 0.1, 0.25, 0.5	4/3	$5D, 6D$

110 $H_T = 0.1H$ for all turbines. The domain size in the x-direction is increased to 1.25π to accommodate larger axial spacings for the cases with $S_X = 5D$ or $6D$.

2.3 Analytical Model

An analytical modeling framework based on the model by Bastankhah and Porté-Agel (2014) is evaluated for the multi-rotor configuration in this paper. The model assumes that the velocity deficit in the wake decays in the streamwise (x) direction, and

115 follows a Gaussian profile in the radial directions. The deficit due to turbine rotor i located at (x_i, y_i, z_i) at a downstream point (x, y, z) is given as

$$\frac{\Delta \bar{u}_i(x,y,z)}{\bar{u}_{up}(z)} = C(x) \times \exp\left(-\frac{(y-y_i)^2 + (z-z_i)^2}{2\left(k^*(x-x_i) + \sigma_0\right)^2}\right), \tag{1}$$

$$C(x) = \left(1 - \sqrt{1 - \frac{C_T}{8\left(k^*(x-x_i)/d_0 + \sigma_0/d_0\right)^2}}\right), \tag{2}$$

for $x > x_i$. The length scale d_0 equals D for 1-rotor and d for 4-rotor cases. The argument of the square-root in eq. (2) is set to

120 zero whenever it is less than zero, which happens very close to the turbines.

The combined effect of multiple turbine rotors has been modeled in the past using several empirical techniques. Primary among these are addition of velocity deficits (implying linear addition of momentum deficit), square-root of sum of squares (implying addition of kinetic energy deficit; also termed as quadratic merging), and considering the largest deficit to be dominant. In this study, a hybrid between the first two approaches is found to give best results. The hybrid approach involves linear

125 merging of wakes originating at the same x location, with quadratic merging of wakes originating at different x locations. This
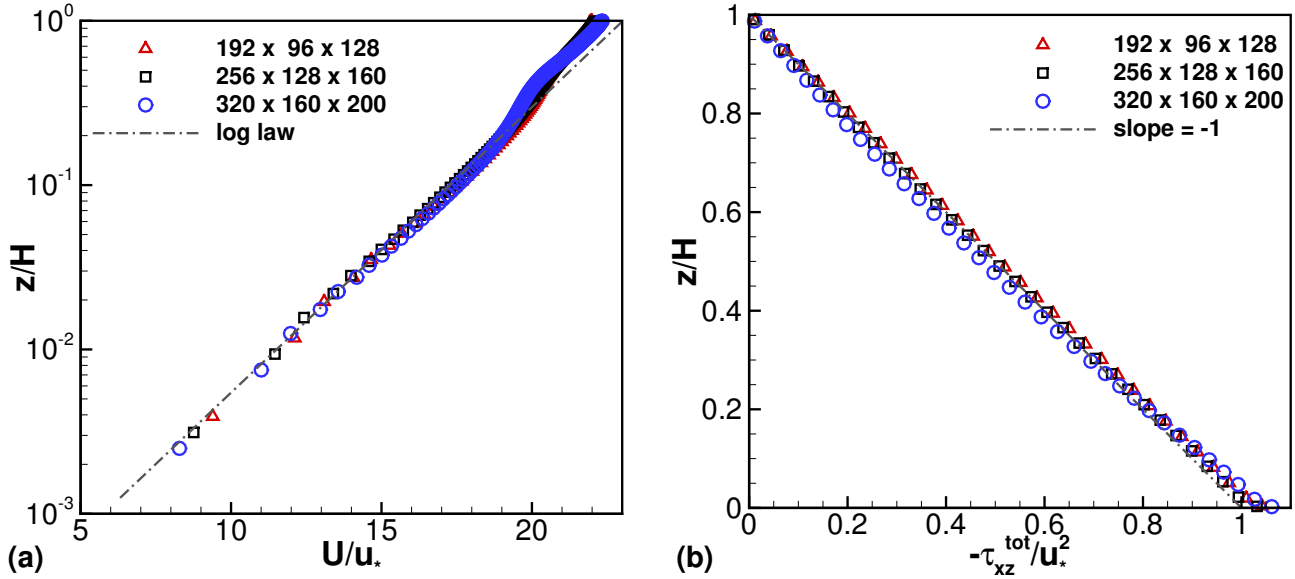
Figure 2. Profiles of time- and horizontally-averaged (a) streamwise velocity, and (b) negative of total shear stress from the ABL (precursor) simulations with varying grid sizes. Total shear stress is the sum of resolved, subgrid-scale and wall-modeled components.

can be written as

$$\Delta \bar{u}_{tot}(x,y,z) = \left[\sum_{i=1}^{N_{xt}} (\Delta \bar{u}_{lin})_i^2 \right]^{1/2}; \quad (\Delta \bar{u}_{lin})_i = \sum_{j=1}^{N_r(x_t)} \Delta \bar{u}_j(x,y,z). \tag{3}$$

N_{xt} is the number of unique axial locations where a turbine is located. $N_r(x_t)$ is the number of rotors at the location x_t. In this paper, N_{xt} is 1 for the isolated turbine cases, and 5 for the wind farm cases. Furthermore, since we only consider either an
130 isolated turbine or a wind farm with one column of turbines, N_r is 1 for the 1-rotor cases and 4 for the 4-rotor cases. Finally, the mean velocity at each point in the domain is calculated according to

$$\bar{u}(x,y,z) = \bar{u}_{up}(z) - \Delta \bar{u}_{tot}(x,y,z). \tag{4}$$

The upstream velocity is assumed to follow the logarithmic profile, $\bar{u}_{up}(z) = (u_*/\kappa)\ln(z/z_0)$, with $\kappa = 0.4$.

This modeling framework involves two empirical parameters, k^* and σ_0. Comments regarding selecting these parameters
135 are provided in the appropriate sections below.

3 Isolated Turbine Results

3.1 Grid Convergence and Baseline Cases

Precursor ABL simulation results are shown first in Fig. 2. These results are averaged over time and the horizontal directions. As expected, the mean streamwise velocity profiles follow the logarithmic law of the wall, particularly in the lower 20% of

140   the domain. The total shear stress profiles also follow the expected line with slope equal to -1. This indicates that the vertical

transport of momentum by the ABL is correctly represented by the numerical method and AMD subgrid-scale model, and that

the ABL simulations are statistically stationary. Figure 2 also shows that the spatial resolution employed is adequate for these

ABL simulations, since the results are almost independent of the grid size.

    Results of an isolated 1-rotor turbine and an isolated 4-rotor turbine with $s/d = 0.05$ are shown in Fig. 3. Vertical profiles in

145   the mid-span planes at several locations downstream of the turbine are shown. The mid-span plane is located at $Y_{cen} = L_Y/2$

for the 1-rotor configuration. The 4-rotor configuration has two mid-span planes, $Y_{cen} = L_Y/2 \pm (1 + s_h)d/2$. Results at only

one of these, at $L_Y/2 - (1 + s_h)d/2$, are shown, since both planes are statistically identical.

    Figure 3(a) shows that the velocity deficit profiles for the 1-rotor turbine have a single peak close to $z/H = 0.1$. Two

distinct peaks, close to $z/H = 0.1 \pm (1 + s_v)d/2$, are seen for the 4-rotor turbine wake in Fig. 3(b) only at $x/D = 2$. Further

150   downstream, at $x/D = 4$ and 6, two distinct peaks are not easily discernible, indicating that the wakes have merged. The added

turbulent kinetic energy (TKE) profiles in Figs. 3(c-d) show similar evidence of a single large wake for the 1-rotor turbine and

two distinct wakes at $x/D = 2$, which merge further downstream, for the 4-rotor turbine.

    Simulations with varying grid sizes (the IT*-s cases) show that the differences between the intermediate (G2) and finest (G3)

grids employed are almost negligible for the velocity deficits. The added TKE profiles show greater sensitivity to the resolution

155   than the mean velocity deficits. The added TKE normalized by the TKE is computed for different grid sizes. The variation of

this quantity between the grids G2 and G3 is found to be at most 9 % close to the top tip height, $z/H = 0.1 + (1 + s_v)d/2$,

at $x/D = 4$. The variation of added TKE averaged over the rotor area to the TKE averaged over the rotor area between grids

G2 and G3 is found to be at most 2.1 %, again at $x/D = 4$. Similar magnitude of variation with grid size was observed for the

4-rotor turbine with $s/d = 1$, and the results are not repeated here for brevity. The variation with grid size of 1-rotor turbine

160   wakes is smaller than that of 4-rotor turbine wakes. Similar results were also reported in Ghaisas et al. (2018). We conclude that

the intermediate grid, with $256 \times 128 \times 160$ grid points, is sufficient for obtaining converged results, and all further simulations

are carried out using this resolution. For the wind farm cases with domain size increased to 1.25π in the x direction, the number

of points in the x direction is increased to 320 to retain the same resolution. This grid is labeled as G2A in Table 1.

3.2   Effect of Tip Spacing

165   Isolated 4-rotor turbines with varying tip spacings, $s_h = s_v = s$, are studied in this subsection (IT2-s cases). Contours of

the mean streamwise velocity deficit and the TKE (Fig. 4) in the mid-span planes show that one large wake immediately

downstream of a 1-rotor turbine is replaced by four smaller wakes immediately downstream of the four rotors of the 4-rotor

turbines. Comparing Figures 4(a,c,e), it is clear that the wake of a 4-rotor turbine at any downstream location (e.g. at $x/D = 4$),

is weaker in magnitude than that of the 1-rotor turbine. This is also seen in the profiles shown in Fig. 5. In other words, the

170   wake of a 4-rotor turbine is seen to recover faster than the wake of a 1-rotor turbine with the same thrust coefficient and rotor

area. Figure 5 also shows that greater the tip spacing of the 4-rotor turbine, faster is the wake recovery. This is also indicated

by the shortening of the contour lines corresponding to $\Delta u/u_* = 1$ and 2.5 in Fig. 4 with increasing tip spacing.
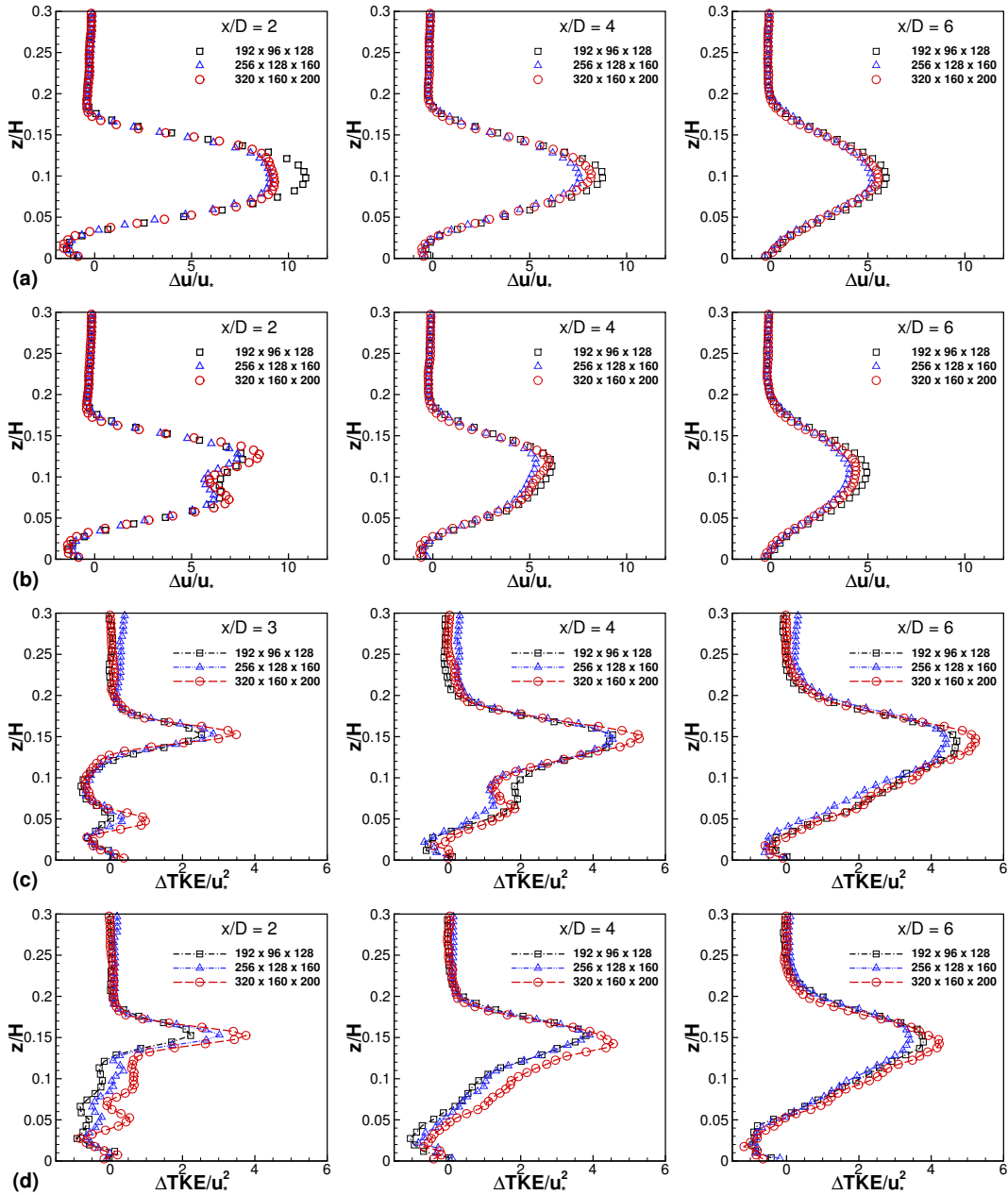
Figure 3. Profiles of mean velocity deficit at the centerline and downstream of an isolated (a) 1-rotor turbine and (b) 4-rotor turbine with $s = 0.05d$ for three different grid resolutions. Profiles of added turbulent kinetic energy (TKE) downstream of (c) 1-rotor turbine and (d) 4-rotor turbine with $s = 0.05d$. Mean velocity deficit and added TKE are defined as $\Delta u(x,z) = u(-1D, Y_{cen}, z) - u(x, Y_{cen}, z)$ and $\Delta TKE(x,z) = TKE(x, Y_{cen}, z) - TKE(-1D, Y_{cen}, z)$, respectively. Y_{cen} is $L_Y/2$ for 1-rotor turbine and $L_Y/2 - (1 + s_h)d/2$ for 4-rotor turbine.
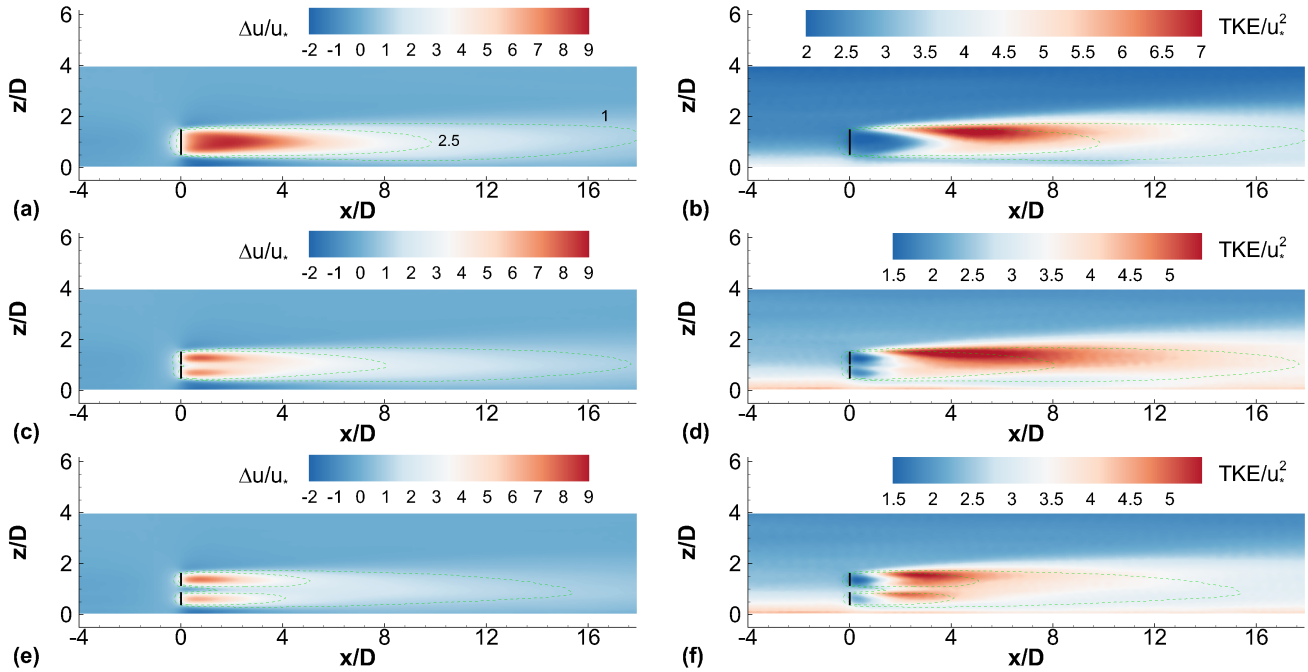
Figure 4. Contours of (a,c,e) mean velocity deficit and (b,d,f) TKE at the centerline, for (a,b) 1-rotor turbine, and 4-rotor turbines with tip spacings (c,d) $s = 0.1d$ and (e,f) $s = 0.5d$. Centerline Y_{cen} varies with turbine configuration. Black solid lines denote turbine rotors. Dashed lines are velocity deficit contours corresponding to the levels $\Delta u/u_* = 1$ and 2.5.

An intuitive explanation for the increasing rate of wake recovery with increasing tip spacing is as follows. The characteristic length scale of the wake of the 1-rotor turbine is diameter D, while that for the individual wakes of the 4-rotor turbines is

175 the smaller diameter d. Furthermore, the spacing between the tips of the 4-rotor turbine allows for greater entrainment of low-momentum fluid into the 4-rotor turbine wakes. As a result, the rate of wake recovery is larger for the 4-rotor turbine as compared to the 1-rotor turbine, and increases with increasing s.

The wakes of the individual rotors of a 4-rotor turbine expand with downstream distance, and eventually merge to form a single wake. The axial distance where individual wakes of the four rotors may be considered to have merged increases with

180 increasing s. This is seen clearly in Fig. 5, where two peaks in the velocity deficit profiles are not seen at $x/D = 4$ for the $s/d = 0.1$ turbine, while two peaks are clearly visible at $x/D = 6$ for the $s/d = 0.5$ turbine.

The contour plot of TKE shown in Fig. 4(b) is strikingly similar to those reported previously (e.g. Fig. 18 in Abkar and Porté-Agel (2015)) for an isolated 1-rotor turbine. The TKE contours in Fig. 4(b) are similar in shape to those in Fig. 4(d) beyond approximately $x/D = 4$, but are quite dissimilar to the contours in Fig. 4(f). This is further evidence for the observation that the

185 wake-merging distance increases with increasing s. The rotors of the 4-rotor turbine behave independently up to increasingly larger downstream distances with increasing s.

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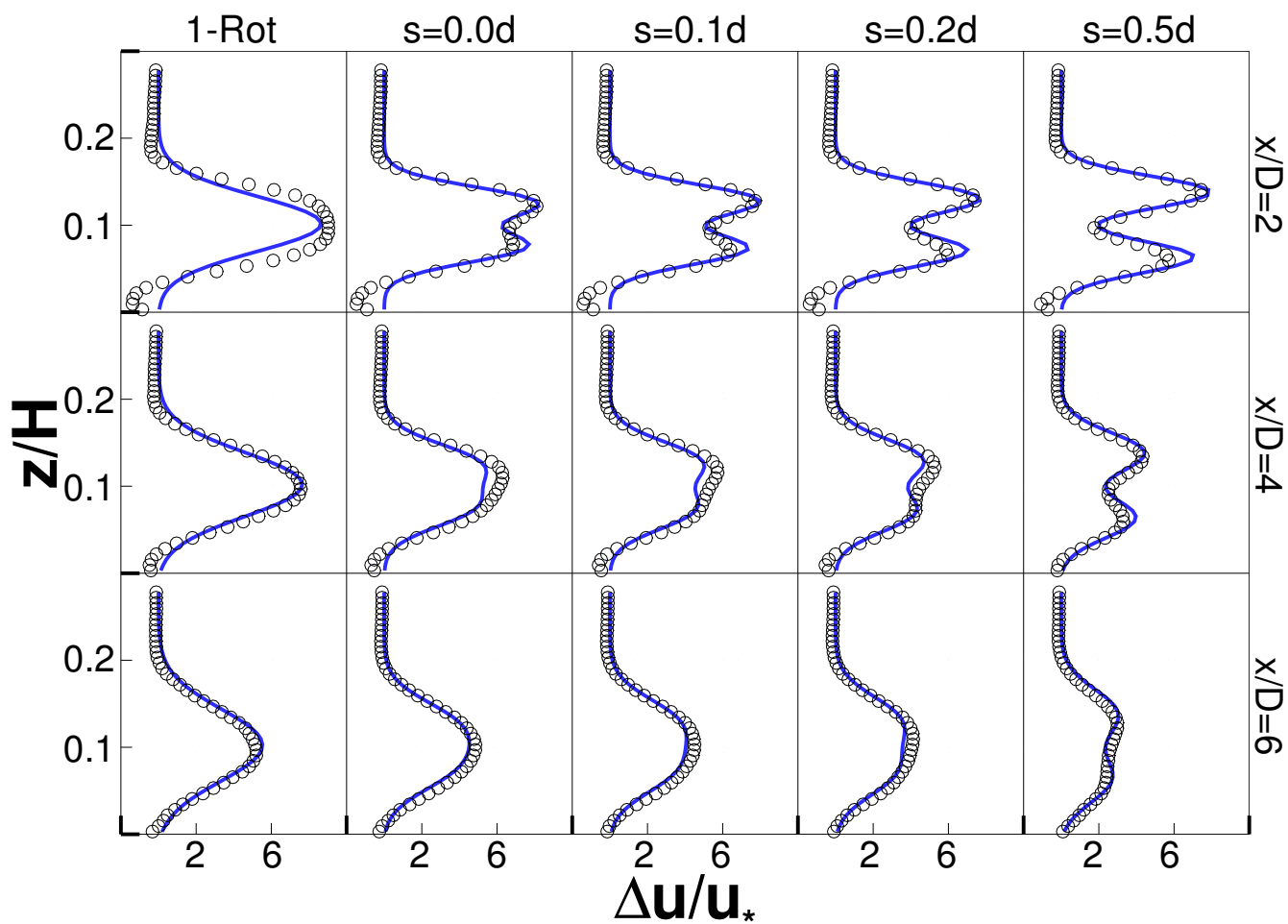
Figure 5. Velocity profiles downstream of an isolated 1-rotor turbine and isolated 4-rotor turbines with different tip spacings, $s/d = 0, 0.1, 0.2$ and 0.5. Black symbols are LES results and blue lines are analytical model results.

A succinct representation of the effect of tip spacing on the wake of an isolated 4-rotor turbine with respect to that of an isolated 1-rotor turbine is shown in Fig. 6, where rotor-disk-averages of three quantities is plotted as a function of the axial distance . The rotor-disk averages are calculated at each axial (x/D) location and over different regions in the $y - z$ plane
190   depending on the turbine configuration. The averages are computed over one disk of diameter D, centered at $(L_Y/2, 0.1H)$ for the 1-rotor turbine, and over four disks of diameters d each, centered at $(L_Y/2 \pm (1+s_h)d/2, 0.1H \pm (1+s_v)d/2)$, for the 4-rotor turbines. The disk averaged TI is actually the ratio of the disk-averaged TKE and disk-averaged mean streamwise velocity, being slightly different from the disk-average of the point-wise turbulence intensity.

Figure 6(a) shows that the streamwise velocity deficits are always smaller for a 4-rotor turbine than for a 1-rotor turbine,
195   and that deficits decrease monotonically with increasing tip spacing. Interestingly, the 4-rotor turbine with grazing rotor blades (tip spacing $s/d = 0$) also shows reduced velocity deficits in the intermediate downstream region, i.e. $x/D = 4$ and $x/D = 6$. The curves corresponding to the $s/d = 0$ turbine and the $s/d = 1$ turbine act as bounds to the curves corresponding to intermediate tip spacings. The disk-averaged added TKE and TI_{disk} curves do not show a monotonic behavior at all downstream locations with increasing s. The curves corresponding to the $s/d = 0$ and $s/d = 1$ turbines do not act as bounds for the curves
200   corresponding to the intermediate tip spacings. However, in general, the second order turbulent statistics show a decrease in magnitude with increasing tip spacing.

3.3   Effect of Thrust Coefficient

The IT2-C'_T cases, along with two cases from the IT2-s set of simulations, are compared to study the effect of thrust coefficient. Only one 4-rotor configuration, with tip spacing $s/d = 0.1$, is considered here. Figure 7 shows that the trends observed for
205   $C'_T = 4/3$ hold for the other two thrust coefficients studied as well. The disk-averaged velocity deficits are smaller for the 4-rotor turbine than for the corresponding 1-rotor turbine. The added TKE (not shown) and TI_{disk} are also smaller for the 4-rotor turbine than for the 1-rotor turbine for all the thrust coefficients studied.

3.4   Analytical Model

The analytical modeling framework predicts the mean velocity deficits of the 1-rotor and 4-rotor turbines accurately. Empirical
210   parameters values $k_* = 0.03$ and $\sigma_0/d_0 = 0.28$ were found to lead to accurate predictions for all the cases investigated. Here, d_0 equals D for the 1-rotor cases and equals d for the 4-rotor cases. These values of k_* and σ_0 are slightly different from those proposed in Bastankhah and Porté-Agel (2016), but within the range mentioned in Bastankhah and Porté-Agel (2014). In particular, Fig. 5 shows that the radial profiles of the velocity deficit at several downstream locations, and for turbines with different tip spacings, are predicted quite accurately. Slight under-predictions or over-predictions are observed very close to the
215   turbine, but the overall predictions are accurate, particularly beyond $x/D = 2$. Disk-averaged velocity deficit profiles are also predicted accurately, but are not shown on Fig. 6(a) to avoid clutter. Figures 7(a-b) show that the Gaussian analytical model is reasonably accurate at predicting the disk-averaged velocity deficit for all thrust coefficients beyond the very-near-wake region, i.e. approximately beyond $x/D = 2$.
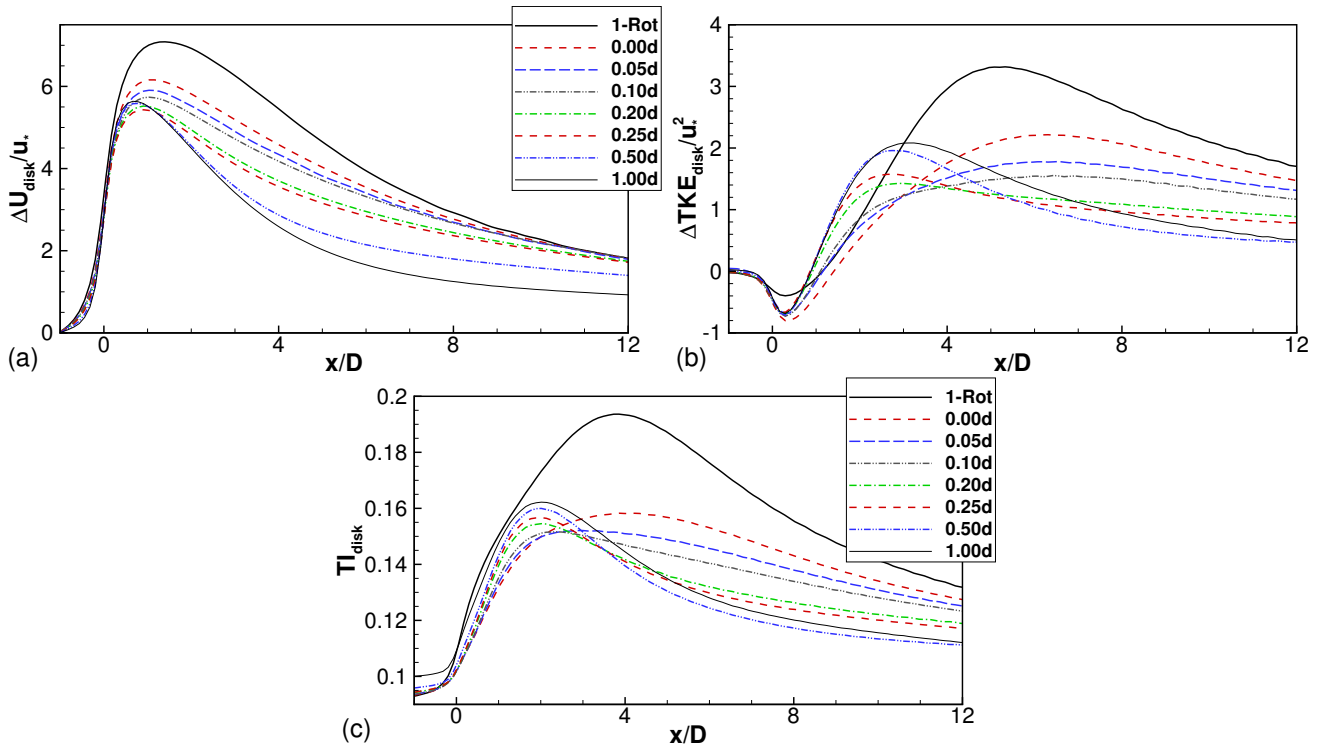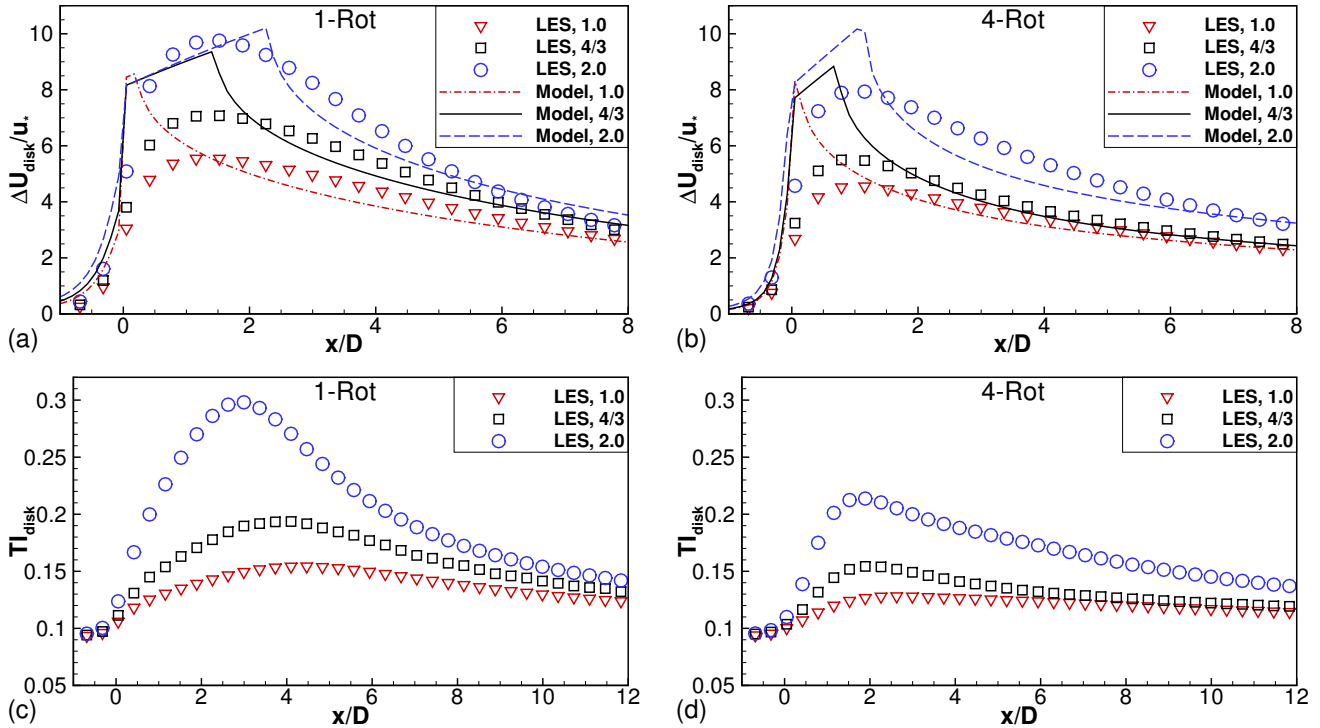
Figure 6. Effect of tip spacing on disk-averaged (a) mean velocity deficit, (b) added TKE and (c) turbulence intensity. Disk averages are computed over rotor disk area(s) corresponding to each turbine configuration. Disk-averaged turbulence intensity is the ratio of the disk-averaged velocity deficit to the disk-averaged velocity $TI_{disk} = \Delta U_{disk}/U_{disk}$.

4   Multi-Turbine Simulation Results

220   Wind farms comprised of a line of five turbines aligned with each other and with the mean wind direction are studied here. These cases are labeled WF* in Table 1.

4.1   Effect of Tip Spacing

The effect of tip spacing on the contours of velocity deficit and TKE are seen in Fig. 8. The axial spacing between different turbines in the wind farm is kept fixed at $4D$ and the thrust coefficient is $4/3$ for all rotors of all turbines. It is clear that the

225   velocity deficits are significantly different between the 1-rotor and 4-rotor wind farms, as well as between 4-rotor wind farms with different tip spacings. The single wake behind the turbines in the 1-rotor wind farm are replaced by four smaller wakes behind the turbines in the 4-rotor wind farms. The wakes move further apart in the radial directions as the tip spacing increases. Similar to the TKE distribution behind an isolated 1-rotor turbine, the TKE values are largest around the top-tip height of the turbines.

Figure 7. Effect of thrust coefficient on disk-averaged (a,b) velocity deficit and (c,d) turbulence intensity for (a,c) 1-rotor turbine and (b,d) 4-rotor turbine with $s/d = 0.1$.

The effect of tip spacing on 4-rotor wind farms is quantified in Fig. 9. Focusing on Fig. 9(a-b), the profiles of the velocity deficits averaged over the rotor disk and TI_{disk} have local maxima close to the turbine locations, i.e. at $x/D = 0, 4, 8, 12$ and 16. The velocity deficit profile for the 1-rotor wind farm has a maximum close to turbine 2 (located at $x/D = 4$), as seen in Figures 9(a) and 8(a). The velocity deficit profile saturates from turbine 3 onward, i.e. the local maxima at $x/D = 8, 12$ and 16 have approximately equal magnitudes. The TI_{disk} profiles in Fig. 9(b) show similar behavior for the 1-rotor wind farm.

The velocity deficits of the 4-rotor turbines are seen in Fig. 9(a) to be smaller than those of the 1-rotor turbine for the first two turbines ($x/D = 0, 4$). In this region, $x/D < 8$, the deficits decrease with increasing tip spacing, which is consistent with the observations for isolated turbines (Fig. 6(a)). The deficits accumulate and the disk-averaged profiles for all 4-rotor wind farms are almost equal to that for the 1-rotor wind farm for turbine rows 3 onward (for $x/D > 8$). The turbulent intensity profiles are smaller for the 4-rotor wind farms than for the 1-rotor wind farm, and decrease with increasing s/d. This sensitivity to the tip spacing persists downstream of all turbines, unlike the velocity deficits, which are sensitive only downstream of the first two turbines.

The relative powers of the turbines are shown in Fig. 9(c). The power of the first (or front) turbine is used for normalization in each wind farm. Thus, the relative power for turbine i is calculated as $P_i/P_1 = \overline{u_i^3}/\overline{u_1^3}$, where the overhead bar represents time-
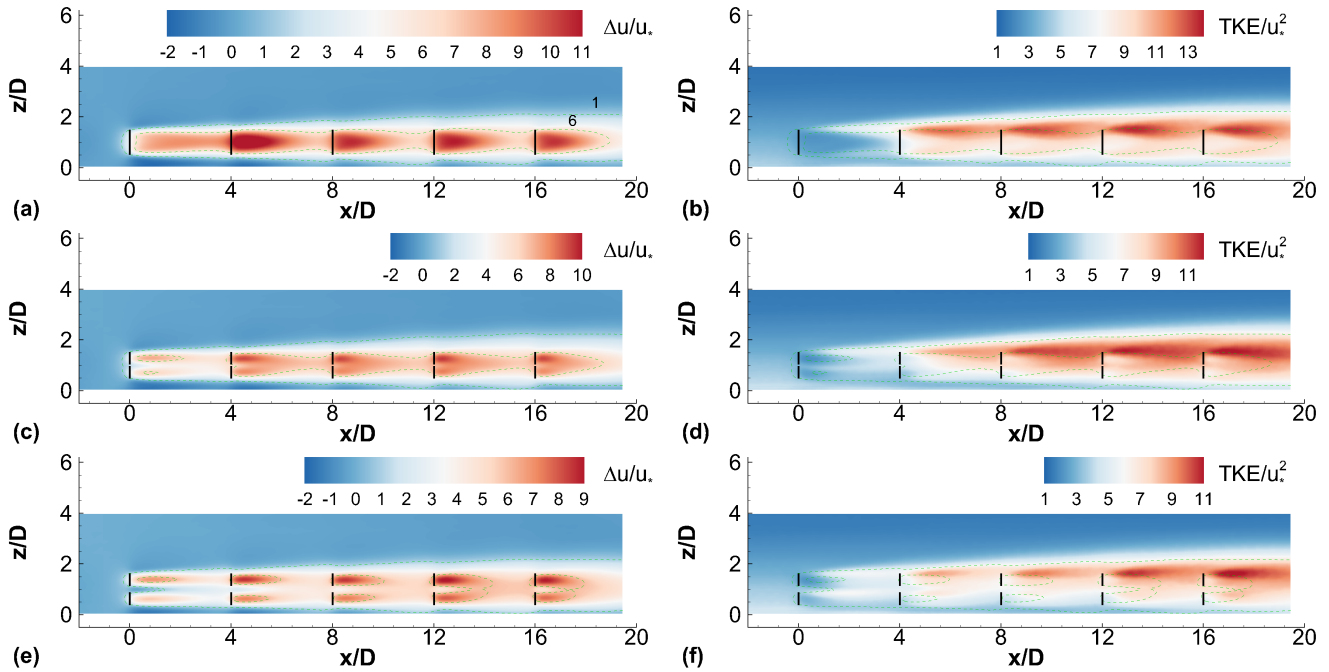
Figure 8. Contours of (a,c,e) streamwise velocity deficit and (b,d,f) TKE at the centerline for (a,b) 1-rotor wind farm, and 4-rotor wind farms with tip spacings (c,d) $s/d = 0.1$ and (e,f) $s/d = 0.5$. Axial spacing is $4D$ in each wind farm. Black lines denote turbine rotors. Dashed lines are velocity deficit contours corresponding to levels $\Delta u/u_* = 1$ and 6.

averaging and subscript i denotes the location of the turbine within the wind farm. The relative power of turbine 2 ($x/D = 4$) in the 1-rotor wind farm is minimum, and the relative power profile shows a slight recovery for turbines 3-5. This is consistent with the maximum for the velocity deficit at turbine 2, seen in Fig. 9(a). The relative powers of turbines in the 4-rotor wind farms are sensitive to the tip spacing as well as the turbine location. For $s/d = 0.1$, only turbine 2 has larger relative power than turbine 2 of the 1-rotor wind farm, while for $s/d = 0.5$, turbines 2-4 have larger relative powers than the corresponding turbines of the 1-rotor wind farm. All these trends are consistent with the velocity deficit profiles seen in Fig. 9(a).

4.2 Effect of Axial Spacing and Thrust Coefficient

The effect of axial spacing on the performance of 4-rotor wind farms can be studied by comparing Figures 9(d-f) to Figures 9(a-c). While the same qualitative trends are seen for axial spacings of $S_X = 4D$ and $6D$, there are significant quantitative differences. The larger spacing between turbines in the $6D$ wind farms allows the wakes to recover to a greater extent before another turbine is encountered. Thus, the disk averaged velocity deficits and turbulence intensities are, in general, smaller in the wind farms with axial spacing of $6D$. Consequently, comparing Figures 9(c) and (f), the relative power values are larger for wind farms with larger axial spacing.

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Interaction between the effects of tip spacing and axial spacing are also seen on comparing Figures 9(c) and (f). For instance, the relative powers of turbines $2, 3$ and 4 of the wind farm with $s/d = 0.5$ are appreciably larger than the corresponding turbines of the 1-rotor wind farm, when the axial spacing is $4D$. However, relative powers of only turbines 2 and 3 of the wind farm

260 with tip spacing $s/d = 0.5$ are appreciably larger than those of the corresponding 1-rotor wind turbines, when the axial spacing is increased to $6D$. Thus, tip spacing has a greater effect on the relative power in a closely spaced wind farm.

Figure 10 shows that the trends observed for $C_T' = 4/3$ hold for other values of thrust coefficient as well. The velocity deficit and turbulence intensity are larger for cases with larger thrust coefficient. For each value of C_T', the velocity deficit of the 4-rotor wind farm is smaller than that of the 1-rotor wind farm downstream of the first two turbines (for $x/D < 8$) and

265 almost equal beyond this. Since the tip spacing of the 4-rotor wind farm is $s/d = 0.1$, only turbine 2 shows a larger relative power in the 4-rotor wind farm compared to the 1-rotor wind farm, consistent with observations made in Fig. 9.

The effect of all governing parameters (s, S_X, C_T') on the wake losses in multi-rotor wind farms is presented succinctly in Fig. 11. Figure 11(a) shows the average power of turbines 2 through 5, normalized by the power of the front turbine in each wind farm. It is seen that P_{2-5}/P_1 is larger for all 4-rotor wind farms than the corresponding 1-rotor wind farm with the same

270 thrust coefficient and axial spacing. The benefit increases with increasing tip spacing.

Each data point in Fig. 11(a) is normalized by the power of the front turbine in the respective wind farm. The front turbine power is expected to be similar to that of an isolated turbine, and hence, is expected to be dependent on the thrust coefficient. This is seen to be the case in Fig. 11(b), where the power of the front turbine extracted from the different wind farm cases are shown. For comparison across cases with different thrust coefficients, all powers are normalized by the power of the front

275 turbine in the 1-rotor wind farm with the same thrust coefficient. Figure 11(b) also shows that the front turbine power in 4-rotor wind farms is weakly dependent on the tip spacing. As the tip spacing varies over $s/d = 0.1$ to 0.5, the front turbine power varies by 3.5%, 2.7% and 3.2%, with the thrust coefficients fixed at $1, 4/3$ and 2, respectively.

To account for the differences in the front turbine power, the average power of turbines 2 through 5 is replotted in Fig. 11(c), with only the 1-rotor front turbine powers used for normalization. The same qualitative conclusions can be drawn from

280 Fig. 11(c), as were drawn from Fig. 11(a), although the magnitudes of the benefit are larger. Finally, the differences between the relative powers of the 4-rotor and 1-rotor configurations are plotted in Fig. 11(d). This plot is directly derived from Fig. 11(c) by subtracting the data points corresponding to the 1-rotor wind farm from the 4-rotor wind farm data, i.e. $\Delta P_{2-5} = P_{2-5} - P_{2-5}^{1-Rot}$. This quantity measures the extent by which wake losses in a 4-rotor wind farm are smaller than wake losses in a 1-rotor wind farm with the same inter-turbine spacing and with all rotors operating with the same thrust

285 coefficient. The benefit of 4-rotor wind farms increases with increasing tip spacing and with decreasing thrust coefficient. The effect of axial spacing on the benefit is slightly ambiguous. For a fixed thrust coefficient and tip spacing, the benefits are largest for $S_X = 4D$, and are almost equal for $S_X = 5D$ and $6D$.

4.3 Analytical Model

Predictions of the analytical modeling framework for wind farms comprised of a line of five turbines are examined in this

290 section. The parameter k_*, which controls the growth rate of the wake, is extracted from all the 1-rotor wind farm LES. First,

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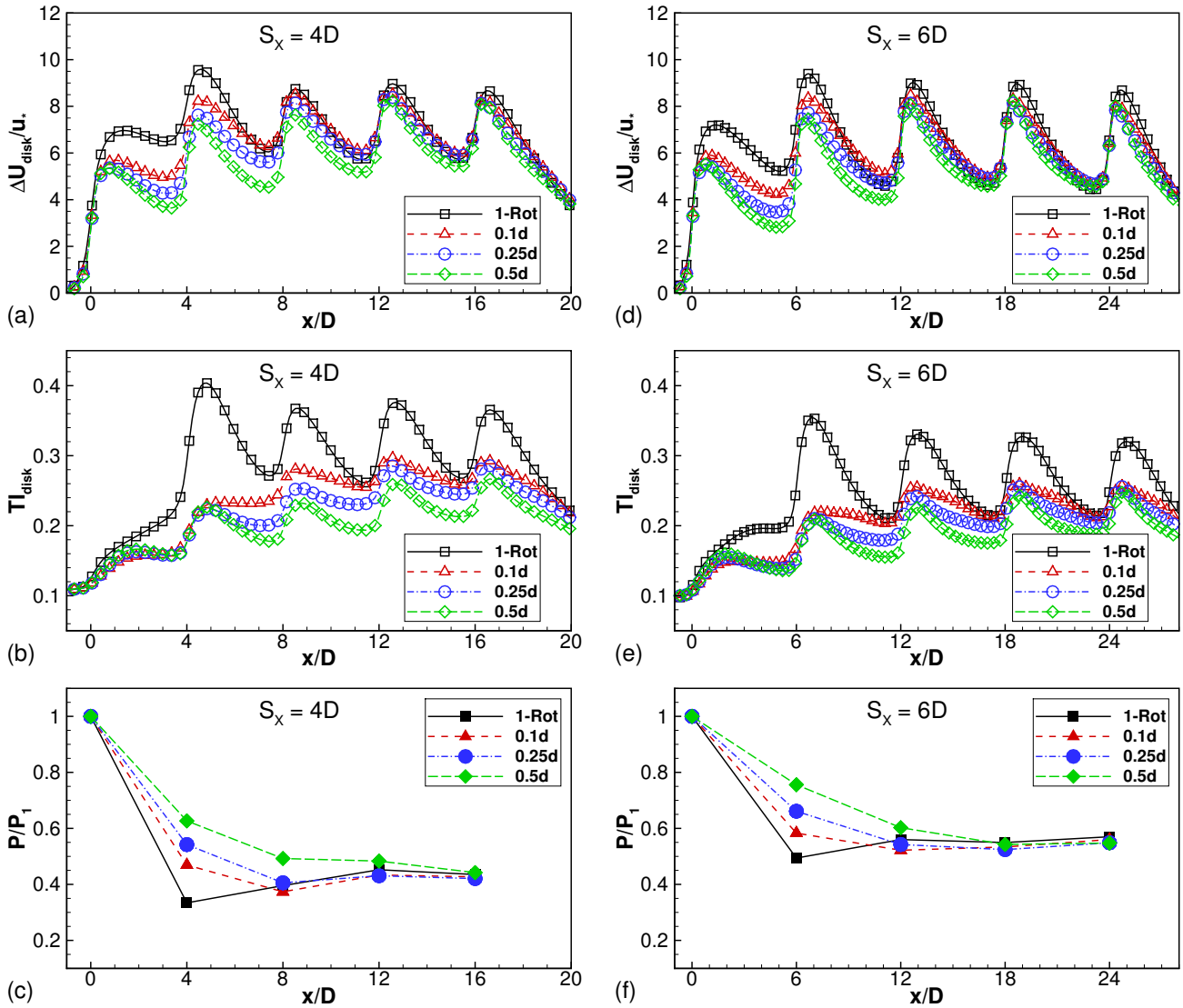
**Figure 9.** Disk-averaged (a) velocity deficit and (b) turbulence intensity, and (c) relative power for wind farms with $S_X = 4D$ and varying tip spacings. Power is normalized by front turbine in each wind farm to compute relative power. (d,e,f) Corresponding results for wind farms with $S_X = 6D$.
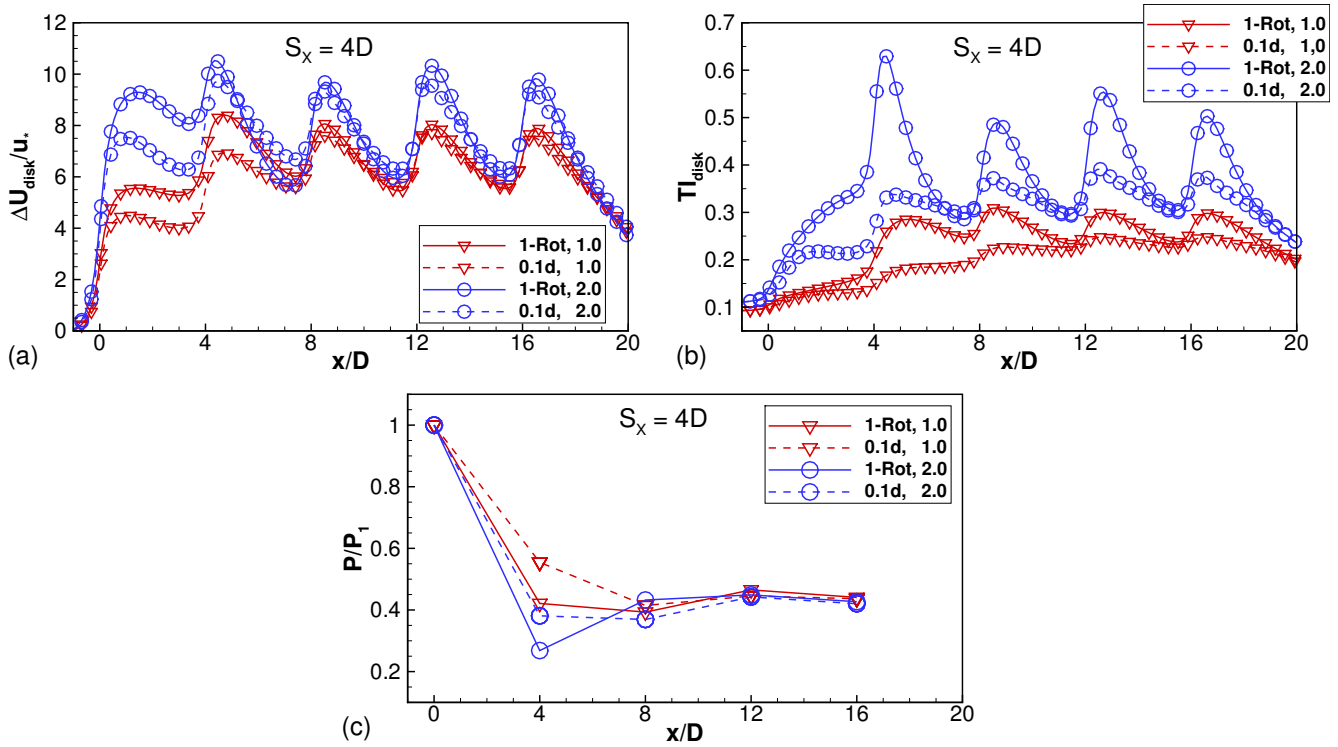
Figure 10. Disk-averaged (a) velocity deficit and (b) turbulence intensity, and (c) relative power for 1-rotor and 4-rotor wind farms with $S_X = 4D$ and varying thrust coefficient.

the wake widths in the y and z directions are calculated using the definition outlined in Bastankhah and Porté-Agel (2016).

$$\sigma_y(x) = \frac{1}{\sqrt{2\pi}\Delta\bar{u}_{max}(x)}\int_{-\infty}^{\infty}\Delta\bar{u}(x,\hat{y},Z_{cen})\,d\hat{y}, \qquad (5)$$

$$\sigma_z(x) = \frac{1}{\sqrt{2\pi}\Delta\bar{u}_{max}(x)}\int_{-\infty}^{\infty}\Delta\bar{u}(x,Y_{cen},\hat{z})\,d\hat{z}, \qquad (6)$$

where $(Y_{cen}, Z_{cen}) = (L_Y/2, 0.1H)$ are the mid-span and mid-vertical planes of the 1-rotor wind turbine wakes, and $\Delta\bar{u}_{max}(x)$

295 is the maximum of the velocity deficit at location x. The wake width is then calculated as the geometric mean of the wake widths in the two transverse directions, $\sigma = \sqrt{\sigma_y\sigma_z}$.

Wake widths extracted from three 1-rotor LES with fixed $S_X = 4D$ and varying thrust coefficient are shown in Fig. 12(a). Turbines are located at $x/D = 0, 4, 8, 12$ and 16 in this plot. Moving downstream from one turbine location, the wake widths generally increase, until the effect of the next downstream turbine is felt. The wake width profiles show dips close to the turbine

300 locations, followed by regions of growth. Regions where the wake widths grow approximately linearly are identified with black solid lines in Fig. 12(a). These black solid lines are linear fits to the data, and the extents of the linear fitting region are identified visually. The slopes of these lines yield the wake growth rate parameter, k_*.

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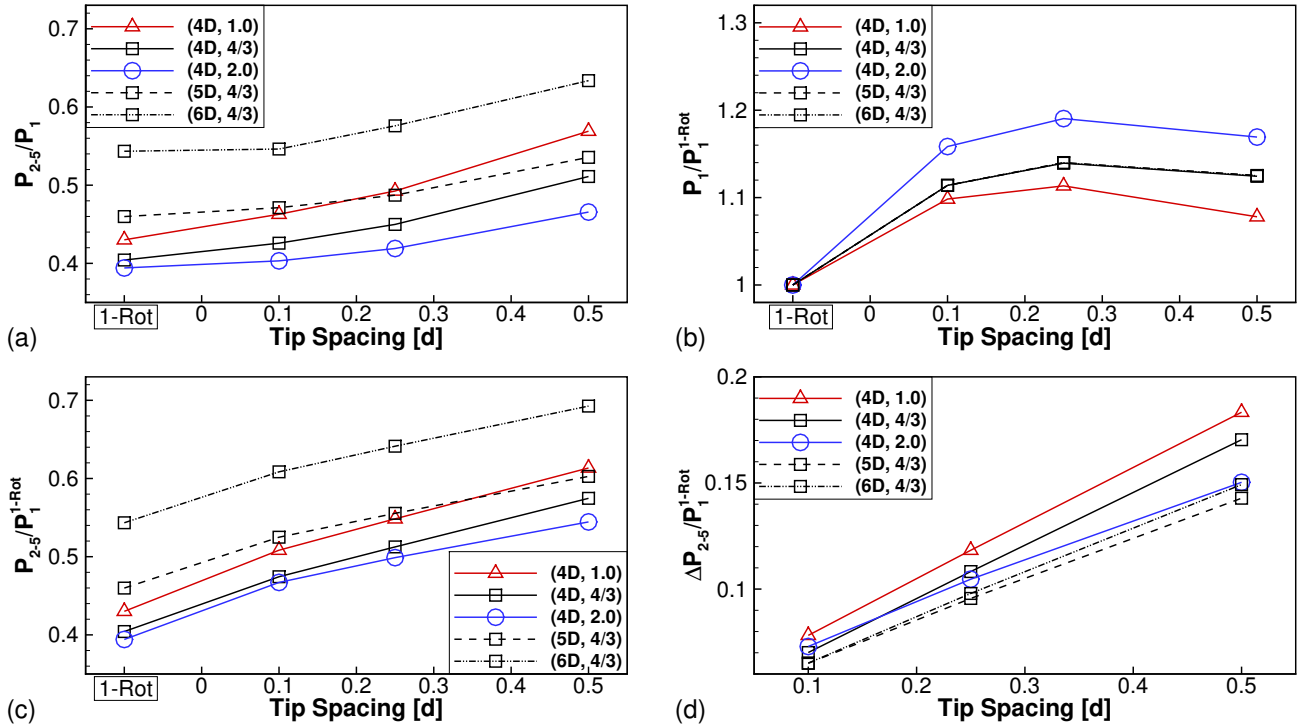
Figure 11. Effect of tip spacing, thrust coefficient and axial spacing on (a) power of turbines 2 through 5 normalized by power of front turbine, (b) power of front turbine and (c) power of turbines 2 through 5 normalized by power of front turbine of corresponding 1-rotor wind farm. (d) Benefit of 4-rotor wind farms over corresponding 1-rotor wind farm.
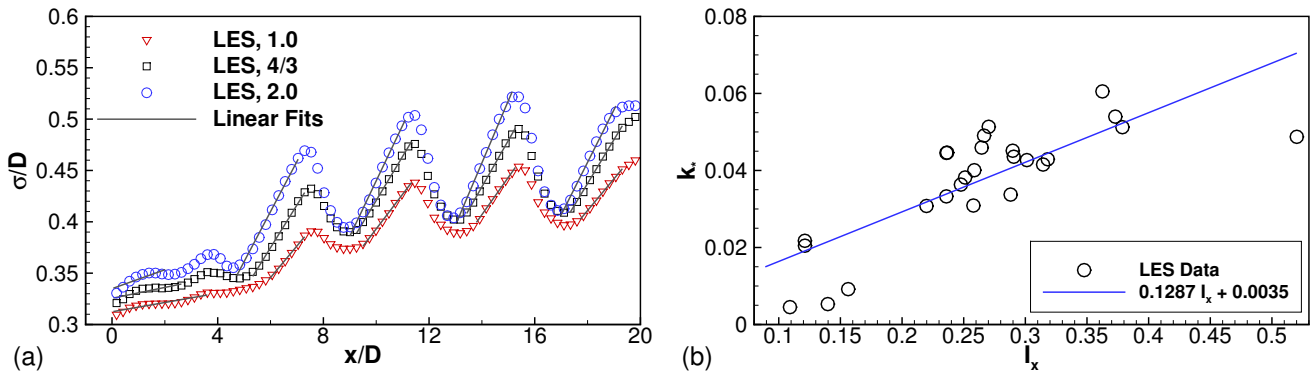
Figure 12. (a) Wake width, σ/D, extracted from LES of 1-rotor wind farms with axial spacing $S_X = 4D$ and varying thrust coefficient indicated in the legend. Slopes of black fitting lines give wake growth rate parameter k_*. (b) Wake growth rate parameter as a function of disk-averaged streamwise turbulence intensity extracted from all LES of 1-rotor wind farms. Blue line is the linear fit to the LES data.

The wake growth rate parameter values for all turbines in the 1-rotor wind farm simulations are compiled in Fig. 12(b). The k_* values are plotted against the streamwise turbulence intensity, I_x, at each turbine rotor disk. As observed in previous

305 studies, the wake growth rate increases with increasing turbulence intensity. The solid blue line fits the data with a correlation coefficient of 0.8. In subsequent model runs for 1-rotor and 4-rotor wind farms, this linear regression model is used to determine k_*, with I_x extracted from the LES results.

Model predictions are compared to LES results for two cases in Fig. 13. The sensitivity of the model predictions to the second tunable parameter, the initial wake width σ_0, is seen in this figure. Figure 13(a) shows that the disk-averaged velocity deficit is

310 over-predicted by the analytical model with $\sigma_0/D =$ very close to the turbines, while it is under-predicted with $\sigma_0/D = 0.32$. Farther away from the turbines, approximately between $1D$ to $3D$ downstream of each turbine, using $\sigma_0/D = 0.28$ yields good agreement with the LES results, while using $\sigma_0/D = 0.32$ continues to yield under-predictions. The power predictions shown in Fig. 13(b) also show similar sensitivity to the value of σ_0. The relative power of turbine 2 is captured accurately with $\sigma_0/D = 0.28$, while the relative powers of further downstream turbines are under-predicted by around 10%. With $\sigma_0/D = 0.32$,

315 the relative power of turbine 2 is over-predicted, while that of further downstream turbines is in better agreement with the LES results. Similar conclusions can be drawn from the results of the 4-rotor turbine with $s/d = 0.1$, shown in Figures 13(c) and (d). In conclusion, the combination of model parameters which leads to accurate predictions in the wake ($1D$ to $3D$ downstream, in the present examples), does not necessarily lead to accurate predictions of power, for which, the values at and very close to the turbines need to be predicted accurately.

320 Relative power predictions for all the wind farm cases are compared to LES results in Figures 14 and 15. The average error in predicting the relative powers of turbines 2 through 5 are shown in each case. The k_* values are obtained as outlined above, while $\sigma_0/d_0 = 0.28$ is used for all cases, where d_0 equals D for the 1-rotor cases and equals d for the 4-rotor cases. The absolute errors in relative power averaged over turbines 2 through 5 are shown in Figures 14 and 15. It should be noted that this level of accuracy is similar to that observed in previous studies (Stevens et al., 2015, 2016) of wind farms that are finite in

325 axial as well as spanwise directions, and where the wind is directed along only one direction, or averaged over a very narrow (less than $2°$) sector.

The errors are seen to be smallest for the 1-rotor cases. For 1-rotor wind farms, typically, the power of the second turbine is smallest, and there is a slight recovery for turbines 3, 4 and 5. This behavior is reproduced well by the analytical model. On the other hand, in the 4-rotor cases, particularly those with $s/d = 0.5$, the power falls gradually and typically saturates by turbine 3

330 or 4. The model predictions fail to reproduce this gradual decrease in power, and, instead, saturate by turbine 2 onward. Thus, the errors are largest for turbine 2, although the relative power level of turbines 4 and 5 is typically well captured.

In conclusion, the analytical modeling framework is capable of reproducing LES results of 1-rotor and 4-rotor wind farms with reasonable accuracy, comparable to previous results for 1-rotor turbines (Stevens et al., 2015). Improved prediction of the region very close to the turbine is needed to further improve the accuracy of the model at predicting the power degradation and

335 wake losses in wind farms.
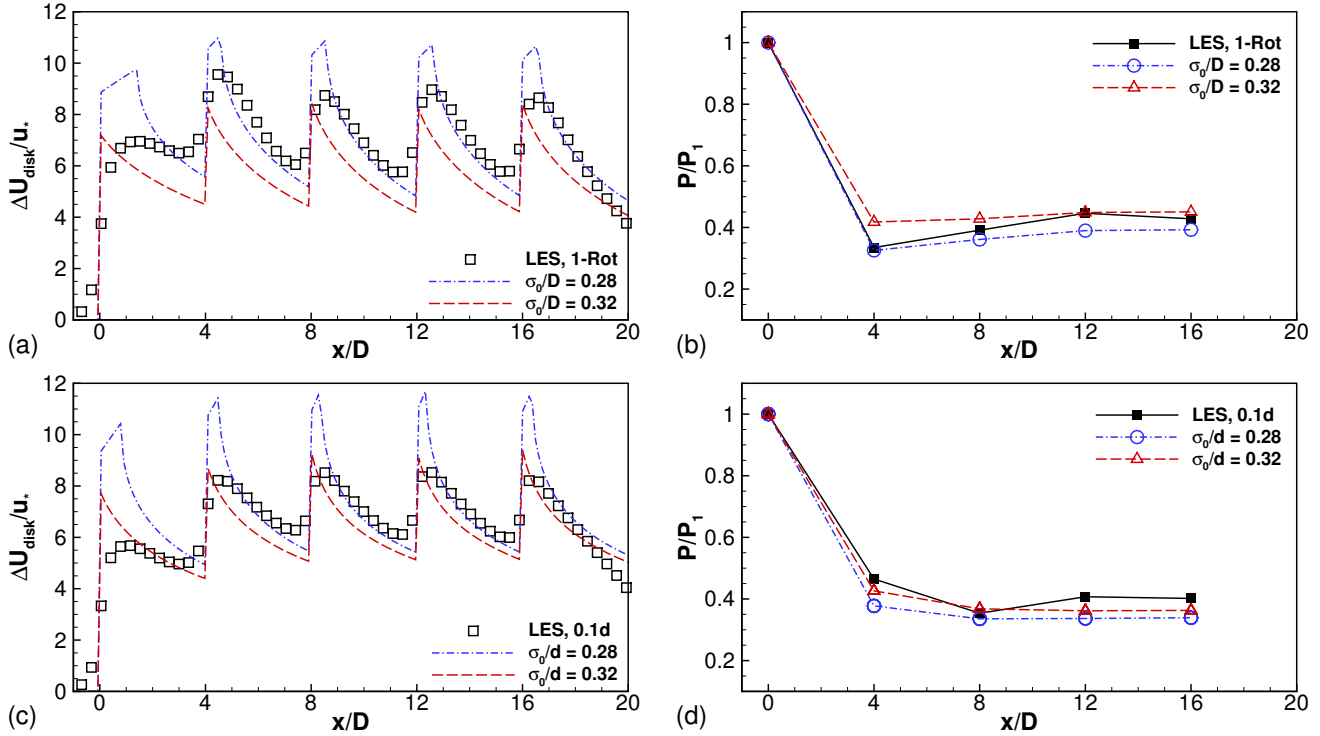
Figure 13. LES results and model predictions of (a) disk-averaged velocity deficit and (b) relative power for 1-rotor wind farm with $S_X = 4D$ and $C'_T = 4/3$. (c,d) Corresponding results for 4-rotor wind farm with $s = 0.1d$.

5 Discussion and Summary

This paper is devoted to studying the turbulent wake of a novel multi-rotor wind turbine configuration, and to comparing it with a conventional single-rotor wind turbine wake. The potential benefits offered by this configuration, with four rotors (with diameters $d = D/2$) mounted on a single tower, over the conventional single-rotor turbine (with diameter D) are studied in

340 detail for the first time. Large eddy simulation is used as the primary tool for this work, Applicability of an analytical modeling framework based on the assumption of Gaussian radial profiles of velocity deficits to the multi-rotor configuration is also examined.

The LES results outlined in Sect. 3 show that an isolated 4-rotor turbine wake recovers faster compared to an isolated 1-rotor turbine wake. The isolated 4-rotor turbine wake also shows smaller TKE levels in the rotor disk region. A simple physical

345 reason for this faster wake recovery and lower TKE levels is that the greater perimeter-to-area ratio of the multi-rotor turbine allows for greater entrainment of low momentum fluid into the wake. The behavior of the wake is sensitive to the tip spacing (s/d), with faster wake recovery seen for for larger s/d. This is consistent with the simple physical reasoning presented above, since if s/d is very large, each rotor of the multi-rotor turbine behaves independently of other rotors, and the wake of each rotor is characterized by the smaller length scale, d. For realistic values of $s/d \sim 0.1 - 0.5$, the rotor wakes do not act entirely
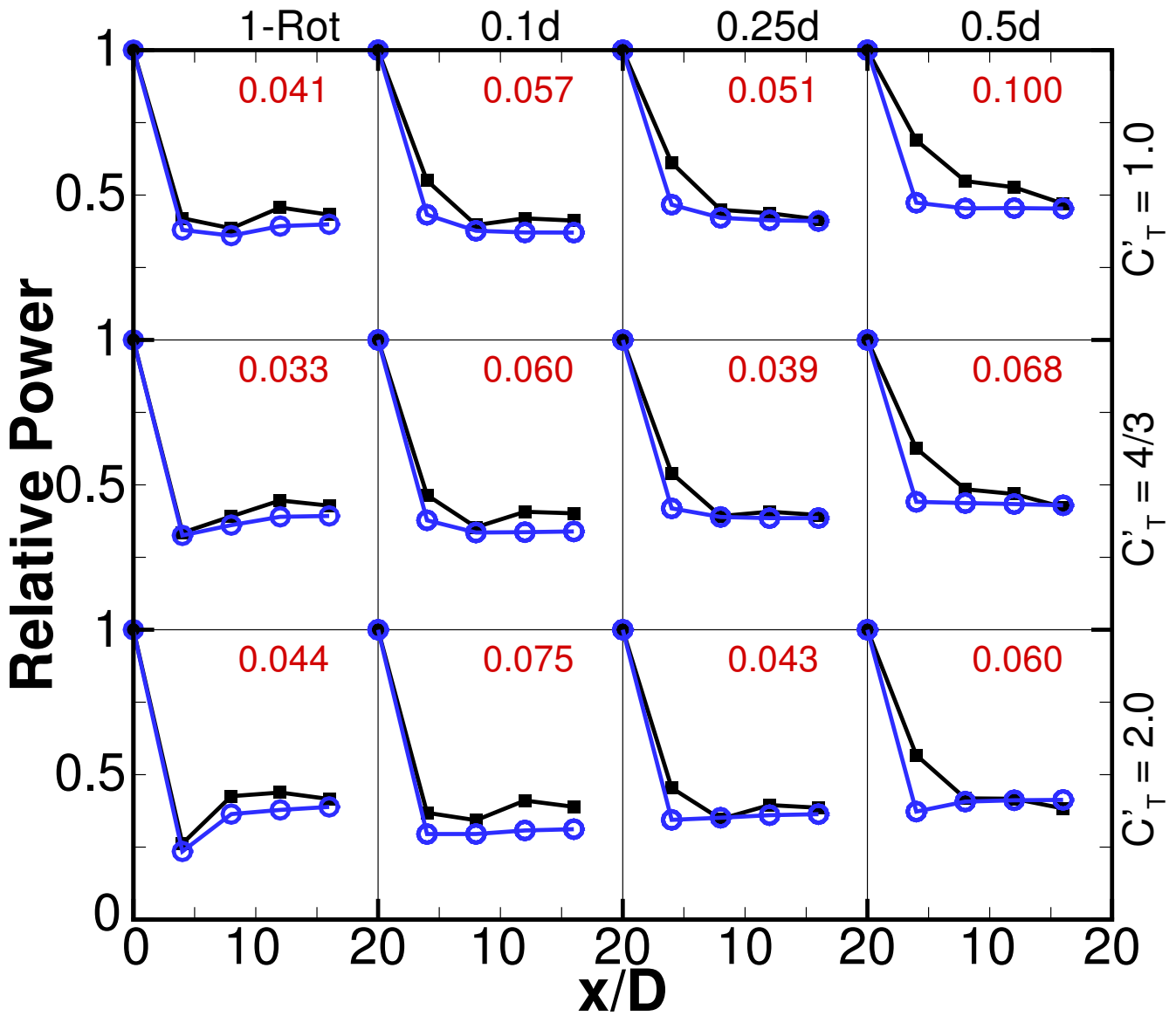
Figure 14. Relative power for 1-rotor and 4-rotor wind farms with fixed $S_X = 4D$, and varying tip spacing and thrust coefficient. Black squares are LES results. Blue circles are model predictions.
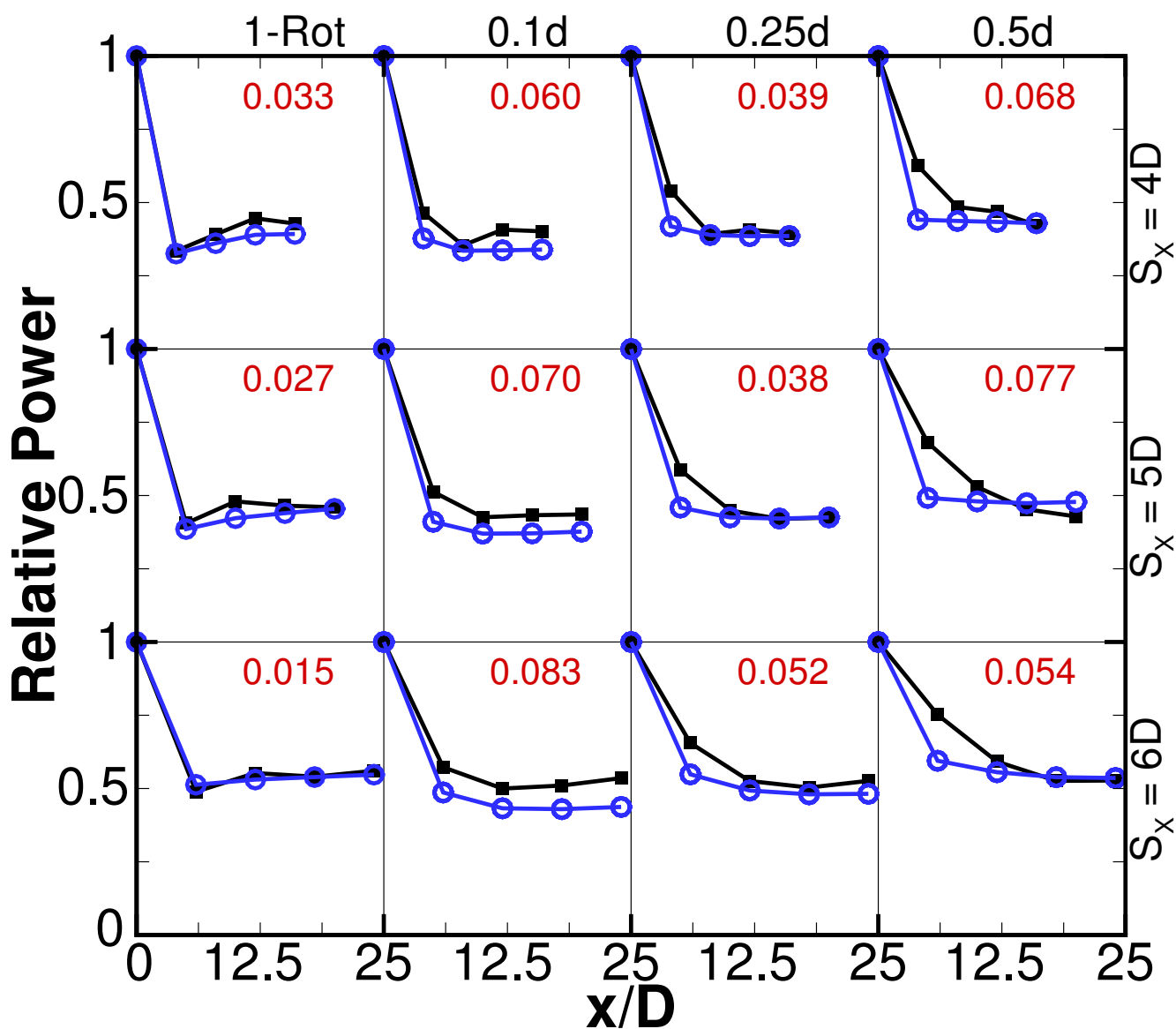
Figure 15. Relative power for 1-rotor and 4-rotor wind farms with fixed $C'_T = 4/3$, and varying tip spacing and axial spacing. Black squares are LES results. Blue circles are model predictions.

350 independent of each other, and the wakes do interact and merge with each other beyond a certain downstream distance. The reduced TKE levels suggest potential for reduced fatigue loads on the blades. These results for an isolated turbine are shown to be consistent for all thrust coefficient (C_T') values evaluated.

In Sect. 4, a line of 5 turbines is evaluated to study the interaction between several multi-rotor wind turbines. For these wind farm simulations, the axial spacing (S_X) between different units is an important parameter, in addition to the tip spacing

355 and the thrust coefficient. Consistent with the results of the isolated turbine LES, the velocity deficits are smaller in 4-rotor wind farms than in the corresponding 1-rotor wind farms until a certain distance into the wind farm. This distance increases with increasing s/d and decreasing S_X. The turbulence intensity levels are significantly smaller for all downstream locations, which indicates potentially smaller fatigue loads for downstream turbines, for all combinations of s/d and S_X. These results are, again, consistent for all C_T' values evaluated using LES.

360 The effect of smaller velocity deficits is reflected in the relative powers, or equivalently, the wake losses experienced by wind farms. Wind farms comprised of multi-rotor turbines always show benefits over similar wind farms comprised of 1-rotor turbines. The benefit increases with increasing tip spacing, and decreasing thrust coefficient. The effect of the axial spacing on the benefit is ambiguous, since it is non-monotonic. This effect should be investigated in more detail in the future. The analytical model predictions are sensitive to the tunable parameters. The results in Sect. 3.4 and 4.3 show that with appropriate choices,

365 reasonably accurate predictions of the LES results can be obtained. The predictions are quite accurate beyond approximately $2D$ downstream of an isolated 1-rotor or 4-rotor turbine. In multi-turbine cases, the predictions are accurate for 1-rotor wind farms, and most 4-rotor wind farms. The model, however, fails to reproduce the trend of gradual decrease in relative power with turbine row, which is particularly pronounced for wind farms with larger s/d. The difficulties in accurately reproducing these trends are partly due to the fact that the Gaussian wake model is valid only beyond a certain distance downstream of a

370 turbine, and is not valid immediately upstream and immediately downstream of a turbine. Thus, this study points to the need for better analytical modeling of the region very close (upstream as well as downstream) to the turbine.

The actuator drag-disk model provides a crude representation of the processes occurring very near the turbine disks. While this crude representation is sufficient for the purposes of capturing the interactions between the turbines and the atmospheric boundary layer, future studies should focus on using the actuator disk/line models with rotation of the blades included. Potential

375 benefits associated with co-rotation and counter-rotation of the rotors in the multi-rotor configuration can be studied. Fatigue loads on individual blades of isolated multi-rotor turbines as well as multi-rotor turbines downstream of other units should also be studied in the future. Finally, developing better analytical models for both, 1-rotor and multi-rotor, configurations continues to be a persistent challenge in wind energy research, and will be pursued in future work.

Appendix A: C_T-Matched 1-Rotor Wind Farms

380 Single-rotor and multi-rotor turbines with the same rotor area and same thrust coefficient have been considered to be equivalent in the main body of this paper. This equivalence was based on the 'local' thrust coefficient, C_T'. Assuming validity of the inviscid actuator-disk theory, imposing a local thrust coefficient implies imposing an induction factor, a, and a thrust coefficient,
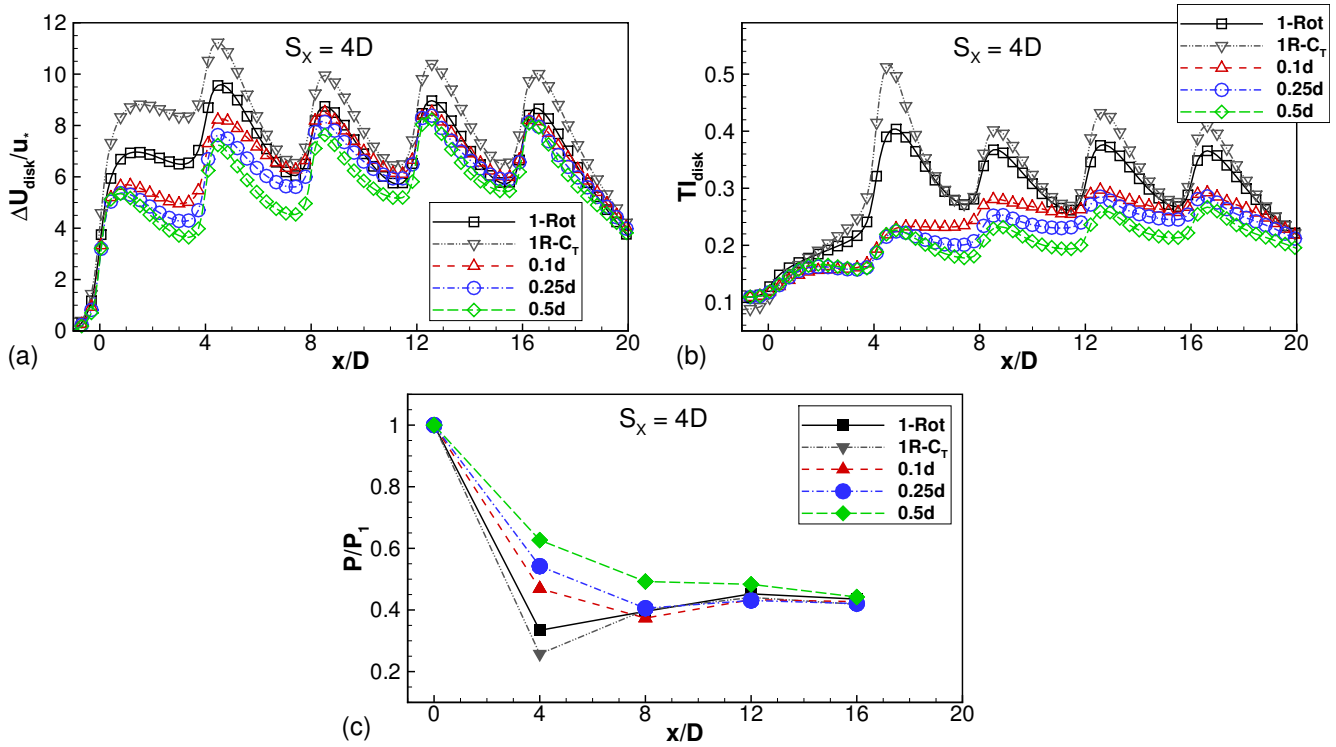
Figure A1. Adding results of 'C_T-matched' run to Figures 9(a-c). Disk-averaged (a) velocity deficits, and (b) turbulence intensity, and (c) relative power for wind farms with axial spacing $S_X = 4D$. $C_T' = 1.61$ for simulation labeled $1R - C_T$ and $C_T' = 4/3$ for all other simulations.

C_T. These quantities are related by

$$C_T' = \frac{C_T}{(1-a)^2}, \quad a^2 - a + \frac{C_T}{4} = 0. \tag{A1}$$

385 The classical actuator-disk theory, however, is not valid for the turbine disks subjected to the sheared, turbulent boundary layer inflow in this study. Consequently, given a value of C_T', the implied values for a and C_T are different from those predicted by eq. (A1). Furthermore, since the single rotor in a 1-rotor turbine and the four individual rotors in a 4-rotor turbine are subjected to different values of shear and turbulence intensity, the implied values of a and C_T are different for the 1-rotor and 4-rotor turbines. As seen in Fig. 11(b), the power of the front turbine in 1-rotor and 4-rotor wind farms is different although identical

390 C_T' values are used for all rotors.

In this appendix, three additional 1-rotor wind farm simulations are reported with $S_X = 4D$ and with C_T' adjusted such that the resulting C_T is closer to those of the corresponding 4-rotor turbines. Through a trial-and-error approach, $C_T' = 1.14$, 1.61 and 2.47 were found to yield C_T values that are within 1.5% of those of the 4-rotor wind farms with $C_T' = 1$, $4/3$ and 2, respectively. These simulations are denoted as 'C_T-matched' runs, and are labeled as $1R - C_T$ in Figures A1 and A2 here.
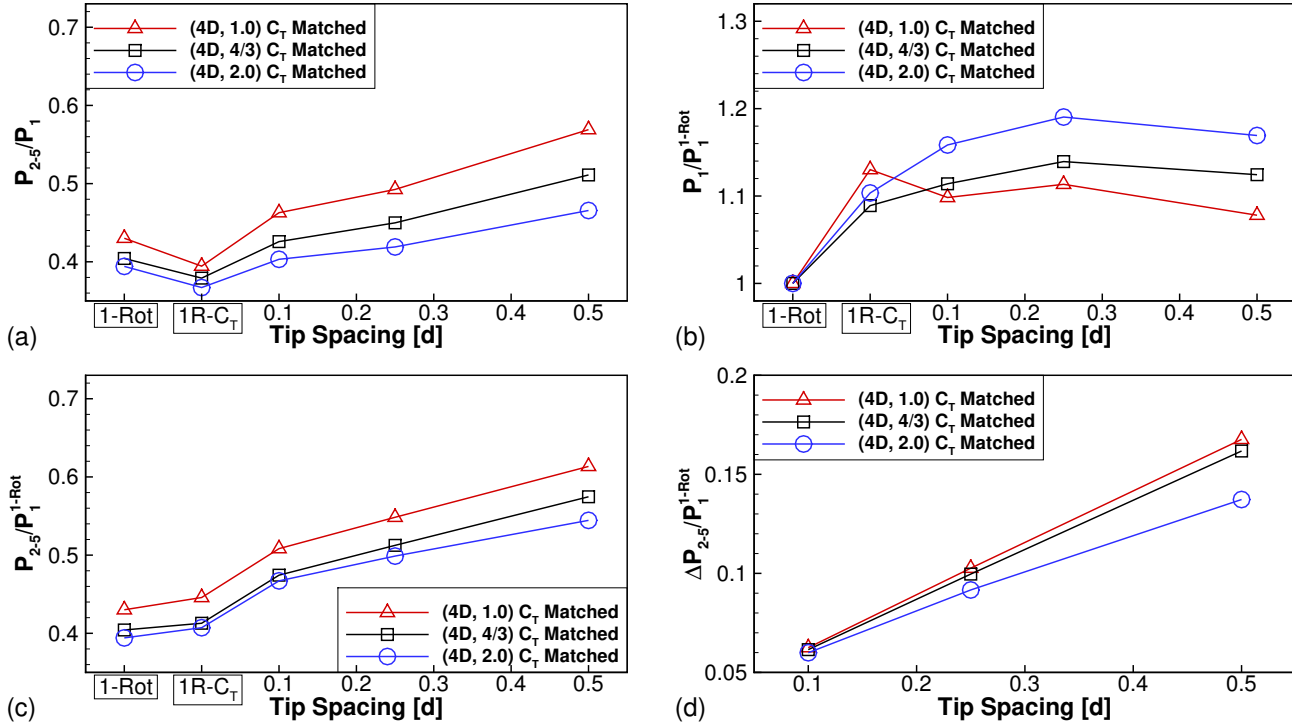
Figure A2. Adding results of 'C_T-matched' runs to Fig. 11. Effect of tip spacing and thrust coefficient on (a) power of turbines 2 through 5 normalized by power of front turbine, (b) power of front turbine and (c) power of turbines 2 through 5 normalized by power of front turbine of corresponding 1-rotor wind farm. (d) Benefit of 4-rotor farms over corresponding C_T-matched 1-rotor wind farm. Labels indicate (S_X, C_T') pairs. $C_T' = 1.14, 1.61$ and 2.47 for the runs labeled $1R - C_T$, corresponding to $C_T' = 1, 4/3$ and 2 respectively.

Figure A1 is a reproduction of Fig. 9(a-c) appended with the additional 1-rotor wind farm simulation with $C_T' = 1.61$. The disk-averaged velocity deficit and turbulence intensity profiles are larger than for the 1-rotor wind farm, particularly at $x/D = 4$ (turbine 2). The resulting power degradation (Fig. A1c) is more severe at turbine 2, and almost identical to the 1-rotor wind farm for further downstream turbines.

Figure A2 is a reproduction of Fig. 11 appended with results from all three 'C_T-matched' runs. Focusing on the black line with squares in Fig. A2(b), it is seen that the power of the front turbine in the additional 1-rotor wind farm simulation (labeled '1R-C_T') is much closer to the powers of the front turbines in the three 4-rotor wind farms, than the front-turbine power in the 1-rotor simulation. In particular, the front-turbine power of the 4-rotor wind farm with $s/d = 0.25$ exceeds the front-turbine power of the 'C_T-matched' wind farm by only 4.4%, while it exceeds the front-turbine power of the 1-rotor wind farm by almost 14%. Similarly, the front-turbine powers of the '1R-C_T' runs are much closer to those of the corresponding 4-rotor wind farms, than the front-turbine powers of the corresponding 1-rotor wind farm. Figures A2(a), (c) and (d) show the same qualitative behavior as Figures 11 (a), (c) and (d). In particular, the benefits of 4-rotor wind farms over the corresponding C_T-matched 1-rotor wind farms are seen in Fig. A2(d). This figure is derived from Fig. A2(c) by subtracting corresponding

'1R-C_T' data point values from each of the 4-rotor data points. Although the numerical values are slightly different from Fig. 11(d), it is clear that the qualitative conclusions do not change, viz. the benefits of 4-rotor wind farms increase with

410 increasing tip spacing and decreasing thrust coefficient.

In summary, this appendix ensures that the qualitative conclusions regarding the benefits of the 4-rotor wind farms remain unchanged, regardless of whether '1-Rot' (C'_T-matched) or '1R-C_T' (C_T-matched) 1-rotor wind farms are used for reference.

415 *Author contributions.* All authors jointly designed the numerical experiments and interpreted the results. NSG and ASG wrote the code and performed the simulations. NSG prepared the manuscript with contributions from all authors.

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