
Supplementary Materials for

A New Way to Estimate Maximum Power from Wind Turbines: Linking Newtonian with Action Mechanics

Estimating maximum power from wind turbines

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This PDF file includes:

1. Figure S1. . Outline of computer program
2. Table S1. Data output for Vevor 1.24 m diameter turbine. Variation in wind angle of incidence.
3. Table S2. Data output for vevor 1.24 m diameter turbine. Variation in tip-speed ratio.
4. Table S3. Data sets for GE 1.5MW output turbine, variation in angle of incidence and in tip-speed ratio.
5. Turbine5/cal: Coding for TRS32 Astrocal Vevor Turbine (cgs units).
Coding for GE scale turbines is similar except in SI units for inputs and outputs.
6. Python code for estimation of wind power
7. Mathematica notebook for simple estimation of wind power

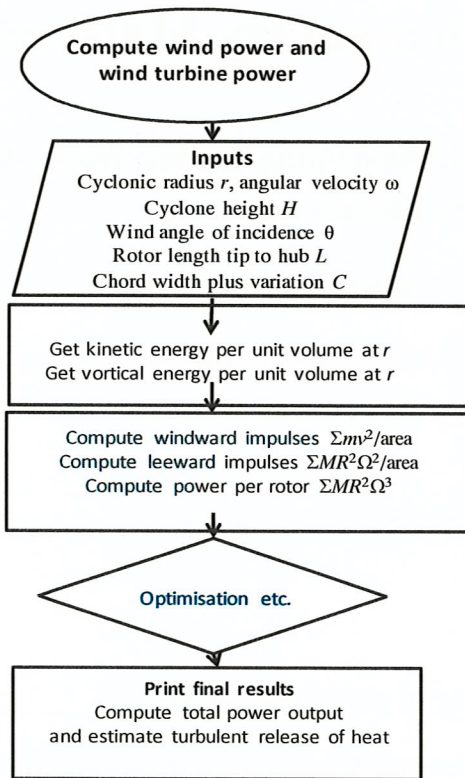


Figure S1. Outline of computer programs.

V=10 m/sec

Venor 1.24 m diameter

Area = 358.8 cm²

L=52 cm

D=1.225x10⁻³

R=10^o8 cm

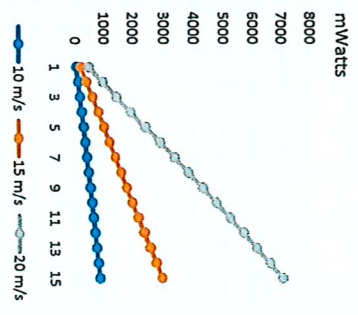
H=10^o5 cm

Table S1

	5	10	15	20	25	30	35	40	45	50	55	60	65	70	75	80	85	90	
S=3																			
Tw x10 ⁷	0.062243	0.244256	0.532217	0.904152	1.33145	1.780834	2.216664	2.603405	2.908089	3.102617	3.165724	3.084494	2.855304	2.48414	1.986262	1.385243	0.711438	0	0
Tb x10 ⁷	0.080049	0.079134	0.077617	0.075509	0.072826	0.069589	0.065823	0.061555	0.056819	0.051651	0.04609	0.040177	0.033959	0.027483	0.020797	0.013953	0.007003	0	0
Tw-Tb x10 ⁷	-0.01781	0.165122	0.454601	0.828644	1.258624	1.711245	2.150841	2.541849	2.85127	3.050966	3.119635	3.044317	2.821345	2.456657	1.965465	1.371289	0.704435	0	0
Q	57.6923	57.6923	57.69231	57.6923	57.69231	57.69231	57.69231	57.6923	57.69231	57.69231	57.69231	57.69231	57.69231	57.69231	57.69231	57.69231	57.69231	57.69231	57.69231
P mW	-1.02728	9.526199	26.22695	47.80635	72.61294	98.72566	124.0872	146.6451	164.4963	176.0173	179.9789	175.6337	162.7699	141.7302	113.3922	79.11284	40.64047	0	0
V=15																			
S=3																			
Tw x10 ⁷	0.140046	0.549575	1.197489	2.034344	2.995763	4.006876	4.987494	5.857661	6.543201	6.980888	7.122879	6.940113	6.424435	5.589315	4.469089	3.116796	1.600736	0	0
Tb x10 ⁷	0.120074	0.118701	0.116425	0.113263	0.109239	0.104384	0.098734	0.092333	0.085229	0.077477	0.069134	0.060266	0.050939	0.041224	0.031196	0.02093	0.010505	0	0
Tw-Tb x10 ⁷	0.019973	0.430874	1.081064	1.921081	2.885524	3.902492	4.88876	5.765328	6.457972	6.903411	7.053745	6.879847	6.373496	5.54809	4.437893	3.095866	1.590231	0	0
Q	86.5384	86.5385	86.5385	86.53846	86.53846	86.53846	86.53846	86.53846	86.53846	86.53846	86.53846	86.53846	86.53846	86.53846	86.53846	86.53846	86.53846	86.53846	86.53846
P mW	1.728414	37.28721	93.55365	166.2474	249.7953	337.7156	423.0657	498.9226	558.863	597.4106	610.4202	595.3713	551.5525	480.1232	384.0484	267.9115	137.6161	0	0
V=20																			
S=3																			
Tw x10 ⁷	0.248971	0.977023	2.128869	3.616612	5.325801	7.123335	8.866656	10.41362	11.63236	12.41047	12.6629	12.33798	11.42112	9.93656	7.945048	5.540971	2.845753	0	0
Tb	0.160098	0.158268	0.155233	0.151018	0.145652	0.139179	0.131646	0.123111	0.113639	0.103302	0.090218	0.080355	0.067919	0.054966	0.041595	0.027907	0.014007	0	0
Tw-Tb	0.088873	0.818755	1.973636	3.465595	5.180149	6.984156	8.735011	10.29051	11.51872	12.30717	12.57268	12.25762	11.3532	9.881594	7.903453	5.513064	2.831746	0	0
Q	115.3846	115.3846	115.3846	115.3846	115.3846	115.3846	115.3846	115.3846	115.3846	115.3846	115.3846	115.3846	115.3846	115.3846	115.3846	115.3846	115.3846	115.3846	115.3846
P mW	10.25459	94.47168	227.7272	399.8763	597.7094	805.8642	1007.886	1187.366	1329.083	1420.058	1450.694	1414.341	1309.985	1140.184	911.9369	636.1228	326.7399	0	0
10 m/s																			
15 m/s																			
20 m/s																			

angle

		Table 52														
		Varying tip-speed ratio														
		Vewor 1.24 m diameter														
		1	2	3	4	5	6	7	8	9	10	11	12	13	14	15
10 m/sec	λ	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15
	$T_w \times 10^7$	3.165724	3.165724	3.165724	3.165724	3.165724	3.165724	3.165724	3.165724	3.165724	3.165724	3.165724	3.165724	3.165724	3.165724	3.165724
	Tb $\times 10^7$	0.015363	0.030726	0.04609	0.061453	0.076816	0.092179	0.107542	0.122906	0.138269	0.153632	0.168995	0.184358	0.199722	0.215085	0.230448
	$T_w - T_b \times 10^7$	3.150361	3.134998	3.119635	3.104271	3.088908	3.073544	3.058182	3.042819	3.027455	3.012092	2.996729	2.981366	2.966003	2.950639	2.935276
Ω rad/sec		19.23077	38.4615	57.69231	76.9231	96.15385	115.3846	134.6154	153.8462	173.0769	192.3076	211.5385	230.7692	250	269.2307	288.4615
P w		60.58386	120.5768	179.9789	238.7901	297.0104	354.6398	411.6783	468.1259	523.9827	579.2485	633.9234	688.0074	741.5006	794.4029	846.7143
15 m/sec	λ	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15
	$T_w \times 10^7$	7.12288	7.12288	7.12288	7.12288	7.12288	7.12288	7.12288	7.12288	7.12288	7.12288	7.12288	7.12288	7.12288	7.12288	7.12288
	Tb $\times 10^7$	0.02305	0.04609	0.06913	0.09218	0.115223	0.138269	0.161314	0.184358	0.207403	0.230448	0.253493	0.276538	0.299582	0.322627	0.345672
	$T_w - T_b \times 10^7$	7.09983	7.076789	7.05374	7.0307	7.007655	6.984611	6.961566	6.938521	6.915476	6.89243	6.869387	6.846342	6.823297	6.800252	6.777207
Ω rad/sec		29	57.6923	86.5385	115.3846	144.2307	173.0769	201.923	230.7692	259.6153	288.4614	317.3077	346.1538	375	403.8462	432.6923
P w		204.803	408.2763	610.4202	811.2346	1010.72	1208.875	1405.701	1601.197	1795.364	1988.201	2179.709	2369.888	2558.736	2746.256	2932.446
20 m/sec	λ	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15
	$T_w \times 10^7$	12.6629	12.6629	12.6629	12.6629	12.6629	12.6629	12.6629	12.6629	12.6629	12.6629	12.6629	12.6629	12.6629	12.6629	12.6629
	Tb $\times 10^7$	0.030726	0.061453	0.092179	0.122906	0.153632	0.184358	0.215085	0.245811	0.276538	0.307264	0.33799	0.368717	0.399443	0.43017	0.460896
	$T_w - T_b \times 10^7$	12.63217	12.60144	12.57072	12.53999	12.50926	12.47854	12.44781	12.41709	12.38636	12.35563	12.32491	12.29418	12.26345	12.23273	12.202
Ω rad/sec		38.46154	76.92308	115.3846	153.8462	192.3077	230.7692	269.2308	307.6923	346.1538	384.6154	423.0769	461.5385	500	538.4615	576.9231
P w		485.8527	969.3418	1450.467	1929.229	2405.628	2879.6626	3351.334	3820.642	4287.586	4752.166	5214.383	5674.237	6131.173	6586.853	7039.616



3647	046	.	3660	046	.
3648	046	.	3661	046	.
3649	046	.	3662	046	.
3650	046	.	3663	046	.
3651	046	.	3664	046	.
3652	046	.	3665	046	.
3653	046	.	3666	046	.
3654	046	.	3667	046	.
3655	046	.	3668	046	.
3656	046	.	3669	046	.
3657	046	.	3670	046	.
3658	046	.	3671	046	.
3659	046	.			
Turbine5/cal Vevor Turbine cgs			0593	119	print
KE = IW ² /2 I=MR ² /2 28/08/21 08/21			0594	002	HALT
Revised 14/9/21 e ^{0.67} TbD			Chord width of rotor blade cm		
Estimate [Tw-Tb]xRadian/sec=Power			0635	096	LABEL
Radius of air cell say 1000 km 10 ⁸ B			0636	067	C
Tapered blade 2.3 to 11.5 cm			0637	107	STO MEM
Radius is R+10 cm + 1.0 cm up to 52 cm			0638	048	0
Chord from 2.30 to 11.504 +0.1769/cm			0639	054	6
Get KE and Vortical energy throughput			0640	119	print
Revised 3/10/21 mv ² for kT in vort en			0641	002	HALT
0400	096	LABEL	Angle of attack degrees		
0401	082	R	0682	096	LABEL
0402	107	STO MEM	0683	079	0
0403	048	0	0684	107	STO MEM
0404	049	1	0685	048	0
0405	119	print	0686	055	7
0406	002	HALT	0687	119	print
Cell height H say 5000 m 5.10 ⁵ cm			0688	002	HALT
0447	096	LABEL	Length of rotor input L cm		
0448	072	H	0729	096	LABEL
0449	107	STO MEM	0730	076	L
0450	048	0	0731	107	STO MEM
0451	050	2	0732	048	0
0452	119	print	0733	056	8
0453	002	HALT	0734	119	print
Angular velocity W radians/sec			0735	002	HALT
0494	096	LABEL	Tip speed/wind ratio lambda		
0495	087	W	0776	096	LABEL
0496	107	STO MEM	0777	083	S
0497	048	0	0778	107	STO MEM
0498	051	3	0779	048	0
0499	119	print	0780	057	9
0500	002	HALT	0781	119	print
Mean air density g/cm ³			0782	002	HALT
0541	096	LABEL	0783	096	LABEL
0542	068	D	0784	088	X
0543	107	STO MEM	Calculate max wind speed rw cm/sec		
0544	048	0	0825	040	(
0545	052	4	0826	109	RCL MEM
0546	119	print	0827	048	0
0547	002	HALT	0828	049	1
Area of wind blades A=LxC cm ²			0829	042	x
0588	096	LABEL	0830	109	RCL MEM
0589	065	A	0831	048	0
0590	107	STO MEM	0832	051	3
0591	048	0	0833	041)
0592	053	5	Print max wind speed cm/sec		

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0782 002 HALT
0783 096 LABEL
0784 088 X
Calculate max wind speed rw cm/sec
0825 040 (
0826 109 RCL MEM
0827 048 0
0828 049 1
0829 042 x
0830 109 RCL MEM
0831 048 0
0832 051 3
0833 041 )
Print max wind speed cm/sec
0874 119 print
0875 107 STO MEM
0876 049 1
0877 048 0
Calculate KE
Calculate volume cm^3
0958 040 (
0959 112 pi
0960 042 x
0961 109 RCL MEM
0962 048 0
0963 049 1
0964 113 Square
0965 042 x
0966 109 RCL MEM
0967 048 0
0968 050 2
0969 041 )
Print vol cm^3 of air cell, stow m11
1010 119 print
1011 107 STO MEM
1012 049 1
1013 049 1
1014 040 (
1015 109 RCL MEM
1016 049 1
1017 049 1
1018 042 x
1019 109 RCL MEM
1020 048 0
1021 052 4
1022 041 )
Print mass of air cell, stow m12 g
1063 119 print
1064 107 STO MEM
1065 049 1
1066 050 2
1067 040 (
1068 109 RCL MEM
1069 049 1
1070 050 2
1071 042 x
1072 109 RCL MEM
1073 048 0
1074 049 1
1075 113 Square
1076 047 /
1077 052 4
1078 041 )
Print I and stow in m13
1119 119 print
1120 107 STO MEM
1121 049 1
1122 051 3
1123 040 (
1124 109 RCL MEM
1125 049 1
1126 051 3
1127 042 x
1128 109 RCL MEM
1129 048 0
1130 051 3
1131 113 Square
1132 041 )
Print KE IW^2 of air cell, stow m14
1173 119 print
1174 107 STO MEM
1175 049 1
1176 052 4
Calculate KE, power of turbine volume
1217 040 (
1218 040 (
1219 040 (
1220 040 (
1221 109 RCL MEM
1222 049 1
1223 048 0
1224 042 x
1225 112 pi
1226 042 x
1227 109 RCL MEM
1228 048 0
1229 056 8
1230 113 Square
1231 041 )
1232 042 x
1233 040 (
1234 109 RCL MEM
1235 048 0
1236 052 4
1237 042 x
1238 109 RCL MEM
1239 049 1
1240 048 0
1241 113 Square
1242 041 )
1243 041 )
1244 047 /
1245 050 2
1246 041 )
Print rate kinetic energy P=DAv^30.6
1287 119 print
1288 042 x
1289 048 0
1290 046 .
1291 054 6

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1292 041 )
Print adjustment for Cp
1333 119 print
Get vortical energy
1374 107 STO MEM
1375 049 1
1376 052 4
1377 040 (
1378 040 (
1379 040 (
1380 112 pi
1381 042 x
1382 109 RCL MEM
1383 048 0
1384 056 8
1385 113 Square
1386 041 )
1387 047 /
1388 040 (
1389 051 3
1390 042 x
1391 109 RCL MEM
1392 048 0
1393 053 5
1394 041 )
1395 041 )
1396 119 print
1397 114 1/x
1398 042 x
1399 109 RCL MEM
1400 049 1
1401 052 4
1402 041 )
1403 119 print
1404 040 (
1405 040 (
1406 040 (
1407 109 RCL MEM
1408 048 0
1409 049 1
1410 113 Square
1411 042 x
1412 109 RCL MEM
1413 048 0
1414 051 3
1415 041 )
1416 119 print
1417 042 x
1418 040 (
1419 050 2
1420 057 9
1421 042 x
1422 049 1
1423 046 .
1424 054 6
1425 055 7
1426 101 EE
1427 050 2
1428 052 4
1429 111 +/-

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1430 041 )
1431 041 )
Print action @v
1472 119 print
1473 047 /
1474 049 1
1475 046 .
1476 048 0
1477 053 5
1478 052 4
1479 101 EE
1480 050 2
1481 055 7
1482 111 +/-
1483 041 )
Print nv
1524 119 print
1525 107 STO MEM
1526 056 8
1527 048 0
Get mv^2lnvv
1568 040 (
1569 040 (
1570 109 RCL MEM
1571 056 8
1572 048 0
1573 110 LOG e
1574 119 print
1575 042 x
1576 040 (
1577 040 (
1578 050 2
1579 057 9
1580 042 x
1581 049 1
1582 046 .
1583 054 6
1584 055 7
1585 101 EE
1586 050 2
1587 052 4
1588 111 +/-
1589 041 )
1590 042 x
1591 109 RCL MEM
1592 049 1
1593 048 0
1594 113 Square
1595 041 )
1596 107 STO MEM
1597 057 9
1598 049 1
Print mv^2
1639 119 print
1640 041 )
Print mv^2lnnv
1681 119 print
1682 041 )
1683 107 STO MEM
1684 056 8

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1685	049	1		1862	050	2
1686	040	(1863	052	4
1687	109	RCL MEM		1864	111	+/-
1688	056	8		1865	041)
1689	049	1		1866	041)
1690	047	/		Print N per cm^3		
1691	109	RCL MEM		1907	119	print
1692	056	8		1908	042	x
1693	048	0		1909	109	RCL MEM
1694	041)		1910	056	8
Print hvv vortical quantum				1911	049	1
1735	119	print		1912	041)
1736	047	/		1913	119	print
1737	054	6		Print ergs/cm^3 or vortical pressure		
1738	046	.		1954	107	STO MEM
1739	054	6		1955	056	8
1740	050	2		1956	049	1
1741	054	6		1957	040	(
1742	049	1		1958	109	RCL MEM
1743	101	EE		1959	056	8
1744	050	2		1960	049	1
1745	055	7		1961	042	x
1746	111	+/-		1962	109	RCL MEM
1747	041)		1963	049	1
Print v frequency				1964	048	0
1788	119	print		1965	041)
1789	114	1/x		Print power pe/cm^2		
1790	042	x		2006	119	print
1791	050	2		2007	107	STO MEM
1792	046	.		2008	056	8
1793	057	9		2009	050	2
1794	057	9		Get power for 3 blades and area		
1795	055	7		2050	040	(
1796	057	9		2051	109	RCL MEM
1797	101	EE		2052	056	8
1798	049	1		2053	050	2
1799	048	0		2054	042	x
1800	041)		2055	109	RCL MEM
Print Y wavelength				2056	048	0
1841	119	print		2057	053	5
1842	040	(2058	042	x
1843	040	(2059	051	3
1844	049	1		2060	041)
1845	046	.		Print vortical power total blade area		
1846	050	2		2101	119	print
1847	050	2		2102	107	STO MEM
1848	053	5		2103	056	8
1849	101	EE		2104	051	3
1850	051	3		Get T		
1851	111	+/-		2145	040	(
1852	047	/		2146	109	RCL MEM
1853	040	(2147	057	9
1854	050	2		2148	049	1
1855	057	9		2149	047	/
1856	042	x		2150	040	(
1857	049	1		2151	049	1
1858	046	.		2152	046	.
1859	054	6		2153	051	3
1860	055	7		2154	056	8
1861	101	EE		2155	048	0

2156	054	6	2411	109	RCL MEM
2157	101	EE	2412	048	0
2158	049	1	2413	052	4
2159	054	6	2414	042	x
2160	111	+/-	2415	109	RCL MEM
2161	042	x	2416	049	1
2162	051	3	2417	048	0
2163	041)	2418	041)
2164	041)	Print, stow total impact SumMvsin0/sec		
Print Torque/Temperature			2459	119	print
2205	119	print	2460	107	STD MEM
Get thrust per rotor blade @ 0 degrees			2461	049	1
2246	040	(2462	055	7
2247	109	RCL MEM	Calculate Torque and anti-Torque-heat		
2248	048	0	2503	040	(
2249	053	5	2504	040	(
2250	042	x	2505	109	RCL MEM
2251	040	(2506	049	1
2252	109	RCL MEM	2507	055	7
2253	048	0	2508	047	/
2254	055	7	2509	109	RCL MEM
2255	115	Sine	2510	048	0
2256	041)	2511	053	5
2257	042	x	2512	041)
2258	109	RCL MEM	2513	107	STD MEM
2259	049	1	2514	049	1
2260	048	0	2515	055	7
2261	041)	2516	041)
Print A sin 0 x v = volume per sec			2517	119	print
2302	119	print	2518	107	STD MEM
2303	107	STD MEM	2519	049	1
2304	049	1	2520	055	7
2305	053	5	2521	040	(
2306	040	(2522	050	2
2307	109	RCL MEM	2523	046	.
2308	049	1	2524	051	3
2309	053	5	2525	048	0
2310	042	x	Print tip chord		
2311	109	RCL MEM	2566	119	print
2312	048	0	2567	107	STD MEM
2313	052	4	2568	048	0
2314	042	x	2569	054	6
2315	109	RCL MEM	Print tip width		
2316	049	1	2610	119	print
2317	048	0	2611	109	RCL MEM
2318	113	Square	2612	048	0
2319	047	/	2613	056	8
2320	050	2	2614	107	STD MEM
2321	041)	2615	048	0
Print M A sin 0 v^2/2 KE impact on blade			2616	048	0
2362	119	print	2617	119	print
2363	107	STD MEM	Get Tw and Tb		
2364	049	1	2658	096	LABEL
2365	054	6	2659	084	T
Calculate SumMvsin0/sec force/blade			2660	040	(
2406	040	(2661	040	(
2407	109	RCL MEM	2662	048	0
2408	049	1	2663	046	.
2409	053	5	2664	049	1
2410	042	x	2665	055	7

2666	054	6		2921	057	9	
2667	057	9		Get integrated torque or rate action			
2668	117	SUM MEM		SumMvsinDcos20/metre sec;SumFxR=Tb			
2669	048	0		Get rotor reverse torque, action rate			
2670	054	6		3042	040	(
2671	041)		3043	040	(
2672	040	(3044	109	RCL MEM	
2673	109	RCL MEM		3045	049	1	
2674	049	1		3046	048	0	
2675	055	7		3047	042	x	
2676	042	x		3048	109	RCL MEM	
2677	109	RCL MEM		3049	048	0	
2678	048	0		3050	057	9	
2679	054	6		3051	041)	
2680	041)		3052	047	/	
2681	041)		3053	109	RCL MEM	
Print MvsinD/sec force/metre blade				3054	048	0	
2722	107	STO MEM		3055	056	8	
2723	049	1		3056	041)	
2724	056	8		Stow rotor W			
2725	040	(3097	107	STO MEM	
2726	040	(3098	050	2	
2727	050	2		3099	048	0	
2728	042	x		3100	040	(
2729	109	RCL MEM		3101	109	RCL MEM	
2730	048	0		3102	050	2	
2731	055	7		3103	048	0	
2732	041)		3104	042	x	
2733	115	Sine		3105	040	(
2734	042	x		3106	109	RCL MEM	
2735	109	RCL MEM		3107	048	0	
2736	049	1		3108	048	0	
2737	056	8		3109	043	+	
2738	041)		3110	049	1	
SMvsinDcos20/metresec, action/area/sec				3111	048	0	
2779	107	STO MEM		3112	041)	
2780	049	1		3113	041)	
2781	056	8		Get RW of blade at R and stow			
Label Torque or rate of Action				3154	113	Square	
Get SumRotor Windward Torque Tw				3155	041)	
2862	040	(RVdD/sec			
2863	109	RCL MEM		3196	107	STO MEM	
2864	049	1		3197	051	3	
2865	056	8		3198	048	0	
2866	042	x		3199	040	(
2867	040	(3200	040	(
2868	109	RCL MEM		3201	040	(
2869	048	0		Get blade area dRcost segment at R			
2870	048	0		3242	109	RCL MEM	
2871	043	+		3243	048	0	
2872	049	1		3244	055	7	
2873	048	0		3245	099	Cosine	
2874	041)		3246	042	x	
2875	042	x		3247	109	RCL MEM	
2876	051	3		3248	048	0	
2877	041)		3249	054	6	
2878	041)		3250	041)	
Get cumulative Tw				3251	042	x	
2919	117	SUM MEM		Get inertia MR/sec, power MR^20m^2			
2920	049	1		Use 2-D density D^2/3 for Tw output			

3332	040	(3548	051	3
3333	109	RCL MEM	3549	109	RCL MEM
3334	048	0	3550	048	0
3335	052	4	3551	054	6
3336	091	Exponent	3552	119	print
3337	048	0	Clear	cumulative memories for reruns	
3338	046	.	3593	048	0
3339	054	6	3594	107	STO MEM
3340	055	7	3595	048	0
3341	042	x	3596	054	6
3342	109	RCL MEM	3597	107	STO MEM
3343	051	3	3598	049	1
3344	048	0	3599	057	9
3345	041)	3600	107	STO MEM
3346	041)	3601	050	2
3347	047	/	3602	049	1
3348	109	RCL MEM	3603	107	STO MEM
3349	050	2	3604	050	2
3350	048	0	3605	050	2
3351	041)	3606	002	HALT
3352	042	x	3607	046	.
3353	051	3	3608	046	.
3354	041)	3609	046	.
Get	Sum $MV^2 \cos \theta = MRVd\theta/dt = MV^2 = HEAT$		3610	046	.
3395	117	SUM MEM	3611	046	.
3396	050	2	3612	046	.
3397	049	1	3613	046	.
Print	total back torque, action/sec		3614	046	.
3438	039	Dec JPNZ	3615	046	.
3439	084	T	3616	046	.
3440	040	(3617	046	.
3441	109	RCL MEM	3618	046	.
3442	049	1	3619	046	.
3443	057	9	3620	046	.
3444	119	print	3621	046	.
3445	045	-	3622	046	.
3446	109	RCL MEM	3623	046	.
3447	050	2	3624	046	.
3448	049	1	3625	046	.
3449	119	print	3626	046	.
3450	041)	3627	046	.
Print	and stow 3-rotor net Torque		3628	046	.
3491	119	print	3629	046	.
3492	107	STO MEM	3630	046	.
3493	050	2	3631	046	.
3494	050	2	3632	046	.
3495	040	(3633	046	.
3496	109	RCL MEM	3634	046	.
3497	050	2	3635	046	.
3498	050	2	3636	046	.
3499	042	x	3637	046	.
3500	109	RCL MEM	3638	046	.
3501	050	2	3639	046	.
3502	048	0	3640	046	.
3503	119	print	3641	046	.
3504	041)	3642	046	.
Print	and stow net Power per windmill		3643	046	.
3545	119	print	3644	046	.
3546	107	STO MEM	3645	046	.
3547	050	2	3646	046	.

Python coding

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        "from matplotlib import cm\n",
        "import seaborn as sns\n",
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    "In python, the keyword `lambda` creates an \"anonymous function\".
    This is a function (a subroutine) that doesn't have a name, but can be
    executed like any other function. Since it doesn't have a name, the only
    way to keep track of it is to assign it to a variable.\n",
    "\n",
    "Historically this comes from the lambda-calculus (Alonso Church in
    the 1930s), and it is fascinating and very expressive. Unfortunately, it
    isn't used as a fundamental design unit for code in many languages...\n",
    "\n",
    "Mathematica is an exception: it uses these so extensively that they
    have quite a compact shorthand (&, #1, #2, ...): \n",
    "```\nlambda x, y, z: x + y * z```\n in python can be written as
    ``\n#1+#2*#3&``\n, where `#1` corresponds to `x`, `#2` to `y`, etc.\n",
    "\n",
    "These lambdas are functions in the sense meant by mathematicians.
    They calculate an output from only the arguments: \n",
    "$f(x, y, z) = x + y z$ \n",
    "is written as..."
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    "f(2, 3, 4)"
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```

```
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    "scipy.integrate.quad(sq, 1, 3)"
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    "An important feature of anonymous functions in python is that they  
can capture values from where they are defined."
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    "three_plus(4)"
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    "It is considered very poor form to modify a captured value.\n",
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function without redefining it, which makes code very difficult to read  
and manage. Lots of lanugages prohibit this:"
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```

```

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    "### Vectorising\n",
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argument and produces another function that does the same operation over
a vector/array, elementwise:"
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    "The `numpy.vectorize` function actually does this to all the
arguments in a function, so that they behave according to the conventions
of numpy arrays (there is a lot of detail about all that). I use this
feature when I create the grid-based results for the 3D plot later on."
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    "f_vec = np.vectorize(f) # the function f from above: f(x, y, z) = x
+ y * z\n",
    "f_vec(x = np.array([1, 2, 3]),\n",
    "      y = np.array([4, 5, 6]),\n",
    "      z = np.array([7, 8, 9]))"
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```



```

    "---\n",
    "# The Equations\n",
    "The integrand functions take numerical values in all arguments.\n",
    "\n",
    "The torque functions require `c_fn` and `theta_fn` as inputs. These
functions calculate the chord width and the angle of incidence at a given
radius.\n",
    "\n",
    "For example, this `c_fn` defines a constant chord width of 1.3m (the
radius argument is unused): \n",
    "```lambda r: 1.3```\n",
    "\n",
    "And this `theta_fn` defines the angle of incidence on a twisted
blade that goes from `theta0` at the base, to `theta0` plus 10 degrees at
the 8m tip: \n",
    "```lambda theta0, r: theta0 + 10 * deg_to_rad * r / 8.0```\n"
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"deg_to_rad = 2.0 * np.pi / 360.0\n",
"rad_to_deg = 360.0 / (2 * np.pi)\n",
"radsec_to_revmin = 60/(2*np.pi)\n",
"\n",
"rho = 1.225"
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```

```

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  "    return (rho\n",
  "            * c\n",
  "            * np.sin(theta)\n",
  "            * np.sin(2*theta)\n",
  "            * np.power(v, 2)\n",
  "            * r)\n",
  "\n",
  "def tb_integrand(r, omega, c, theta):\n",
  "    return (np.power(rho, 2/3)\n",
  "            * c\n",
  "            * np.cos(theta)\n",
  "            * np.power(r, 2)\n",
  "            * omega)\n",
  "\n",
  "def torque_windward(v, c_fn, theta_fn, R0, RL, theta0):\n",
  "    integrand = lambda r: tw_integrand(r, v, c_fn(r),\n",
theta_fn(theta0, r))\n",
  "    res, _ = scipy.integrate.quad(integrand, R0, RL)\n",
  "    return res\n",
  "\n",
  "def torque_backward(omega, c_fn, theta_fn, R0, RL, theta0):\n",
  "    integrand = lambda r: tb_integrand(r, omega, c_fn(r),\n",
theta_fn(theta0, r))\n",
  "    res, _ = scipy.integrate.quad(integrand, R0, RL)\n",
  "    return res"
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    "### Power"
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    "def power(v, omega, c_fn, theta_fn, R0, RL, theta0):\n",
    "    tw = torque_windward(v, c_fn, theta_fn, R0, RL, theta0)\n",
    "    tb = torque_backward(omega, c_fn, theta_fn, R0, RL, theta0)\n",
    "    return (tw - tb) * omega\n",
    "\n",
    "def maximise_power(v, omega, c_fn, theta_fn, R0, RL):\n",

```

```

    "    ''' Maximises power output over theta0.\n",
    "    \n",
    "    Returns the maximum power output of the blade and the theta
that produces it\n",
    "    '''\n",
    "    # scipy implements only minimisation\n",
    "    # the max of a function's values occurs at the min of the
negative of the function's values\n",
    "    def minimisation_objective(theta0):\n",
    "        return -power(v, omega, c_fn, theta_fn, R0, RL, theta0)\n",
    "    res = scipy.optimize.minimize_scalar(minimisation_objective)\n",
    "    max_power = -res.fun # negate the minimum to recover the maximum
value\n",
    "    max_theta = res.x\n",
    "    return max_power, max_theta"
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"L = 38.75\n",
"r0 = 0.00\n",
"rL = L\n",
"c_fn = lambda r: 1.0\n",
"theta_fn = lambda theta0, r: theta0"
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"# Vevor blade definition\n",
"# \ "Tapering from 2.30cm at the tip 62.0cm from the hub to 11.50cm
at the base,\n",
"# supported on a 10cm stalk to the hub"\n",
"\n",
"L = 0.620\n",

```

```

    "r0 = 0.100\n",
    "rL = L\n",
    "c0 = 0.115\n",
    "cL = 0.023\n",
    "\n",
    "c_fn = lambda r: (r - r0) * (cL - c0) / (rL - r0) + c0\n",
    "theta_fn = lambda theta0, r: theta0"
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    "c0 = 3.025\n",
    "cL = 0.100\n",
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    "c_fn = lambda r: r * (cL - c0) / L + c0\n",
    "theta_fn = lambda theta0, r: theta0"
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    "r0 = 0.000\n",
    "rL = L\n",
    "c0 = 3.025\n",
    "cL = 0.100\n",
    "phiL = 15 * deg_to_rad # The twist at the tip of the blade, relative
to the base\n",
    "\n",
    "c_fn = lambda r: r * (cL - c0) / L + c0\n",
    "theta_fn = lambda theta0, r: theta0 + phiL * r / L"
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```

```

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    "λ = 8.0\n",
    "v = 15.0\n",
    "Ω = v*λ/L\n",
    "display(f'Ω = {Ω} rad/s')\n",
    "display(f'Ω = {Ω*radsec_to_revmin} rev/min')\n",
    "\n",
    "# varying the angle of incidence\n",
    "N = 80\n",
    "θ = np.linspace(5, 90, N) * deg_to_rad\n",
    "\n",
    "power_vec = np.vectorize(power)\n",
    "torque_windward_vec = np.vectorize(torque_windward)\n",
    "torque_backward_vec = np.vectorize(torque_backward)\n",
    "\n",
    "p = power_vec(v, Ω, c_fn, theta_fn, r0, rL, theta0=θ) # using an array in the theta0 argument, so that the function produces an array result\n",
    "tw = torque_windward_vec(v, c_fn, theta_fn, r0, rL, theta0=θ)\n",
    "tb = torque_backward_vec(Ω, c_fn, theta_fn, r0, rL, theta0=θ)\n",
    "\n",
    "max_p, θ_max_p = maximise_power(v, Ω, c_fn, theta_fn, r0, rL)"
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    "display(f'θ_max_p: {θ_max_p*rad_to_deg} deg')\n",
    "display(f'max_p: {3*max_p*1e-6} MW)"
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```

```

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    "ax.plot( $\theta$ *rad_to_deg, 3*tw*1e-6, linewidth=1.0)\n",
    "ax.plot( $\theta$ *rad_to_deg, 3*tb*1e-6, linewidth=1.0)\n",
    "\n",
    "ax.axvline( $\theta_{\max_p}$ *rad_to_deg, linewidth=0.5, color='k')\n",
    "ax.axhline(3*max_p*1e-6, linewidth=0.5, color='k')\n",
    "\n",
    "ax.set_xlabel(r'Angle of incidence,  $\theta$  (deg))\n",
    "ax.set_ylabel(r'Power,  $p$  (MW))\n",
    "ax.grid()"
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    "idx_theta = np.linspace(5, 90, N)*deg_to_rad\n",
    "\n",
    "grid_v, grid_theta = np.meshgrid(idx_v, idx_theta)\n",
    "grid_omega = grid_v*lambda/L"
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    "                        linewidth=0, antialiased=True)\n",
    "\n",
    "ax.view_init(15, 220)\n",
    "\n",
    "ax.set_xlabel(r'Wind speed, $v$ (m/s))\n",
    "ax.set_ylabel(r'Angle of incidence, $\theta$ (deg))\n",
    "ax.set_zlabel(r'Power, $p$ (MW))\n",
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```


In[]:= Canvas [□]

In[]:= Radial action estimation of windward and leeward torques for wind turbines;
Tw and Tb and potential power output (P) in Watts

In[]:=
vV = 20; (* Wind velocity m/sec *)
dD = 1.225 (* Air density kg/m^3 *)
θ = 50Degree (* Wind angle of incidence *)
cC = 5 (* Chord width m *)
lL = 80 (* Blade length m *)
λ = 10 (* Tip speed ratio RΩ/vV *)
Ω = λ * vV / lL (* Rotor angular velocity rad/sec *)
Nu = 3 (* Number rotors *)
Tw = dD cC Sin[θ] Sin[2 θ] vV^2 lL^2 / 2 (* Windward torque *)
Tb = dD cC Cos[θ] Ω lL^3 / 3 (* Leeward or Back torque *)
Pt = (Tw - Tb) Ω Nu

Out[]:= 1.225

Out[]:= 50 °

Out[]:= 5

Out[]:= 80

Out[]:= 10

Out[]:= $\frac{5}{2}$

Out[]:= 3

Out[]:= 5.91455×10^6

Out[]:= 1.67982×10^6

Out[]:= 3.17605×10^7

In[]:=

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