# A fully integrated optimization framework for designing a complex geometry offshore wind turbine spar-type floating support structure

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**Abstract.** Spar-type platforms for floating offshore wind turbines are considered suitable for commercial wind farm deployment. To reduce the hurdles of such floating systems to become competitive, a fully integrated optimization framework and fully modular optimization problem setup is presented which can allow handling of many problems at various levels of fidelity. Thus, at first, a conceptual design study can shed the light on innovative floater configurations, which can, in a second step,

- 5 be further optimized, utilizing the same multi-fidelity framework, to obtain detailed designs. In this paper, a spar floater for a 5 MW wind turbine is used as basis. The approach for generating an initial but very innovative floater design comprises the segmentation of the floating cylinder into three parts, the specification of a freer optimization formulation with less restrictions on the floater geometry, and the allowance for alternative ballast materials. The optimization of the support structure focuses primarily on cost reduction, expressed in terms of the objective to minimize the floater structural material. The opti-
- 10 mization results demonstrate significant potential for cost savings when alternative structural and manufacturing strategies are considered.

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Abbreviations: AEP, Annual Energy Production; ALPSO, Augmented Lagrangian Particle Swarm Optimization; BC, Base Column;  $BC_{low}$ , Base Column lower part;  $BC_{mid}$ , Base Column middle part;  $BC_{up}$ , Base Column upper part; CapEx, Capital Expenditure; COBYLA, Constrained Optimization BY Linear Approximation; DLC, Design Load Case; Dymola, Dynamic Modeling Laboratory; FOWT, Floating Offshore Wind Turbine; IWES, Institute for Wind Energy Systems; LCoE, Levelized Cost of Energy; MoWiT, Modelica library for Wind Turbines; NREL, National Renewable Energy Laboratory; NSGAII, Non-dominated Sorting Genetic Algorithm II; OC3, Offshore Code Comparison Collaboration; OC4, Offshore Code Comparison Collaboration Continuation; OpEx, Operational Expenditure; Rkfix4, Runge-Kutta fixed-step and 4th order method; RNA, Rotor-Nacelle Assembly; SWL, Still Water Level; TI, Turbulence Intensity; TP, Tapered Part; UC, Upper Column

#### 1 Introduction

- 15 With floating support structures for offshore wind turbines, more offshore wind resources can be captured and used for power generation, as around 60% to 80% of the ocean areas cannot be exploited with bottom-fixed structures, which are limited to water depths of up to around 50 m (European Wind Energy Association, 2013). The floating offshore wind technology is no longer in its infancy. Over the last decade, the technology readiness level of floating offshore wind turbine (FOWT) systems has significantly increased so that "floating offshore wind is coming of age", as WindEurope states in its floating offshore
- 20 wind vision statement (WindEurope, 2017, p.4). The large number of research studies, research projects, scaled model tests, prototype developments, and full scale model test phases paved the way towards this current status. More than 40 floating foundation concepts exist and are under development, of which only a few are already used in small floating wind farms (Quest Floating Wind Energy, 2020; Future Power Technology, 2019; James and Ros, 2015; Mast et al., 2015). For further speed-up of the market uptake of floating offshore wind farms, significant cost-reductions are still required.
- From the survey-based study by Leimeister et al. (2018) the conclusion is drawn that the spar-buoy concept is the most mature and has the highest technology readiness level, being already very convenient for volume production and certification due to its simple geometry. However, in order to enhance its suitability for multi-MW wind farm deployment, this technology has to be further advanced in order to benefit from a wider range of possible installation sites, simplified handling (both construction, assembly, transport, and installation), reduced levelized cost of energy (LCOE), as well as improved system motion performance
- 30 (Leimeister et al., 2018). The advancement can be realized in a number of ways. Both Hirai et al. (2018) and Yamanaka et al. (2017) use a three-segmented advanced geometry spar, where a larger diameter column makes up the middle part to allow for shortening the overall length of the spar and reducing the system cost. In contrast, Zhu et al. (2019) utilize the three elements just in an opposite way (the spar element as middle part, interconnecting a slightly larger bottom column and a large upper column), focusing on increased restoring and improved motion performance. Within the Fukushima Floating Offshore Wind
- 35 Farm Demonstration Project FORWARD an advanced spar-type support structure, consisting of a spar with a column each at the bottom, in the middle, and at the upper end, for a floating substation (Fukushima Kizuna) allows for utilization already at around 110 m water depth, improved motion performance, and reduced installation cost (Wright et al., 2019; Yoshimoto and Kamizawa, 2019; Yoshimoto et al., 2018; James and Ros, 2015; Matsuoka and Yoshimoto, 2015; Main(e) International Consulting LLC, 2013). A similarly structured advanced spar, equipped with damping fins for stabilization in sway and heave
- 40 direction, was initially used for a 5 MW wind turbine (Fukushima Hamakaze); however, after some investigations and studies by Matsuoka and Yoshimoto (2015), the middle column was removed to optimize the system restoring, motion performance, and construction cost (Yoshimoto and Kamizawa, 2019; James and Ros, 2015; Main(e) International Consulting LLC, 2013). Other advancements are inspired by the oil and gas industry and deal with, for example, truss spar platforms, in which a truss section connects a bottom tank with the floating platform and heave plates can be included similarly to Fig. 1(a) (Chen et al.,
- 45 2017; Perry et al., 2007; Bangs et al., 2002), or added helical strakes, as shown in Fig. 1(b), for improving the dynamic response of the FOWT system (Ding et al., 2017b, a). The advanced spar-type floater by the Massachusetts Institute of Technology (Lee, 2005), on the other hand, has a relatively shallow draft and gets stability support from a two-layered taut-leg mooring system,

which goes beyond a common delta or so called crowfoot connection of the mooring lines to the spar-buoy structure (Butterfield et al., 2007). Furthermore, an additional tuned mass damper (He et al., 2019) or moonpool (Fig. 1(c)) (Pham and Shin, 2019) can advance the common spar floater.

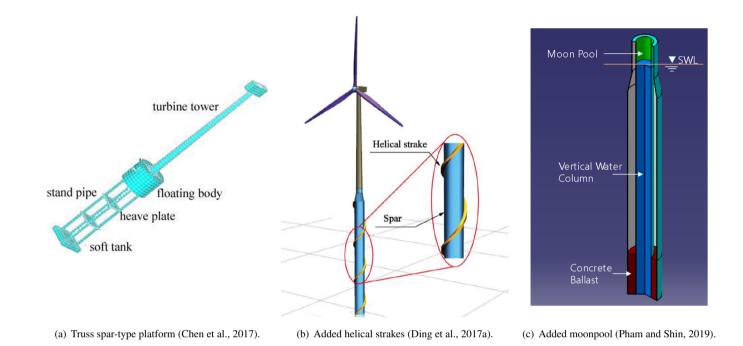


Figure 1. Inspirations for advanced spar-type floaters.

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To obtain an advanced spar-type floater through optimization, most research approaches are based on the common cylindrical spar-type floater shape and utilize gradient-based methods (Dou et al., 2020; Hegseth et al., 2020; Berthelsen et al., 2012; Fylling and Berthelsen, 2011) or genetic algorithms (Karimi et al., 2017; Choi et al., 2014). Some applications are purely dealing with the support structure—focusing on basic hydrodynamic analyses, maximum system stability, and minimum ma-

- 55 terial cost (Choi et al., 2014), reduced draft, weight, and cost with at the same time increased power output (Lee et al., 2015), or optimized floater cost and power generation (Gao and Sweetman, 2018)—while other design optimization approaches are highly complex and account for optimizing several components of the FOWT system, such as the tower, mooring system, power cable, and/or blade-pitch controller in addition to the floating platform, and focus on extreme loads, structural strength, fatigue life, or power quality in addition to costs and global system responses (Dou et al., 2020; Hegseth et al., 2020; Sandner
- 60 et al., 2014; Fylling and Berthelsen, 2011) or distinguish also between different floater types (Karimi et al., 2017; Sclavounos et al., 2008). Even if a reduced draft is often aimed and obtained (Hegseth et al., 2020; Gao and Sweetman, 2018; Lee et al., 2015; Sandner et al., 2014) and sometimes the spar-buoy floater is subdivided into several cylindrical sections (Hegseth et al., 2020; Berthelsen et al., 2012; Fylling and Berthelsen, 2011) or a broad range of allowable values is considered for the design

variables (Karimi et al., 2017; Sclavounos et al., 2008), always common spar-type platform designs are considered, meaning a

- 65 structure consisting of welded sections, for which reason even Hegseth et al. (2020) limit the maximum allowable taper angle. Thus, this paper aims to demonstrate that through a freer optimization formulation more potential solutions for an advanced spar-type floater design with a higher degree of innovation can be captured. The conceptual design study and optimization approach, applied in this work, focus on hydrodynamic and system-level analyses. Due to the conceptual character of this study, which precisely targets to explore novel design spaces, not that stringent limitations on the structure and dimensions are
- 70 yet required. The optimization approach followed in this paper bases on an initial design optimization example by Leimeister et al. (2020b), which, however, is quite simple and does not include any aspects and goals for going beyond and advancing the common spar-buoy floater design but only focuses on optimizing the global system performance. While global system performance criteria still have to be fulfilled but are only incorporated as constraints, the main objective of this study is cost reduction—expressed in terms of the material used—and the optimization problem is specified in such a way that ad-
- 75 vancements, which go beyond just obtaining a reduced draft, can be achieved. Hence, by allowing design variables out of a wider range of values, contemplating different ballast materials, and considering novel structural realization approaches for the resulting optimized geometries, new alternatives of potential and innovative floater design solutions are opened up. All these requirements regarding design variables and optimization criteria are—together with specific environmental conditions and the fully coupled aero-hydro-servo-elastic dynamic characteristics of a FOWT system—incorporated into a fully mod-
- 80 ular optimization framework. Its current capability is sufficient for this conceptual design study; however, due to its close interlinking with the—as well—fully modular and multi-fidelity numerical modeling environment, the framework can easily be extended for serving more holistic FOWT system design optimization problems of higher fidelity, including a subsequent detailed design development.
- In order to figure out in detail the required characteristics of such a floating platform, first (Sect. 2), a reference FOWT system with corresponding modifications and assessment criteria towards more innovative design solutions is specified, on which basis the optimization problem—consisting of design variables, objective function, and constraints—can be defined. Subsequently, the automated design optimization of the advanced spar-type FOWT system is performed in Sect. 3, including some preprocessing automated design load case (DLC) simulations, as well as the characterization of the automated optimization framework and the iterative optimization approach. The results of the optimization simulations are presented in Sect. 4 and further discussed in Sect. 5. Finally, some conclusions are drawn in Sect. 6.

#### 2 Forming the basis for innovative floater configurations

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According to the survey conducted by Leimeister et al. (2018), industry professionals and scientific experts judge the advanced spar-type floater—compared to the common spar-buoys, semi-submersibles, tension leg platforms, barges, or any hybrid, multi-turbine, or mixed-energy floating system—to be the most suitable wind turbine support structure for deployment in floating offshore wind farms. To prepare the fully modular optimization problem setup, by which means problems at various levels of fidelity—such as, in this first instance, the conceptual design study on innovative floater configurations—can be addressed,

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both a reference FOWT system (Sect. 2.1) and the advancement options, with necessary design modifications (Sect. 2.2) and associated assessment criteria (Sect. 2.3), need to be defined. From these specifications, the modifiable design variables  $x_i$ collated in the design variables vector X, the objective functions  $f_i$  to be minimized, as well as the equality  $(h_i)$  and inequality

100  $(q_i)$  constraints to be fulfilled, are derived to set out the optimization problem according to the following formal expressions, in which the functions are either directly dependent on the design variables or also on the resulting fully coupled FOWT system, denoted with system(X).

find  $X = \{x_1, ..., x_k\}$ to minimize  $f_i(X, system(X))$ , i = 1, ..., l $h_i(X, system(X)) = 0$ , i = 1, ..., msubject to  $q_i(X, system(X)) < 0$ , i = 1, ..., nsubject to

#### Reference floating offshore wind turbine system 2.1

105 As starting point of the design optimization towards an innovative floating platform for an offshore wind turbine, a traditional spar-buoy FOWT system, taken from phase IV of the OC3 (Offshore Code Comparison Collaboration) project (Jonkman, 2010) with the main properties provided in Appendix A, is used. This is further modified, as explained in Sect. 2.2, to allow the realization of advanced features and the development of an innovative spar-type structure.

#### 2.2 Design modifications for facilitating advancements and innovative floater configurations

- An aero-hydro-servo-elastic coupled model of dynamics of the reference spar-buoy FOWT system (Sect. 2.1) is developed and 110 verified by Leimeister et al. (2020a), using MoWiT<sup>1</sup> (Modelica library for Wind Turbines), which is developed at Fraunhofer Institute for Wind Energy Systems (IWES) (Leimeister and Thomas, 2017; Thomas et al., 2014; Strobel et al., 2011). The modeling approach in MoWiT utilizes the object-oriented, equation-based, and component-based modeling language Modelica<sup>2</sup> and, therefore, follows a hierarchical structure with interconnected main components and subcomponents to represent the 115 complex wind turbine system and corresponding fully coupled system dynamics in accompanying time-domain simulations.
- This multibody approach provides high flexibility to model various wind turbine system types, environmental conditions, and simulation settings by simply modifying single model components and, hence, forms the basis for an approach of a framework towards multi-fidelity.

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This MoWiT model of the OC3 phase IV spar-buoy FOWT system is used as basis and modified to facilitate advancements so that innovative floater designs can be obtained through automated optimization. As this work focuses on the design of the floating structure and not on the mooring system, a shorter, less heavy, and, hence, cheaper platform design shall be obtained by changing the floater geometry and allowing for further advanced features, which will be addressed in detail in Sect. 2.3. Different characteristic shapes of advanced spar-type floating platforms are pointed out in Sect. 1. In this study, a similar

<sup>&</sup>lt;sup>1</sup>www.mowit.info (Accessed: 25 May 2021)

<sup>&</sup>lt;sup>2</sup>https://www.modelica.org/ (Accessed: 22 January 2020)

concept as presented by Zhu et al. (2019) and realized in the Fukushima Hamakaze FOWT system (Matsuoka and Yoshimoto,

- 125 2015; Yoshimoto and Kamizawa, 2019) is applied. Thus, the long cylindrical element below the tapered part is divided into three partitions: the base column upper part  $BC_{up}$ , which shall serve for gaining buoyancy; the base column middle part  $BC_{mid}$ , which mainly provides the separation of the other two parts to deepen the position of part 3; and the base column lower part  $BC_{low}$ , which can be filled with ballast and this way shall shift the center of gravity downwards.
- This partitioning is schematically represented in Fig. 2, showing the unchanged geometric parameters and dimensions for 130 the upper column (UC) and tapered part (TP) in a light shade (gray) and indicating the three sections of the base column (BC) together with the ballast filling in the base column lower part  $BC_{low}$ . Based on this derivation of the modified spar-buoy floater model for enabling the incorporation of the considered advancements, seven design variables are defined as presented in Table 1.

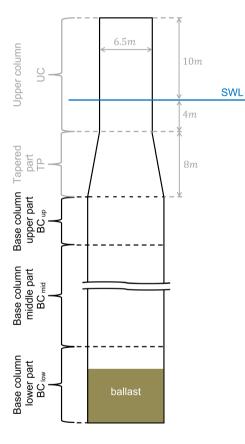


Table 1. Definition of the seven design variables.

Design variable	Formal expression	Description
$x_1$	$D_{ m BC_{up}}$	Diameter of $BC_{up}$
$x_2$	$D_{ m BC_{mid}}$	Diameter of $BC_{\rm mid}$
$x_3$	$D_{ m BC_{low}}$	Diameter of $BC_{\rm low}$
$x_4$	$H_{\rm BC_{up}}$	Height of $BC_{\rm up}$
$x_5$	$H_{ m BC_{mid}}$	Height of $BC_{\rm mid}$
$x_6$	$H_{ m BC_{low}}$	Height of $BC_{low}$
$x_7$	$H_{\mathrm{ballast}}$	Height of the ballast

**Figure 2.** Geometrical definitions of the advanced spar-type floating platform.

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Apart from these modifications, which are directly related to advancements of the geometric configuration, also the material density of the support structure and the wall thickness of the cylindrical spar-buoy elements are changed. As the material density of the OC3 phase IV spar-buoy is not explicitly stated in the definition document (Jonkman, 2010), a value of 10,000 kg/m<sup>3</sup>

is derived in the model verification study (Leimeister et al., 2020a). However, to better match the common steel properties of offshore structures, a material density of 7,850 kg/m<sup>3</sup> is used in this study. Furthermore, the wall thickness of the spar<sup>3</sup> is changed from the fixed value of 0.0314 m, which is derived by Leimeister et al. (2020a), to a wall thickness that is adaptable

to the specific floater configuration. To obtain an appropriate wall thickness for a corresponding floater design, a fixed ratio of the support structure structural mass to the displaced mass of water is deployed. This ratio is for a spar-type floating platform 0.13—according to representative values from research designs and academic studies and excluding designs, such as the Hywind demonstrator, which are for safety reasons heavily oversized (Bachynski, 2018). Hence, the equivalent structural mass of the spar-type floater (meaning the mass of the spar-type steel structure, excluding tower, wind turbine, and ballast)
with certain outer dimensions (diameters D<sub>i</sub> and heights H<sub>i</sub>) and corresponding displaced volume can be determined following

### Eq. (1).

$$\frac{\text{floater structural mass}}{\text{buoyancy mass}} = 0.13\tag{1}$$

This results for the reference spar-buoy floating system into a structural mass of 1.070E+06 kg, which is a bit lower than the original structural mass of 1.150E+06 kg (Leimeister et al., 2020b). The corresponding appropriate wall thickness, which is
150 kept the same for all parts of the specific floater geometry, is computed by means of Eq. (2). This equation is derived from the expression for the spar-type floater structural mass, using a material density of 7,850 kg/m<sup>3</sup> as explained earlier. In Eq. (2), *H<sub>i</sub>* and *D<sub>i</sub>* are the heights and diameters of each element, meaning UC, TP, BC<sub>up</sub>, BC<sub>mid</sub>, and BC<sub>low</sub>. The diameter of the tapered part *D*<sub>TP</sub>, however, is determined according to Eq. (3) as mean of the diameters of UC and BC<sub>up</sub>.

wall thickness = 
$$\frac{\sum_{i} (H_i D_i) - \sqrt{\left[\sum_{i} (H_i D_i)\right]^2 - \frac{4}{\pi} \frac{\text{floater structural mass}}{\text{material density}} \sum_{i} H_i}}{2\sum_{i} H_i}$$
(2)

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$$D_{\rm TP} = \frac{D_{\rm UC} + D_{\rm BC,up}}{2}$$
 (3)

This way, a wall thickness of 0.0372 m is obtained for the original OC3 phase IV spar-buoy with reduced material density (7,850 kg/m<sup>3</sup>) and adopted structural mass to displaced mass ratio of 0.13. This thickness value lies within the acceptable range, based on available data for the OC4 (Offshore Code Comparison Collaboration Continuation) phase II semi-submersible floater.

As the conceptual design (optimization) study does not focus on the mooring system, as already mentioned, and due to the

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fact that the mooring system itself could be covered in a separate or subsequent detailed design optimization task, any change in the restoring system characteristics due to shifted fairlead positions is prevented by utilizing constant (the original) resulting mooring system properties. This means that independent of possible attachment points to the reshaped floating platform the resulting stiffness of each mooring line is taken from the system motion, assuming the original fairlead positions as defined in Sect. 2.1. A realistic mooring system design for the finally obtained optimized floating platform, which represents the

<sup>&</sup>lt;sup>3</sup>Referring here purely to the circumferential walls of the hollow cylindrical or conical elements, as for base and lid a fixed marginal cap thickness of 0.0001 m is applied, according to the implemented model in the verification study (Leimeister et al., 2020a).

- 165 considered resulting mooring system properties, can then afterwards be obtained through a subsequent optimization. This might even happen manually—depending on the degree of complexity—as it is applied in studies for designing equivalent mooring systems (Molins et al., 2015; Udoh, 2014). However, having not included the mooring system as design variable within the optimization of the spar-type floating platform, further system performance improvements due to modified mooring system parameters or fairlead positions—in addition to an optimized support structure design—are limited. This, however,
- 170 leaves open the possibility of subsequent fine tuning of the conceptual design solution obtained through optimization based on hydrodynamic and system-level analyses. By addressing the mooring system in the subsequent detailed design optimization or even in a successive but separate optimization algorithm, the dynamic response of the FOWT system, as well as the mooring line tension itself, can be significantly improved by considering an advanced and more complex optimization problem, in which—apart from various line diameters and lengths—different mooring line arrangements and distribution forms can be
- utilized, the optimum number of lines within the mooring system and best fairlead position elaborated, different mooring types used or even mixed within segmented lines, and also clump weights incorporated (Tafazzoli et al., 2020; Barbanti et al., 2019; Men et al., 2019; Chen et al., 2017).

#### 2.3 Assessment criteria towards more innovative floater design solutions

Within this study, the advancements for achieving a conceptual design of an innovative floating platform go beyond the main

- 180 objectives to reduce the draft of the floater and the cost of the overall system. Further advanced features comprise the investigation of alternative materials, which are from an economic point of view comparative to currently used materials, however, positively influence the final floater design due to their different material properties and characteristics, as well as the consideration of novel structural approaches which might be more promising than the common approach of welding cylindrical and tapered sections together and allow a widening of the design space for such innovative floater shapes. In this conceptual design
- 185 study, any detailed structural integrity checks are not yet addressed. However, due to the multi-fidelity model, optimization problem, and framework setup, these can be added easily for a more extensive optimization approach in a subsequent detailed design study. The advantage of focusing right now only on hydrodynamics and global system performance without defining any restrictions regarding structural aspects is that floater designs, which would have been discarded when performing structural integrity checks and as they would be unfeasible to be realized with conventional structural approaches, can still be captured as potential solutions when considering different structural realization approaches.

The only structural related focus, considered in this approach, is the minimization of the structural cost. This is represented through the one and only objective function  $f_1$  (Eq. 4), which aims to minimize the structure material volume of the advanced spar-type floating platform.

$$f_1 = \frac{\text{floater structural mass}}{7,850 \text{ kg/m}^3} \tag{4}$$

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In order to achieve a shortened length of the floater, the allowable system draft is limited to the original draft of the OC3 phase IV FOWT as maximum value, as well as to a recommended minimum value of 15.0 m (Ng and Ran, 2016). The resulting allowable total height of the BC (between 3.0 m and 108.0 m) has to be distributed to the three partitions; however,

no restrictions prevail and also the option of utilizing not all three BC parts is possible. Thus, the minimum allowable value for the height of each of the BC parts is machine epsilon  $(10^{-15} \text{ m})$ —as a zero value is unfeasible from a modeling point of view. These requirements with regard to the overall system draft and the heights of each partition are represented in Table 2 by the inequality constraints  $g_{18}$  and  $g_{19}$ , as well as  $g_7$  to  $g_{14}$ , respectively. For the ballast height, it additionally has to be guaranteed

that it does not exceed the actual BC<sub>low</sub> height, what implies inequality constraint  $g_{20}$ .

	Table 2.	Definition	of the 25	inequality	constraints.
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Inequality constraint	Formal expression	Description
$g_1$	$10^{-15} \mathrm{m} - x_1$	Minimum allowable value of $x_1$
$g_2$	$x_1 - 120.0 \text{ m}$	Maximum allowable value of $x_1$
$g_3$	$10^{-15} \mathrm{m} - x_2$	Minimum allowable value of $x_2$
$g_4$	$x_2 - 120.0 \text{ m}$	Maximum allowable value of $x_2$
$g_5$	$10^{-15} \mathrm{m} - x_3$	Minimum allowable value of $x_3$
$g_6$	$x_3 - 120.0 \text{ m}$	Maximum allowable value of $x_3$
$g_7$	$10^{-15} \mathrm{m} - x_4$	Minimum allowable value of $x_4$
$g_8$	$x_4 - 108.0 \mathrm{~m}$	Maximum allowable value of $x_4$
$g_9$	$10^{-15} \mathrm{m} - x_5$	Minimum allowable value of $x_5$
$g_{10}$	$x_5-108.0~{ m m}$	Maximum allowable value of $x_5$
$g_{11}$	$10^{-15} \mathrm{m} - x_6$	Minimum allowable value of $x_6$
$g_{12}$	$x_6 - 108.0 \mathrm{~m}$	Maximum allowable value of $x_6$
$g_{13}$	$10^{-15} \mathrm{m} - x_7$	Minimum allowable value of $x_7$
$g_{14}$	$x_7 - 108.0 \mathrm{~m}$	Maximum allowable value of $x_7$
$g_{15}$	$\max(\text{total inclination angle}) - 10.0^{\circ}$	Maximum total inclination angle
$g_{16}$	$\max(\mathrm{horizontal}\ \mathrm{nacelle}\ \mathrm{acceleration}) - 1.962\ \mathrm{m/s^2}$	Maximum horizontal nacelle acceleration
$g_{17}$	mean(translational motion) $- 64.0 \text{ m}$	Mean translational motion
$g_{18}$	$3.0 \mathrm{m} - (x_4 + x_5 + x_6)$	Minimum draft
$g_{19}$	$x_4 + x_5 + x_6 - 108.0 \text{ m}$	Maximum draft
$g_{20}$	$x_7 - x_6$	Ballast filling height within $BC_{low}$
$g_{21}$	-ballast density	Minimum allowable value of the ballast density
$g_{22}$	ballast density $-5.0\mathrm{E}{+}03 \mathrm{kg/m}^3$	Maximum allowable value of the ballast density
$g_{23}$	$0.5 \cdot 10^{-15} \text{ m} + \text{wall thickness} - 0.5x_1$	Wall thickness and diameter of $BC_{\rm up}$
$g_{24}$	$0.5 \cdot 10^{-15} \text{ m} + \text{wall thickness} - 0.5x_2$	Wall thickness and diameter of $BC_{\rm mid}$
$g_{25}$	$0.5\cdot 10^{-15} \mathrm{\ m+wall} \mathrm{\ thickness} - 0.5 x_3$	Wall thickness and diameter of $BC_{\rm low}$

The applied concept of a three-segmented spar-type floater with elements for buoyancy, distance, and ballast shall not only allow different heights but also different diameters of these elements. Thus, the allowable value range for the diameter of each of the BC parts is set from machine epsilon—due to the same modeling feasibility reason—to 120.0 m, leading to the inequal-

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ity constraints  $g_1$  to  $g_6$  specified in Table 2. The maximum diameter is chosen deliberately large—corresponding to the total maximum draft of the floating system—to ensure that the border of feasible solutions is well captured. From a manufacturing point of view, cylindrical offshore structures with diameters of more than 10.0 m are realistic: Various sources<sup>4,5</sup> state a value of 11.0 m, the reference semi-submersible floating platform from phase II of OC4 has an upper column diameter of 12.0 m

- 210 (Robertson et al., 2014), and the diameter of the spar-buoy utilized in the Hywind Scotland floating wind farm<sup>6,7</sup> is even up to 14.5 m large. However, looking at other floating platform solutions, such as the Damping Pool<sup>®</sup> floater by Ideol<sup>8</sup> with outer dimensions of 36 m x 36 m and a resulting diagonal length of almost 51 m or again the OC4 phase II semi-submersible platform (Robertson et al., 2014) with an overall outer dimension of almost 82 m in diameter, shows that floating structures with a large overall outer diameter can be obtained without being restricted to the manufacturing feasibility limits for pure cylinders.
- 215 Thus, from a hydrodynamic point of view, a cylindrical offshore structure with very large diameter can be realized as well through several smaller diameter cylinders being clustered together in a circle, representing similar hydrodynamic behavior and characteristics. Finally, attention has to be drawn on the minimum possible diameter of the BC parts, which always has to be at least as large as twice the actual wall thickness corresponding to the specific geometric floater configuration. This requirement adds the inequality constraints  $g_{23}$  to  $g_{25}$ , defined as well in Table 2.
- Having modified the diameters and heights of the three BC parts, as well as the ballast filling height, and having adjusted the wall thickness according to the structural mass to displaced mass ratio, as defined in Sect. 2.2, the ballast density has to be adjusted to match the original floating equilibrium between buoyancy force, system weight, and downward mooring force, so that the original hub height is maintained. To exclude unfeasible system solutions, in which material would have to be removed from the system (realized, for example, by reducing the material density) to meet this equilibrium condition, it has to be ensured
- that the actual resulting ballast density carries a positive value, which is reflected through inequality constraint g<sub>21</sub> given in Table 2. However, to account for truly realistic ballast densities, also the uppermost allowable value of the ballast density has to be constrained. Leimeister et al. (2020b) have explored, within a first-stage design optimization application example, densities for common and cheap materials to be used as ballast for a floating spar-buoy. The densest material included is sandstone (or other rocks) with a density of about 2.6E+03 kg/m<sup>3</sup>. Apart from sand, sand mixed with water, concrete, or rocks, MagnaDense (heavyweight concrete) is also used in industry as high-density material<sup>9,10,11</sup>. With MagnaDense, densities of up

MagnaDense (heavyweight concrete) is also used in industry as high-density material  $^{10000}$ . With MagnaDense, densities of up to 5.0E+03 kg/m<sup>3</sup> can be obtained<sup>12</sup> (LKAB Minerals, 2019). Even if minimization of the structure material volume is defined as objective function to represent the structural cost, the cost of the two potential densest ballast materials is elaborated to avoid

<sup>5</sup>https://www.windkraft-journal.de/2019/06/14/steelwind-nordenham-ist-von-wpd-die-gruendungsstrukturen-fuer-den-offshore-wind-park-yunlin-in-

<sup>&</sup>lt;sup>4</sup>https://sif-group.com/en/wind/foundations (Accessed: 13 August 2019)

taiwan-zu-fertigen/136551 (Accessed: 13 August 2019)

<sup>&</sup>lt;sup>6</sup>https://www.equinor.com/content/dam/statoil/documents/newsroom-additional-documents/news-attachments/brochure-hywind-a4.pdf (Accessed: 13 June 2019)

<sup>&</sup>lt;sup>7</sup>https://www.equinor.com/en/news/worlds-first-floating-wind-farm-started-production.html (Accessed: 13 June 2019) <sup>8</sup>https://floatgen.eu/ (Accessed: 13 August 2019)

<sup>&</sup>lt;sup>9</sup>Floating offshore wind project manager at a leading company in offshore industry, personal communication, 6 February 2020.

<sup>&</sup>lt;sup>10</sup>https://www.lkabminerals.com/en/industry-uses/offshore-energy/offshore-wind-structures/ (Accessed: 7 June 2020)

<sup>&</sup>lt;sup>11</sup>https://www.lkabminerals.com/de/floating-offshore-wind-2018/ (Accessed: 7 June 2020)

<sup>&</sup>lt;sup>12</sup>https://www.lkabminerals.com/wp-content/uploads/2019/02/MagnaDense-SDS-12-06INT-19-03.pdf (Accessed: 5 February 2020)

significant larger ballast costs when utilizing MagnaDense instead of the common cheap materials pointed out by Leimeister et al. (2020b). However, when comparing the material prices for sandstone<sup>13</sup> (for the ballast density limit of 2.6E+03 kg/m<sup>3</sup>)

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and MagnaDense<sup>9,14</sup> (for the ballast density limit of  $5.0E+03 \text{ kg/m}^3$ ), it turns out that both ballast materials have a similar cost of around 150 \$ per ton, which is less than 20% of the material cost for structural (raw) steel of about 700 \$ per tonne<sup>15</sup> (Grogan, 2018; Butcher, 2018). Thus, the ballast density is constrained to a maximum of  $5.0E+03 \text{ kg/m}^3$ , corresponding to inequality constraint  $g_{22}$  specified in Table 2, without worrying about any negative impact on the cost situation.

- As particular attention is paid to the global system performance, there are three additional criteria, which the **FOWT** system has to fulfill. For system rotational stability reasons, a maximum total inclination angle of  $10.0^{\circ}$  is allowed in operational conditions (Leimeister et al., 2020b; Katsouris and Marina, 2016; Kolios et al., 2015; Huijs et al., 2013), leading to inequality constraint  $g_{15}$  given in Table 2. Furthermore, due to sensitive components in the nacelle and to prevent any issues with the lubrication, the nacelle acceleration—corresponding to the acceleration at the tower top—is commonly and depending on the specific wind turbine limited to a maximum of 0.2 to 0.3 times the gravitational acceleration constant (Nejad et al., 2017; Huijs et al., 2013; Suzuki et al., 2011); herein the lower value of  $1.962 \text{ m/s}^2$  is used for defining inequality constraint  $g_{16}$ , following
- et al., 2013; Suzuki et al., 2011); herein the lower value of 1.962 m/s<sup>2</sup> is used for defining inequality constraint  $g_{16}$ , following a conservative approach (Leimeister et al., 2020b). Finally, the static translational displacement of a (non TLP-type) FOWT system, corresponding to the mean of the translational motion, is—based on experience—restricted to 0.2 times the water depth (Leimeister et al., 2020b). With a water depth of 320.0 m in the case of the OC3 phase IV spar-buoy floating system, the mean translational motion is limited in this application to 64.0 m, which implies inequality constraint  $g_{17}$ .

### 250 3 Fully modular and automated design optimization

The final automated design optimization of the modified reference spar-type FOWT system described in Sect. 2.2 consists of preprocessing automated system simulations for identifying the simulation conditions to be considered within the optimization (Sect. 3.1), as well as the actual iterative optimization approach for obtaining a conceptual innovative spar-type floating platform design (Sect. 3.3). Both steps utilize a framework for automated simulation and optimization developed at Fraunhofer IWES and presented in Sect. 3.2.

#### **3.1** Preprocessing automated system simulations

Standardization and classification bodies give recommendations on DLCs to be considered when designing FOWT systems. Thus, in the technical specification IEC TS 61400-3-2 (International Electrotechnical Commission, 2019b), based on the international standard IEC 61400-3-1 (International Electrotechnical Commission, 2019a), and in the standard DNVGL-ST-0119
(DNV GL AS, 2018), building on the standard DNVGL-ST-0437 (DNV GL AS, 2016), a substantial number of DLCs is listed which cover different operating states at various environmental conditions. When performing an iterative design optimization approach, however, it is not practical to simulate the full set of DLCs for each design considered. This is not only for reasons

<sup>&</sup>lt;sup>13</sup>https://www.alibaba.com/showroom/sandstone-price-per-ton.html (Accessed: 5 February 2020)

<sup>&</sup>lt;sup>14</sup>https://german.alibaba.com/product-detail/magnadense-heavy-concrete-172429386.html (Accessed: 5 February 2020)

<sup>&</sup>lt;sup>15</sup>https://spendonhome.com/structural-steel-fabrication-cost/ (Accessed: 5 February 2020)

of high computational effort, but also due to the fact that not all DLCs may be relevant or design driving for the specified optimization problem.

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Thus, in this work, the same approach as taken by Leimeister et al. (2020b) in another first-stage design optimization application example is adopted. From IEC 61400-3-1 (International Electrotechnical Commission, 2019a), DLC 1.1 at three wind speeds closely around rated wind speed, as well as DLC 1.3 and DLC 1.6—both at below-rated, rated, and cut-out wind speeds—are selected as they are expected to cover the highest thrust load and corresponding system inclination and mean translational displacement at rated wind speed, as well as maximum dynamic responses in extreme turbulent wind conditions 270 or at severe irregular sea states, and, hence, might be critical for the three global performance constraints  $q_{15}$  to  $q_{17}$  defined in Table 2, which need to be checked and adhered to. By taking six different seeds for turbulent wind and irregular waves for each wind speed considered into account to capture the randomness of the environmental conditions, 54 simulation cases are defined—corresponding to 18 distinct environmental settings per DLC—as summarized in Table 3.

DI C		Win	d conditions		Sea conditions <sup>*</sup>			
DLC	Wind speed	Long. $\mathbf{TI}^{\dagger}$	Wind seed	Yaw misalignment	Sign. wave height	Peak period	Current speed	
	10.0 m/s	18.34%	1 6	$-8^\circ, 0^\circ, 8^\circ$	1.74 m	6.03 s	0.074 m/s	
1.1	11.4 m/s	17.38%	7 12	$-8^\circ, 0^\circ, 8^\circ$	1.99 m	6.44 s	0.084 m/s	
	13.0 m/s	16.53%	13 18	$-8^\circ, 0^\circ, 8^\circ$	2.30 m	6.92 s	0.096 m/s	
	8.0 m/s	35.00%	1 6	$-8^\circ, 0^\circ, 8^\circ$	1.44 m	5.48 s	0.059 m/s	
1.3	11.4 m/s	26.97%	7 12	$-8^\circ, 0^\circ, 8^\circ$	1.99 m	6.44 s	0.084 m/s	
	25.0 m/s	16.68%	13 18	$-8^\circ, 0^\circ, 8^\circ$	4.94 m	10.14 s	0.184 m/s	
	8.0 m/s	20.30%	1 6	$-8^\circ, 0^\circ, 8^\circ$	10.37 m	14.70 s	0.059 m/s	
1.6	11.4 m/s	17.38%	7 12	$-8^\circ, 0^\circ, 8^\circ$	10.37 m	14.70 s	0.084 m/s	
	25.0 m/s	13.64%	13 18	$-8^\circ, 0^\circ, 8^\circ$	10.37 m	14.70 s	0.184 m/s	

Table 3. System parameters for preprocessing simulations of selected DLCs (Leimeister et al., 2020b).

\* Please notice that each realization of the turbulent wind with a different wind seed uses a different wave seed as well.

<sup>†</sup> turbulence intensity

These 54 system simulations have already been performed by Leimeister et al. (2020b) with the original OC3 phase IV floating system and are in this study carried out with the modified reference system from Sect. 2.2. The simulations are executed 275 automatically, utilizing a fully modular framework, which is introduced in Sect. 3.2. From the total simulation time of 800 s, the last 600 s (excluding any transients at the beginning) are evaluated with respect to the system performance criteria. For the modified reference spar-type floating system, the five highest values for the three performance parameters and corresponding DLC simulation cases, as well as the position of the most critical DLC for the original OC3 phase IV FOWT (DLC 1.6 at rated wind speed with wind seed number 11 and yaw misalignment angle of 8°), found by Leimeister et al. (2020b), are presented in 280 Table 4. This shows that the same DLC is still of high criticality for the modified reference spar-type floating system. It scores

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not the highest for the performance criteria; however, the total system inclination angle is almost 96% of the highest value obtained in the 54 DLC simulations, the horizontal nacelle acceleration is even almost 99% of the highest value occurring, and the mean translational motion is just less than 1% lower than the maximum value obtained.

Table 4. The highest values for the three performance parameter	s and corresponding DLC simulation cases	s, based on the modified reference
spar-type <mark>FOWT</mark> system.		

Position	DLC	Wind speed	Wind seed	Yaw misalignment	Max(total inclination angle)
1	1.6	11.4 m/s	8	-8°	3.924°
2	1.6	11.4 m/s	10	$0^{\circ}$	3.876°
3	1.6	11.4 m/s	7	$-8^{\circ}$	3.859°
4	1.6	11.4 m/s	11	$8^{\circ}$	3.761°
5	1.6	11.4 m/s	12	8°	3.632°
Position	DLC	Wind speed	Wind seed	Yaw misalignment	Max(horizontal nacelle acceleration
1	1.6	25.0 m/s	16	$0^{\circ}$	2.339 m/s <sup>2</sup>
2	1.6	25.0 m/s	14	-8°	$2.322 \text{ m/s}^2$
3	1.6	8.0 m/s	5	$8^{\circ}$	$2.313 \text{ m/s}^2$
4	1.6	11.4 m/s	7	-8°	$2.312 \text{ m/s}^2$
5	1.6	11.4 m/s	11	8°	2.311 m/s <sup>2</sup>
Position	DLC	Wind speed	Wind seed	Yaw misalignment	Mean(translational motion)
1	1.6	11.4 m/s	9	$0^{\circ}$	19.533 m
2	1.1	11.4 m/s	9	$0^{\circ}$	19.455 m
3	1.3	11.4 m/s	9	$0^{\circ}$	19.455 m
4	1.6	11.4 m/s	12	$8^{\circ}$	19.430 m
5	1.6	11.4 m/s	8	$-8^{\circ}$	19.351 m
6	1.6	11.4 m/s	11	$8^{\circ}$	19.345 m

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Thus, this DLC 1.6 at 11.4 m/s wind speed with wind seed number 11 and yaw misalignment angle of 8° is used for defining the environmental conditions for the system simulations throughout the iterative optimization approach, which is specified in detail in Sect. 3.3. As, however, it is not ensured that the outcome of the DLC results comparison—based on the reference spar-type FOWT system—does not change for the optimized floater design, the 54 environmental conditions will be simulated subsequent to the design optimization process and the criticality of the DLCs will be assessed again, as covered in Sect. 4.4.

#### Automated optimization framework 290 3.2

The preprocessing DLC simulations (Sect. 3.1), as well as the actual iterative optimization approach (Sect. 3.3), are executed in an automated manner by means of a Python-Modelica framework developed at Fraunhofer IWES (Leimeister et al., 2021; Leimeister, 2019). The structure and components of this framework for automated simulation and optimization are presented in Fig. 3. The framework consists of three modules: a modeling environment, a simulation tool, and a programming framework.

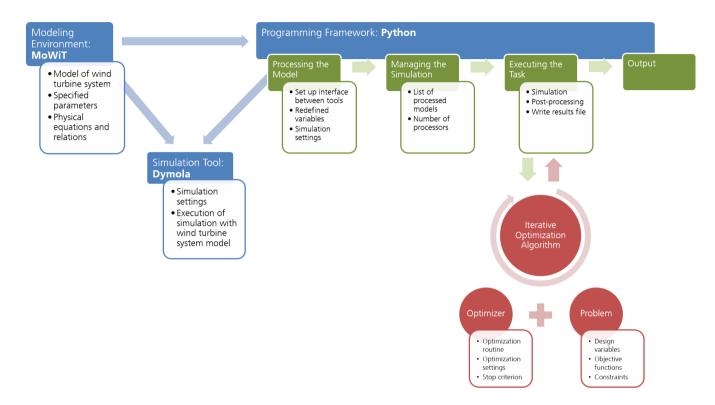


Figure 3. The Python-Modelica framework for automated simulation and optimization, adapted from Leimeister et al. (2021).

295 Having modeled the reference spar-type FOWT system, described in Sects. 2.1 and 2.2, in MoWiT, system and environmental parameters, as well as the underlying physical equations and relations are specified. From the aero-, hydro-, control, and structural dynamic approaches available in MoWiT and covered in detail by Leimeister et al. (2020a), blade-elementmomentum theory including dynamic stall and dynamic wake; linear Airy wave theory, Wheeler stretching, and MacCamy-Fuchs approach; built-in operating control; as well as modal reduced anisotropic beams for blades and rigid bodies for tower and floating structure, are utilized in this application. 300

As simulation tool, Dymola (Dynamic Modeling Laboratory) by Dassault Systèmes<sup>16</sup>, which is capable of time-domain simulations of complex Modelica models, is used.

<sup>&</sup>lt;sup>16</sup>http://www.dymola.com/ (Accessed: 4 February 2020)

The programming framework is coded in Python. The implemented scripts follow—as detailed in Fig. 3—a four-step process for the automated execution of simulations, such as the 54 environmental conditions from Sect. 3.1, and facilitate as well the option to embed an iterative optimization algorithm, as described in Sect. 3.3. More detailed information on the Python-Model-

ica framework, both regarding the theory and structure, as well as its capabilities and some application examples, can be found in the publications by Leimeister et al. (2021); Leimeister (2019).

## 3.3 Specification and execution of the iterative optimization approach

As displayed in Fig. 3, the iterative optimization algorithm (Sect. 3.3.3) coupled to the Python-Modelica framework requires in addition to the model and simulation information also the definition of the optimization problem (Sect. 3.3.1) and specification of the optimizer (Sect. 3.3.2).

#### 3.3.1 Optimization problem

The optimization problem comprises the specification of design variables, objective functions, as well as constraints. This is defined and implemented according to the details provided in Table 1, Eq. 4, and Table 2, respectively.

#### 315 3.3.2 Optimizer

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From the broad range of available algorithms and methods (Leimeister et al., 2021), only gradient-free optimization algorithms can be chosen for the application to fully coupled wind turbine systems modeled by means of MoWiT. In general, for such a complex engineering system, as a FOWT is, evolutionary algorithms are highly suited to find the global optimum of a defined optimization problem (Mishra et al., 2017). Based on a comparative analysis of genetic and evolutionary algorithms from

- 320 Platypus<sup>17</sup>, NSGAII (Non-dominated Sorting Genetic Algorithm II) is found to be the most suitable optimizer for the multi-objective optimization problem in the first-stage design optimization application example on a common floating spar-buoy wind turbine system (Leimeister et al., 2020b). As a genetic algorithm can deal with both formulations of an optimization problem (single-objective and multi-objective) and, hence, also with the optimization problem considered in this study (which holds only one objective function as defined in Eq. 4), and as the system simulations with the iterative optimization algorithm
- 325 based on NSGAII can be also parallelized in a highly efficient manner, it is stuck to the well-performing—both with respect to the convergence speed and the compliance rate concerning the constraints—optimizer NSGAII.

For the genetic algorithm NSGAII, which follows the principle of Darwin's theory of evolution—meaning having individuals which develop further and further each generation towards performing better with respect to the fitness (objective) function—the number of individuals in each generation (the population size), the strategies for representing the evolution, and the stop

330 criterion for terminating the iterative optimization algorithm have to be defined. Due to the complex optimization problem with seven design variables and 25 constraints, the population size is set equal to the maximum possible number of processors, on which simulations can be run simultaneously. On an AMD Ryzen Threadripper 2990WX 32-Core Processor with 64-bit

<sup>&</sup>lt;sup>17</sup>https://platypus.readthedocs.io/en/latest/ (Accessed: 6 April 2020)

system and 64 virtual processors 60 processors could be used for parallel simulations. Hence, 60 individuals are considered in each generation. The individuals are randomly generated. When evaluating the objective function and constraints, the dominant

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individuals—each selected based on a comparison of two individuals—form the basis for the next generation, which is created without using any variator. These are the default generator, selector, and variator settings of NSGAII in Platypus. The stop criterion for terminating the iterative optimization algorithm is defined through the total number of simulations to be performed, while the convergence is checked separately when post-processing the simulation results. As the convergence speed is not known ahead of the execution of the specific optimization problem, the experience from the first-stage design optimization

340 application example (Leimeister et al., 2020b) is used and the total number of simulations is increased to account for the much more complex optimization problem considered in this study. Hence, the resulting number of generations being simulated is roughly tripled, so that a total number of simulations of 10,000 is chosen, corresponding to more than 166 full generations with 60 individuals each.

#### 3.3.3 Optimization algorithm

Having modeled the FOWT system (Sect. 2.2), stated the simulation settings (Table 5), defined the optimization problem 345 (Sect. 3.3.1), and specified the optimizer (Sect. 3.3.2), the iterative optimization algorithm can be executed by means of the Python-Modelica framework for automated simulation and optimization (Sect. 3.2).

Table 5. Simulation settings.

Simulation variable	Value	Note
Simulation interval	from 0 s to 800 s	The first 200 s are accounted for as pre-simulation time to exclude any transients.
Output interval length	0.05 s	
Solver	Rkfix4	(Runge-Kutta fixed-step and 4th order method)
Fixed integrator step-size	0.01 s	

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Within the iterative optimization algorithm, the values of the design variables for the 60 individuals of the first generation (number 0) are selected by the optimizer based on the specified allowable value ranges. All individuals are simulated in parallel on the available 60 processors and analyzed afterwards by the optimizer with respect to their fitness—meaning the objective function—and their compliance with the constraints based on the resulting time series, evaluated between 200 s and 800 s. As also simulations may have failed (due to too bad performance of instable floating system designs which demonstrate a negative metacentric height), the simulated time is checked against the specified simulation stop time (800 s according to Table 5). In case of an unsuccessful simulation and, hence, incomplete time series, the parameters of interest addressed in the 355 constraints  $g_{15}$  to  $g_{17}$  for the system performance are not taken by evaluating the time series but are set equal to twice the maximum allowable value, as given in Eqs. 5 to 7. This way, it can be ensured that unsuccessful simulations do not comply with all constraints and, hence, are undesirable design solutions, which the optimizer then discards from further selection of well-performing individuals.

 $\max(\text{total inclination angle})|_{\text{failing system}} = 2 \cdot 10.0^{\circ} = 20.0^{\circ}$ 

$$\Rightarrow g_{15}(system(X)|_{\text{failed}}) = 20.0^{\circ} - 10.0^{\circ} = 10.0^{\circ} \nleq 0$$

(5)

(6)

(7)

360

$$\max(\text{horizontal nacelle acceleration})|_{\text{failing system}} = 2 \cdot 1.962 \text{ m/s}^2 = 3.924 \text{ m/s}^2$$

$$\Rightarrow g_{16}(system(X)|_{\text{failed}}) = 3.924 \text{ m/s}^2 - 1.962 \text{ m/s}^2 = 1.962 \text{ m/s}^2 \nleq 0$$

mean(translational motion) $|_{\text{failing system}} = 2 \cdot 64.0 \text{ m} = 128 \text{ m}$ 

$$\Rightarrow g_{17}(system(X)|_{\text{failed}}) = 128 \text{ m} - 64.0 \text{ m} = 64.0 \text{ m} \nleq 0$$

Having evaluated the simulated individuals of generation 0, the optimizer selects the design variables for the individuals of the next generation (number 1), again in accordance with the specified allowable value ranges, but also based on the fitness and constraints compliance rate of each of the previous individuals, using the tournament selector for evaluating the dominance.
365 Then, the loop of simulating individuals, evaluating each system with respect to the objective function and constraints, and reselecting values for the design variables of the individuals of the next generation is repeated as long as the number of executed simulations is still below the specified total number of simulations of 10,000. This iterative optimization algorithm ends when the stop criterion is reached—the final results are now available.

#### 4 Results

370 **T**he optimization run takes effectively about 31 days and eleven hours and comprises 10,011 individuals simulated in total, ranging from generation 0 up to generation 166, with full populations up to and including generation 165.

#### 4.1 Developments throughout the iterative optimization process

Figure 4 shows in light blue for all simulated individuals the values for the design variables  $x_1$  to  $x_7$ , as defined in Table 1. The original values belonging to the reference spar-type FOWT system, covered in Sect. 2.2, are plotted additionally as red lines

- 375 for comparative purposes. Post-processing of the simulation results and checking the constraints yield the dark blue recolored individuals which comply with all specified constraints. The finally selected optimum, which is presented in Sect. 4.3, is marked with a yellow filled circle framed in orange. The developments of the design variables throughout the iterative optimization process show that in the first generations, the optimizer selects individuals covering the entire design space; however, none of the first is meeting all requirements. With more generations, the compliance rate is significantly increased. Overall, the spread
- in the design variables is decreased for more generations being simulated and for some design variables the change in their values is even very limited for the individuals which comply with all constraints. This indicates that the optimization algorithm is converging, though it has not yet fully converged, which is underlined by the fact that the optimum originates from the last generation.

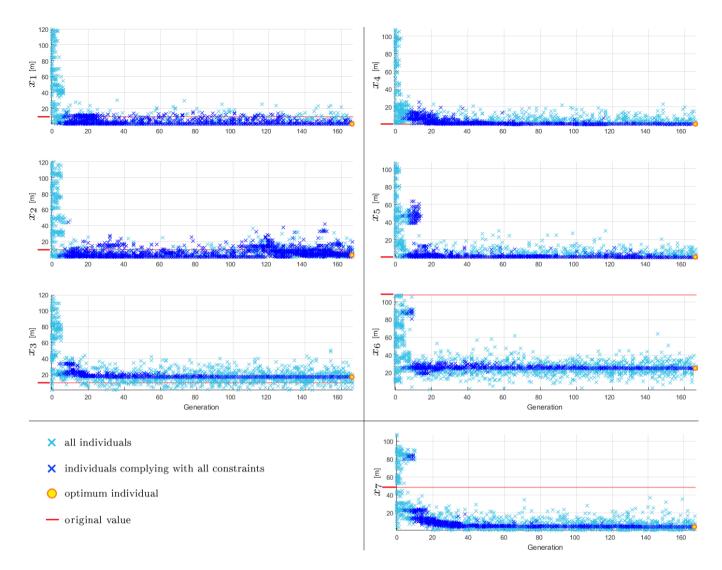


Figure 4. Developments of the design variables throughout the iterative optimization process.

Similarly, the developments of the constraints throughout the iterative optimization process are analyzed. g1 to g14 are taken
into account ahead of the simulations when the optimizer selects the design variables for the new individuals and, hence, are never violated. This can clearly be seen in Fig. 4, where all individuals lie within the allowable value ranges of the design variables. The developments of the remaining constraints g15 to g25 are presented in Fig. 5. The light cyan crosses indicate the results for all simulated individuals, while the individuals which simultaneously comply with all constraints are recolored in dark bluish green. The limits of the inequality constraints, which should all be less or equal to zero, are indicated in red. The finally selected optimum is marked again with a yellow filled circle framed in orange. For g21 and g22 it has to be noted that the ordinate is limited to [-1E+4, 1E+4] for reasons of clarity, as a few more individuals yield values in the order of magnitude

of six. For  $g_{18}$  to  $g_{20}$  and  $g_{23}$  to  $g_{25}$ , which are directly related to and dependent on the design variables, the developments of the constraints and the corresponding design variables are similar. For the other constraints, the trend is rather different, having a large spread in the results throughout the simulated generations. The fact that for  $g_{15}$  to  $g_{17}$  only a few distinguishable individuals are plotted in the first generations is caused by the large number of unstable design solutions that are selected by the optimizer in the first trials. Due to the unsuccessful simulations, the performance variables are set to undesired values, as explained in Sect. 3.3.3, and, hence, are all the same for all failing systems. This is as well visible throughout the generations, as there is a line at the specified undesired value formed by the individuals that do not complete the simulations successfully.

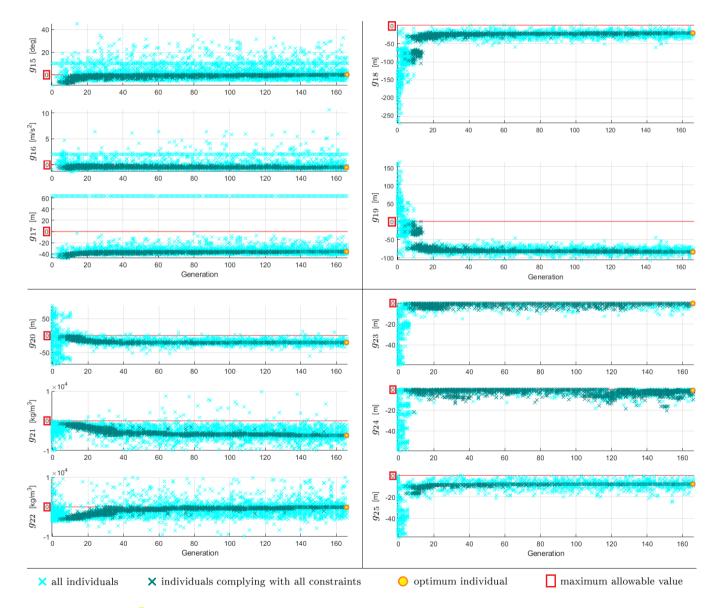


Figure 5. Developments of the constraints throughout the iterative optimization process.

#### 4.2 Innovative floater geometries in the design space

400 As presented and mentioned in Sect. 4.1, the individuals of the first generations cover the entire design space, while the individuals that comply with all constraints are in a much narrower area. The geometric design variables  $x_1$  to  $x_6$  of these individuals, setting height and diameter of each BC part in correlation, are plotted in light blue unfilled circles in Fig. 6. The original and optimum designs are highlighted by red and blue, respectively, filled circles. From these individuals, which comply with all constraints, seven examples are selected to demonstrate the diversity of potential (meaning successful but maybe not

405 yet optimum) innovative floater geometries. These examples are schematically drawn with black lines in Fig. 6 together with the original shape in red and having represented the ballast heights in dashed lines. The corresponding figures for design variables, performance parameters, objective function, and further resulting geometrical and structural parameters are outlined in Table 6. These numbers also underline that—when evaluating  $g_1$  to  $g_{25}$ —none of the inequality constraints is violated.

Table 6. Key figures of the exemplary potential innovative floater geometries.

Ex.	Gen.	Ind.	x <sub>1</sub> [m]	x <sub>2</sub> [m]	x <sub>3</sub> [m]	x <sub>4</sub> [m]	x <sub>5</sub> [m]	x <sub>6</sub> [m]	x <sub>7</sub> [ <b>m</b> ]	Ballast density [kg/m <sup>3</sup> ]	Wall thickness [m]	Draft [m]
1	115	45	0.116	13.410	16.612	6.930	0.002	25.903	4.573	4.585E+03	0.0578	44.836
2	14	15	8.899	1.528	31.100	5.551	1.183	19.518	17.774	1.003E+03	0.1052	38.252
3	78	32	15.253	0.164	16.612	0.018	1.109	25.033	10.709	2.156E+03	0.0580	38.160
4	8	6	14.755	0.172	20.090	6.970	4.665	91.993	84.016	1.037E+03	0.0797	115.628
5	9	45	10.550	43.919	33.605	13.896	1.798	89.776	84.684	1.008E+03	0.1344	117.470
6	10	8	5.158	2.331	34.015	6.997	46.270	25.683	22.727	1.022E+03	0.1135	90.950
7	9	57	0.523	2.331	33.154	6.159	62.944	25.683	22.727	1.013E+03	0.1106	106.786

Ex.	Max(tot. inclination angle) [°]	Max(hor. nacelle acceleration) [m/s <sup>2</sup> ]	Mean(transl. motion) [m]	$f_1$ [ <b>m</b> <sup>3</sup> ]	Steel mass [kg]	Ballast mass [kg]
1	9.9	1.337	28.155	99.1	7.778E+05	4.544E+06
2	5.0	1.231	22.241	266.2	2.090E+06	1.355E+07
3	9.3	1.724	27.308	107.7	8.455E+05	5.004E+06
4	2.6	1.955	17.503	530.1	4.162E+06	2.761E+07
5	1.6	1.664	21.089	1428.6	1.121E+07	7.570E+07
6	3.9	1.447	21.109	407.9	3.202E+06	2.111E+07
7	4.6	1.159	22.138	384.8	3.021E+06	1.987E+07

Looking at the floater geometries presented in Fig. 6, it becomes clear that not all of these shapes can be realized with conventional manufacturing solutions, where cylindrical sections are welded together. It has to be emphasized that these results 410 are solely based on the hydrodynamic and system-level analyses, as specified within the optimization problem, as well as on

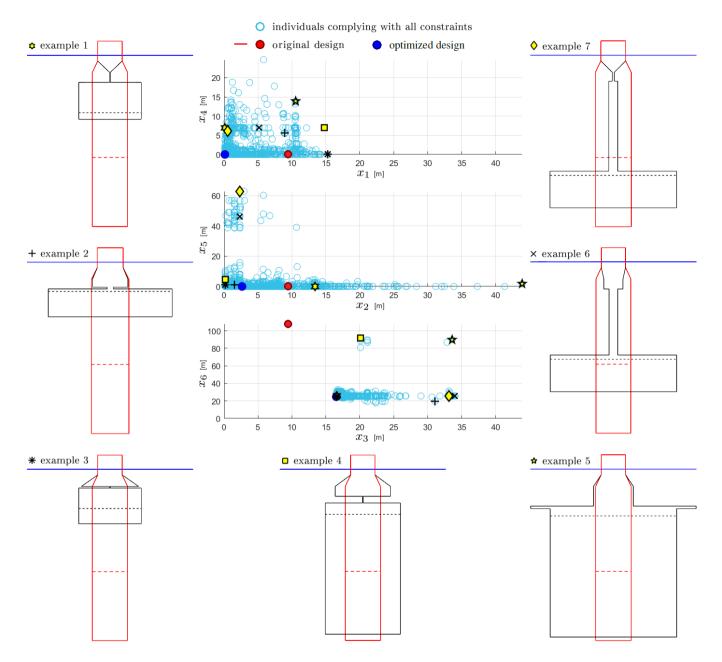


Figure 6. Exemplary potential innovative floater geometries selected from the individuals complying with all constraints.

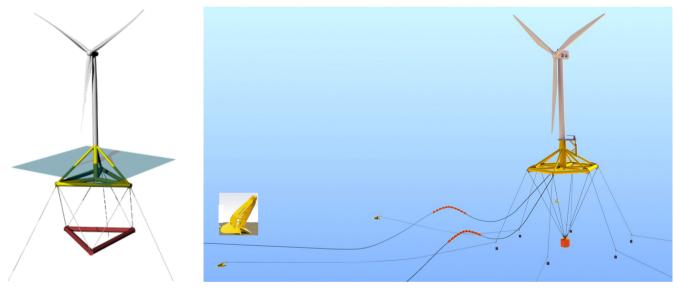
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the advancements taken into account in Sect. 2.3 which clearly intend the utilization of alternative and innovative structural realization approaches. Other additional types of analyses—addressing structural integrity, manufacturability, and localized design—can, hence, deem some of the presented potential design solutions unfeasible, which is discussed in some more detail in Sect. 5. However, the advantage of this methodology—by focusing only on the global system performance—is that a new

range of potential floater designs is opened up and shapes like these presented in Fig. 6 can still be considered as feasible solutions when different structural realization approaches are applied. These approaches can range from truss structures to tendons to realize large diameter changes, as well as very thin "distance" elements, without utilizing tapered sections or having issues with the structural integrity. Idea and impulse provider for such alternative structural realization approaches can be,

420 for example, the oil and gas industry with truss spar platform design solutions (Fig. 1(a)) (Chen et al., 2017; Perry et al., 2007; Bangs et al., 2002) or innovative floating platform concepts like the TetraSpar by Stiesdal Offshore Technologies A/S (Fig. 7(a)) (Borg et al., 2020; Stiesdal, 2019) or the pendulum-stabilized Hexafloat floater (Fig. 7(b)) by Saipem, realized in the AFLOWT project (Ribuot, 2019; Richard, 2019). Similarly to these two innovative pendulum-stabilized floating platform concepts, example 6 in Fig. 6 would suggest to implement a configuration with a separate ballast body, connected to the main

425 body through cables or tendons. Therefore, this kind of structure cannot be realized with conventional spar manufacturing, but the approach adopted in this paper can shed the light on such innovative shapes, that may require alternative structural strategies.



(a) Stiesdal's TetraSpar (Borg et al., 2020).

(b) Saipem's Hexafloat (Ribuot, 2019).

Figure 7. Pendulum-stabilized innovative floating platform concepts.

#### 4.3 The optimized conceptual floater design

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Due to the single-objective nature of the optimization problem, the selection of the optimum solution happens directly through evaluating the one and only objective function. This means that from all individuals that comply with all constraints, this is chosen as optimum which exhibits the lowest value for the structure material volume of its floater design solution. First, looking at the development of the objective function  $f_1$  throughout the iterative optimization process, as presented in Fig. 8, the trend of all simulated individuals (plotted in light green) shows a significant minimization of the objective function clearly below the original value of 136.3 m<sup>3</sup>, indicated in Fig. 8 by a red line—after a large spread in the first generations.

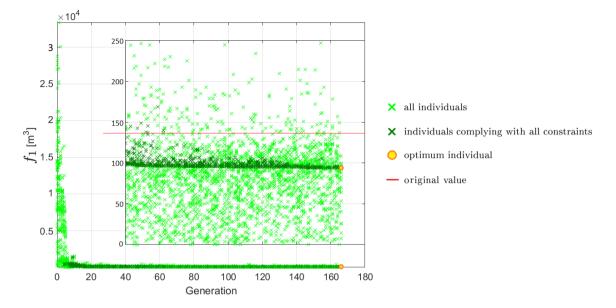


Figure 8. Development of the objective function throughout the iterative optimization process.

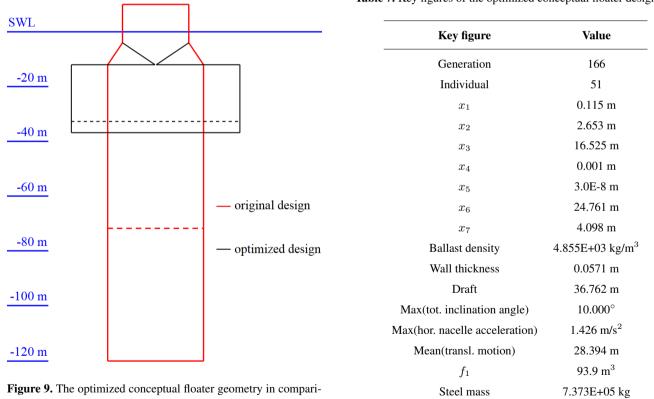
Zooming into the objective function results from generation 40 on, as included in Fig. 8, provides a much clearer indication of the development of the structure material volume for the individuals which comply with all constraints (recolored in dark green): they aggregate to an asymptote. This is already visible in early generations; however, the spread in the objective function results is decreasing with more generations being simulated. This asymptotic clustering of the individuals which comply with all constraints to a minimum objective function value, on the one hand, states the convergence of the iterative optimization process and, on the other hand, portends that there will be several—more or less similar (elaborated in the following)—design solutions, which yield comparable low structure material volumes that are all very close to the minimum value observed.

The individual with the minimum structure material volume is pointed out in Fig. 8 by means of a yellow filled circle framed in orange. This design solution yields a reduction of the structure material volume of more than 31% compared to the original (modified) reference spar-type floating platform, for which it must be noted that it neither has been designed

- 445 with the same design requirements, nor has it yet been optimized. The fact that this optimum solution is just found in the last generation simulated states that full convergence is not yet reached, despite the converging trend in most of the design variables and constraints, as well as in the objective function. Nevertheless, due to the asymptotic aggregation of the individuals mentioned above, the first ten minimum objective function results from the individuals which comply with all constraints are evaluated. This results—as some individuals yield the same objective function value—into 16 individuals with a just by 2.84E-
- 450 4% increased structure material volume, comparing the tenth lowest with the minimum value, and shapes that are difficult to

distinguish from each other. This proves the above mentioned anticipation and leads to the conclusion that—if, in addition to the maximum number of simulations, a reasonable convergence tolerance had been specified as supplementary stop criterion— the optimization algorithm would not have required all 10,000 simulations and would have stopped much earlier. However, due to the strong similarity of the last design solutions, no significant differences in the results would have been perceived.

- 455 The geometry of the optimized conceptual floater shape (black line) is shown schematically in Fig. 9 in comparison to the original floating platform drawn in red, while the key figures of the optimized floater geometry are presented in Table 7. The global system performance points out that the maximum total inclination angle is the most critical criterion, as the value obtained from the optimized design is equal to the specified upper limit of 10°. With respect to the design development within the optimization process, both Fig. 9 and Table 7 indicate the following advancements: To reduce structure material volume, the overall length of the floating platform is significantly decreased (the draft of the optimized floater, however, is still some way
  - from the minimum allowable draft of 15 m), the width of the bottom part of the support structure is enlarged, while the upper and middle parts are almost left out (leading to this significant constriction in the tapered part) and a very low ballast volume is obtained through a significantly increased ballast density, utilizing MagnaDense or high-density concrete as ballast material.



son to the original shape.

**Table 7.** Key figures of the optimized conceptual floater design.

Ballast mass

4.267E+06 kg

Overall, the shape of the optimized conceptual floater design resembles rather a submerged thick barge-type floater, hanging

- 465 below the upper column element. The constriction in the tapered part is significant and would not directly be technically feasible, both from a manufacturing point of view and with respect to structural integrity. The reason for the current shape obtained is the connection of the upper column to the upper BC part, which, however, is, as well as the middle BC part, negligible. Thus, for this floater configuration, the tapered part could directly connect the end of the upper column with the top of the lower BC part. The change in required structure material would be not that significant; however, the related change
- 470 in the displaced water volume has to be taken into account by adjusting the structure mass and by carefully evaluating the system performance due to the shifted center of buoyancy. This realization by means of a tapered section, however, comes with a large diameter change and corresponding large taper angle, which may be critical for both hydrodynamic load calculations and manufacturing, as discussed in more detail in Sect. 5. However, the structural issues due to the geometrical configuration of the optimized floater as presented in Fig. 9, or as well due to the large diameter change when utilizing a tapered section,
- become void when, instead, connecting the upper column and lower BC part by means of a number of rigid slender braces or some tendons, in combination with plated partial bulkheads for load transfer. These manufacturing solutions go beyond the conventional structural realization approach of welding cylindrical sections together, but they make the found optimized floater design solution feasible and are expected to represent similar system performance. The fitness of the floater solution proposed by the optimizer is underlined due to its similarity (with respect to the innovative structural realization approach) to the most novel and alternative solutions suggested by the research community, such as the Stiesdal's TetraSpar (Fig. 7(a)) (Borg et al.,
- 2020; Stiesdal, 2019) or the Hexafloat by Saipem (Fig. 7(b)) (Ribuot, 2019; Richard, 2019).

#### 4.4 Performance of the optimized system in different environmental conditions

With the conceptual design solution for the innovative **FOWT** platform obtained from the optimization run, finally, the DLCs that are selected for the preprocessing automated system simulations for choosing the most critical DLC (as presented in Sect. 3.1) are rerun to check whether a shift in the most critical DLC happens. The criticality is again assessed by evaluating the fully coupled system performance criteria and analyzing the corresponding constraints  $g_{15}$  to  $g_{17}$ . The highest values and corresponding DLC simulation cases, as well as the values obtained with the selected critical DLC 1.6 at rated wind speed with wind seed number 11 and yaw misalignment angle of 8°, are presented in Table 8.

- For the optimized design solution, a shift in the criticality of the DLCs is observed. The smallest change in the order of criticality of the 54 environmental conditions happens in the horizontal nacelle acceleration. Still, the cases from DLC 1.6 at cut-out wind speed, as well as around rated wind speed, are most critical, but the DLC used within the iterative optimization algorithm is still among the first ten with an acceleration value that is almost 12% lower compared to the maximum obtained from all simulated DLCs. This, however, is itself still more than 17% below the maximum allowable horizontal nacelle acceleration and, hence, uncritical, which—on a side note—is not the case for the original floating spar-buoy wind turbine system. A
- 495 significant increase in the resulting performance values and considerable change in the order of criticality of the environmental conditions is obtained for the mean translational motion. Here, the selected DLC for the optimization process drops down from the originally sixth position to the 22nd, while it is just 10% below the highest value achieved, which is still less than half of

**Table 8.** The highest values for the three performance parameters and corresponding DLC simulation cases, based on the optimized floating system.

Position	DLC	Wind speed	Wind seed	Yaw misalignment	Max(total inclination angle)
1	1.1	13.0 m/s	18	8°	12.061°
2	1.1	11.4 m/s	10	$0^{\circ}$	12.011°
3	1.3	11.4 m/s	10	$0^{\circ}$	12.011°
4	1.1	11.4 m/s	7	-8°	11.903°
5	1.3	11.4 m/s	7	-8°	11.903°
30	1.6	11.4 m/s	11	8°	10.000°
Position	DLC	Wind speed	Wind seed	Yaw misalignment	Max(horizontal nacelle acceleration
1	1.6	25.0 m/s	17	8°	$1.620 \text{ m/s}^2$
2	1.6	25.0 m/s	18	$8^{\circ}$	$1.618 \text{ m/s}^2$
3	1.6	25.0 m/s	13	-8°	$1.550 \text{ m/s}^2$
4	1.6	25.0 m/s	16	$0^{\circ}$	$1.521 \text{ m/s}^2$
5	1.6	25.0 m/s	15	$0^{\circ}$	$1.480 \text{ m/s}^2$
10	1.6	11.4 m/s	11	8°	1.426 m/s <sup>2</sup>
Position	DLC	Wind speed	Wind seed	Yaw misalignment	Mean(translational motion)
1	1.1	13.0 m/s	15	0°	31.564 m
2	1.1	11.4 m/s	9	$0^{\circ}$	31.375 m
3	1.3	11.4 m/s	9	$0^{\circ}$	31.375 m
4	1.1	13.0 m/s	17	$8^{\circ}$	30.631 m
5	1.1	11.4 m/s	12	$8^{\circ}$	30.337 m
22	1.6	11.4 m/s	11	8°	28.394 m

the maximum allowable value and, hence, again uncritical. However, the most sever shift in the criticality of the DLCs happens for the total inclination angle of the system. As indicated in Sect. 4.3, the maximum allowable value is already reached in the
environmental condition considered for the optimization approach. This DLC, however, is for the optimized design solution no longer prevailing but just on the 30th position, meaning that 29 other environmental conditions (mostly from DLC 1.1 and DLC 1.3, as well as some others from DLC 1.6) exceed the specified upper limit by up to more than 20%. In these environmental conditions, the floater design obtained from the optimization run would have to stop operation, while the overall system stability is not expected to be critical, as commonly much higher values for a parked FOWT system in extreme environmental conditions are acceptable, such as 15° considered by Hegseth et al. (2020). However, to avoid reduced system availability, the

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occurring changed criticality of the DLCs has to be addressed already during the optimization—by, for example, considering safety factors for such critical and design-driving performance criteria. Alternatively or additionally, the performance in all environmental conditions can be further improved by subsequent optimization of the currently unaltered mooring system. These options are discussed in more detail in Sect. 5.

#### 510 5 Discussion

In addition to the results presented, analyzed, and discussed in Sect. 4, more details on these results are addressed in the following and further aspects are discussed.

First of all, the duration of the optimization simulations needs to be dealt with. If an additional stop criterion based on a realistic convergence tolerance had been specified, not the full 10,000 simulations would have to be simulated as the conver-515 gence tolerance would have been reached maybe already after around 40 generations. Thus, the conceptual design study would have required maybe just less than a quarter of the actually spent time. However, even around 181 hours—which is more than a week—is still too long for just a conceptual design study. The reason behind the currently quite long time required does not lie in the multi-fidelity framework and fully modular optimization problem setup, but rather in the developmental stage of the numerical model for a FOWT system. Thus, a 800 s load case simulation with a FOWT in irregular sea state and with turbulent

- 520 wind conditions takes about four and a half hours, which is about 20 times as much as the time to be simulated. This is a known issue and part of the current development work at Fraunhofer IWES. While for bottom-fixed wind turbine systems real-time capability of the numerical models based on MoWiT has already been achieved (Feja and Huhn, 2019), the optimization of the code for floating systems is still at an early stage of development. However, based on the experience with the bottom-fixed numerical wind turbine system models, it is expected to achieve as well real-time capability for floating numerical wind turbine
- 525 system models based on MoWiT after some further advancements of the code. At that time, the full simulation of the specified optimization problem will just require about one and a half days. This would then be very promising both for conceptual design studies and detailed design optimization tasks, which—due to the fully modular and multi-fidelity approach applied—can be realized with the same numerical modeling and optimization environment.
- Based on the findings of the DLC simulations with the optimized conceptual FOWT system design, it is recommended to take
  some safety factors for the maximum allowable performance values into account. If the horizontal nacelle acceleration had been exceeded in some of the 54 environmental conditions, it would not have been that critical, as a maximum allowable value of up to 0.3 times the gravitational acceleration constant—and not only 0.2 times as applied—is often accepted, as already mentioned in Sect. 2.3. The specific maximum allowable values for an operating FOWT system have to be provided by the turbine manufacturer or operator. Thus, maybe a higher inclination angle is still acceptable; however, if 10° are really the uppermost total inclination angle can as well afterwards be applied in the post-processing of the results and, this way, a floater design performing well in all 54 environmental conditions can be obtained. The downside of this approach, however, is that a larger structure material volume would be required and that this design would not represent an optimized solution. A profitable option,

hence, is to adjust the—currently excluded and unchanged—mooring system properties and layout design. By modifying these

- 540 in a subsequent optimization task, the optimized floater design can be retained and at the same time the performance of the FOWT system in all considered environmental conditions improved—in this case especially the system inclination. Apart from the considered 54 environmental conditions, however, the optimized FOWT system design has to prove to withstand any potential environmental and operational condition during its design life. Thus, for a subsequent more realistic and detailed design analysis, the entire set of DLCs recommended by standards has to be considered—at least in the pre-selection and final
- 545 reassessment of the selected critical load case. The full set of DLCs should consider more realistic environmental conditions by accounting for various natural periods per considered sea state, utilize longer simulation times to capture the low frequency dynamics of the FOWT system, and include also load cases with occurrence of a fault—such as grid loss—or with other transient loads—due to, for example, gusts—which might cause high accelerations and extreme loads.

Considering the wide design space—especially the broad allowable value ranges for the structural diameters—and the ex-550 treme environmental conditions included in the DLC simulations, some refinements in the model with respect to the hydrodynamic calculations are suggested.

- For an accurate representation of the hydrodynamic loads on the floating structure, the hydrodynamic coefficients have to be recalculated for each specific diameter. While the horizontal added mass coefficient, as well as the total inertia force, are already determined in dependency of the actual structural diameter and wave number, as the MacCamy-Fuchs approach is applied for each column element separately, the horizontal drag coefficient is currently not altered from the original value of 0.6. This is a valid assumption for large diameters already at low flow velocities; however, for small diameter structures, which can occur within the optimization algorithm, an around twice as large horizontal drag coefficient might be applicable (Clauss et al., 1992). In the vertical (heave) direction, both added mass and drag coefficients are currently unchanged, while a vertical Froude-Krylov excitation force is considered, accounting for the difference between UC diameter and the diameter at the floater base. Especially for geometries with large diameter changes, as well as with large diameters, which can be regarded as heave plates, the hydrodynamic coefficients will differ from the original values for a continuous cylinder as the OC3 phase IV spar-buoy. Furthermore, the vertical Froude-Krylov excitation force would have to be adjusted to the specific geometry, when the lower BC part is connected by means of trusses or tendons to the upper column, to account for the differences between each upper and lower surfaces. Both changes in the hydrodynamic coefficients in heave direction and adjusted vertical Froude-Krylov excitation force—will mainly affect the heave motion of the floating system, as well as the roll and pitch motions in some respect. With the geometry obtained from the optimization, however, it is expected to experience less strong system responses if the hydrodynamic coefficients are adjusted accordingly—which would benefit, for example, the system inclination—while the system responses will increase slightly if the vertical Froude-Krylov excitation force is determined accurately for the considered geometry.
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- For more extreme environmental conditions with extreme waves and structures similar to those obtained with the optimization run that tend to have a large diameter directly at or close to the top of the BC, the event that the upper surface 575

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of such a large diameter cylinder becomes dry has to be accounted for when calculating the added mass and damping coefficients in order to not overestimate the heave and pitch added mass and, thus, to not underestimate the horizontal nacelle acceleration in case of more energetic sea states. Furthermore, having a horizontal surface close to the water surface—in the presented settings with a minimum distance of 12 m—could be as well critical structurally or maybe due to the impossibility of common service vessels to approach the wind turbine. However, it has to be noted that it is aimed to establish a conceptual innovative floating platform design optimized with respect to the hydrodynamics. This, then, needs to be compromised imposing other prevailing constraints, such as structural limits—as discussed later in more detail again—or accessibility, for which, for example, walk-to-work solutions with a gangway can be exploited.

- The applied MacCamy-Fuchs approach is in principle just valid for cylinders with vertical walls and not for cylinders with abrupt diameter changes, leading to conical sections or even large horizontal surfaces anywhere along the column (the latter one, however, is considered again by means of the vertical Froude-Krylov excitation force, as discussed previously). If the MacCamy-Fuchs approach is applied to conical structures, the wave load from especially waves with low periods will be underestimated. This could be in the order of magnitude of up to 8% or 14% for a cone angle of around  $6.7^{\circ}$  or  $12.2^{\circ}$ , respectively, and could affect wave periods of 3 s to 6 s or 3.5 s to 7 s, according to investigations on a tapered bottom-fixed offshore wind turbine support structure (Leimeister et al., 2019). Thus, this potential underestimation of the hydrodynamic loading is mostly relevant for the environmental conditions of DLC 1.1, as well as for the below and at rated wind speed cases of DLC 1.3. For the design solution proposed in Sect. 4.3, in which the bottom end of the upper column is directly connected with the large diameter lower BC part, the taper angle would amount 32°. Any hydrodynamic calculations based on the MacCamy-Fuchs approach would no longer be meaningful if the design solution is realized by means of a solid tapered part. Thus, the alternative suggestion of having instead a number of rigid slender braces would be favored. In order to ensure valid computation of the hydrodynamics already within the optimization approach, another constraint on the maximum taper angle shall be added, as implemented with a limit of  $10^{\circ}$  by Hegseth et al. (2020). This aspect is, however, less critical when allowing for different structural solutions—as done in this conceptual design study—where trusses or tendons prevent any utilization of strongly tapered sections.

Expectedly and as addressed and discussed in Sects. 4.2 and 4.3, the geometrical configurations of the potential (Fig. 6) and optimized (Fig. 9) innovative floater designs may not be technically feasible from a structural integrity and manufacturability point of view, adopting the standard manufacturing solutions. For obtaining a high detail structural design, further localized analyses and assessments regarding the manufacturability have to be performed subsequently. However, structural integrity checks for buckling or stress concentration and for accounting for a realistic and adjustable base and lid thickness, which is currently just set to a fixed marginal value, can—due to the multi-fidelity character of the problem setup and framework—as well directly be integrated in the definition of the optimization problem for the subsequent detailed design study. Nonetheless, based on the assumptions and focus of this study, which is on a conceptual design development, as well as hydrodynamic and system-level analyses, a significantly improved and more cost-efficient floater with a highly innovative shape can be achieved.

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This is as well feasible when considering different structural realization approaches, such as braces and truss structures or tendons, as already used in the oil and gas industry (Fig. 1(a)) (Chen et al., 2017; Perry et al., 2007; Bangs et al., 2002) or utilized in innovative floater concepts (Fig. 7) (Richard, 2019; Stiesdal, 2019), instead of following purely the conventional structural approach of welding cylindrical and tapered sections together.

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Finally and admittedly, for really considering an optimization of the wind turbine system cost, the ratio of CapEx (Capital Expenditure) to AEP (Annual Energy Production) or even the LCoE, which additionally takes OpEx (Operational Expenditure) and sometimes also costs of decommissioning—into account, would have to be considered to be minimized. This way, a real trade-off between saved material costs, changed expenditure of manufacturing and maintenance of the system, and different system performance, and, hence, affected AEP can be found. However, this requires a more holistic and complex approach,

615 system performance, and, hence, affected AEP can be found. However, this requires a more holistic and complex approach, considering annual environmental distributions at the location of interest, calculations for the full life-time of the system, as well as knowledge of possible manufacturing processes and related costs. The fully modular and multi-fidelity numerical modeling and optimization environment facilitates further extension of the present work to take into account these steps and aspects in the future.

#### 620 6 Conclusions

In this paper, an automated optimization approach is applied to a spar-type **FOWT** system to develop a conceptual innovative floating platform design, which is optimized with respect to the change in hydrodynamics and their impact on the main system performance, while structural, manufacturability, or other constraints are not considered, whereas other advancements are facilitated. This approach, following a freer optimization formulation, is taken in order to be able to explore novel design spaces which can be better from a hydrodynamic point of view and show potential for more cost-efficient design solutions, but may require novel structural approaches, as actively investigated by the community (e.g. Stiesdal's TetraSpar and Saipem's Hexafloat). The application is based on the OC3 phase IV reference spar-buoy FOWT system. This, however, is modified by dividing the spar-buoy base column into three distinct partitions, so that sufficient buoyancy, as well as a deep center of gravity can be obtained. Furthermore, the wall thickness is adjusted based on a common ratio of the support structure's struc-

- 630 tural mass to the displaced mass of water. The optimization focuses on the minimization of the steel volume of the floater, which represents an approximation of the CapEx of the floating platform. In addition, constraints regarding the outer dimensions (meaning the allowable value ranges of the design variables), the global fully coupled system performance, the system draft, the ballast, and the geometric integrity are defined, whereby advanced features—such as alternative ballast materials or novel structural approaches—are incorporated in the definition of the value ranges of the design variables and ballast density.
- 635 Having selected, based on preprocessing automated system simulations, one DLC which is most critical for the constrained system performance criteria, the iterative optimization algorithm run is performed, utilizing the Python-Modelica framework for automated simulation and optimization, as well as using the genetic algorithm NSGAII as optimizer. The analysis of the optimization simulation results shows that the individuals which comply with all prescribed constraints aggregate as for their objective function values to an asymptote. The applied iterative optimization algorithm presented in this study yields a con-

- 640 ceptual floating support structure design, which has a by more than 31% reduced structure material volume compared to the original floating platform, meets all global performance criteria for the considered critical DLC, has an overall draft of 36.8 m, utilizes MagnaDense or high-density concrete as ballast material, and resembles a submerged thick barge-type floater. Based on the applied hydrodynamic and system-level analyses an optimized initial innovative floater design is obtained, which has to be further refined by incorporating structural checks into the optimization process, but can be realized by means of alternative
- 645 structural approaches, which utilize trusses or tendons instead of solely welding cylindrical sections together. Due to the fully modular and multi-fidelity character of the numerical modeling and optimization environment, the same framework can be utilized for a detailed design development and optimization, which is based on manufacturers' feedback on potential structural realization approaches, includes structural checks, follows a more holistic approach by allowing also the modification of other components (such as the mooring system), and potentially utilizes more detailed hydrodynamic calculations as well.

### 650 Appendix A: Main properties of the OC3 phase IV floating offshore wind turbine system

The OC3 phase IV FOWT system consists of the NREL (National Renewable Energy Laboratory) 5 MW reference wind turbine (Jonkman et al., 2009), an offshore adapted tower, the spar-buoy floater, and three evenly spaced catenary mooring lines (Jonkman, 2010). The main properties of the FOWT system, which is designed for a water depth of 320.0 m, are summarized in Table A1. This OC3 phase IV spar-buoy FOWT system was defined as a reference design for code-to-code verifications and code-to-experiments validation and, hence, was not necessarily vet optimized.

**Table A1.** Main properties of the OC3 phase IV FOWT system (Jonkman, 2010; Jonkman et al., 2009).

Wind	turbine	Offshore adapted tower		
Rated power	5 MW	Top elevation above $SWL^{\dagger}$	87.6 m	
Rotor diameter	126.0 m	Bottom elevation above $SWL^{\dagger}$	10.0 m	
Hub height	90.0 m	Top diameter	3.87 m	
Cut-in wind speed	3.0 m/s	Bottom diameter	6.5 m	
Rated wind speed	11.4 m/s	Top wall thickness	0.019 m	
Cut-out wind speed	25.0 m/s	Bottom wall thickness	0.027 m	
RNA <sup>*</sup> mass	350,000 kg	Structural mass	249,718 kg	

Spar	-buoy floater	Mooring lines		
Elevation range of upper column	4.0 m below SWL $^{\dagger}$ to 10.0 m above SWL $^{\dagger}$	Line length	902.2 m	
Elevation range of base column	120.0 m to 12.0 m below $SWL^{\dagger}$	Line diameter	0.09 m	
Upper column diameter	6.5 m	Fairlead position below $SWL^{\dagger}$	70.0 m	
Base column diameter	9.4 m	Anchor position below $SWL^{\dagger}$	320.0 m	
Mass including ballast	7,466,330 kg	Radius to anchor from floater centerline	853.87 m	

<sup>\*</sup> rotor-nacelle assembly

<sup>†</sup> still water level

*Author contributions.* ML: Conceptualization, Data curation, Methodology, Software, Validation, Formal analysis, Investigation, Project administration, Visualization, Writing—original draft, Writing—review; MC: Conceptualization, Methodology, Supervision, Writing—review & editing; AK: Supervision, Writing—review & editing, Funding acquisition.

Competing interests. The authors declare that they have no conflict of interest.

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