OC6 ~~Pp~~roject ~~Pp~~hase IV: Validation of ~~Nn~~umerical ~~Mm~~odels for ~~Nn~~ovel ~~Ff~~loating ~~Oo~~ffshore ~~Ww~~ind ~~Ss~~upport ~~Ss~~tructures

Roger Bergua[1], Will Wiley[1], Amy Robertson[1], Jason Jonkman[1], Cédric Brun[2], Jean-Philippe Pineau[2], Quan Qian[3], Wen Maoshi[3], Alec Beardsell[4], Joshua Cutler[4], Fabio Pierella[5], Christian Anker Hansen[5], Wei Shi[6], Jie Fu[6], Lehan Hu[6], Prokopios Vlachogiannis[7], Christophe Peyrard[7], Christopher Simon Wright[8], Dallán Friel[8], Øyvind Waage Hanssen-Bauer[9], Carlos Renan dos Santos[9], Eelco Frickel[10], Hafizul Islam[10], Arjen Koop[10], Zhiqiang Hu[11], Jihuai Yang[11], Tristan Quideau[12], Violette Harnois[12], Kelsey Shaler[13], Stefan Netzband[14], Daniel Alarcón[15], Pau Trubat[15], Aengus Connolly[16], Seán B. Leen[17], Oisín Conway[17]

[1]National Wind Technology Center, National Renewable Energy Laboratory, Golden, CO 80401, USA
[2]Research Department, Bureau Veritas – Marine Division, Saint-Herblain, 44818, France
[3]Research Institute, CSSC Haizhuang Windpower Co. Ltd, Chongqing, 401122, China
[4]Turbine Engineering, DNV, Bristol, BS2 0PS, UK
[5]Department of Wind Energy and Energy Systems, Technical University of Denmark, Lyngby, 2800, Denmark
[6]State Key Laboratory of Coastal and Offshore Engineering, Dalian University of Technology, Dalian, 116024, China
[7]R&D National Hydraulics and Environment Laboratory (LNHE), EDF, Chatou, 78400, France
[8]Gavin & Doherty Geosolutions Ltd., Dublin, D14 X627, Ireland
[9]Institute for Energy Technology, Kjeller, 2007, Norway
[10]Maritime Research Institute Netherlands (MARIN), Wageningen, 6708 PM, The Netherlands
[11]School of Engineering, Newcastle University, Newcastle upon Tyne, NE1 7RU, UK
[12]Floating Offshore Group, PRINCIPIA, La Ciotat 13600, France
[13]Power and Technology Department, Shell International Exploration and Production, Houston, TX 77079, USA
[14]Fluid Dynamics and Ship Theory, Hamburg University of Technology, Hamburg, 21073, Germany
[15]Department of Civil and Environmental Engineering, Universitat Politècnica de Catalunya, Barcelona, 08034, Spain
[16]Department of Offshore Wind, Wood Group Kenny, Galway, H91 R9YR, Ireland
[17]Department of Mechanical Engineering, School of Engineering, University of Galway, Galway, H91 HX31, Ireland

Correspondence to: Roger Bergua (roger.bergua@nrel.gov)

Abstract. This paper provides a summary of the work done within Phase IV of the Offshore Code Comparison Collaboration, Continued, with Correlation and unCertainty (OC6) project, under International Energy Agency Wind Technology Collaboration Programme Task 30. This phase focused on validating the loading on and motion of a novel floating offshore wind system. Numerical models of a 3.6-MW horizontal-axis wind turbine atop the TetraSpar floating support structure were compared using measurement data from a 1:43 Froude-scale test performed in the University of Maine's Alfond Wind-Wave (W2) Ocean Engineering Laboratory. Participants in the project ran a series of simulations, including system equilibrium, surge offsets, free-decays, wind-only conditions, wave-only conditions, and a combination of wind and wave conditions. Validation of the models was performed by comparing the aerodynamic loading, floating support structure motion, tower base loading, mooring line tensions, and keel line tensions. The results show a relatively good estimation of the aerodynamic loading and a reasonable estimation of the platform motion and tower base fore-aft bending

moment. However, there is a significant dispersion in the dynamic loading for the upwind mooring line. Very good agreement was observed between most of the numerical models and the experiment for the keel line tensions.

1 Introduction

The objective of Phase IV of the Offshore Code Comparison Collaboration, Continued, with Correlation and unCertainty (OC6) project is to evaluate the accuracy of load predictions and motions by modeling tools for a novel floating offshore wind turbine (FOWT).

The OC6 project is part of an ongoing effort under the International Energy Agency Wind Technology Collaboration Programme (IEA Wind) Task 30 to verify and validate offshore wind turbine modeling tools (IEA Wind, 2023). The foundational OC3 (Offshore Code Comparison Collaboration) project originated ~~back~~ in 2005 as a verification of simulation tools capable of predicting the coupled dynamic loads and responses of FOWTs. Modeling tools that can accurately predict the loading are necessary to enable more reliable and optimized designs. The OC3 project (Jonkman and Musial, 2010) and its extension, the OC4 project (Popko et al., 2012; Robertson et al., 2013), focused on code-to-code comparisons for several fixed-bottom (monopile, tripod, jacket) and floating (spar buoy, semisubmersible) designs. The OC5 project (Robertson et al., 2015; Robertson et al., 2016; Robertson et al., 2017; Popko et al., 2018; Popko et al., 2019) compared simulation results to tank test data and measurements from a wind turbine in the Alpha Ventus offshore wind farm. The OC6 project focused on differences observed in previous projects between model predictions and measurements or phenomena not well understood. The first phase (Robertson et al., 2020; Wang et al., 2021) studied the loads associated ~~to~~with the slow-period surge and pitch motions in FOWTs, which are excited through nonlinear wave loading. The second phase (Bergua et al., 2022a) focused on incorporating a more accurate soil/structure interaction model into the simulation tools~~, able to~~that better represents the boundary conditions and damping in fixed-bottom systems. The third phase (Bergua et al., 2022b) focused on validating the rotor aerodynamic loading for a FOWT undergoing large motions in the surge and pitch directions caused by the floating support structure. The implications of those large motions were also investigated for the near and far wake behavior (Cioni et al, 2023). The fourth phase is focused on validating the coupled dynamics of a novel floating wind turbine design with a streamlined floating support structure, different from traditional FOWT designs. It is also the first time, in these projects, that load predictions for the internal loading within the floating support structure are provided and compared to measurements.

Participants in OC6 Phase IV modeled a 1:43 scaled version of a 3.6-MW wind turbine atop the TetraSpar floating support structure designed by Stiesdal Offshore ~~Technologies~~. The scaled model tested is representative of the full-scale demonstration project that was installed in Norway in July 2021 (Stiesdal Offshore, 2022 ~~Technologies~~). The testing campaign was performed by the University of Maine (Allen & Fowler, 2019). The OC6 Phase IV project followed a stepwise validation approach where the complexity was increased one step at a time to identify and understand potential

differences between the experiment and the numerical models (validation) or differences between the numerical models (verification).

The group ran a series of simulations, including system equilibrium, surge offsets, free-decays, wind-only conditions, wave-only conditions, and a combination of wind and wave conditions. This paper summarizes the work done within the OC6 Phase IV project.

The organization of the remainder of the paper is as follows: Section 2 provides a description of the scaled model and the testing performed. Section 3 provides a description of the active participants involved in the OC6 Phase IV project and their modeling approach. Section 4 then summarizes the load cases that were performed for the verification and validation. Finally, Sections. 5 and Sect. 6 provide some example results from the project and the conclusions drawn, respectively.

2 Model Ddefinition

To validate the accuracy of predicting the loading and motion of a novel FOWT, measurement data were used from an experimental campaign conducted at the Harold Alfond Wind-Wave (W2) Ocean Engineering Laboratory of the Advanced Structures and Composites Center at the University of Maine (University of Maine, 2022) in December 2018 (Borg, 2019) was used. All the quantities in this section are given at full scale, except when specified otherwise. The basin is 30 m long, 9 m wide, and 5 m deep (model scale), and is equipped with a 16-paddle wave maker opposite a beach and a bank of fans 7 m wide and 3.5 m tall (model scale). The testing used a 1:43 Froude-scale thrust matched model of the 3.6 -MW Siemens Gamesa wind turbine, with a rotor diameter of 129 m, atop the TetraSpar floating support structure (see Figure. 1a and Figure. 2a). The scaled model was thrust-matched, providing rotor thrust characteristics similar to the full-scale system at certain operating conditions. Theis configuration studied is representative of the one used in the full-scale demonstration project (Stiesdal Offshore, 2022 Technologies).

(a)   (b)
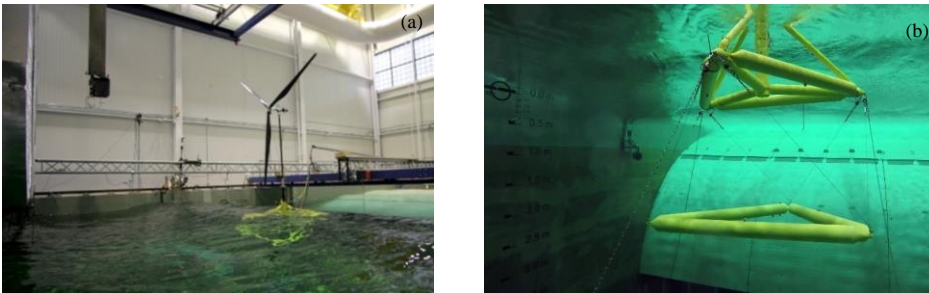
Figure 1: (a) TetraSpar 1:43 scaled model during testing at the University of Maine. (b) Underwater view of the floating support structure. Pictures courtesy of University of Maine.

The TetraSpar is made of two separate structures: hull and keel (see Fig~~ure.~~ 1b and Fig~~ure.~~ 2b). The hull consists of a vertical central column ~~(CC)~~ directly beneath the wind turbine tower. At the base of the central column, there are three radial braces ~~(RB)~~ in the horizontal plane spaced 120° apart. Rigidity is given to this base with hull tri- braces ~~(HT)~~ and diagonal braces ~~(DB)~~. The keel is made of three tri- braces ~~(KT)~~ in the horizontal plane. The hull has a tetrahedral shape while the suspended keel has a triangular shape and acts as a system counterweight. Six taut cables, ~~denoted as~~called keel lines ~~(KL)~~, ~~were~~are used to link the buoyant hull and the ballasted keel. The floating support structure behaves as a rigid body when the keel lines remain in tension. The system features a spar-like stability with the whole system center of mass located below the center of buoyancy. The lower the center of mass, the higher the gravitational restoring moments in roll and pitch degrees of freedom. This behavior is achieved thanks to the location of the keel (37.9 m below the hull) and its large mass (80 % of the mass for the complete floating support structure).

All members are cylindrical tubes and most of them are cone-shaped at one end (e.g., ~~RB~~radial braces) or both ends (e.g., ~~DB~~diagonal braces, ~~HT~~hull tri-braces, and ~~KT~~keel tri-braces). For reference, the ~~HT~~hull tri-braces, ~~KT~~keel tri-braces, and ~~CC~~central column cylinders are 4.3 m in diameter. The ~~CC~~central column is 32.15 m long, the ~~HT~~hull tri-braces ~~is~~are 52.18 m long, and the ~~KT~~keel tri-braces ~~is~~are 64.30 m long. The hull center of mass is located 13.5 m below the mean sea level while the keel center of mass is located 56.6 m below the mean sea level. The center of buoyancy of the floating support structure is located around 33.8 m below the mean sea level along the ~~CC~~central column longitudinal axis. In the physical construction of the system (both at full -scale and model- scale), some members are linked by means of pin joints at one end (e.g., ~~RB~~radial braces in the connection with the central column) or both ends (e.g., ~~DB~~diagonal braces, and ~~HT~~hull tri-braces). The pin joints allow one rotational degree of freedom. This implies that bending moments in one direction would not be transferred through the kinematic joint, which would be an important. ~~This~~ consideration ~~would be important~~ to study the loading within the hull. However, for this validation campaign, the structural properties (e.g., members' thickness distribution) were not known, and, therefore, it was not possible to assess such internal loading. Accordingly, participants considered the hull and keel as distinct rigid bodies and only included the flexibility of the keel lines within the floater. The definition document of the OC6 Phase IV project (Wiley et al., 2023) provides information about the length and external diameter for all members (necessary to characterize the hydrostatic and hydrodynamic loads), the equivalent lumped mass and inertia for the hull and the keel to be included as rigid bodies, and the material properties for the keel lines. The total mass of the floating support structure (including the fairlead tension sensors) is 5.66×10^6 kg, and the buoyancy is 6.13×10^7 N. Due to the small waterplane area, the draft of the platform is very sensitive to the system mass and the support structure volume.

(a) (b) (c~~b~~)
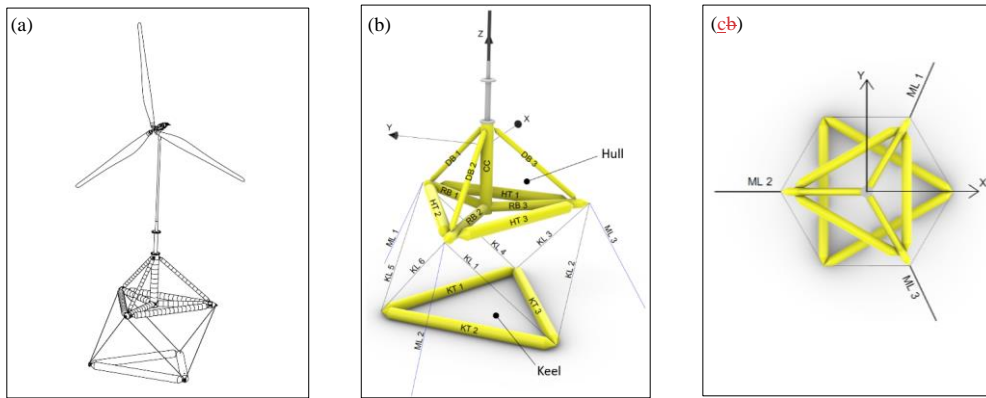
Figure 2: (a) Schematic representation of the model-scale system. (b) Perspective V~~v~~iew of the TetraSpar floating support structure, with the nomenclature and coordinate system used in the project: CC = central column, DB = diagonal brace, HT = hull tri-brace, KL = keel line, KT = keel tri-brace, ML = mooring line, RB = radial brace.~~.~~ (c) Top view of the TetraSpar floating support structure, with the coordinate system used in the project.

The mooring system consisted of three catenary mooring lines (ML1, ML2, and ML3 in Fig~~ure.~~ 2b and Fig. 2c). Each of the three lines was made of two different sections: a lower section made of heavier chain and an upper section of lighter chain. The connection between the upper and lower chains for the upwind line (M~~m~~ooring L~~l~~ine 2) was near the wave tank floor. This condition may be challenging for some numerical models (e.g., quasi-static mooring line models) because it makes the catenary equations more difficult to solve. The two downwind lines (M~~m~~ooring L~~l~~ines 1 and 3) had to be truncated in the testing due to space limitations in the wave tank. The same configuration was replicated in the numerical models. The water depth considered is 193.5 m. The upwind mooring line has a total length around 700 m while the two downwind lines are around 300 m in length~~, each~~.

During the testing, one umbilical cable was used to transfer data and power between the system and the carriage in the wave basin. The presence of this umbilical had a large impact on the surge restoring force relative to that of the mooring system. This resulted~~s~~ in a significant shift of the resting position, surge eigenfrequency, and ~~change of the~~ system dynamics. ~~Therefore, o~~One additional line was included in the numerical models to account for the umbilical. Detailed dimensions and properties for the mooring lines as well as the umbilical can be found in the definition document (Wiley et al., 2023).

The tower in the model test was made of aluminum and carbon fiber and was built to match a Froude-scaled first tower-bending eigenfrequency of the full-scale design. The sectional properties as well as the material properties are provided in the definition document (Wiley et al., 2023). The tower length ~~i~~was around 76 m, and its mass ~~i~~was 1.51 $\times$ 10^5 kg. The resultant hub height was 88.27 m from the mean sea level. Hammer impact tests were performed to check the first tower-bending eigenfrequency. The rotor-nacelle~~-~~ assembly (RNA) was present in these tests. The first tower-bending frequency

for a cantilevered condition (fixed-free) was 0.34 Hz in the fore-aft direction and 0.35 Hz in the side-to-side direction. For the floating body (free-free) boundary condition, it is expected that a higher tower bending eigenfrequency (but this was not tested directly).

The blades used in the testing were made of carbon fiber and were considered to be rigid in the numerical models. The blades were 61.1 m long and had a mass of 1.9×10^4 kg. Information about the blade properties (e.g., twist, chord length, and airfoil thickness-to-chord ratio along the blade), airfoil polars (i.e., lift and drag coefficients for different angles of attack), blades mass, and blades center of gravity were provided in the definition document (Wiley et al., 2023). Properties of the nacelle as well as the complete RNA are also available in the definition document. The rotor shaft tilt angle was 0° and the RNA mass was 2.62×10^5 kg. The total mass for the system (RNA, tower, and floating support structure including the fairlead tension sensors) was 6.08×10^6 kg and the center of mass was located around 39.9 m below the mean sea level. Table 1 shows the system damped natural frequencies for the floating system including the umbilical. These frequencies were obtained from free-decay tests performed in the platform six degrees of freedom. The surge equilibrium position of the platform during the free-decay tests was significantly different (around 14 m more upwind) than the one observed in the wind and wave load cases, likely impacting the mooring stiffness. The reason for the different resting position is not known. When comparing the free-decay tests with and without umbilical, the surge, sway, and yaw directions experienced significant changes in the damped natural frequencies (Wiley et al., 2023). These changes are due to the presence of the umbilical and the different resting positions with and without umbilical. When comparing these two free-decay tests configurations, it can be observed that the umbilical pulls the system around 17 m downwind.

Table 1: System damped natural frequencies

Direction	Frequency [Hz]
Surge	0.0073
Sway	0.0044
Heave	0.025
Roll	0.029
Pitch	0.030
Yaw	0.0070

The instrumentation that was used for this validation campaign measured structural loads (e.g., tower base bending moments, keel line tensions, and fairlead tensions in the mooring lines), tracking motion (e.g., keel and hull six degrees of freedom), and environmental conditions (e.g., wave elevation and wind speed). Table 2 provides a summary of the instrumentation used. A complete list of the sensors used during the experimental campaign is also available in the definition document (Wiley et al., 2023).

Table 2: Summary of instrumentation used within the OC6 Phase IV project

Channel	Number of sensors	Sensor type	Sampling frequency[*]	Notes
Wind speed calibration	3	Direction ultrasonic anemometer	32 Hz	Removed after calibration
Wind speed reference	1	Magnitude hotwire anemometer	50 Hz	Available in calibration and testing
Wave elevation calibration	3	Wave probe	50 Hz	Removed after calibration
Wave elevation reference	6	Wave probe	50 Hz	Available in calibration and testing
Hull and keel positions	58	Optical tracking camera	50 Hz	Markers in hull and keel are tracked
Fairlead tensions	3	Load cell	50 Hz	
Keel line tensions	6	Load cell	50 Hz	
Tower base bending moment	1	Strain gauge array	50 Hz	Signal DC component not reliable

170 [*]Sampling frequency at model scale.

3 Participants and Mmodeling Aapproach

A total of 17 academic and industrial partners from 10 different countries participated in the OC6 Phase IV project. Those actively involved were Bureau Veritas (BVMO, France), China State Shipbuilding Corporation (CSSC, China), Det Norsk Veritas (DNV, United Kingdom), Technical University of Denmark (DTU, Denmark), Dalian University of Technology
175 (DUT, China), Électricité de France (EDF, France), Gavin & Doherty Geosolutions Ltd (GDG, Ireland), Institute for Energy Technology (IFE, Norway), Maritime Research Institute Netherlands (MAR, The Netherlands), National Renewable Energy Laboratory (NREL, United States of America), Newcastle University (NU, United Kingdom), PRINCIPIA (PRI, France), Shell (SHELL, United States of America), Hamburg University of Technology (TUHH, Germany), Universitat Politècnica de Catalunya (UPC, Spain), and Wood Group Kenny and University of Galway (together referred to as W&UG, Ireland).
180 The system studied in OC6 Phase IV requires a coupled aero-hydro-elastic approach. The participants used modeling approaches of different fidelity for the structural dynamics, aerodynamics, and hydrodynamics. Some participants decided to use more than one modeling approach, and some used different codes. A total of 19 numerical models were involved in this verification and validation campaignstudy. A list of the participants, codes, and the structural approach adopted is provided in Table 32. Most participants included the tower and keel lines flexibility in their numerical models and canould obtain the
185 internal loading. Participants modelleddd the tower using a finite-element model (e.g., by means of a beam theory). Several participants then performed a modal reduction (e.g., Craig-Bampton method (Craig and Bampton, 1968)) to improve the computational efficiency. Different approaches can be adopted to model the keel lines depending on the code capabilities. – Ffor example, from higher to lower fidelity: cable elements, nonlinear springs, linear springs, or slender beams. Cable elements yield forces in tension and account for the proper mass distribution and line sagging. The nonlinear springs
190 approach can potentially reproduce the same stiffness behavior, but the cable mass distribution is not included. Unlike the nonlinear springs approach, the linear springs yield forces in tension and compression (which is unphysical). The slender

beam approach can provide the desired linear axial stiffness under tension and distributed mass. However, it also yields forces at compression and introduces some small stiffness in undesired directions (e.g., shear, bending, and torsion) not seen in the keel lines. Slack events in the keel lines require the use of a nonlinear approach (e.g., cable elements or nonlinear springs). However, potential slack in the keel lines havewas not been observed during the testing (Borg, 2019), and any of the above proposed approaches should provide similar results.

NREL used two numerical models (NREL1 and NREL2) with the only difference being the structural approach adopted. Similar to OC6 Phase II (Bergua et al., 2022a), NREL1 models the tower in the ElastoDyn module and the substructure (i.e., floater) in the SubDyn module while NREL2 models the tower and substructure in the SubDyn module. The SubDyn module of OpenFAST (Jonkman et al., 2023) makes use of a modal reduction. When performing a modal reduction by means of the Craig–Bampton approach for a fixed-bottom system (e.g., system studied in the OC6 Phase II project) in SubDyn, the Guyan modes (also known as boundary modes) provide information about the static deflection of the body and the Craig–Bampton modes (also known as internal or normal modes) provide information about the inherent dynamics. The Guyan modes are obtained with the interface node free, whereas while the Craig–Bampton modes are obtained with the interface node fixed. When computing the Guyan and Craig–Bampton modes, the bottom (reaction node) is fixed or it accounts for the foundation compliance. For a floating system, the formulation is different (see Jonkman et al., 2023 for more details). In this case, the Guyan modes capture the rigid–body motion because there is no reaction node, and the Craig–Bampton modes (interface node fixed) capture the elastic modes. For a floating system, the elastic modes and the applied loads (e.g., gravity acceleration gravitational loading, hydrostatic loading, hydrodynamic loading) in OpenFAST are expressed in a floating frame of reference. Currently, Ffor a floating system in OpenFAST, the hydrostatic and external gravitational loads are It is important to note that the applied loads are computed based on the rigid-body motion (Guyan modes); (rigid body motion for a floating system)the hydrodynamic loads are computed based on the rigid-body and elastic velocities and accelerations. Onl; andy the mooring loads in OpenFAST arare computed based on the combined Guyan and Craig–Bampton modes (i.e., rigid–body motion and elastic deflections in the substructure). For the above reasons, wWhen modeling floating systems in OpenFAST, it is recommended to avoid including in SubDyn bodiesstructural parts in SubDyn that experience significant elastic deflections. Accordingly, the NREL2 model can be considered of lower fidelity than NREL1. Other participants using the OpenFAST code (e.g., DUT1 and SHELL) adopted a similar approach to NREL1. W&UG included the structural part in Flexcom rather than OpenFAST. Flexcom does not perform a modal reduction, andbut instead computes the solution in the time domain using a direct integration method.

Most participants used a dynamic approach to account for the mooring lines in their numerical models (Table 23). The mooring line dynamic approach is usually based on a lumped mass approach where the lines are discretized into concentrated masses connected by massless springs with dampers in parallel. Only four participants (BVMO, DNV, NU, and UPC) used a quasi-static approach. For BVMO, NU, and UPC, the quasi-static approach relies on catenary formulations to compute the mooring line shape and tension at every time step, assuming instantaneous static equilibrium. DNV useds a pre-computed lookup table with the quasi-static fairlead forces based on horizontal and vertical displacements. The main

disadvantage of the quasi-static approach is that it neglects the mooring line inertial forces and the hydrodynamics (e.g., drag and added mass). The quasi-static approach may have difficulties reproducing the proper behavior when the line experiences significant motions or dynamic events (e.g., snap loads when a line becomes slack and it is suddenly under tension again, as observed in the OC5 Phase II project (Robertson et al., 2017)).

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Table 32: Summary of participants, codes, and structural approach

Participant	Code	Structural flexibility		Mooring lines	
		Tower	**Keel Llines**	**Quasi-static**	**Dynamic**
BVMO	Opera		✓	✓	
CSSC	HAWC2	✓	✓		✓
DNV	Bladed	✓	✓	✓	
DTU	HAWC2	✓	✓		✓
DUT1	OpenFAST	✓	✓		✓
DUT2	SIMA	✓	✓		✓
EDF	DIEGO	✓	✓		✓
GDG	OrcaFlex	✓	✓		✓
IFE	3DdFloat	✓	✓		✓
MAR1	aNySIM-XMF	✓	✓		✓
MAR2	ReFRESCO & aNySIM-XMF				✓
NREL1	OpenFAST	✓	✓		✓
NREL2	OpenFAST	✓	✓		✓
NU	DARwind	✓		✓	
PRI	Deeplines Wind	✓	✓		✓
SHELL	OpenFAST	✓	✓		✓
TUHH	panMARE				✓
UPC	FloaWwDyn	✓	✓	✓	
W&UG	Flexcom & OpenFAST	✓	✓		✓

Table 43 provides a list of the participants, codes, and the aerodynamic approach used. Participants in the project used models of different fidelity: blade element momentum (BEM) theory, free vortex wake (FVW) methods, and computational fluid dynamics (CFD). All BEM and FVW models used by participants are based on the lifting line theory. The airfoil polar data provided in the definition document wasere used as input for these numerical models. The BEM approach relies on

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several corrections (e.g., dynamic inflow, skewed wake, blade-root and blade-tip losses, unsteady airfoil aerodynamics) to address the rotor aerodynamics and subsequent loads in different wind turbine operating conditions. Higher–fidelity models like FVW and CFD inherently account for these effects and are better suited to study situations like skewed flow caused by yawed inflow or rotor tilt. Moreover, these higher–fidelity models can provide insights about the wake behavior. MAR2

240 used a blade-resolved CFD approach and ~~it used~~ a surface mesh based on the blade geometry provided by the University of Maine.

The numerical model used by SHELL ~~i~~was the same numerical model ~~as~~used by NREL1. The only difference ~~wa~~is in terms of the aerodynamic model. SHELL use~~s~~d a FVW approach, whereas ~~while~~ NREL use~~d~~s a BEM approach. For wind turbines in operating conditions, it ~~i~~was expected that SHELL and NREL1 would provide different responses due to the different

245 aerodynamic approach. When the wind turbine is in idling or parked conditions, the aerodynamic induction model is disabled, and SHELL and NREL1 models should provide the same response.

Table 43: Summary of participants, codes, and aerodynamic approach

Participant	Code	Aerodynamic approach		
		BEM	**FVW**	**CFD**
BVMO	Opera	✓		
CSSC	HAWC2	✓		
DNV	Bladed	✓		
DTU	HAWC2	✓		
DUT1	OpenFAST	✓		
DUT2	SIMA	✓		
EDF	DIEGO	✓		
GDG	OrcaFlex	✓		
IFE	3DdFloat	✓		
MAR1	aNySIM-XMF	✓		
MAR2	ReFRESCO & aNySIM-XMF			✓
NREL1	OpenFAST	✓		
NREL2	OpenFAST	✓		
NU	DARwind	✓		
PRI	Deeplines Wind	✓		
SHELL	OpenFAST		✓	
TUHH	panMARE		✓	

| UPC | FloaWwDyn | ✓ |
| W&UG | Flexcom & OpenFAST | ✓ |

The TetraSpar floating support structure ~~design~~ is made of slender members. Due to the slender nature of the system, it is possible to use strip theory (e.g., using the Morison's equation ~~(ME)~~) to model the hydrodynamic loading. However, the Morison equation is more suitable for simple geometries, and the different junctions at the end of the braces could challenge the load estimation. For a floating system, it is necessary to account for the relative form of the Morison~~'s~~ equation. Assuming, for the sake of simplicity, that the fluid and structure velocity are colinear and normal to the member, the relative form of Morison equation can be expressed as Eq. (1).

$$F = \frac{1}{2} \cdot C_d \cdot \rho \cdot D \cdot (u_w - u_s) \cdot |u_w - u_s| + C_p \cdot \rho \cdot \frac{\pi \cdot D^2}{4} \cdot \dot{u}_w + C_a \cdot \rho \cdot \frac{\pi \cdot D^2}{4} \cdot \dot{u}_w - C_a \cdot \rho \cdot \frac{\pi \cdot D^2}{4} \cdot \dot{u}_s \tag{1}$$

where F is the force per unit length, u_w is the fluid velocity, u_s is the structure velocity, $\dot{u}_w$ is the fluid acceleration, $\dot{u}_s$ is the structure acceleration, D is the cylinder outer diameter, ρ is the fluid density, C_d is the drag coefficient, C_p is the wave dynamic pressure coefficient (1 for circular members), and C_a is the added mass coefficient. The inertia coefficient (C_m) is related to the added mass and the wave dynamic pressure coefficients as follows: $C_m = C_a + C_p$.

The first term in Eq. (1) corresponds to the drag force (that includes wave excitation forcing and damping), the second term corresponds to the Froude-–Krylov force, the third term is the scattering force, and the fourth term is the added mass component. The combination of Froude-–Krylov and scattering forces can also be referred as the fluid inertia force.

Alternatively, it is possible to study the system by means of the boundary element method based on the potential flow ~~(PF)~~ theory. In general, this method is used for large-–volume structures and assumes small motions around the equilibrium position. The hydrodynamic properties are obtained in frequency-domain in tools like WAMIT (Lee and Newman, 2006), NEMOH (Babarit and Delhommeau, 2015), or HAMS (Liu, 2019) that compute the wave diffraction and radiation for three-dimensional floating structures.

Table 4̶5 shows a comparison between ~~the~~ strip theory and ~~the PF~~potential flow theory. The ~~PF~~potential flow theory does not include viscous drag effects. To overcome this limitation, it is possible to use a hybrid model that accounts for the radiation-diffraction solution from the ~~PF~~potential flow theory augmented with ~~nd~~ Morison-based elements for the drag forces.

Table 5̶4: Comparison between strip and potential flow theories

Physics		Strip theory	Potential flow theory
Drag forces		Constant drag coefficient	None
Inertial forces	Froude-–Krylov	Constant dynamic pressure coefficient	Frequency dependent
	Scattering forces	Constant added mass coefficient	
	Added mass	Constant added mass coefficient	Frequency dependent
Damping	Linear	None	Frequency dependent radiation

	Quadratic	Viscous damping from drag forces	None
forces			
Hydrostatic restoring		Linear or nonlinear	Linear

Table 5~~6~~ provides a list of the participants, codes, and the hydrodynamic approach used. Most participants modeled the TetraSpar system by means of the ~~ME~~Morison equation. Participants using the potential flow~~PF~~ theory, except NU, included the viscous drag component from Morison-based members. Some participants (BVMO) used the linear ~~PF~~potential flow data for radiation and diffraction~~,~~ and added the second-order sum and difference frequency forces. These second-order terms (2nd PF) are nonlinear effects able to excite the floating system out of the wave linear region (covered by the 1st PF). The difference-frequency forces account for the low frequency range, including the mean and slow drift. The sum-frequency forces can excite the floating system above the linear wave region. It is also important to note that the TetraSpar design has two bodies: hull and keel. It is common practice for the potential flow method to assume rigid-body motion and apply the hydrostatic and hydrodynamic loads lumped at a point. However, ~~T~~to get the loads at the keel lines, it is necessary to discretize the system into at least two bodies. Some participants using the ~~PF~~potential flow method included two ~~potential flow~~potential flow bodies in their numerical models while others included the hull as a ~~potential flow~~potential flow body and the keel as Morison elements.

Some participants used a linear hydrostatic stiffness while others used a nonlinear hydrostatic stiffness. Participants using the potential flow theory (NU) and some participants using a hybrid approach (BVMO) included a linear hydrostatic stiffness computed at the undisplaced platform position. Some participants using the hybrid approach (EDF, TUHH) or the strip theory (DUT1, DUT2) computed the nonlinear hydrostatics at the instantaneous platform position up to the mean sea level. Most participants using the strip theory (CSSC, DNV, DTU, GDG, IFE, MAR1, NREL, PRI, SHELL, UPC, W&UG) and CFD (MAR2) computed the nonlinear hydrostatics at the instantaneous platform position up to the wetted surface.

Regarding the wave theory used, some participants used a linear superposition of Airy waves while others also included second-order wave kinematics (Sharma and Dean, 1981). In addition, ~~W~~wave stretching allows for the wave kinematics and hydrodynamic loads to be computed at all nodes up to the instantaneous free surface, unlike linear models without wave stretching, which compute wave kinematics and loads at nodes up to the mean sea level regardless of a wave crest or trough at a given time. Second-order wave kinematics and wave stretching are an extension to the strip-theory solution, and it is only considered by the Morison equation and ~~ME and~~ hybrid models ~~(ME+PF)~~. Participants with Morison-based elements in their models used different wave stretching theories: vertical (EDF, GDG, IFE, NREL, SHELL), extrapolation (W&UG), and Wheeler (CSSC, DNV, DTU, MAR1, PRI, UPC).

Some participants (IFE, MAR1, NREL, PRI, SHELL, TUHH, and UPC) prescribed the wave elevation time series recorded during the experiment in their simulations while other participants prescribed the wave spectrum or used statistical information to generate the waves (potentially with random phasing that d~~oes~~id not match the experiment).

Table 6~~5~~: Summary of participants, codes, and hydrodynamic approach (ME = Morison equation, PF = potential flow, CFD = computational fluid dynamics)

Participant	Code	Hydrodynamics				Wave theory		Wave stretching
		ME	1st PF	2nd PF	CFD	1st order	1st & 2nd order	
BVMO	Opera	✓	✓	✓		✓		
CSSC	HAWC2	✓					✓	✓
DNV	Bladed	✓				✓		✓
DTU	HAWC2	✓				✓		✓
DUT1	OpenFAST	✓				✓	✓	
DUT2	SIMA	✓				✓		
EDF	DIEGO	✓	✓			✓		✓†
GDG	OrcaFlex	✓				✓		✓
IFE	3dDFloat	✓					✓	✓
MAR1	aNySIM-XMF	✓‡					✓	✓
MAR2	ReFRESCO & aNySIM-XMF				✓	✓		
NREL1	OpenFAST	✓					✓	✓
NREL2	OpenFAST	✓					✓	✓
NU	DARwind		✓					
PRI	Deeplines Wind	✓					✓	✓
SHELL	OpenFAST	✓					✓	✓
TUHH	panMARE	✓	✓				✓	✓
UPC	FloaWwDyn	✓				✓		✓
W&UG	Flexcom & OpenFAST	✓				✓		✓

†Only included in irregular wave conditions.

‡Morison-based elements for the hull only account for the drag and Froude-Krylov forces. The added mass, scattering forces, and linear damping are obtained from a hydrodynamic database based on a potential flow solution. It can be considered a hybrid model.

The information provided in Tables 32, 34, and 56 indicates the modeling approach adopted by each participant. Many of tThe codes usedmay have other capabilities not appliedused in this project.

Numerical models were built at full scale, and the results, as well as the measurements and discussion, are presented at full-scale, using Froude scaling to upscale the measurements. MAR2, the only participant using a CFD code, simulated the system at model -scale and upscaled the results using Froude scaling before providing them for comparison. This may allow the CFD approach to capture physics that are scale-dependent.

4 Load cCases

A stepwise validation procedure was performed in the OC6 Phase IV project ~~taking~~that took advantage of the experimental
315 campaign carried out by the University of Maine. The testing campaign used an open-loop control approach, where the rotor
speed and the blade pitch angle were held constant. This configuration disregards potential dynamic effects due to the
interactions between the rotor, the platform motions, and the wind turbine controller that could be present in a closed-loop
approach.

Table 67 provides a summary of the simulations that are presented in Sect~~ion~~. 5, including the system equilibrium (Load
320 Case 1.1), wind-only condition for rated thrust considering the floating system (Load Case 3.1) and a fixed boundary
condition at the tower base (Load Case 3.4), wave-only conditions considering regular waves (Load Case 4.1), wave-only
conditions considering irregular waves (Load Cases 4.2, 4.3, and 4.4), and a combination of wind and wave conditions (Load
Cases 5.2, 5.3, and 5.4). The description provided in Table 67 for the wave-only conditions (e.g., rated, post-rated condition)
corresponds to the representative sea state for a given wind condition.

325 Load case numbering is consistent with past phases of the OC3–OC6 projects. Some numbers are skipped because Table 67
only includes the load cases presented in Sect~~ion~~. 5. The list of all load cases studied can be found in the definition document
(Wiley et al., 2023).

Table 76: ~~Offshore Code Comparison Collaboration, Continued, with Correlation and unCertainty (~~OC6~~)~~ Phase IV load case
simulations (summary)

	Load Ccase	Description	Wind Cconditions	Marine Cconditions	Comparison Ttype
Static Analysis	1.1	Equilibrium	None	Still water	Static response
Wind Only	3.1	Rated wind (floating platform)	Steady wind V_{hub} = 9.89 m s^{-1} Ω = 12.2 rpm, β = -6.2°	Still water	Steady response
	3.4	Rated wind (fixed platform)	Steady wind V_{hub} = 9.89 m s^{-1} Ω = 12.2 rpm, β = -6.2°	Still water	Steady response
Waves Only	4.1	Post-rated condition	None Ω = 0 rpm, β = 0°	Regular waves: H = 8.31 m, T = 12.41 s	Time series (t = 3,934 s)
	4.2	Rated condition	None Ω = 0 rpm, β = 0°	Irregular waves: JONSWAP wave spectrum H_s = 1.46 m, T_p = 6.73 s, γ = 2.3	Time series (t = 10,977 s)
	4.3	Post-rated condition	None Ω = 0 rpm, β = 0°	Irregular waves: JONSWAP wave spectrum H_s = 8.00 m, T_p = 12.20 s, γ = 2.2	Time series (t = 10,977 s)
	4.4	50-y~~ear~~ storm	None Ω = 0 rpm, β = 0°	Irregular waves: JONSWAP wave spectrum H_s = 12.81 m, T_p = 15.79 s, γ = 3.3	Time series (t = 10,977 s)
Wind and	5.2	Rated condition	Unsteady wind V_{hub} = 9.89 m s^{-1}, TI = 2.4 %	Irregular waves: JONSWAP wave spectrum H_s = 1.46 m, T_p = 6.73 s, γ = 2.3	Time series (t = 10,977 s)

	Load Case	Description	Wind Conditions	Marine Conditions	Comparison Type
Waves			Ω = 12.2 rpm, β = -6.2°		
	5.3	Post-rated condition	Unsteady wind V_{hub} = 24.05 m s^{-1}, TI = 2.5 % Ω = 13.3 rpm, β = 18.7°	Irregular waves: JONSWAP wave spectrum H_s = 8.00 m, T_p = 12.20 s, γ = 2.2	Time series (t = 10,977 s)
	5.4	50-year storm	Unsteady wind V_{hub} = 44.62 m s^{-1}, TI = 8.9 % Ω = idling, β = 89°	Irregular waves: JONSWAP wave spectrum H_s = 12.81 m, T_p = 15.79 s, γ = 3.3	Time series (t = 10,977 s)

H: regular wave height
H_s: significant wave height
T: regular wave period
T_p: peak-spectral wave period

γ: peak-enhancement factor
V_{hub}: average hub-height wind speed
TI: turbulence intensity
Ω: rotor speed

β: blade pitch angle
t: time

330

The system equilibrium was studied based on the initial measurements during the testing when the wind or wave loading was not yet applied. During the testing, significant differences in the surge and heave resting positions were observed before applying wind and wave loads, depending on the load case (between -6.7 m and +6.4 m in surge and 4 m range in heave from the origin of the coordinate system used). This introduces a significant uncertainty in the experimental results that may also impact the tension observed in the mooring lines, especially in mooring line 2. It is known that the umbilical used during the testing significantly moved the system downwind (around 31 m compared to the resting position without umbilical during the free-decay tests). This behavior is also replicated in the numerical models when including the umbilical in the system. However, this large surge offset is less than ideal. The umbilical, as well as the friction of the mooring lines with the bottom of the wave tank, could be introducing some hysteresis that may result in the different system resting conditions observed during the testing. The different position of the floating system also impacts the mooring system stiffness. For Load Case 1.1, participants considered still water conditions in their numerical models and reported the static equilibrium of the system.

Load Cases 3.X were used to characterize the wind turbine aerodynamic thrust force. In Load Cases 3.1 and 3.4, the mean wind speed at the hub height was 9.89 m s^{-1}, and the turbulence intensity was 2.4 %. The goal in the testing was to achieve a uniform inflow wind in the rotor area. The wind field generated did not cover the region next to the water (e.g., floater) due to the location of the fans. For Load Cases 3.X, participants considered spatially uniform steady winds and did not account for the aerodynamic drag in the floating support structure. During the testing, the rotor speed was kept constant at 12.2 rpm, and the blade pitch angle was set to -6.2° to match the target aerodynamic rotor thrust using the tower base bending moment as a proxy sensor. Likely, the resultant aerodynamic thrust force during the testing in the wave basin with the blade pitch at 0° was not enough to match the target thrust, and it was decided to decrease the blade pitch angle. For a given wind speed and rotor speed, decreasing the blade pitch angle results in higher angles of attack. For the blade pitch angle of -6.2°, the angles of attack along the blade

(e.g., from 25 % blade span to the blade tip) are mainly in the airfoil polars nonlinear region (e.g., angles of attack between 9º and 14º). This means that the aerodynamic working point is around the stall region.

355 Load Cases 4.X were used to characterize the hydrodynamics of the system. These load cases were wave-only conditions with the system being loaded by waves, no inflow wind, and with a parked wind turbine condition. Load Case 4.1 considered a regular wave for a severe sea state that can be considered representative of the wind turbine operating in the post-rated region. The regular wave height considered was 8.31 m~~.~~, and the regular wave period was 12.41 s. Load Cases 4.2, 4.3, and 4.4 considered irregular waves. Load Case 4.2 can be considered a moderate sea state representative of the wind turbine

360 operating at the rated region while Load Case 4.3 can be considered a severe sea state representative of post-rated conditions. Load Case 4.4 addressed the system response for an extreme storm condition with a 50-year return period. The significant wave height and peak-spectral wave period for these three irregular wave conditions are available in Table ~~6~~7. In the definition of the testing campaign, the Torsethaugen wave spectrum was used. The Torsethaugen ~~wave~~ spectrum is a double-peaked wave spectrum model, best suited to North Sea conditions ~~often features two peak periods~~ (Torsethaugen and Haver,

365 2004)~~.~~,. When computing the wave spectrum from the measured waves, it was observed that the spectrum mainly featured one peak. Given the popularity and availability of the JONSWAP wave spectrum in the codes used by participants, it was decided to also provide the ~~but in this case a single peak was observed in the measurements. Accordingly,~~ peak-enhancement factor ~~factors chosen~~ to reproduce the proper wave elevation distribution ~~by means of a JONSWAP wave spectrum were provided to the participants~~ by means of the JONSWAP spectrum (see Table ~~6~~7). For reference, the definition

370 document (Wiley et al., 2023) shows the comparison between the measured waves spectrum, the Torsethaugen spectrum, and the JONSWAP spectrum. Participants using statistical information to generate the irregular waves could use the JONSWAP or the Torsethaugen spectrum parameters available in the definition document.

Load Cases 5.X dealt with the system response for the combined wind and waves that were studied separately in Load Cases 3.X and Load Cases 4.X. For these load cases, participants considered spatially uniform unsteady winds based on the

375 measured hub-height wind speed in the X-direction.

Testing the 50-year storm condition with wind (~~ei.~~eg., Load Case~~s 3.3 and~~ 5.4) presented some challenges due to ~~a~~ large platform sway and yaw offsets ~~motion of~~in the system. This behavior was not representative of the physical full-scale response. Froude scaling results in low Reynolds numbers and poor aerodynamic performance, often requiring larger chord

380 lengths than the geometrically scaled values would suggest. It may be that the large chord lengths used along the scaled blades resulted in higher-than-expected resultant rotor ~~yaw~~ radial loads in idling conditions. To limit the sway and yaw ~~yaw~~ motion during the testing in the wave basin, ~~the University of Maine added~~ a yaw stiffness bridle was added for these load cases. Additional insights about this arrangement can be found in the definition document (Wiley et al., 2023). When simulating the system with the rotor in idling conditions and the yaw bridle in place, several participants (e.g., DNV, DTU,

385 NREL, W&UG) reported instability issues in their numerical models. The wind turbine experienced a coupled motion that involved the sway, roll, and yaw degrees of freedom. During idling conditions (89° blade pitch angle), it is recommended to

disable the aerodynamic induction model. Moreover, it is also recommended to disable the unsteady airfoil aerodynamics because the angles of attack along the blade are significant, placing the aerodynamic model in deep dynamic stall conditions where the conventional unsteady aerodynamic theory is not valid. The instabilities observed in some numerical models decreased when following these recommended practices, but they did not completely disappear. Some participants (e.g., DNV, DTU, NREL, DNV) decided to impose a constant rotor speed of 0.7 rpm to get rid of this instability. W&UG applied a higher axial stiffness to the yaw bridle (2two orders of magnitude stiffer) to alleviate these effects.

During the testing, the wind speed was measured by means of a hotwire anemometer. The hotwire was located around 1.5 rotor diameters in front of the wind turbine rotor, at one-third of the hub height, and half a blade radius sideways. When looking at the recorded wind speed during Load Case 3.1, it was observed that the mean wind speed was 9.37 m s^{-1} (-5.0 %) instead of the expected 9.86 m s^{-1} from the calibration process performed without the wind turbine (ultrasonic anemometer placed at the hub-height location). Similarly, the recorded mean wind speed in Load Case 5.2 was 10.67 m s^{-1} (+8.2 %). In Load Case 5.3, the measured wind speed also deviated from the expected value (27.18 m s^{-1} (+8.2 %) instead of 24.05 m s^{-1}). Participants in the project used the wind speed values recorded during the calibration process (Table 7) as input for their numerical models.

5 Results

In this section, a comprehensive overview of the studied load cases shown in Table 67 is presented. Results are presented for the aerodynamic loading, floating support structure motion, tower base loading, upwind fairlead tension, and keel line tensions are discussed.

5.1 Aerodynamic loading

Load Cases 3.X focus on ensuring that the aerodynamic models were implemented correctly. During the experimental testing in Load Case 3.1, the blade pitch angle was adjusted to match the rated aerodynamic rotor thrust. The blade pitch angle used during the testing (-6.2°) was also used by the participants, as it returned a similar output. Participants considered steady spatially uniform wind in their numerical models while the experiment containsed a residual amount of turbulence.

Figure 3 shows the aerodynamic rotor thrust, the fFairlead 2 tension, the tower base fore-aft bending moment, and the hull pitch motion. For the experiment, a gray rectangle indicative of two2 times the standard deviation (2σ) is also included.

The aerodynamic rotor thrust was not measured or derived during the testing. The results from the numerical models awere compared with each other as a verification. In addition, the aerodynamic rotor thrust was approximated by means of an analytical approach (moment balance around the tower base location) using the measured hull pitch rotation and the tower base fore-aft bending moment. The analytical approach accounted for the RNA and tower center of mass locations, disregarded the influence from the umbilical (relatively small due to the proximity between tower base and the umbilical

attachment point in the tower) and assumed rigid-body motion. First, the tower base fore-aft bending moment was computed analytically considering the bending contributions from the RNA and tower weight according to the measured hull pitch rotation in resting conditions (close to -2°). The measured tower base bending moment in resting conditions was not used because the mean value (signal DC component) during the testing campaign was not reliable (see Table 2). Then the relative measured tower base bending moment between the resting condition and the loaded condition was considered, as well as the bending contributions from the RNA and tower weight contributions according to the measured hull pitch rotation for the loaded condition (close to 4.8°). The analytical approach returned an aerodynamic rotor thrust equal to 6.03 x 10^5 N for this condition. As observed in Fig. 3, most numerical models tend to slightly overpredict the thrust force. Moreover, it is expected that the real thrust force is slightly lower than 6.03 x 10^5 N when the tower flexibility is considered due to the lever arm distance contributions in tower and especially RNA weights with regards to the tower base.

The fFairlead 2 tension corresponds to the load measured in the upwind mooring line. As observed in Fig. 3, the numerical models predict a lower tension than the experiment. However, the numerical models had excellent agreement when looking at the nonlinear relationship between force and imposed static surge offsets. The higher-than-expected tension for the experiment is likely due to the large surge position of the system in resting conditions before the wind was applied (+6.4 m from the system origin). Regarding the tower base fore-aft bending moment, it was decided a decision was made to analysze relative magnitudes between the resting condition and the loaded system for the reasons aforementionedbecause the mean bending moment during the testing campaign was not reliable. The same postprocessing was applied for the participant results (i.e., the mean value reported for the equilibrium condition in Load Case 1.1 was subtracted from the reported results). The numerical models tend to slightly overpredict the tower base fore-aft bending moment. This is aligned with the observation that the would indicate that the aerodynamic rotor thrust observed byin the numerical models is slightly higher than in the experiment. Similarly, the relative hull pitch rotation is compared in Fig. 3. The hull motion is tracked at its center of mass. The experiment and most numerical models experience a hull pitch rotation of -2° for the equilibrium condition (Load Case 1.1). When the rated wind condition is applied, the relative pitch rotation is close to 7° or 8°. This means that the hull pitch moves from -2° to 5° or 6°. Similar to the tower base fore-aft bending moments, most numerical models predict a slightly larger rotation than the experiment.

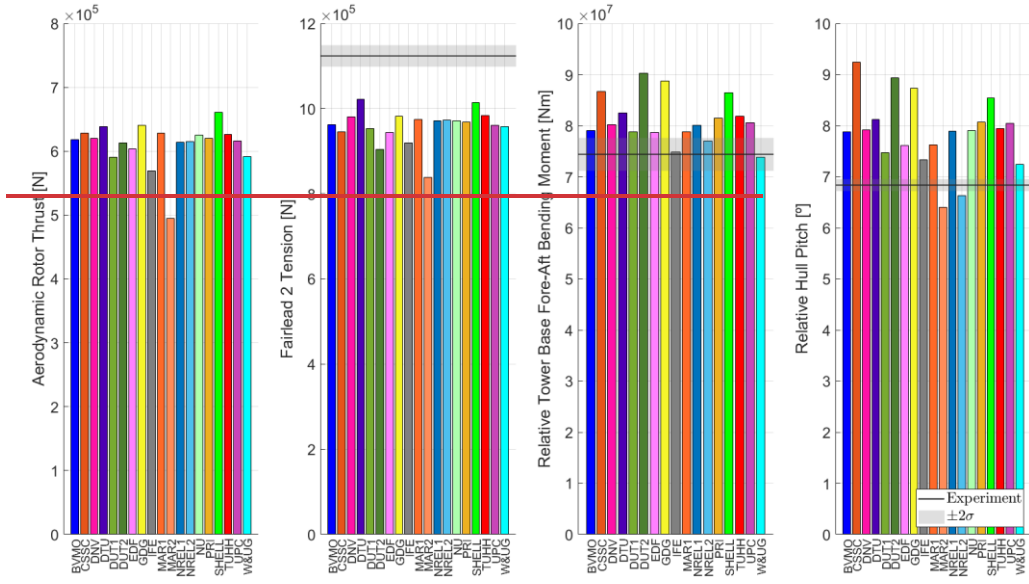
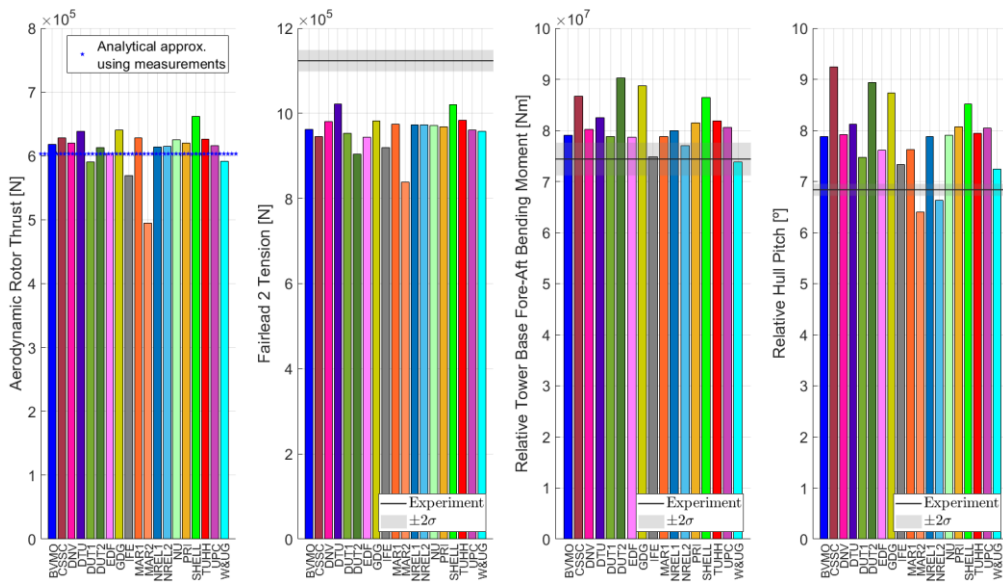



Figure 3: Aerodynamic rotor thrust, ~~f~~Fairlead 2 tension, tower base fore-aft bending moment, and hull pitch rotation for rated wind conditions (Load Case 3.1).

The unusual blade pitch angle (-6.2°) used during the experiment and replicated by the participants, provide~~s~~d an aerodynamic rotor thrust similar to the target ~~one~~thrust for most numerical models based on the tower base bending moment shown in Fig~~.~~ ~~ure~~ 3. However, when looking at the aerodynamic rotor torque (not shown), a negative value was ~~it was~~ observed ~~that the value is negative~~ for the experiment and the numerical models. This means that the wind turbine had to be powered (i.e., system acting as a motor instead of a generator) to maintain the rotational speed of 12.2 rpm.

Figure 4 compares the aerodynamic rotor thrust when considering the floating system (Load Case 3.1) and a fixed boundary condition at the tower base (Load Case 3.4). For reference, Fig~~ure.~~ 4 also includes the median from participant~~s~~ results for the two boundary conditions. The scaled system and the numerical models do not have a wind turbine tilt angle. In Load Case 3.1, the rotor is tilted according to the hull pitch angle (see Fig~~ure.~~ 3) ~~and~~and the tower compliance, whereas ~~while~~ in Load Case 3.4 the rotor is solely tilted according to the tower compliance. Most numerical models predict a very similar aerodynamic rotor thrust for these two boundary conditions.

Some higher-~~ ~~fidelity models (i.e., FVW and CFD) show ~~some~~ sensitivity to the hull rotation. For example, MAR2 (CFD) experiences a significant reduction (-11.8 %) in the aerodynamic rotor thrust for the floating system. This may be related to some physical effects ~~due to~~caused by high angles of attack occurring in the simulation. In those conditions, flow separation at the leading edge and stall may occur~~,~~ ~~leading to~~resulting in lower thrust values. ~~However, i~~It should be noted that the

Reynolds numbers at model- scale are low. Flow separation at large angles of attack and the transition from laminar to turbulent flow may not be captured accurately in the blade-resolved approach at model- scale using the standard k-omega SSTshear stress transport turbulence model. An opposite trend is observed in Fig. 4 for SHELL (FVW). In this case, the rotor tilt due to the hull pitch rotation results in a slightly higher (+3.64 %) rotor aerodynamic thrust compared to the fixed platform condition.

Comparing NREL1 and SHELL (same numerical model with different aerodynamic theory) for the fixed platform condition, it can be observed that SHELL (FVW) results in a slightly higher (+3.58 %) rotor aerodynamic thrust than NREL1 (BEM). This is aligned with the behavior observed in the OC6 Phase III project (Bergua et al., 2022b). This difference is likely due to the slightly different induction factors in the rotor.

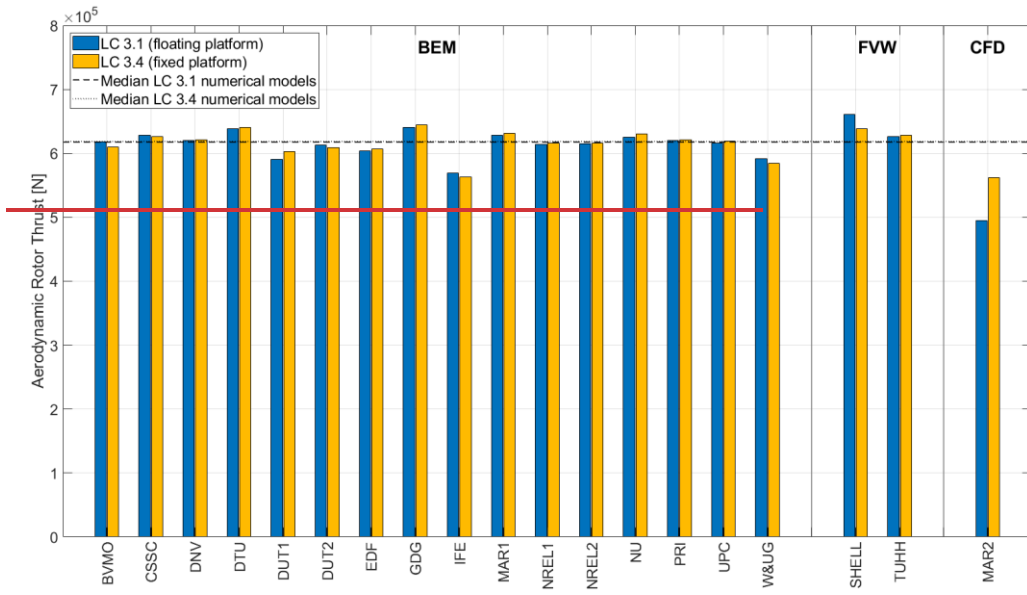
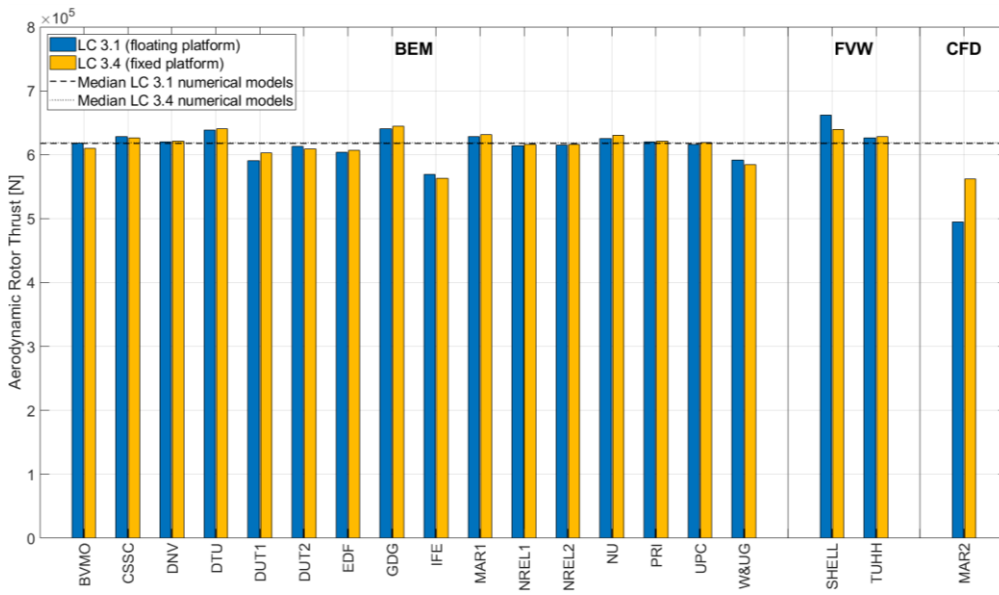



Figure 4: Comparison of aerodynamic rotor thrust for floating (Load Case 3.1) and fixed (Load Case 3.4) conditions. Outputs sorted ~~out~~ according to the aerodynamic theory: blade element momentum (BEM), free vortex wake (FVW), and computational fluid dynamics (CFD).

5.2 Floating sSupport Sstructure ~~M~~motion

One regular wave-only test was used to examine the wave-structure response of the system. A response amplitude operator (RAO) for a regular wave is the ratio between the system motion response amplitude to the wave excitation amplitude at the wave ~~natural~~ frequency. The regular wave studied in the basin is nonlinear (i.e., it contains wave harmonics, and it does not describe a perfect sinusoid)~~.~~ as such, the RAO is computed based on the fast Fourier transform ~~(FFT)~~ amplitude obtained at the wave fundamental ~~natural~~ frequency. Figure 5 shows the normalized ~~RAO~~ response amplitude at the wave frequency for the surge, heave, and pitch degrees of freedom at the hull center of mass~~-~~ in Load Case 4.1 (~~using a regular wave for a~~ severe sea state) ~~(Load Case 4.1)~~. The figure also includes the experimental results from three repeats performed during the testing. Most numerical models tend to slightly overpredict the surge and heave motions, ~~while~~whereas the pitch motion is slightly underpredicted. The CFD approach from MAR2 seems to be able to accurately predict the response in the three directions. Participants initially tuned the hydrodynamic coefficients (i.e., drag and inertia coefficients for ~~ME~~Morison equation models and drag coefficients for the hybrid models) based on free decays in still water. However, it is expected that the ~~viscous drag~~

hydrodynamicviscous damping –is dependent on the sea state (Pegalajar-Jurado et al., 2019). Nevertheless, participants in this project used the same hydrodynamic coefficients for all the cases studied.

NREL performedBased on a sensitivity analysis performed over the MEorison equation approach and observed that , it was determined that the hull motion response experienced in Load Case 4.1 was mainly driven by the hydrodynamic inertia coefficientmponent. Higher hydrodynamic drag coefficients resulted in slightly higher damping for the system, but the model was not very sensitive. The most slender members (e.g., diagonal braces) have contributions from both drag and inertia, and while the largest members (e.g., central column) are dominated by inertia. These observations are also aligned with previous studies on the TetraSpar design (Thomsen et al., 2021). Members near the mean sea level (i.e., central column and diagonal braces) drive the system response, as the magnitude of the wave kinematics areis strongesthigher in that region and decay exponentially with increasing depth. For reference, most numerical models based on the Morison Eequation approach used an inertia coefficient equal to 2. This value is also in good agreement with previous studies on the TetraSpar design (Thomsen et al., 2021).

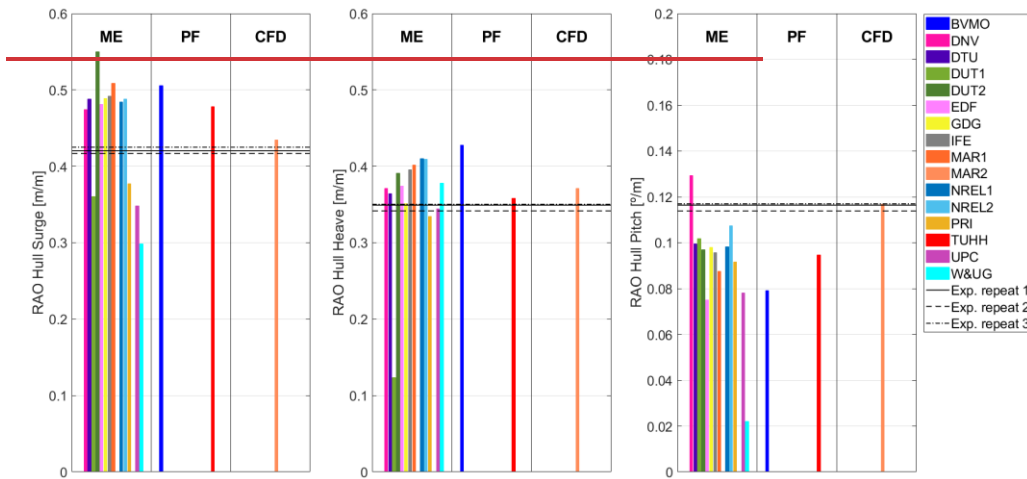
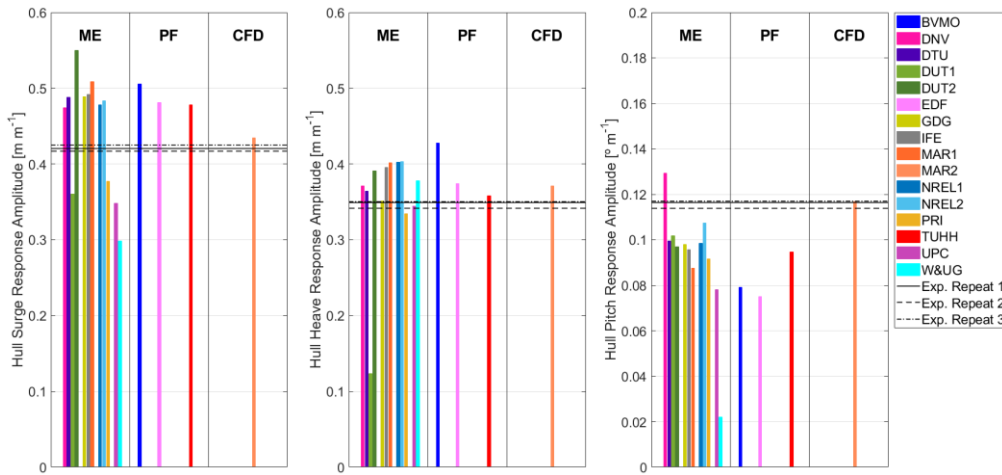
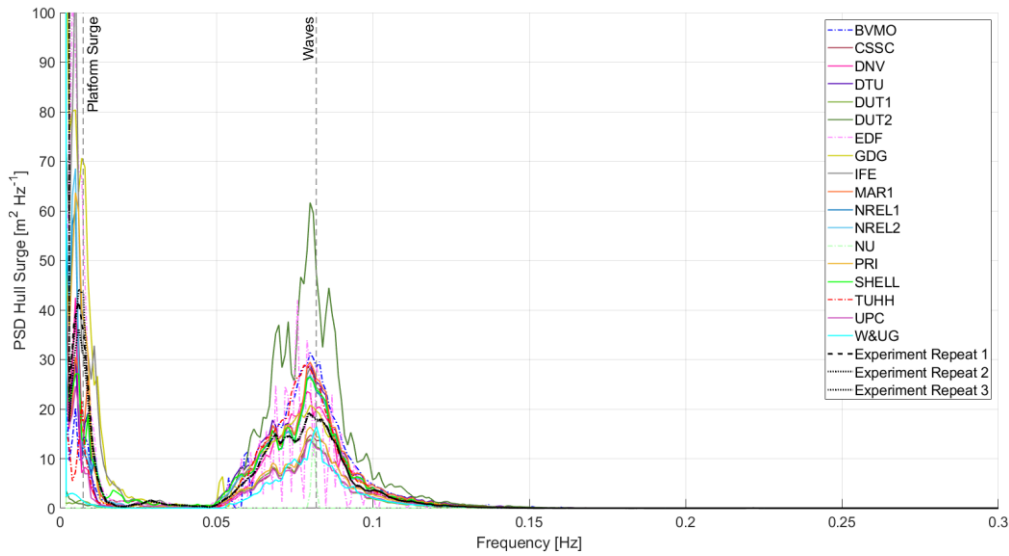
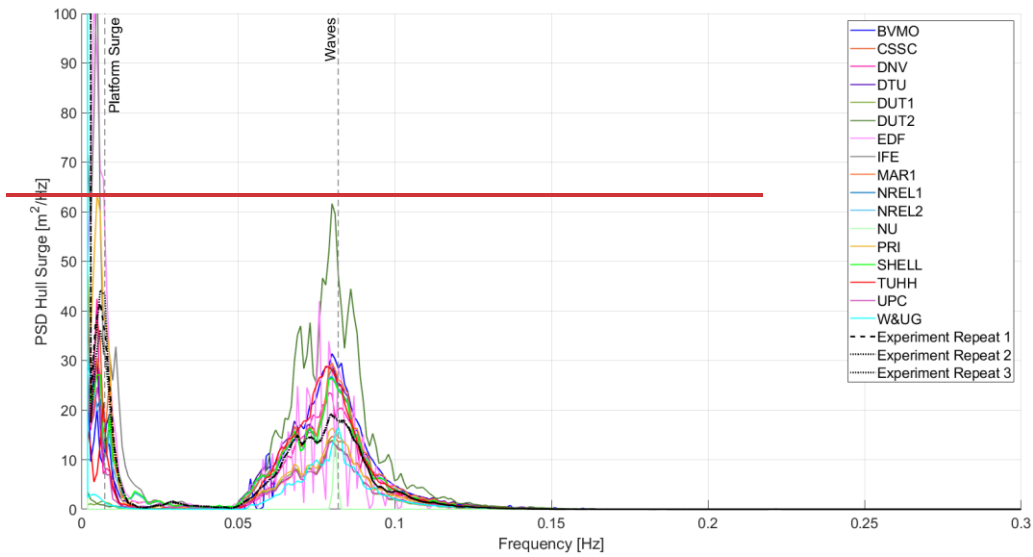



Figure 5: Normalized response amplitude at the wave frequency in Load Case 4.1 ~~Response amplitude operators (RAO)~~ for the surge, heave, and pitch degrees of freedom ~~for a regular wave with a height of 8.31 m and period of 12.41 s (Load Case 4.1). O.~~ Outputs sorted ~~out~~ according to the hydrodynamic theory: Morison equation (ME), potential flow (PF) augmented with viscous drag, and computational fluid dynamics (CFD).

Figure 6 shows the power spectral density (PSD) of the hull surge motion for the combined unsteady wind and wave condition in Load Case 5.3. Participants using a Morison equation approach are denoted with a solid line, and participants using a potential flow approach or potential flow augmented with viscous drag are denoted with a dash-dotted line. Similar patterns are used in Fig. 10 and Fig. 14. The significant wave height and peak-spectral wave period considered in Load Case 5.3 ~~is~~are comparable in magnitude to the regular wave analy~~z~~sed in Load Case 4.1. In the very low--frequency range, rigid--body (slow drift) motion activity can be observed around the platform surge eigenfrequency. The response at this frequency is driven by the wind excitation and nonlinear hydrodynamic forces able to excite the system outside the linear wave excitation region (e.g., second-order terms in the ~~PF~~potential flow solution or viscous drag forces, especially when including second-order wave kinematics and wave stretching ~~for the numerical models using Morison-based elements~~). In this case, the impact of the wave stretching is significantly larger than the impact of including second-order wave kinematics. Participants using the vertical wave stretching theory obtained the largest slow drift response. This was also observed by NREL when comparing the response between vertical and Wheeler wave stretching theories. Figure 6 ~~also~~ shows a group of participants (DUT1, DUT2, NU, and W&UG) with a very small response around the platform surge frequency. These participants defined a steady wind condition and~~,~~ therefore~~,~~ do not include the wind excitation ~~due~~from ~~to~~ the wind turbulence in the very low--frequency region. For NU, using a steady wind condition together with aonly linear potential

flow data 1st PF approach for the hydrodynamics, resulteds in a lack of system response in the platform surge eigenfrequency.

For the linear wave region (around 0.08 Hz in Figure. 6), the results are aligned with the response observed for the regular wave condition. For example, participants slightly overpredicting the response in the regular wave condition (Figure. 5) are also showing similar trends for the irregular wave condition (Figure. 6). Figure 6 also shows a group of participants (DUT1, DUT2, NU, and W&UG) with a very small response around the platform surge frequency. These participants defined a steady wind condition and, therefore, do not include the wind excitation due to the wind turbulence in the very low frequency region. For NU, using a steady wind condition together with a 1st PF approach for the hydrodynamics, results in a lack of system response in the platform surge eigenfrequency.

535 **Figure 6: Power spectral density (PSD) of the hull surge motion for the combined wind and wave condition in Load Case 5.3. Solid lines denote participants using a Morison equation approach, and dash-dotted lines denote participants using a potential flow approach or potential flow augmented with viscous drag.**

For the rated and post-rated conditions, the turbulence intensity during the testing was very small (see Table 7). The waves were the main excitation of the system. To compare the system response between participants and against the measurements, 540 the PSD can be integrated ~~(labelled in this paper as a PSD sum)~~ for the linear wave excitation region. This metric is equivalent to the variance (standard deviation squared~~)~~), and it is related to fatigue loading over th~~e~~at frequency range ~~of interest and it is related to fatigue loading~~. Table 7~~8~~ shows the lower and upper cut-off frequencies for the three irregular waves considered in Load Cases 4.X and 5.X.

Table 8~~7~~: Lower and upper cut-off frequencies for the linear wave excitation region to compute the PSD ~~sum~~integral
545

Load ~~C~~case	Linear wave excitation region	
	Lower frequency [Hz]	Upper frequency [Hz]
4.2 & 5.2	0.060	0.19
4.3 & 5.3	0.046	0.15
4.4 & 5.4	0.040	0.15

Figure 7 shows the hull surge PSD ~~sum~~integral for the different numerical models categorized according to the hydrodynamic theory used (e.g., M~~orison E~~equation, ~~PF~~potential flow). In this case there are no outputs available from the CFD model (MAR2). Figure 7 also includes the results from three repeats performed during the testing. Similar to Load Case 4.1, the system response in the wave linear region in Load Cases 5.2, 5.3, and 5.4 is governed by the hydrodynamic 550 inertia component. In these load cases the hydrodynamic drag component has minor effects in the response. In Load Case 5.2, the hydrodynamic drag slightly increases the wave loading excitation, while in Load Cases 5.3 and 5.4, it provides some damping to the system. All participants using the ~~PF~~potential flow approach, except NU, accounted for the viscous drag contribution by means of Morison-based members. However, the different behavior observed in Fig~~ure.~~ 7 for NU is not due to the lack of viscous drag but rather something not properly set up in the numerical model.

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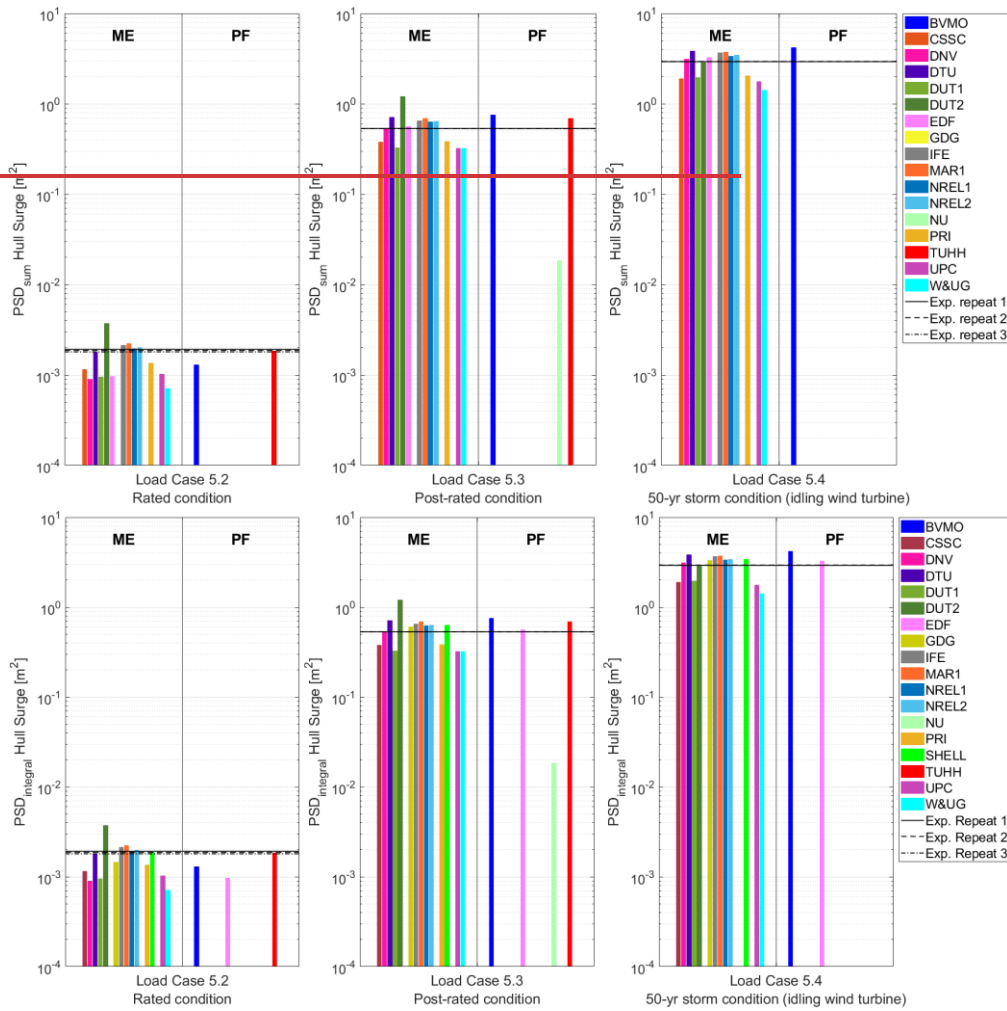
Figure 7: Power spectral density (PSD) sumintegral inof the hull surge motion for the combined wind and wave conditions in Load Cases 5.2, 5.3, and 5.4. Outputs sorted out according to the hydrodynamic theory: Morison equation (ME) and potential flow (PF). Vertical axis in logarithmic scale.

560 Figure 8 and Fig~~ure.~~ 9 provides the PSD ~~sum~~integral for the hull heave and hull pitch motions, respectively. Similar to the regular wave condition (Fig~~ure.~~ 5), most participants tend to underpredict the hull pitch motion when considering the irregular wave conditions (Fig~~ure.~~ 9).

As Fig~~ure.~~ 7, Fig~~ure.~~ 8, and Fig~~ure.~~ 9 show~~s~~, Load Case 5.4 (50-y~~ea~~r storm) results in the largest dynamic loading in the linear wave region and Load Case 5.2 (rated condition) in the smallest dynamic loading. As expected, the PSD ~~sum~~integral is

565 larger for the higher waves.

Although not shown, the system response in the wave linear region is very similar between the conditions without wind (Load Cases 4.X) and with wind (Load Cases 5.X) with only some relatively small differences in the pitch degree of freedom.
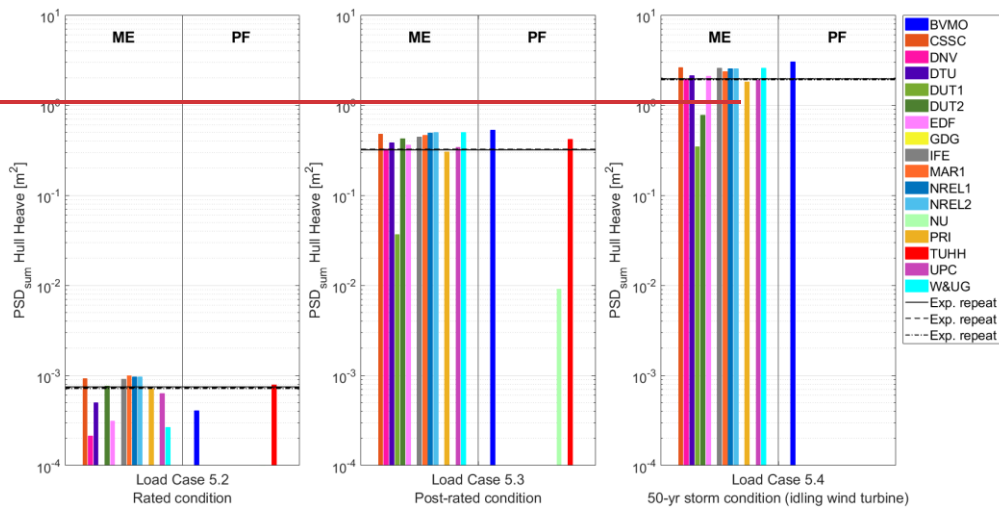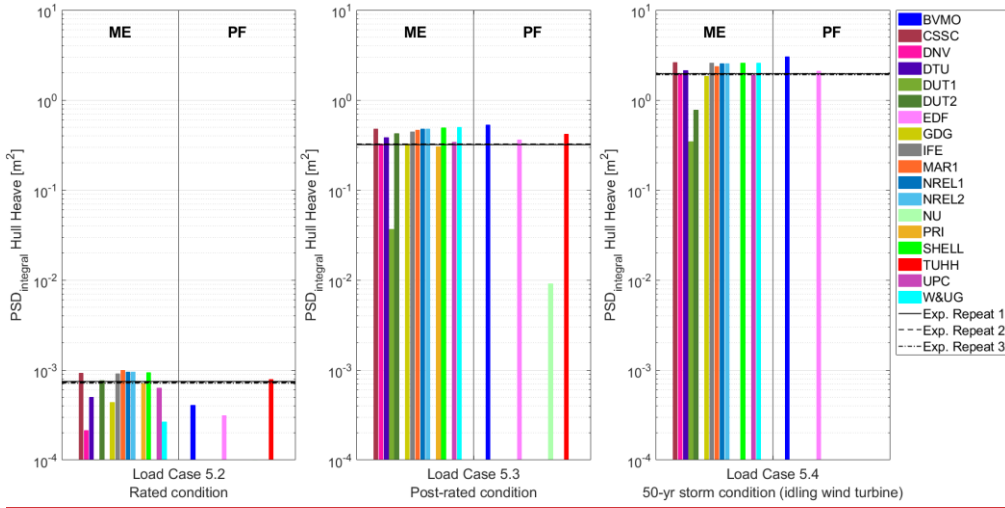
Figure 88: Power spectral density (PSD) suintegralm inof the hull heave motion for the combined wind and wave conditions in Load Cases 5.2, 5.3, and 5.4. Outputs sorted out according to the hydrodynamic theory: Morison equation (ME) and potential flow (PF). Vertical axis in logarithmic scale.
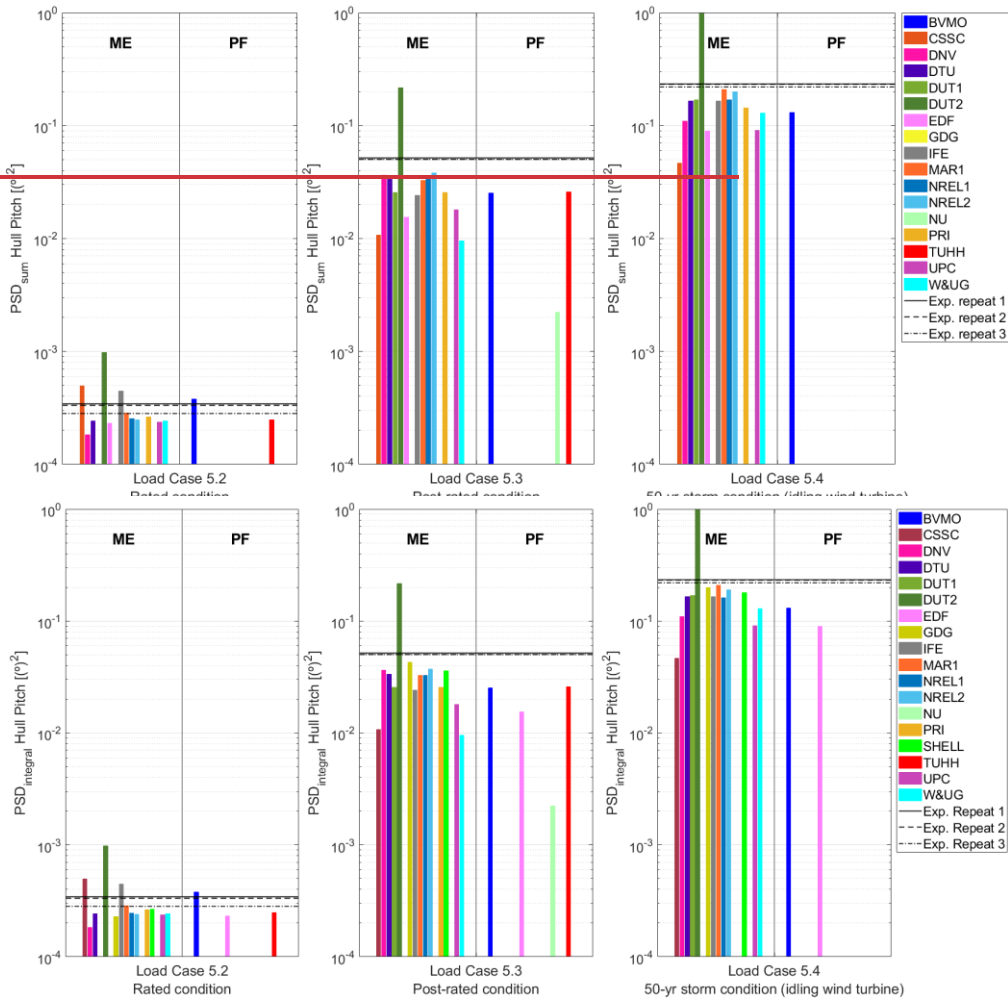
Figure 9: Power spectral density (PSD) ~~sum~~integral ~~in~~of the hull pitch motion for the combined wind and wave conditions in Load Cases 5.2, 5.3, and 5.4. Outputs sorted ~~out~~ according to the hydrodynamic theory: Morison equation (ME) and potential flow (PF). Vertical axis in logarithmic scale.

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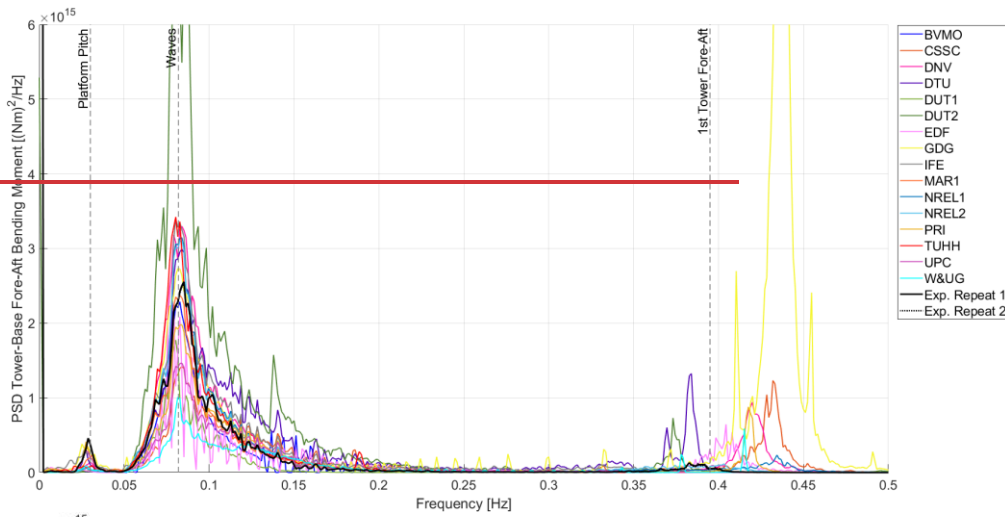
5.3 Tower ~~Bb~~ase ~~Ff~~ore-~~a~~Aft ~~b~~Bending ~~Mm~~oment

The fore-aft and side-to-side bending moments at the tower base were measured during the testing. The wind and wave load excitations occur at a frequency significantly lower than that of the first tower ~~-~~bending mode. Figure ~~6~~10 shows the PSD of the tower ~~-~~base fore-aft bending moment for an irregular wave-only condition (Load Case 4.3).

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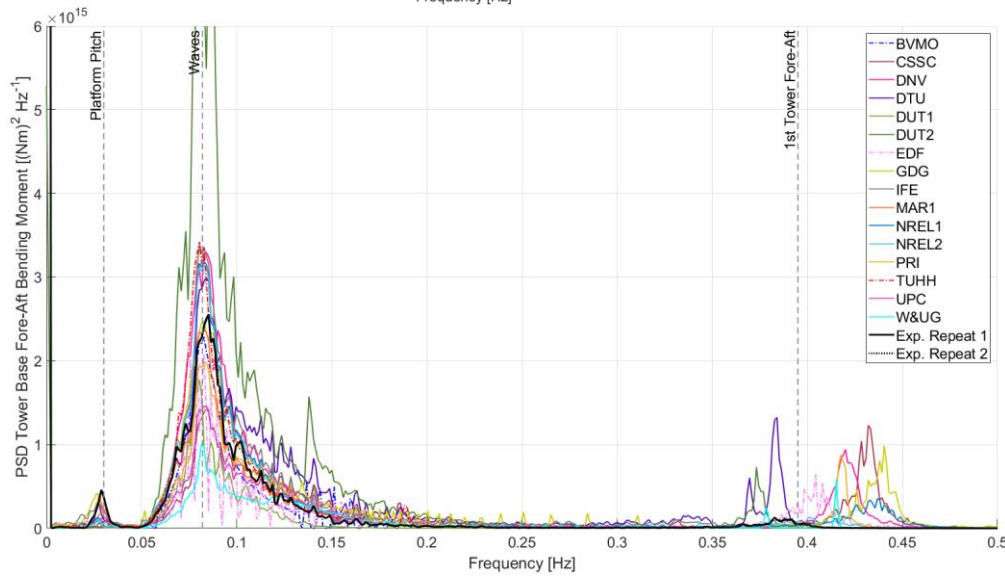
Figure 10: Power spectral density (PSD) of the tower- base fore-aft bending moment for the wave-only condition in Load Case 4.3.

For the experiment, the first tower fore-aft bending mode in free-floating conditions occurs at around 0.39–0.40 Hz. Most numerical models properly reproduced the expected frequency for the first tower-bending eigenfrequency for the
590 cantilevered condition (0.34–0.35 Hz). However, when moving to the free-floating conditions, most numerical models tend to slightly overestimate the tower-bending eigenfrequency.

Figure 10 also shows some Morison-only models overestimating the response at the first tower-bending mode. The Morison equation is only valid for diameter-to-wavelength ratios smaller than 0.2 (Faltinsen, 1993). Otherwise, finite-wavelength diffraction effects become significant. To avoid this, some participants (NREL, UPC) used the MacCamy and Fuchs
595 diffraction correction of the inertia coefficient ~~(C_m) can be applied~~ (MacCamy and Fuchs, 1954). For the TetraSpar design studied in this project, the largest diameter (4.3 m) corresponds to the central column, hull tri–braces, and keel tri–braces. Considering deep-water conditions and the largest diameter being 4.3 m, the Morison equation without any corrections would overestimate the loading above 0.27 Hz. Some participants (DUT2, GDG, NREL) also ~~Alternatively, if this capability is not available in the code used,~~ applied a low-pass filter ~~to~~over the irregular wave spectrum ~~could be used~~. According to the
600 recommended practice DNV-RP-C205 (Det Norske Veritas, 2010), when using a measured input record, it is advised to use a low-pass filter with a cutoff frequency equal to 4 times the peak-spectral wave frequency. This low-pass filter cutoff frequency is 0.33 Hz in Load Case 4.3.

When using Wheeler wave stretching and the MacCamy and Fuchs correction, the response from the NREL1 model was aligned with the experiment around the first tower eigenfrequency. However, when vertical wave stretching was used
605 (shown in Fig. 10), the response at the first tower-bending mode eigenfrequency was significant (despite using the MacCamy and Fuchs correction). This significant response was observed with and without second-order wave kinematics. ~~For the TetraSpar design studied in this project, the largest diameter (4.3 m) corresponds to the central column, hull tri braces, and keel tri braces. Considering deep-water conditions and the largest diameter being 4.3 m, the Morison equation without any corrections would overestimate the loading above 0.27 Hz.~~

610 Figure 11 shows the PSD ~~sum~~integral for the different numerical models categorized according to the hydrodynamic theory used (e.g., M~~o~~Eorison equation, ~~PF~~potential flow). In this case there are no outputs from the CFD model because the numerical model from MAR2 does not ~~account~~s for structural flexibility, ~~for rigid body motion~~ and the loading at the tower base is not available. Figure 11 also includes the results from two repeats performed during the testing.

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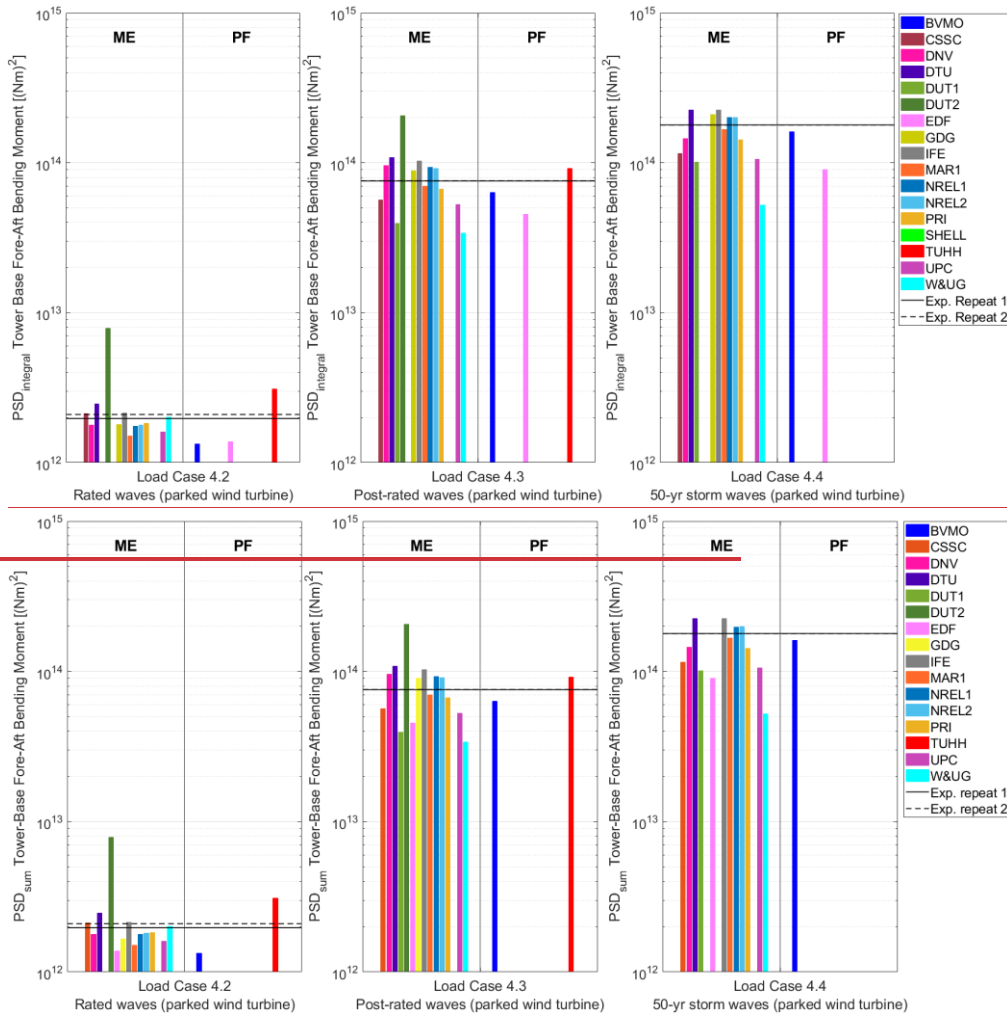
Figure 11: Power spectral density (PSD) ~~sum~~integral ~~in~~of the fore-aft tower -base bending moment for the wave-only conditions in Load Cases 4.2, 4.3, and 4.4. Outputs sorted ~~out~~ according to the hydrodynamic theory: Morison equation (ME) and potential flow (PF) with augmented viscous drag. Vertical axis in logarithmic scale.

5.43 Upwind Ffairlead Ttension

In slack catenary mooring lines (i.e., when part of the line rests on the seabed), the vertical component of the fairlead mean tension is equal to the suspended line net weight, and the total mean ~~the fairlead mean~~ tension is determined by the angle described at the fairlead connection.~~suspended line net weight.~~ When an external force is applied over the wind turbine, the floating system experiences a motion that changes the suspended weight at each line. This results in a new fairlead orientation and tension ~~relative to the unloaded state~~. By projecting the three fairlead tensions in the horizontal plane, the resultant force that opposes the external forces over the system in that plane could be determined.

Figure 12 shows the fairlead tension for the three mooring lines in the system equilibrium condition (Load Case 1.1). As Fig~~ure~~g. 12 shows, the fFairlead 2 tension is significantly higher (+80 %) than the tension in the other two fairleads. The reason is that the umbilical ~~cable~~ pulls the system significantly downwind, loading mMooring lLine 2 and unloading mMooring lLines 1 and 3. This behavior is properly reproduced by the numerical models. Figure 12 also shows a large dispersion for the fFairlead 2 measurement ~~as a consequence of~~due to the different system resting positions experienced during the testing.
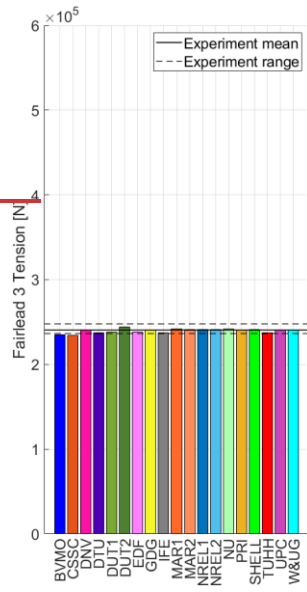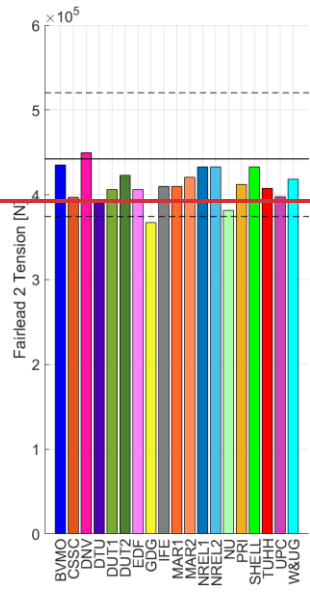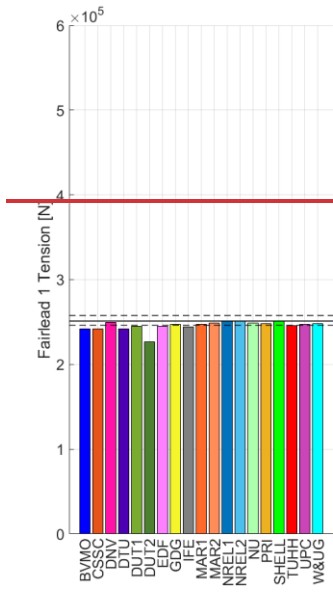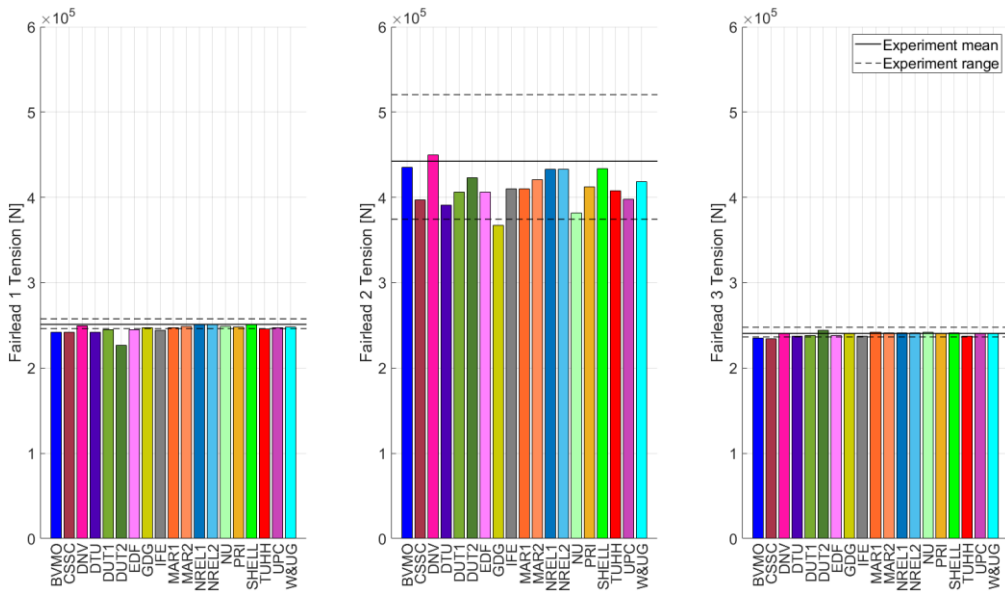
Figure 12: Fairlead tension in the three mooring lines for the system equilibrium condition (Load Case 1.1).

In Load Cases 5.X, the system is loaded by wind and waves. For these conditions, the new fairlead tensions are mainly driven by the aerodynamic thrust force that translates into a platform surge offset. In Load Cases 5.X, the wind is applied along the X-direction. This results in a significant increase in the fFairlead 2 tension (located upwind), whereas while the fairlead tension in lines 1 and 3 (located downwind) tend to slightly decrease. Figure 13 shows the mean fairlead tension for mMooring lLine 2 in Load Cases 5.2, 5.3, and 5.4. As expected, Load Case 5.2 (rated condition) results in the highest fFairlead 2 mean tension because the aerodynamic rotor thrust is at a maximum.

For themost numerical models, the fFairlead 2 mean tension in Load Case 3.1 (steady wind-only condition) shown in Figure. 3 is similar or slightly lower aligned withthan the fFairlead 2 mean tension in Load Case 5.2 (combined unsteady wind and waves) shown in Figure. 13. This is the expected behavior and confirms that the fFairlead 2 mean tension is mainly driven by the aerodynamic loading. Interestingly, the experiment shows lower Ffairlead 2 mean tensions in Load Case 5.2 (see Figure. 13) compared to Load Case 3.1 (see Figure. 3) and certain discrepancypersion between the repeats during the testing. It is known that during Load Case 5.2, the wind in the wave basin was higher than expected (10.67 m s⁻¹) and significantly higher than the wind measured in Load Case 3.1 (9.37 m s⁻¹). Despite this, the experiment shows larger Fairlead 2 mean tensions in Load Case 3.1. This unexpected behavior between Load Case 3.1 and Load Case 5.2 may be related, once again, may be related to the uncertainty in the system resting position between tests.

Figure 13 also shows that in Load Case 5.2, the Fairlead 2 mean tension observed during the testing is higher than for most of the numerical models. This indicates that during the testing, the platform experienced a higher mean surge position when the system was loaded. This could also be related, in part, to the higher mean wind speed observed during the testing (10.67 m s^{-1}) compared to the mean wind speed used by the participants (9.86 m s^{-1}).

The experiment shows similar Fairlead 2 mean tension in Load Case 5.3 (post-rated condition) and Load Case 5.4 (50-year storm condition) while the numerical models tend to have slightly smaller Fairlead 2 mean tension in Load Case 5.4. For reference, during the storm with the wind turbine in idling conditions, the rotor drag is similar to the tower aerodynamic drag. It may be that the umbilical induces some additional aerodynamic drag force that becomes significant due the high wind speed. Unfortunately, there was no information about the umbilical nominal diameter, and the aerodynamic contribution was not accounted for in the numerical models.

As expected, there are no differences between quasi-static or dynamic mooring lines to estimate the mean line tension.
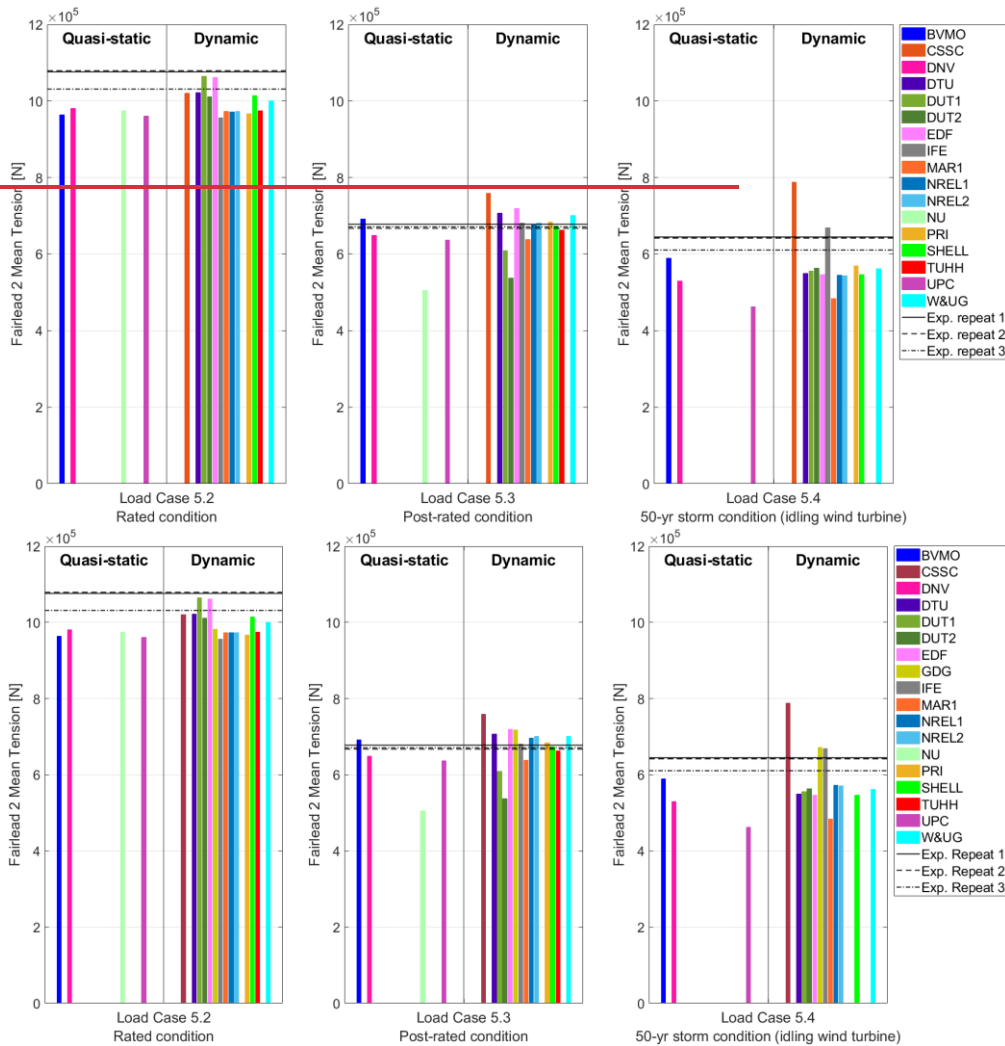
Figure 13: Fairlead 2 mean tension for the combined wind and wave conditions in Load Cases 5.2, 5.3, and 5.4. Outputs sorted ~~out~~ according to the mooring line theory: quasi-static and dynamic.

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During dynamic conditions, the line load variations are determined by the system motion and the corresponding suspended line weight, inertial loads, hydrostatic loads, hydrodynamic loads of the line moving through the water, and hydrodynamic

670 loads due to the wave kinematics. ~~For example, higher hydrodynamic drag and added mass coefficients along the mooring lines would result in higher loads at the fairlead connections.~~ Quasi-static mooring line approaches only capture changes in the suspended line weight, missing the inertial and hydrodynamic load contributions. Therefore, the quasi-static approach should tend to underpredict the dynamic line loads as observed in OC5 Phase II project (Robertson et al., 2017).

Figure 14 shows the PSD of the ~~f~~Fairlead 2 tension for the combined wind and wave condition in Load Case 5.3. Participants

675 using a dynamic approach for the mooring lines are denoted with a solid line, and participants using a quasi-static approach are denoted with a dash-dotted line. In the very low frequency range, activity can be observed around the platform surge, heave, and pitch frequencies. For the linear wave region, large differences between numerical models are observed.

The response from DUT2 shows two peaks around two and three times, respectively, the peak-spectral wave frequency.

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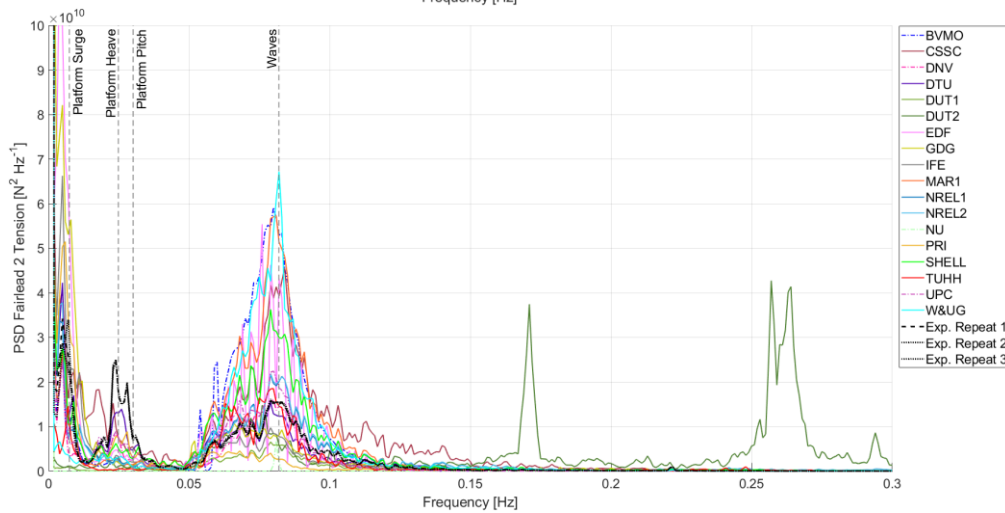
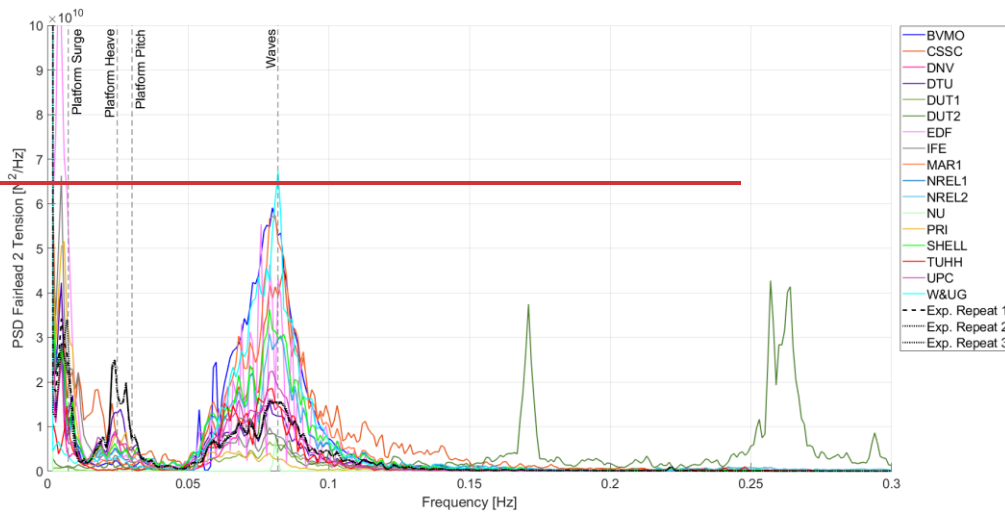


Figure 14: Power spectral density (PSD) of the fairlead 2 tension for the combined wind and wave condition in Load Case 5.3.
~~Solid lines denote participants using a dynamic approach for the mooring lines, and dash-dotted lines denote participants using a quasi-static approach.~~

685   Figure 15 shows the PSD integralsum for the different numerical models categorized according to the mooring line theory used (e.g., quasi-static and dynamic). Figure 15 also includes the results from the three repeats performed during the testing.

As anticipated in Fig~~ure.~~ 14, there is a significant spread in the outputs from the participants. This indicates that the dynamic loading in the mooring lines driven by the platform motion due to the incoming waves is different between the participants. This might be due to a different platform motion (see Fig~~ure.~~ 7, Fig~~ure.~~ 8, and Fig~~ure.~~ 9) or ~~due to~~ a different line response for a similar platform motion. It is also important to note that different platform mean surge positions (influenced by the umbilical cable) will result in a different suspended line ~~weight~~length and orientation, potentially impacting the dynamic loading.~~inertial loading in the mooring lines and resulting in different dynamic forces in the wave region.~~ For example, in Load Cases 5.2 and 5.4, Experiment ~~r~~Repeat 3 ~~during the experiment in Load Cases 5.2 and 5.4 returns~~experienced a smaller platform surge offset that resulted in ~~a~~ lower mean ~~f~~Fairlead 2 tensions than ~~for the~~ for other two repeats~~-~~ (see Fig~~ure.~~ 13). When looking at the corresponding dynamic loading in the wave region (see Fig. 15), it can be observed that ~~This is because repeat 3 during the testing experienced a smaller mean platform surge offset. This lower platform offset results in a smaller suspended line length and, therefore, weight, which in turn is likely to induce a lower inertial load. This can be observed in the lower dynamic loading for the e~~Experiment ~~r~~Repeat 3 returns lower dynamic loading in the wave region for the load cases that experienced the smaller surge offsets ~~shown in Figure 15 for~~ (Load Cases 5.2 and 5.4). Future work could include prescribing the platform six degrees of freedom in the numerical models to determine if the mooring line responses are similar when the platform motion is imposed. It is important to ~~note~~highlight that, in the numerical models, the mooring line~~s~~ lengths of the upper section were ~~corrected~~modified to provide the expected tensions in the static surge offsets (Wiley et al., 2023). The lengths were extended between 12 % and 18 %. It may be that these ~~modifications~~changes were not representative of the physical system.

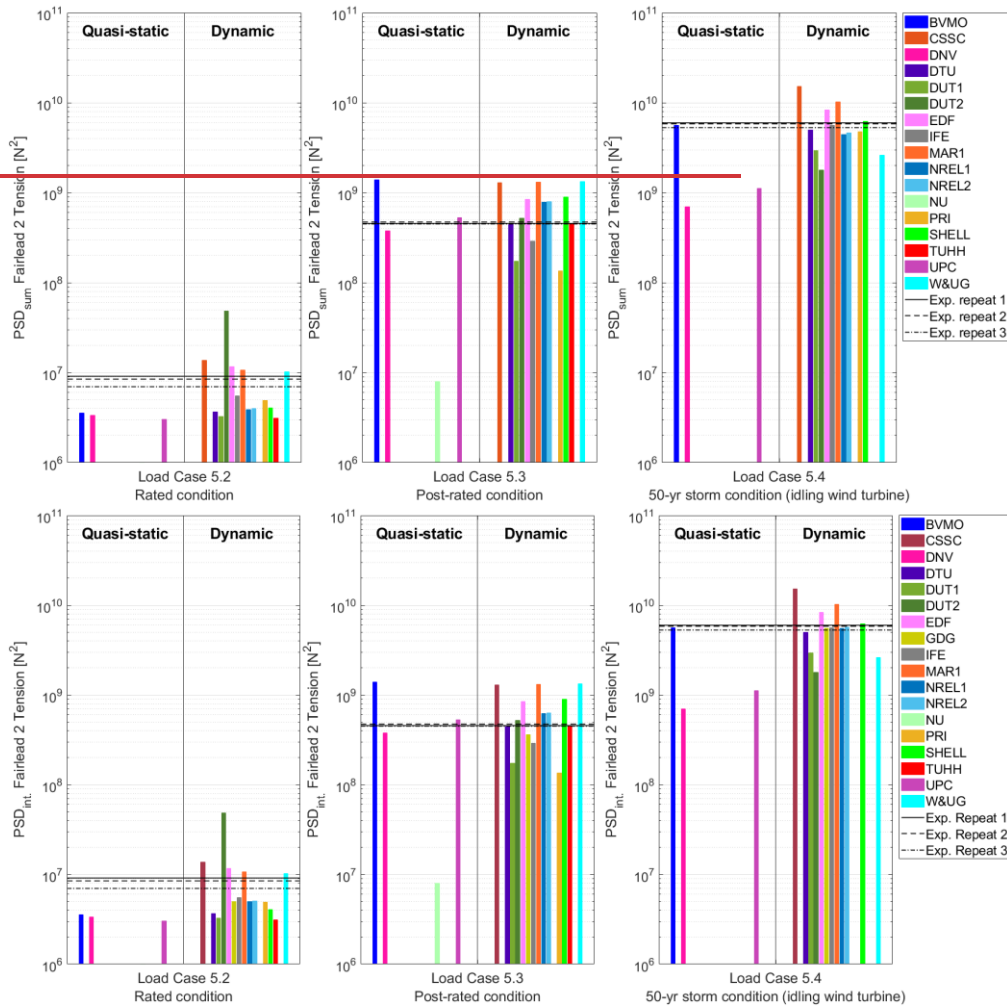



Figure 15: Power spectral density (PSD) integral<s>sum in</s>of the fFairlead 2 tension for the combined wind and wave conditions in Load Cases 5.2, 5.3, and 5.4. Outputs sorted <s>out</s> according to the mooring line theory: quasi-static and dynamic. Vertical axis in logarithmic scale.

5.~~5~~4 Keel ~~L~~line ~~T~~tensions

The suspended keel acts as a counterweight providing floating stability to the system. To ensure this ~~floating~~ stability, the six keel lines must always remain under tension. In these conditions, it is assumed that the two floating support structure bodies (hull and keel) behave as a rigid body. For the load cases studied within OC6 Phase IV project, the experimental data and the response of the numerical models confirm that the two bodies predominantly describe a rigid-body motion. The axial stiffness of the keel lines introduces some flexibility in the floating support structure assembly that results in small relative motions, but the difference in the pitch angle between hull and keel is smaller than 0.8º in all studied conditions. The maximum relative rotation between the two bodies was observed during the 50-year storm condition (Load Case 5.4) in the experiment. Most numerical models experience significantly smaller relative motions between hull and keel. This may denote that the keel line axial stiffness used as input for the numerical models is too high or that there is some uncertainty in the motion tracking system used during the testing.

The keel line system is statically determinate. The keel lines withstand the net keel weight (i.e., the difference between the weight of the keel body acting vertically downward~~s~~ and the hydrostatic buoyancy force acting upward~~s~~). The mean tension at each keel line can be calculated analytically based on the static equilibrium equations (Pereyra, 2018). The keel lines' mean tension distribution is determined by the roll and pitch rotations of the floating support structure. These rotations are mainly driven by the RNA overhang and the aerodynamic loading. For the studied system, the static platform pitch limit where a keel line would become slack is close to 26°. Useful insights can be obtained by performing a parametric analysis of the floater with the analytical model. For example, the keel mass does not change the static platform pitch limit. This is aligned with the observations from previous studies (Pereyra, 2018). By applying scaling factors at the three vertices for the hull base and keel (see Fig. 2c), it can be observed that the static platform pitch limit is only sensitive to the line attachment points at the hull side. Wider spacing between the hull vertices increases the platform pitch limit. The other parameter determining the static platform pitch limit is the keel center of mass location with regard to the hull. Shorter distances between the hull and the keel would increase the platform pitch limit. However, shorter distances between hull and keel would result in smaller gravitational restoring moments. In general, longer distances between hull and keel ensuring that all keel lines remain under tension are preferred. It is also important to note that dynamic variations in the line tension may reduce the maximum platform pitch angle allowed.

Figure 16 shows the tension for each keel line in the equilibrium condition (Load Case 1.1). The disposition of the subplots follows the physical location of the keel lines shown in Fig~~ure.~~ 2b and Fig~~ure.~~ 2c. The subplots include the measured mean tension as well as some small variations that were observed during the testing. Figure 16 also includes for reference the computed tension for each line based on the analytical approach, assuming no platform roll and a platform pitch of -2°. This rotation corresponds to the equilibrium conditions observed during the testing and is reproduced by most numerical models. There are some differences between the tensions obtained in the experiment and the ones from the analytical solution. This denotes that some physical properties in the system may be slightly different.

For the equilibrium condition, the disposition of the floating support structure together with the nacelle oriented along the negative -X direction results in a symmetric system with respect to the XZ vertical plane (see Figure. 2a and Figure. 2bc for reference). This symmetric nature of the system implies that the loading in Kthe keel lLines 1 and 6 are the same, and the same holds true for the pairs of kKeel lLines 2-5 and 3-4, respectively.

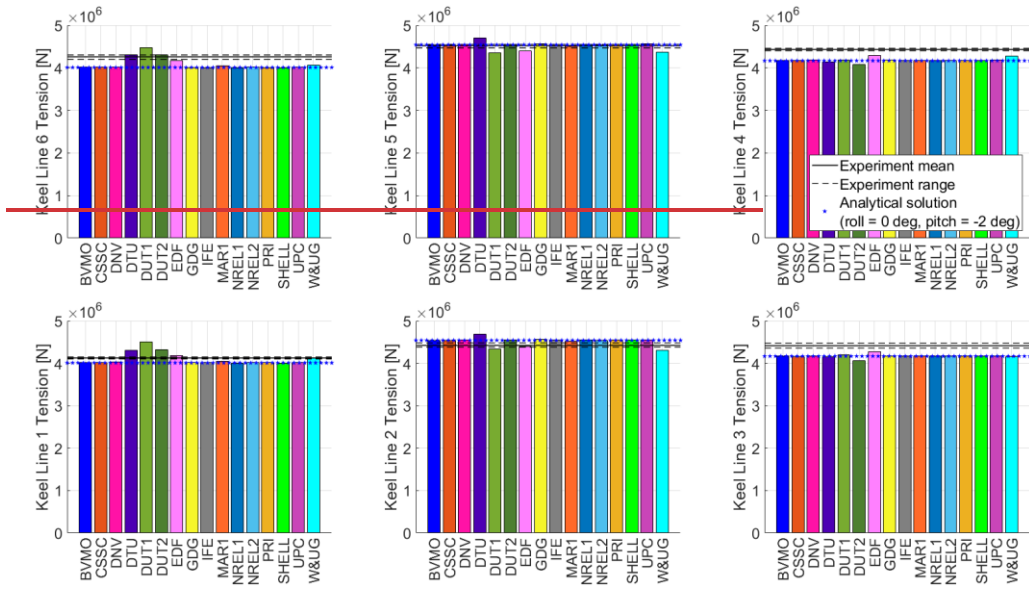
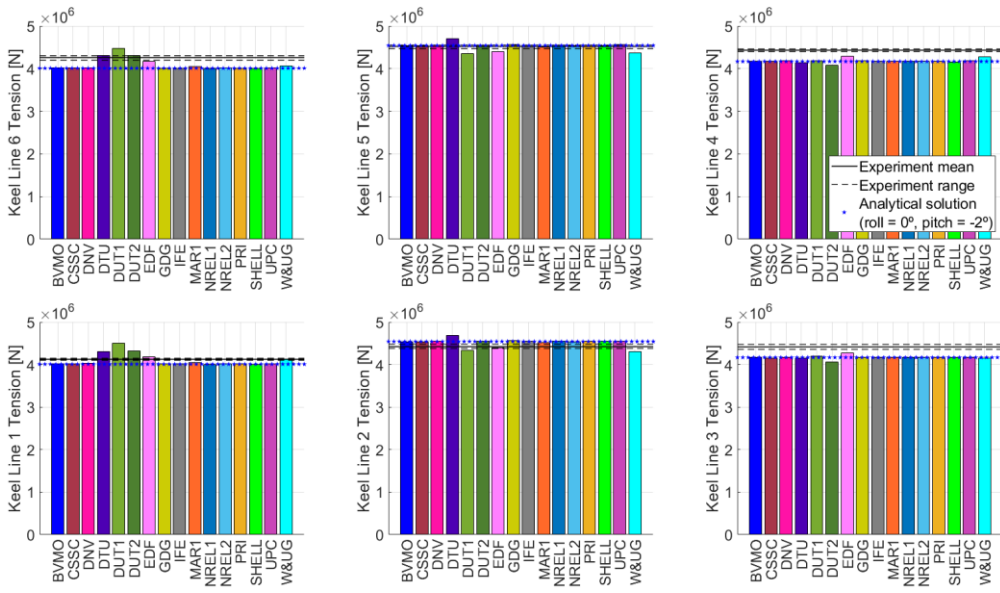
There is excellent agreement between most numerical models and the analytical solution. Some participants (EDF, W&UG) show differences in keel line tensions because the obtained system equilibrium position obtained is slightly different (i.e., different platform pitch and/or roll values) while other participants (DTU, DUT1, DUT2) reproduce the expected system rotation but may not have the proper settings in their numerical models (e.g., slightly offset keel line attachment points, line properties, or keel net weight).

             Figure 16: Tension at the six keel lines in equilibrium conditions (Load Case 1.1).

Figure 17 shows the mean tension for each keel line for the combined wind and wave conditions in Load Cases 5.2, 5.3, and

5.4. The symmetric behavior between keel lines is still present in Load Cases 5.2 and 5.3 due to the keel lines disposition

around the XZ vertical plane, the nacelle being aligned with the X-axis, and the external loading (i.e., wind and waves) being

applied along the X-axis. For these two load cases, the mean keel line tensions are mainly driven by the mean pitch rotation

760   of the floating support structure. Load Case 5.4 (50-year storm condition) does not exhibit the symmetry around the XZ

vertical plane because the system rotates around -2° in roll due to the aerodynamic loading. For this loading condition, the

mean keel lines loading is determined by the combination of roll and pitch angles of the floating support structure. For these

three load cases, the analytical formulation was also able to compute the proper mean keel line tensions based on these

rotations.

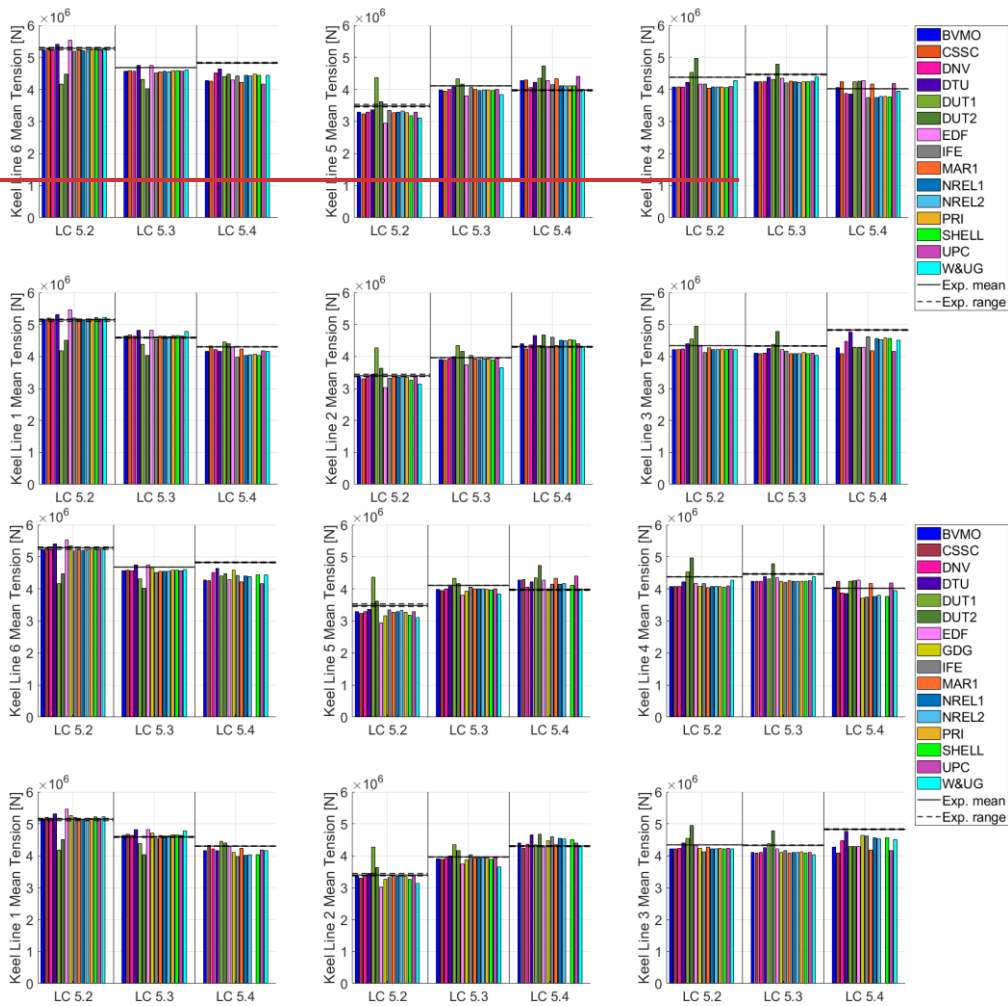



Figure 17: Mean tension at the six keel lines for the combined wind and wave conditions in Load Cases (LC) 5.2, 5.3, and 5.4.

Figure 18 shows the PSD of the six keel lines for the combined wind and waves for the post-rated operating condition (Load Case 5.3). Figure 18 includes the results from the participants as well as the results from three repeats that were performed during the testing. As expected, the keel lines mainly respond at the linear wave excitation region and, similar to the keel

lines mean tension, exhibit the loading by keel pairs around the XZ vertical plane. The two keel lines located at the downwind side (i.e., kKeel lLines 3 and 4) experience the smallest dynamic response. The small peak around 0.22 Hz correspond to the 1P frequency due to a rotor asymmetry (blade mass and blade pitch imbalance).

From a modal analysis performed over the numerical models, it was determined that the translational vibration modes of the suspended keel were located between 1 Hz and 2 Hz, and the keel rotational vibration modes between 2.5 Hz and 3 Hz. Despite these keel natural frequencies being relatively high, the analytical formulation used to estimate the mean keel tensions was not able to capture the dynamic response. The keel line dynamic response is likely driven by the inertial loading with contributions from different degrees of freedom (e.g., surge, heave, and pitch) that the quasi-static analytical formulation does not include. This highlights the importance of using higher-fidelity models like the ones used by the participants in this project to estimate the dynamic keel line tensions.

Most numerical models are well-aligned with the response observed in the experiment. Some participants that had the proper floating support structure rotations but issues with the keel line tensions in the equilibrium condition (e.g., DUT1 and DUT2), also exhibit an unexpected dynamic response. This confirms that something is not properly set up in those numerical models.

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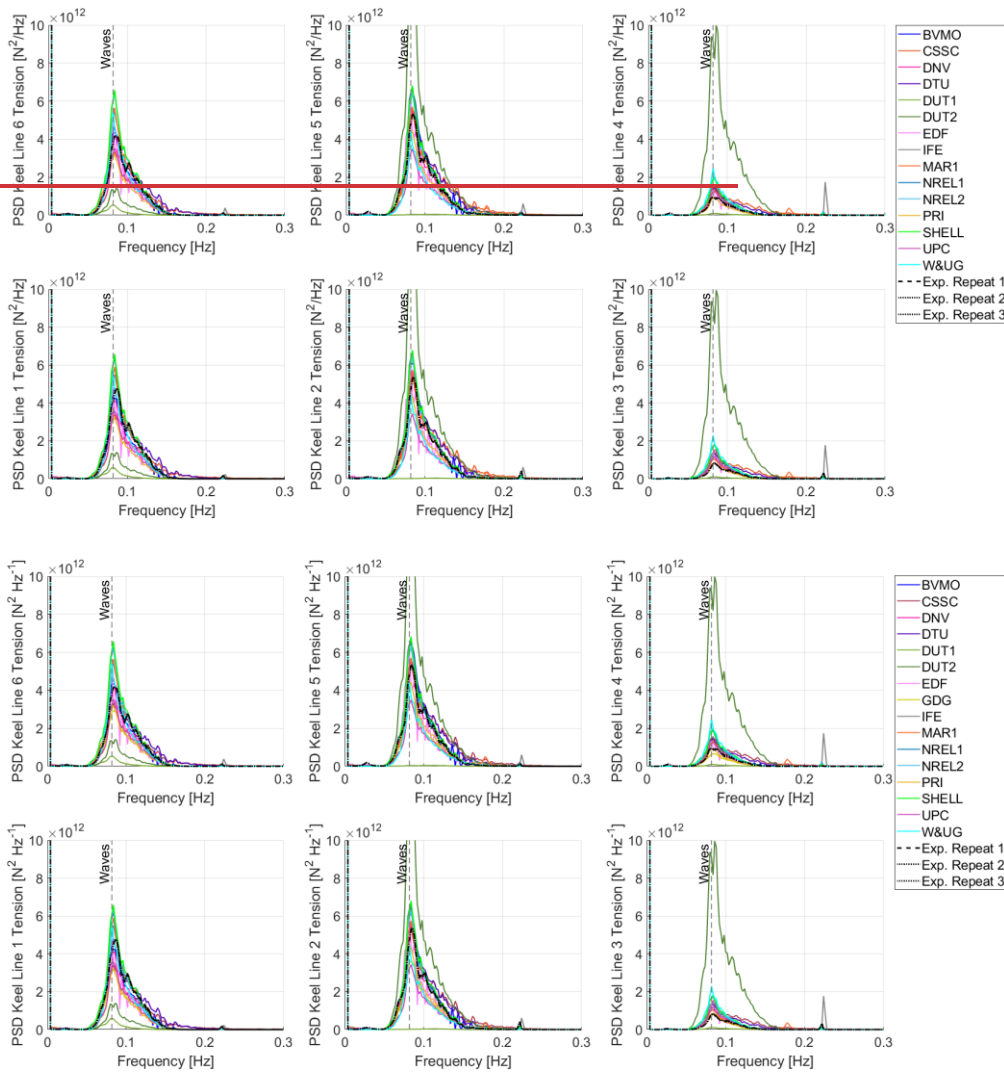
Figure 18: Power spectral density (PSD) for the six keel lines in the combined wind and wave conditions for the post-rated operating condition (Load Case 5.3).

790 Figure 19 shows the PSD sumintegral for each keel line for the combined wind and wave conditions in Load Cases 5.2, 5.3, and 5.4. As expected, the PSD sumintegral is larger for the higher waves. For example, the PSD sumintegral in Load Case 5.2 (rated condition) is the smallest while the PSD integralsum in Load Case 5.4 (50-year storm) is the largest. Similar toAs in Figure. 18, it can be observed that the keel lines located at the downwind side always experience the smallest dynamic response.

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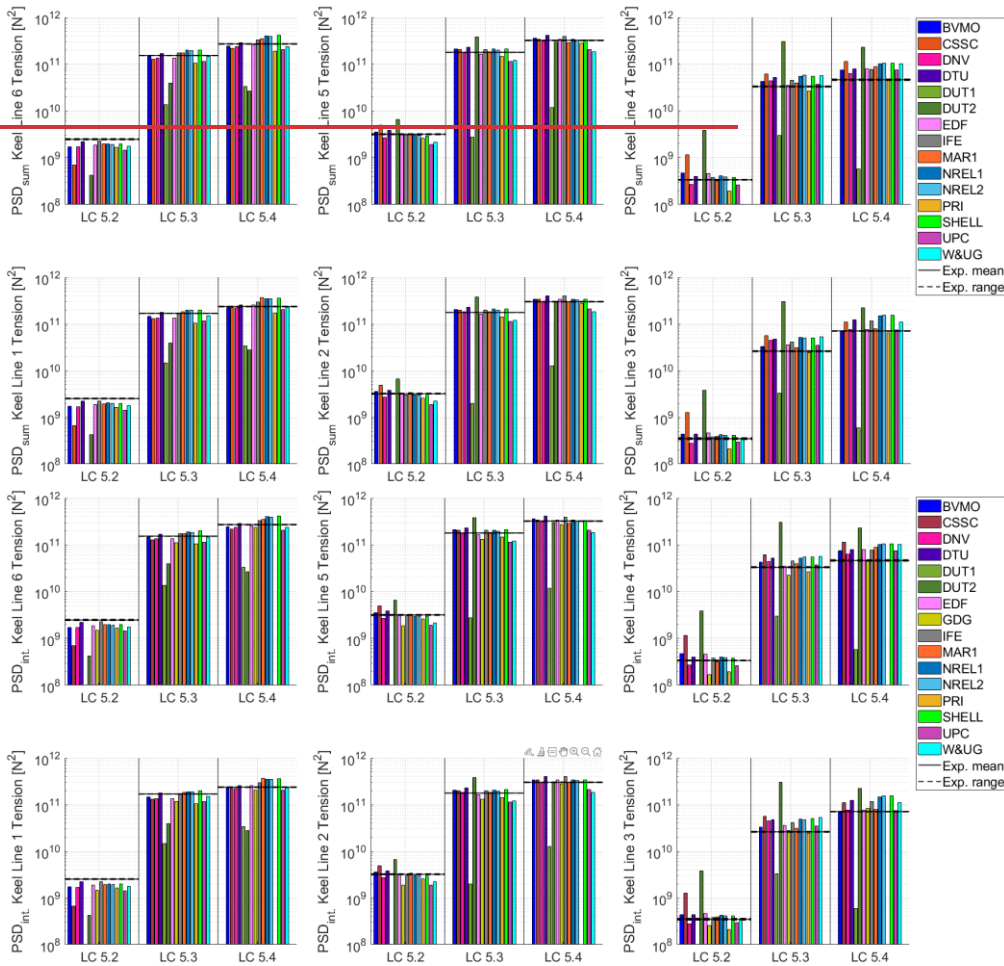
Figure 19: Power spectral density (PSD) integralsum ofin the six keel lines for the combined wind and wave conditions in Load Cases (LC) 5.2, 5.3, and 5.4. Vertical axis in logarithmic scale.

6 Conclusions

In the frame of the OC6 Phase IV project, participants modeled a 3.6- MW wind turbine atop the TetraSpar floating support structure designed by Stiesdal Offshore Technologies. This configuration is representative of the demonstration project installed in Norway in 2021. Numerical results from participants in the project were compared against measurement data from a 1:43 scale test performed by the University of Maine. The system response was studied under wind, wave, and combined wind and wave conditions.

It is important to note that it was the first time that project participants built numerical models for this novel floating support structure design. Due to the multiphysics nature of the system, including aerodynamics, hydrodynamics, and structural dynamics, participants had to use a coupled aero-hydro-elastic approach. The participants defined modeling approaches according to the capabilities of the code(s) used. Setting up these numerical models was quite challenging and prone to user error. This is evidenced by the fact that some participants used the same code and a similar modeling approach but obtained different system responses.

Participants used numerical models of different fidelity levels for the aerodynamics, hydrodynamics, and structural dynamics. During the testing of the floating system, the sensor umbilical used to transfer measurement data and power had an impact in the system equilibrium position and dynamics. Significant changes in the hull surge and heave equilibrium positions were observed between tests (between -6.7 m and +6.4 m in surge and 4 m range in heave from the origin of the coordinate system usedbetween -6.7 m and +6.4 m from the origin of the coordinate system used). Numerical models included the umbilical as an additional line, but there is an some uncertainty associated with it. Moreover, for the 50-year storm condition (idling rotor), the umbilical could be inducing a significant aerodynamic drag force in the system. This force was not included by the participants because the umbilical diameter was not known.

Some differences in the mean wind speed values were also observed between the winds measured during the calibration process and the winds applied during the testing. Participants in the project used the winds measured at hub height during the calibration process as input for the numerical models. However, this may not be fully representative of the actual testing conditions.

Relatively Ggood agreement was observed between the numerical models and the experiment for the aerodynamic loading. The aerodynamic rotor thrust was not directly measured; instead, the platform pitch rotation and the tower base fore-aft bending moment were used as proxy sensors. The pitch rotation experienced by the of the floating support structure due to the aerodynamic thrust force did not impact the mean aerodynamic loading.

The TetraSpar design is made of slender members. When comparing the Morison equation and potential flow (augmented with viscus drag) approaches, in general, no clear differences were observed. The largest differences were due to incorrect settings in some numerical models. For the wave loading conditions studied, the response of the numerical models based on

830   the Morison equation approach were driven by the hydrodynamic inertia component. The Morison equation approach is only valid for diameters-to-wavelength ratios smaller than 0.2. To avoid overestimating the response at the first tower-bending mode, numerical models based on the Morison equation approach had to use the MacCamy and Fuchs diffraction correction of the inertia coefficient and/or low-pass-filter the irregular wave spectrum.

When comparing the Wheeler and vertical wave stretching theories, participants using the vertical stretching obtained the
835   largest slow drift platform motion. NREL also observed increased activity around the first tower-bending mode when using the vertical wave stretching. This resulted in an overestimation for the first tower-bending mode amplitude not observed with the Wheeler stretching theory.

For the combined wind and wave loading, the numerical models showed relatively good agreement for the mean upwind fairlead tension. However, when looking at the dynamic loading, large differences were observed between the numerical
840   models and against the experiment. The response from the numerical models was different even in the linear wave region. The fairlead tensions are determined by the platform motion and the line properties. Differences in the offset platform position under load and dynamic motion around that offset position impact the estimation of the fairlead tensions. Future work may consider prescribing the recorded platform motions during the testing in the numerical models to assess their ability to estimate the mooring line loads. The different hull surge equilibrium positions observed during the testing impact
845   the suspended line weight. This results in different mooring line inertial loading when the platform experiences motions and may explain, in part, the different dynamic forces observed in the wave region.

The TetraSpar design is a novel floating support structure that features unique elements like the keel lines. Characterizing the keel line tensions is important because the suspended keel ensures the floating system stability. These tensions can only be
850   obtained by numerical models with the ability to include structural flexibility within the floating support structure. The potential flow (augmented with viscous drag) approach is a viable option to study the keel line loads if the system is discretized into, at least, two potential flow bodies (hull and keel). For the TetraSpar design, the keel lines mean tension can be determined by means of an analytical approach based on static equilibrium equations. The keel lines mean tension changes observed for the different loading conditions are driven by the floating support structure roll and pitch rotations.
855   Good agreement was observed between most numerical models, the experiment, and the analytical approach. The analytical approach (quasi-static) was not able to capture the dynamic response, but the numerical models and the experiment showed very good agreement between them. This accurate estimation of the keel line tensions enables the computation of the fatigue life for these elements. To account for the loading within the hull and keel, it would be necessary to account for the member-level hydrostatics and hydrodynamics. This is the normal procedure for a Morison equation-based approach, but it would
860   challenge the potential flow approach where, potentially, tens of potential flow bodies may be necessary.

Data Availability

The modeling information, the simulation results, and the experimental data from this project are available to the public will be made available to the public by the end of 2023 through the U.S. Department of Energy Data Archive and Portal, https://a2e.energy.gov/project/oc6 .(Robertson and Bergua, 2023).

865 **Author Contributions**

Amy Robertson: Secured the funding for the OC6 project. Roger Bergua, Will Wiley, Amy Robertson, and Jason Jonkman: Proposed the methodology, formal analysis, and investigation. All authors: Simulated the system and submitted results from their numerical models (detailed in Section. 3). Roger Bergua: Postprocessed and visualized the data from the experiment and the numerical models. Roger Bergua: Wrote the manuscript draft. All authors: Reviewed and edited the manuscript.

870 **Competing Interests**

Some authors are members of the editorial board of Wind Energy Science journal. The peer-review process was guided by an independent editor, and the authors have also no other competing interests to declare.

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