

Controller design for model-scale rotors and numerical-experimental study using prescribed motion

Alessandro Fontanella¹, Elio Daka¹, Felipe Novais^{1,2}, and Marco Belloli¹

¹Mechanical Engineering Department, Politecnico di Milano, Milano, Via La Masa 1, 20156, Italy.

²Maritime Research Institute Netherlands (MARIN), Wageningen, 6708 PM, the Netherlands.

Correspondence: Alessandro Fontanella (alessandro.fontanella@polimi.it)

Abstract. Aerodynamic loads calculation in aero-hydro-servo-elastic modeling tools has been recently validated against experiments for low-frequency platform motions, but without considering any active wind turbine control capability. This work presents a control design framework that to include industry-standard wind turbine control functionalities in a model-scale rotor and its application to a 1:100 scale model of the IEA 15 MW. Wind tunnel tests with fixed foundation and steady wind show the scaled turbine reproduces the steady-state rotor speed-blade pitch-thrust torque characteristics of the IEA 15 MW, confirming the controller design method. Tests with prescribed platform pitch motion are carried out to assess the turbine response and controller modeling in conditions representative of normal operation of floating wind turbines. The blade element momentum model of OpenFAST is verified against the experiment showing aerodynamic thrust and torque are predicted with higher accuracy in the below-rated region than above rated: in our simulation, the decrease of thrust oscillations amplitude due to blade pitch actuation is under predicted. This, combined with uncertainty in modeling the blade pitch actuators, complicates the numerical-experimental simulation of the turbine aerodynamic response in above rated operation.

1 Introduction

Floating wind turbine control has been a topic of research since the introduction of floating wind energy. The main reason is the infamous negative damping problem due to the use of the variable-pitch control strategy of bottom-fixed turbines (Larsen and Hanson (2007), van der Veen et al. (2012)). Most of the research in floating wind turbine control tried to devise new control methodologies to ensure stable operation and reduce fatigue loads for the floating wind turbine components.

Design and verification of control strategies often makes use of aero-hydro-servo-elastic modeling tools to assess the response of the floating system. Accuracy of aerodynamic loads calculation in these tools must be validated to ensure correct modeling of the turbine response. The theme of validation of aerodynamic simulation tools for floating wind turbines has recently been the subject the OC6 Phase III project, which addressed the case of large low-frequency platform surge and pitch motion in a scaled wind turbine (Bergua et al. (2023), Cioni et al. (2023)). The OC6 project used data from the wind tunnel experiment of Fontanella et al. (2021) where no active turbine control strategy was considered. The project has shown the aerodynamic response is quasi-static and is correctly captured by codes of different fidelity in case of low-frequency motion and no active turbine control. Instead, additional verification cases run in the OC6 project have shown aerodynamic unsteadiness

25 may take place when sinusoidal variation of rotor speed or blade pitch is combined with surge motion, but no experimental data were available to validate codes in this scenario.

In the last decade, several scale model experiments about the wind-wave response of floating wind turbines have been carried out, and a review of them is presented by Gueydon et al. (2020). The large majority of tests involving a scaled wind turbine and physical wind generation did not use any active turbine control. The research carried out by the DeepCwind consortium, whose results are summarized by Robertson et al. (2013), investigated the coupled response of three floating wind turbine concepts, but blade pitch and rotor speed were fixed to a constant value (Goupee et al. (2017)). Recently, Mendoza et al. (2022) conducted a scale model experiment of a 15 MW floating wind turbine including active control, but, at the time of writing, only preliminary tests with fixed tower bottom have been used for the validation of offshore modeling tools. Another research effort in this topic is the wave basin experiment of a 10 MW floating wind turbine that was carried out by Madsen et al. (2020). Tests with various wind-wave conditions were compared to two offshore codes by Kim et al. (2023) and the controller used in the simulation study is the same of the experiment. The code validation study of Kim et al. (2023) addressed the floating wind turbine global response, with simultaneous modeling of multiple uncertain phenomena as hydrodynamic viscous loads, turbulent wind field, closed-loop turbine control, rotor aerodynamics with large motion. Overall, the two simulation codes object of validation showed good accuracy with respect to experimental results. Yu et al. (2017) tested a collective blade-pitch controller system in a wind-wave basin and examined the influence of the turbine controller on the platform global response. A small portion of tests has been modeled with an offshore simulation tool, showing good agreement for platform motion at the main wave frequency. However, in the wave-frequency range, platform response is generally driven by linear wave loads and it is difficult to assess the accuracy of aerodynamic loads modeling.

In parallel to experiments with physical modeling of the wind turbine, a large number of floating systems supporting 5 MW-15 MW turbines has been tested in wave basins using a hybrid approach. This approach models rotor loads with a force actuator whose set point is computed in real-time by a numerical model of the full-scale turbine and controller. Gueydon et al. (2018) compared hybrid and physical modeling of wind turbine rotor loads from a tension leg platform floating wind turbine. Thys et al. (2021) investigated the coupling between aerodynamics and hydrodynamics of a 12 MW semi-submersible with a hybrid experiment where aerodynamic loads are computed with a BEM model including the turbine controller and applied to the physical model of the platform and tower with a cable robot. Vittori et al. (2022) tested a 10 MW floating wind turbine reproducing the wind turbine rotor and a state-of-the-art variable-speed control strategy with a multi-fan system. Pires et al. (2020) and Fontanella et al. (2020) utilized the multi-fan system to carry out experiments of the DeepCwind 5 MW and to study the influence on the global floating wind turbine response of two industry-standard control strategies. The hybrid wave basin testing solves the issue related to the downscaling of the controller, but the aerodynamic response of the turbine is reproduced with models that have yet to be validated.

Wind tunnel testing with active control requires control tools for the wind turbine scale model, but there is no consensus or shared practice on how to implement turbine controllers in scale model experiments. Often, controllers of scale model turbines have simplifications with respect to those used in utility-scale machines (e.g., the Reference Open Source Controller ROSCO of Abbas et al. (2022b) or the DTU Wind Energy Controller of Meng et al. (2020)). The purpose of this work is to design

60 a closed-loop controller for a scale model wind turbine with an ad-hoc control development framework. The controller has industry-standard functionalities and, once integrated with the aerodynamic design of the scaled rotor, it allows to replicate in the wind tunnel the aerodynamic loads response of a full-scale machine. Wind tunnel experiments are conducted with the controlled scale model to measure the aerodynamic response with fixed foundation and with prescribed low-frequency platform pitch mimicking a floating wind turbine. The turbine of the experiment with its controller are modeled in the offshore
65 simulation tool OpenFAST and simulations are compared to measurements to identify possible sources of uncertainty in the servo-aero-dynamic response of the scaled wind turbine.

The expected impact of this work is to provide guidance on how to integrate active wind turbine controls in scale model testing using physical wind. The numerical-experimental study using prescribed platform pitch motion verifies the prediction capability of OpenFAST with respect to aerodynamic loading in presence of active turbine control. Data collected in the
70 experiment, as well as the OpenFAST model of the validation study, are shared with the community to promote studies about control of floating wind turbines. The dataset of the experiment and the OpenFAST model can be used for further validation; the ROSCO controller, which has been implemented in Simulink, can be used in future scale model experiments, but also in control studies for utility-scale turbines.

The structure of this article is as follows. Section 2 describes the wind turbine scale model and the experimental setup of
75 the wind tunnel experiment. Section 3 outlines the wind turbine control strategy and the algorithm we followed to down-scale the controller and implement it in the wind turbine scale model. Section 4 presents the methodology we used to investigate the turbine response and verify the controller design procedure. The article is concluded in Sect. 6.

2 Description of the experimental setup

The experimental setup is shown in Fig. 1. It consists of a wind turbine scale model (WTM), which is mounted on top of a
80 six degrees-of-freedom (DOF) robotic platform. Testing is conducted in the atmospheric boundary layer test section of the Politecnico di Milano wind tunnel, which is 13.84 m wide, 3.84 m high, and 35 m long. The wind turbine is placed 20 m downstream the test section inlet. Tests were performed without roughness elements or turbulence generators for a constant inflow velocity and turbulence intensity of 2%.

2.1 Wind turbine aero-servo-dynamic design

85 The wind turbine of the experiment is a $\lambda_L = 1:100$ scaled version of the IEA 15 MW (Gaertner et al. (2020)). It was designed to reproduce the aerodynamic response of the full-scale turbine with wind speed reduced by a factor $\lambda_v = 1:3.5$.

The aim of rotor aerodynamic design is to replicate the blade normal force of the the IEA 15 MW at design tip speed ratio (TSR) of 9 and blade pitch (β) of 0° . The main difficulty in achieving this goal is Reynolds number, which is 350 times lower than for the full-scale turbine. The blade design uses the SD7032 airfoil, which has suitable lift and lift-to-drag characteristics
90 at Reynolds numbers expected for the turbine scale model (i.e., lower than 250k). The blade chord and twist distributions are altered, section by section, to have the lift force and the variation of lift force with angle-of-attack of the IEA 15 MW.

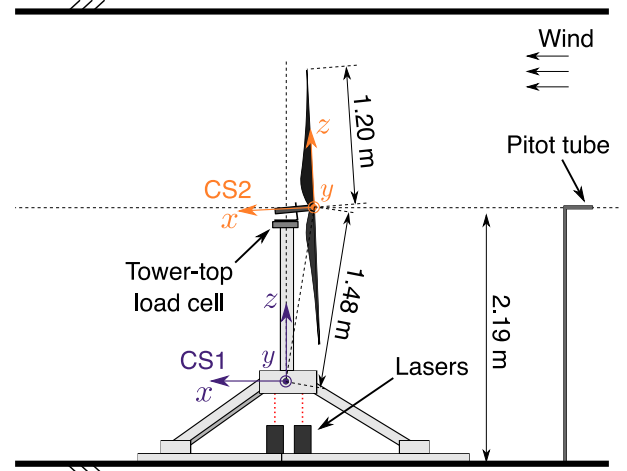
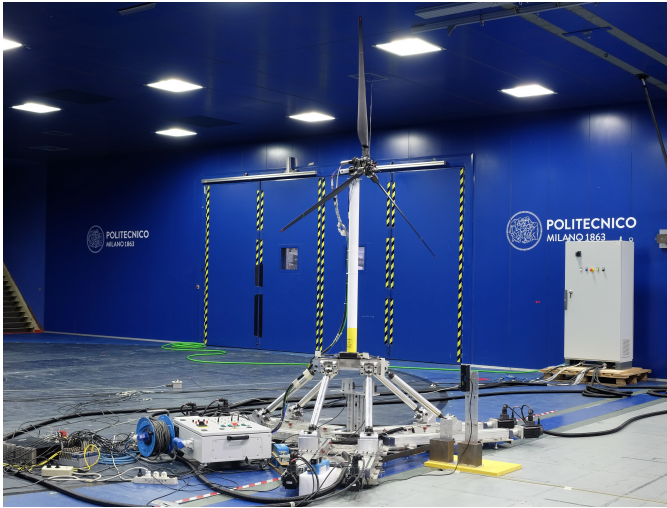


Figure 1. Experimental setup in the Polimi wind tunnel. “CS1” and “CS2” are respectively the coordinate systems for platform motion and rotor forces.

The wind turbine has active generator control and individual blade pitch control. The generator is a brushless DC motor *Maxon EC-4pole-30* with planetary gearbox *Maxon GP32HP* of ratio 21. Generator speed is measured with the encoder *ENC 16 EASY* with 500 pulses per turn, and this signal is the main feedback for closed-loop control of the turbine scale model.

95 The generator high-speed shaft is connected to the rotor low-speed shaft with a toothed belt of transmission ratio equal to 2; the total transmission ratio is $\tau_g = 42$ and the transmission efficiency is $\eta_g = 73.5\%$. The electric motor is driven by a *Maxon ESCON 70/10* controller, it functions as a generator with variable torque set point, which is computed by the variable-speed control strategy of the turbine controller. The tower is an aluminum tube of 75 mm diameter, and the fore-aft mode is at 9.5 Hz (corresponding to 0.33 Hz full-scale).

100 The wind turbine has individual blade pitch actuators, housed inside the hub, that are *Harmonic Drive RSF-5B-30-E050-C*. Each pitch actuator is controlled by a *Maxon EPOS 24/2* drive mounted on the turbine hub. Power and blade pitch set points are transmitted to individual pitch motors with 30 channels slip ring. The wind turbine controller computes generator torque and collective blade pitch set points for the actuators based on generator speed and wind speed measurements. It runs on a National Instrument PXI embedded control system by means of the Veristand interface.

105 The main properties of the turbine model are summarized in Table 1.

2.2 Measurements

Quantities measured in the experiment are rotor forces, platform motion, actual generator speed, collective blade pitch set point, hub-height wind speed. Six-component forces at the tower-nacelle interface are measured with an ATI Mini45 load cell with SI-580-20 calibration. Rotor loads are obtained from the projection in the CS2 reference frame of tower-top loads.

Table 1. Key parameters of the wind turbine model.

| Parameter | Unit | Value |
|----------------|------------------|-------|
| Rotor diameter | m | 2.400 |
| Blade length | m | 1.110 |
| Hub diameter | m | 0.180 |
| Rotor overhang | m | 0.139 |
| Tilt angle | ° | 5.000 |
| Tower-to-shaft | m | 0.064 |
| Tower diameter | m | 0.075 |
| Tower length | m | 1.400 |
| Nacelle mass | kg | 1.975 |
| Blade mass | kg | 0.240 |
| Rotor mass | kg | 2.041 |
| Rotor inertia | kgm ² | 0.279 |
| Tower mass | kg | 2.190 |

110 Platform pitch motion is measured with two MEL M5L/50 lasers placed beneath the robotic platform. Measurement of the undisturbed wind velocity is obtained with a pitot tube placed at centerline, hub-height, 7.15 m upstream the rotor. Generator speed is measured with the generator encoder and reading of this quantity is an output of the generator drive. Measurement of the actual blade-pitch angle is not available and is replaced with the collective blade pitch set point. All measurements are acquired simultaneously with a NI DAQ with sampling frequency of 2000 Hz.

115 **3 Wind turbine control strategy**

The wind turbine controller computes generator torque and collective blade pitch set points based on generator speed and wind speed measurements. It uses the algorithms of the ROSCO introduced by Abbas et al. (2022b) and distributed as a DLL and source Fortran code by Abbas et al. (2022a). In this project, the ROSCO has been implemented in MATLAB Simulink, and the same controller implementation is used for the experiment and for co-simulation with OpenFAST.

120 The logic of the ROSCO implemented in the scaled wind turbine is shown in Fig. 2. It has two main modules: a generator torque controller, controlling generator torque below the rated wind speed to achieve maximum wind-power conversion efficiency; and a collective blade pitch controller, controlling aerodynamic torque above the rated wind speed to limit the extracted power to its nominal value. The generator torque and blade pitch controllers are proportional-integral (PI) controllers with this

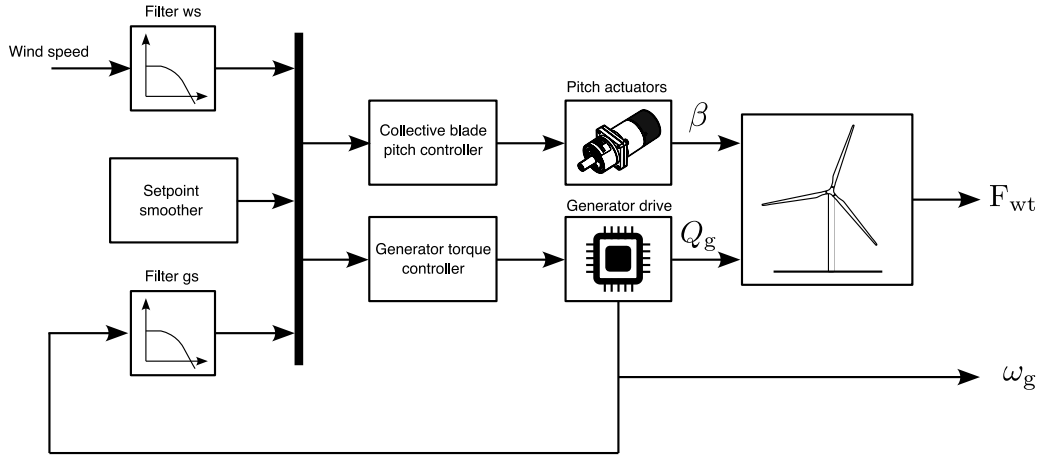


Figure 2. Block diagram showing the ROSCO structure and how it is integrated with the wind turbine scale model. The generator torque controller and collective blade pitch controller are based on the PI controller of Eq. 1, “Filter ws” is the low-pass filter for wind speed, “Filter gs” is the low-pass filter for generator speed, ω_g is the generator speed signal from the generator encoder, \mathbf{F}_{wt} is the 6-components force measured by the tower-top load cell, Q_g is the generator torque, β is the collective blade pitch.

generic structure:

$$125 \quad y = k_{P,y}(\omega_{g,s} - \omega_g) + k_{I,y} \int_0^T (\omega_{g,s} - \omega_g) dt, \quad (1)$$

where y is the control input, either generator torque ($y = g$) or collective blade pitch ($y = \beta$), $k_{P,y}$ and $k_{I,y}$ are the proportional and integral gains, ω_g is generator speed, and $\omega_{g,s}$ is the generator speed set point.

When wind speed is below rated, blade pitch is held constant to the design value of 0° , generator torque is controlled to track a constant TSR set point $\lambda_0 = 9$ and to achieve the maximum power coefficient. In scale model testing, closed-loop
 130 TSR tracking is preferred over the more traditional $k\omega^2$ law, also available in ROSCO, because $k\omega^2$ control does not take into account Reynolds-dependency of aerodynamic torque, which occurs in small-scale turbines (Fontanella et al. (2023a)). With TSR tracking:

$$\omega_{g,s} = \tau_g \frac{\lambda_0 \hat{u}}{R}. \quad (2)$$

where R is rotor radius and \hat{u} is the rotor effective wind speed. In general, this is obtained by means of a wind speed estimator,
 135 but in this case it is measured with the hub-height pitot tube upstream the turbine model. Generator speed is filtered with a second-order low-pass filter, wind speed with a first-order low-pass filter.

When the turbine is in above-rated operation, generator torque is held constant:

$$Q_g = \frac{P_0}{\eta_g \tau_g \omega_{r,0}}, \quad (3)$$

where P_0 is the turbine rated rotor power, and $\omega_{r,0}$ the rated rotor speed. Collective blade pitch is computed with a PI controller
 140 as the one of Eq. 1, where the set point is $\omega_{g,s} = \tau_g \omega_{r,0}$.

When the wind turbine works near the rated wind speed, the set point for the generator torque and the collective blade
 pitch controllers is the same (i.e., $\omega_{g,s} = \tau_g \omega_{r,0}$). This would lead the controllers to conflict with each other with unwanted
 oscillations in the turbine response. To avoid this conflict, the set point smoothing algorithm of Abbas et al. (2022b) is used,
 that progressively lowers the generator speed set point of one of the two controllers to have smooth transition between one
 145 operating regime to the other. Advanced control functionalities available in the ROSCO of Abbas et al. (2022a), such as peak
 shaving or minimum pitch schedule, are implemented in the MATLAB Simulink version of the controller but they are not used
 in this study.

3.1 Scaling of the wind turbine controller

In the experiment, the wind turbine controller is run at model scale. This approach takes a different route with respect to when
 150 the turbine controller is run in real-time in its full-scale version with scaling of input and output signals (e.g., in the work of
 Mendoza et al. (2022)). In Appendix B we prove the full-scale approach does not respect the time scaling.

The controller scaling seeks to: 1) use the same algorithm of the full-scale turbine controller; 2) replicate the aero-servo-
 dynamic response of the IEA 15 MW rotor at model scale given the aerodynamic and structural properties of the wind turbine
 scale model. The tuning procedure is model-based; dynamics of pitch and generator actuators is neglected. The wind turbine
 155 is modeled as a single degree of freedom system corresponding to the rotor-generator, whose equation of motion is:

$$J^* \dot{\omega}_r = Q_a - \tau_g \eta_g Q_g, \quad (4)$$

where $J^* = J_r + \eta_g \tau_g^2 J_g$ is the total inertia of the rotor and generator, Q_a the rotor aerodynamic torque. The aerodynamic
 torque is:

$$Q_a = \frac{1}{2} \rho C_Q(\omega_r, \beta, U) \pi R^3 U^2, \quad (5)$$

160 where ρ is air density and C_Q the torque coefficient. C_Q is assumed to be function of rotor speed, collective blade pitch and
 wind speed U . The expression of Q_a of Eq. 5 is linearized to obtain a linear model of the wind turbine once it is inserted in
 Eq. 4:

$$Q_a \simeq Q_{a,0} + \left. \frac{\partial Q_a}{\partial \omega_r} \right|_0 (\omega_r - \omega_{r,0}) + \left. \frac{\partial Q_a}{\partial \beta} \right|_0 (\beta - \beta_0) + \left. \frac{\partial Q_a}{\partial U} \right|_0 (U - U_0), \quad (6)$$

where $(\cdot)_0$ denotes the steady-state value of a quantity for a given turbine operating point. In a more compact form:

$$165 \quad Q_a \simeq Q_{a,0} + K_{\omega Q} \bar{\omega}_r + K_{\beta Q} \bar{\beta} + K_{U Q} \bar{U}, \quad (7)$$

where $K_{\omega Q}$, $K_{\beta Q}$, $K_{U Q}$ are the aerodynamic torque sensitivities with respect to rotor speed, collective blade pitch and wind
 speed; $\bar{\omega}_r$, $\bar{\beta}$, \bar{U} are the perturbations of rotor speed, blade pitch and wind speed.

Below, we detail the scaling of the controller for the below rated and above rated regions.

3.1.1 Below rated

170 In below-rated operation, blade pitch is fixed ($\bar{\beta} = 0$) and wind speed is assumed constant ($\bar{U} = 0$), thus combining Eq. 4 with Eq. 1 we have:

$$J^* \ddot{\theta}_r - (\tau_g^2 \eta_g k_{P,g} + K_{\omega Q}) \dot{\theta}_r - \tau_g^2 \eta_g k_{I,g} \bar{\theta}_r = 0, \quad (8)$$

where θ_r is rotor azimuth. Gains of the TSR tracking controller are computed from Eq. 8:

$$k_{I,g} = - \left(\frac{J^* \Omega^2}{\tau_g^2 \eta_g} \right)_{sm}, \quad (9)$$

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$$k_{P,g} = - \left(\frac{K_{\omega Q} + 2J^* \Omega h}{\tau_g^2 \eta_g} \right)_{sm}, \quad (10)$$

where $(\cdot)_{sm}$ denotes WTM quantities at model scale; $\Omega = 0.12 \cdot \lambda_v \lambda_L^{-1}$ rad/s, and $h = 0.85$. In below-rated wind speeds $K_{\omega Q}$ is about constant, thus we have a single value for $k_{P,g}$ and $k_{I,g}$. Gains of the generator torque controller for the WTM are $k_{P,g} = -8.3 \times 10^{-3}$ Nm/rad/s, instead of -1.1×10^{-1} Nm/rad/s for the scaled IEA 15 MW, and $k_{I,g} = -1.9 \times 10^{-2}$ Nm/rad
 180 instead of -3.7×10^{-1} Nm/rad.

The set point of the TSR tracking controller is computed with Eq. 2.

3.1.2 Above rated

Above the rated wind speed, generator torque is constant ($\bar{Q}_g = 0$), and assuming constant wind speed, Eq. 4 becomes:

$$J^* \ddot{\theta}_r + (\tau_g K_{\beta Q} K_{P,\beta} - K_{\omega Q}) \dot{\theta}_r + \tau_g K_{\beta Q} K_{I,\beta} \bar{\theta}_r = 0. \quad (11)$$

185 Above rated, aerodynamic sensitivities $K_{\omega Q}$ and $K_{\beta Q}$ depend on wind speed. Gains of the collective pitch controller are computed for discrete wind speeds from rated to cut-out by means of this procedure:

1. we compute the closed-loop frequency and damping ratio of the IEA 15 MW at wind speed U_0 :

$$\Omega_{0,fs} = \left(\sqrt{\frac{\tau_g K_{\beta Q} k_{I,\beta}}{J^*}} \right)_{0,fs}, \quad (12)$$

$$190 \quad h_{0,fs} = \left(\frac{\tau_g K_{\beta Q} k_{P,\beta} + K_{\omega Q}}{2J^* \Omega} \right)_{0,fs}, \quad (13)$$

where $(\cdot)_{0,fs}$ denotes full-scale quantities, that are evaluated at the operating point identified by U_0 in case of wind speed-dependent values;

2. the closed-loop frequency and damping of the WTM are computed by scaling dimensionally those of the IEA 15 MW:

$$\Omega_{0,\text{sm}} = \Omega_{0,\text{fs}} \cdot \lambda_v \lambda_L^{-1}, \quad (14)$$

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$$h_{0,\text{sm}} = h_{0,\text{fs}}; \quad (15)$$

where $(\cdot)_{0,\text{sm}}$ denotes model-scale quantities, that are evaluated at the operating point identified by wind speed $U_0 \lambda_v$ in case of wind speed-dependent values.

3. gains for the WTM are:

$$200 \quad k_{I,\beta}(U_0 \lambda_v) = \left(\frac{J^* \Omega^2}{\tau_g K_{\beta Q}} \right)_{0,\text{sm}}, \quad (16)$$

$$k_{P,\beta}(U_0 \lambda_v) = \left(\frac{K_{\omega Q} + 2J^* \Omega h}{\tau_g K_{\beta Q}} \right)_{0,\text{sm}}. \quad (17)$$

Equations 16-17 result in the gain schedule for the pitch controller, where the scheduling variable is wind speed. However, since at each wind speed corresponds a steady-state value of collective blade pitch, this is used as the scheduling variable (i.e., $U_0 \lambda_v$ is replaced with β_0). In the WTM there is no feedback of the actual pitch angle and it is replaced with the pitch angle set point at previous time step.

Pitch controller gains for the WTM are compared, at model scale, to those of the IEA 15 MW in Fig.3. Gains for the WTM are significantly different than values obtained by scaling gains of the IEA 15 MW, the main reason for this difference being rotor inertia, which is larger for the scale model (0.279 kgm^2) compared to the scaled value of the IEA 15 MW (0.031 kgm^2); aerodynamic sensitivities for the WTM are instead close to the reference wind turbine (see Fig. 7).

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The generator torque set point is computed with Eq. 3, where the generator speed set point is:

$$\omega_{g,s} = \tau_{g,\text{sm}} \left(\left(\frac{\omega_{g,s}}{\tau_g} \right)_{\text{fs}} \right) \frac{\lambda_v}{\lambda_L}. \quad (18)$$

4 Methodology for investigation of the turbine response

The turbine response and the controller scaling are verified with two sets of tests. The steady-state characteristics of wind speed-rotor speed-blade pitch-thrust-torque are assessed running the wind turbine with steady wind and fixed tower bottom. The aero-servo-dynamic response is studied with prescribed platform pitch motion.

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In the experiment, aerodynamic loads are calculated from tower-top load measurements and are compared to two numerical models of the scaled turbine:

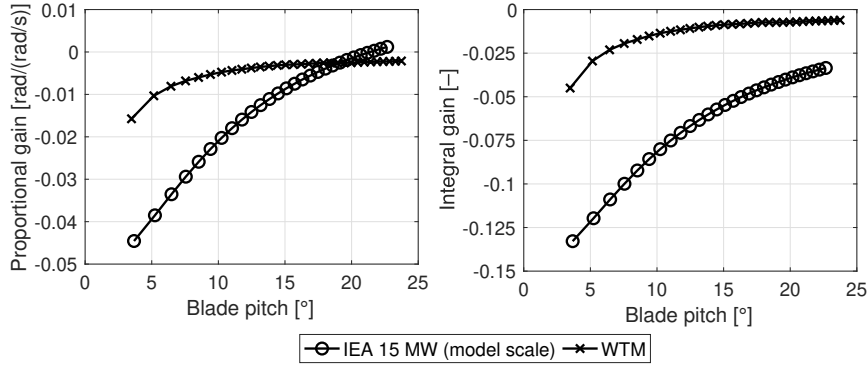


Figure 3. Proportional and integral gains for the collective blade pitch controller of the IEA 15 MW (at model scale) and the wind turbine scale model (WTM) as function of collective blade pitch, which is used for scheduling.

- an OpenFAST model with generator torque and blade pitch controlled with the ROSCO. This model is used to study the coupled aero-servo-dynamic response of the wind turbine.
- the stand-alone AeroDyn model with prescribed platform pitch motion, rotor speed, and blade pitch. Simulations with this tool are used to assess the aerodynamic loads calculations of the aerodynamic module of OpenFAST without including the additional complexity of the closed-loop turbine controller;

This section summarizes the operating conditions and the platform motion of the experiment, it explains the algorithm we used to estimate aerodynamic loads from tower-top forces, and provides a description of the modeling approach adopted in OpenFAST.

4.1 Wind turbine operating conditions and platform motion

The wind turbine in bottom-fixed configuration is run at several wind speeds ranging from 2.5 m/s to 5.8 m/s to measure the steady-state response of rotor torque and thrust, rotor speed and collective blade pitch.

Two operating conditions are selected for tests with platform motion, corresponding to wind speeds of 2.87 m/s and 5.05 m/s. The imposed motion emulates large-amplitude platform tilt oscillations in floating wind turbines. Motion is in the pitch direction (θ), i.e., the rotation about the y-axis of the CS1 reference frame (see Fig. 1), and is sinusoidal:

$$\theta(t) = A_m \sin(2\pi f_m t), \quad (19)$$

where A_m is motion amplitude and f_m is motion frequency. Rotor-level unsteadiness due to the global response of the rotor and its wake, is often associated to the rotor reduced frequency f_r defined as:

$$f_r = \frac{f_m D}{U_0}, \quad (20)$$

where D is rotor diameter. Several combinations of A_m and f_m are run in the experiment to explore the turbine aerodynamic response at various f_r and with different amplitude of apparent wind speed oscillations $\Delta U = 2\pi f_m A_m$. Motions conditions

are summarized in Table 2. Tests with prescribed platform motion are carried out with the turbine controlled by mean of the
 240 ROSCO, but also with fixed rotor speed and blade pitch.

Table 2. Motion conditions (A_m is amplitude of pitch motion, f_m is frequency, ΔU is the apparent wind speed at hub-height, f_r BR is the reduced frequency with below rated wind of 2.87 m/s, f_r AR the reduced frequency with above rated wind of 5.05 m/s).

| A_m [°] | f_m [Hz] | ΔU [m/s] | f_r BR [-] | f_r AR [-] |
|-----------|------------|------------------|--------------|--------------|
| 3.2 | 0.25 | 0.13 | 0.21 | 0.12 |
| 2.2 | 0.25 | 0.09 | 0.21 | 0.12 |
| 1.1 | 0.25 | 0.04 | 0.21 | 0.12 |
| 3.3 | 0.75 | 0.40 | 0.63 | 0.36 |
| 2.2 | 0.75 | 0.26 | 0.63 | 0.36 |
| 1.1 | 0.75 | 0.13 | 0.63 | 0.36 |
| 2.2 | 1.25 | 0.45 | 1.05 | 0.59 |
| 1.7 | 1.25 | 0.34 | 1.05 | 0.59 |
| 1.1 | 1.25 | 0.22 | 1.05 | 0.59 |
| 2.0 | 1.50 | 0.48 | 1.25 | 0.71 |
| 1.7 | 1.50 | 0.41 | 1.25 | 0.71 |
| 1.1 | 1.50 | 0.27 | 1.25 | 0.71 |
| 1.1 | 1.75 | 0.31 | 1.46 | 0.83 |
| 0.8 | 1.75 | 0.23 | 1.46 | 0.83 |
| 0.5 | 1.75 | 0.15 | 1.46 | 0.83 |
| 0.8 | 2.00 | 0.27 | 1.67 | 0.95 |
| 0.5 | 2.00 | 0.18 | 1.67 | 0.95 |
| 0.2 | 2.00 | 0.07 | 1.67 | 0.95 |

4.2 Estimation of rotor aerodynamic loads

A measurement of aerodynamic loads is not available in the experiment, thus aerodynamic thrust and torque are estimated from measurements of tower-top loads, which are processed to remove the force contribution due to inertia and weight of the rotor-nacelle assembly. For every motion condition, two tests are run where the same type of motion (amplitude and frequency)
 245 is prescribed to the wind turbine; in one test there is no wind, the rotor is fixed, and loads measured by the load cell are mostly due to inertia and weight (we assume the contribution due to air drag on the turbine components is negligible); in the test with wind, the load cell measures inertia, weight, and aerodynamic forces. Time series acquired in the two tests are grouped so they have the same integer number of motion periods; time series of forces in the test with no wind are subtracted from time series of forces in the test with wind, after being projected from CS1 to CS2, obtaining the aerodynamic loads. This procedure for

250 estimating aerodynamic forces is reliable when dynamic amplification due to flexibility is negligible (Fontanella et al. (2021)), i.e., when the frequency of motion is significantly lower than the first flexible mode of the wind turbine (in our case it is the fore-aft mode at 9.5 Hz and the maximum f_m is 2 Hz).

Rotor speed is regulated by the wind turbine controller and in general, when the turbine operates in unsteady conditions, rotor speed is not constant, and the inertia torque due to rotor acceleration is present in the load cell measurements. With
 255 sinusoidal platform motion, rotor speed oscillations are dominated by the harmonic component at the motion frequency, as it is shown in Fig. 4. Rotor speed oscillations at frequencies other than f_m are regarded as noise. Aerodynamic torque is computed removing the torque component due to rotor inertia from M_x of CS2:

$$Q(t) = M_x(t) - J_r A_\omega (2\pi f_m)^2 \cos(2\pi f_m t + \phi_\omega), \quad (21)$$

where A_ω and ϕ_ω are the amplitude and phase of the rotor speed spectrum.

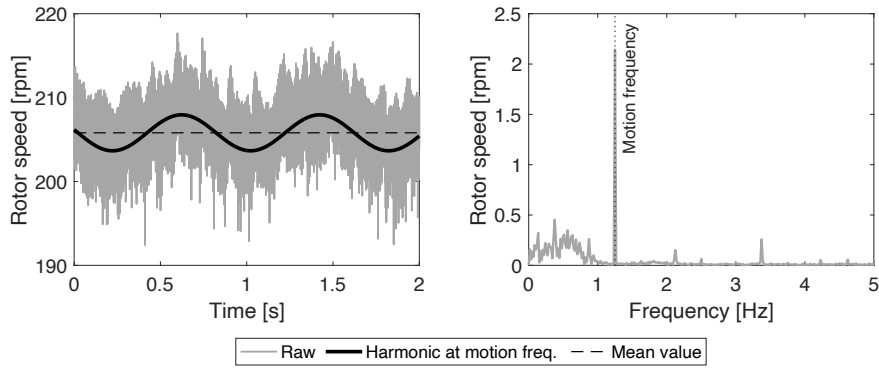


Figure 4. Rotor speed with prescribed platform pitch motion of $f_m = 1.25$ Hz, $A_m = 2.2^\circ$. Left: time series, right: spectrum.

260 4.3 OpenFAST model

An aero-servo-elastic model of the wind turbine scale model of the experiment is created in OpenFAST. Blades and tower are modeled as rigid bodies. A damped oscillator is introduced at the base of the wind turbine and external forces are applied to it to prescribe the platform motion of the experiment. The aerodynamic model is implemented in AeroDyn v15 based on blade twist and chord radial distributions, and Reynolds-dependent polars at 38 radial stations. Calculation of induced velocity in
 265 the AeroDyn BEM is based on wake equilibrium assumption (steady BEM) or on dynamic wake (dynamic BEM). The airfoil model accounts for flow hysteresis during attached flow and dynamic stall.

The wind turbine controller is the same Simulink controller of the experiment which is run in co-simulation with OpenFAST. Pitch actuators are modeled as third-order systems of transfer function:

$$G_{\text{act}}(s) = \frac{b_1 s^2 + b_2 s + b_3}{a_1 s^3 + a_2 s^2 + a_3 s + a_4}. \quad (22)$$

270 Coefficients of $G_{\text{act}}(s)$ are obtained by means of system identification carried out on the WTM before wind tunnel testing. In the frequency range of imposed motion tests, $G_{\text{act}}(s)$ introduces a constant time delay of 0.075 s (i.e., phase is linear with frequency) and unit amplification.

5 Results

This section presents results about the wind turbine response from experimental measurements and numerical simulations.

275 Experiment and simulation are compared to identify differences in the modeling of the servo-aero-dynamic response of the wind turbine. The results section is organized in this way:

- we verify the steady-state response of rotor speed-blade pitch-thrust-torque of the turbine controlled with the ROSCO. In this step, we check the scaling of setpoints and the capability of the aero-servo design of the WTM to replicate the IEA 15 MW at small scale;
- 280 – we examine the steady-state rotor performance coefficients, which are at the base of the linearized aerodynamic model of Sect. 3.1, the tool used to down-scale the wind turbine controller. In particular, the wind turbine controller scaling uses the OpenFAST C_P as a proxy for the experimental C_P , thus we check the consistency of the two;
- we analyze the closed-loop response with unsteady inflow created by platform pitch motion. In detail, we first verify the prediction of the stand-alone aerodynamic model of OpenFAST with prescribed rotor speed oscillations and blade
285 pitch variations; then, we compare the experiment with the OpenFAST model to understand how discrepancies in the modeling of aerodynamics and wind turbine actuation influence the closed-loop response.

5.1 Fixed turbine response

The response of the WTM controlled with the ROSCO is measured at six wind speeds. Figure 5 shows the operating points obtained in the wind tunnel which are compared, at model scale, to the OpenFAST model of the WTM and to the IEA 15 MW.

290 The OpenFAST model of the WTM is perfectly aligned with the IEA 15 MW, whereas rotor speed in the experiment is slightly higher (the maximum error is 10.5 rpm at 2.9 m/s). The discrepancy in the experiment can be due to a small static offset in the generator speed feedback and/or in the wind speed measurement used for calculation of the generator speed set point in the TSR-tracking controller. In below-rated wind speeds, collective blade pitch is 2.3° instead of 0° and this is likely due to misalignment of individual blades and/or incorrect setting of the zero-pitch position. In above rated wind speeds, the rated rotor
295 speed is achieved with values of blade pitch that have an offset of about -3.5° with respect to the IEA 15 MW.

The scale model rotor is designed to match the thrust force of the IEA 15 MW when $\text{TSR} = 9$ and $\beta = 0^\circ$. Below rated, the thrust force in the experiment is lower than the IEA 15 MW due to the blade pitch offset; instead, in OpenFAST, where the blade pitch is 0° , the thrust force is aligned to the one of the IEA 15 MW. Blades misalignment is a well-known problem for model wind turbines: its impact on wind turbine performance and thrust is also investigated by Jüchter et al. (2022), who
300 propose to reduce it with a new blade mounting procedure.

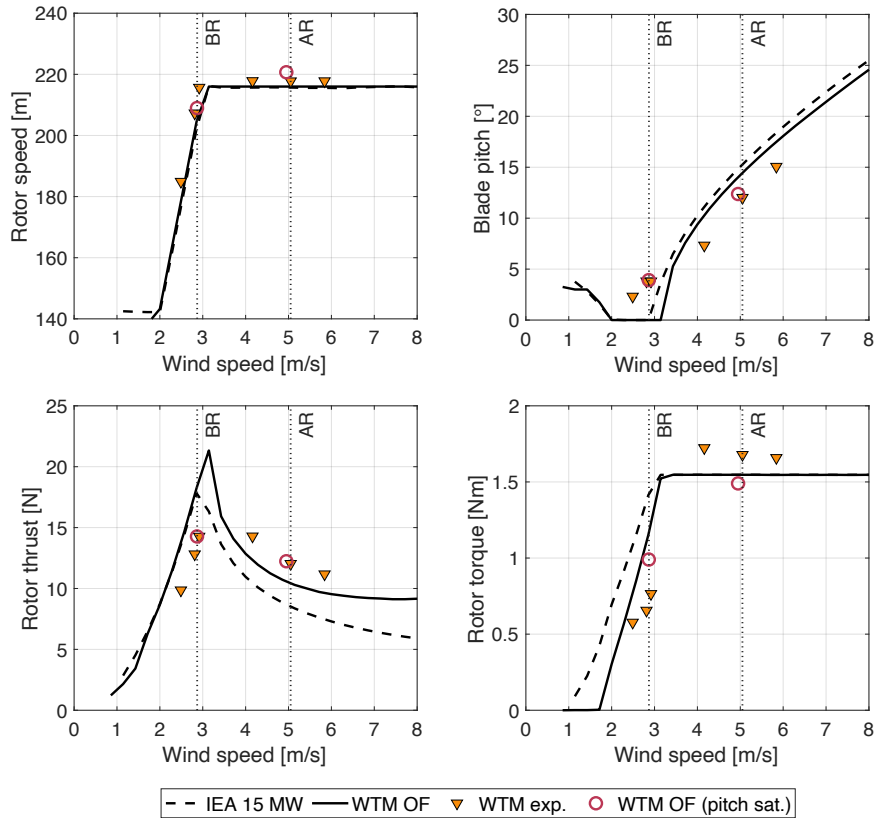


Figure 5. Steady-state operating points for the wind turbine scale model (WTM) obtained in the wind tunnel experiment (exp.) and from the OpenFAST model (OF) are compared to the IEA 15 MW at model scale. Vertical dotted lines identify the below rated (BR) and above rated (AR) operating conditions considered for tests with platform motion. In “WTM OF (pitch sat.)” the minimum pitch in ROSCO is set to the BR pitch of the experiment to simulate the blade pitch offset.

The torque in OpenFAST is lower than the IEA 15 MW for all wind speeds. This is due to the lower efficiency of the SD7032 compared to full-scale airfoils. In the experiment, torque is higher than the rated value for the IEA 15 MW; the cause of this error can be the torque set point obtained with Eq. 3, which requires knowledge of the transmission efficiency, hard to characterize. Rotor thrust has the same trend in the experiment and in the WTM OpenFAST model, but wind tunnel values are higher than in simulations. The difference is attributed to the blade pitch offset that is present in the experiment.

The minimum blade pitch of ROSCO in the WTM OpenFAST model is increased to the collective pitch of the experiment to simulate the blades pitch offset. This model is simulated in the two wind conditions considered in the tests with platform movement and results are closer to those of the experiment. The largest difference is seen for rotor torque in below rated condition, which is 29% higher in OpenFAST compared to the experiment. This difference is attributed to airfoil efficiency which is higher in the OpenFAST model, that is based on polars measured on a 2D airfoil (Fontanella et al. (2021)), than in

the 3D rotor, and it can be due to manufacturing imperfections or unpredicted variations of aerodynamic characteristics that may occur at low Reynolds. The primary function of wind turbine controllers like the ROSCO is to regulate power, thus any difference in airfoil efficiency changes the operating points of the wind turbine making it difficult to reproduce the experiment with a numerical simulation tool for validation tasks. Matching between OpenFAST and the experiment can be improved
315 estimating polars based on experimental data, for example by means of the methodology proposed by Bottasso et al. (2014).

5.2 Performance coefficients and linearized aerodynamic response

Power and thrust coefficients of the wind turbine scale model are measured for various combinations of TSR and blade pitch angle. Wind speed is 4 m/s in all tests and TSR is varied changing rotor speed in open-loop (i.e., without using the ROSCO). The same conditions of the experiment are simulated with the OpenFAST model of the WTM, and results of wind tunnel
320 measurements and simulations are compared in Fig. 5. The maximum C_P in OpenFAST is 0.42 for TSR close to 9 and $\beta = 0^\circ$, whereas the maximum C_P in the experiment is 0.44 for TSR near 9.5 and $\beta = 0^\circ$. The transition from maximum C_P to zero is milder in OpenFAST than in the experiment, where more combinations of (TSR, β) have $C_P \approx 0$; this can be due to the efficiency of the blade that in conditions far from the steady-state characteristic of the turbine is lower in the wind tunnel than in OpenFAST, as already seen in Fig. 5. At TSR = 9, $C_T = 0.8$ in OpenFAST, whereas in the wind tunnel $C_T = 1$. The
325 higher values of C_P and C_T measured in the wind tunnel are likely due to blockage (Robertson et al. (2023)). Despite the low Reynolds and the presence of blockage, the performance of the wind turbine model at its optimal operating point and the shape of the C_P and C_T curves are very close to the IEA 15 MW, and this makes the model sufficiently representative of the full-scale turbine.

Aerodynamic sensitivities are calculated as in Appendix A, from partial derivatives of the performance coefficients of Fig. 6
330 and the steady-state operating points of Fig. 5. Partial derivatives of $C_P(\lambda, \beta)$ and $C_T(\lambda, \beta)$ are obtained from the numerical gradient of rotor performance coefficients which is computed with the central difference method. The aerodynamic sensitivities of the WTM and of the IEA 15 MW are shown in Fig. 7. In above-rated wind speeds, the experiment is in good agreement with the OpenFAST model of the WTM and matches well the IEA 15 MW, whereas larger differences are seen in the below rated region. Discrepancies in below wind speeds are more pronounced for $K_{\beta Q}$ and $K_{\beta T}$, the sensitivities to blade pitch, and they
335 can be due to the blade pitch offset present in the experiment. $K_{\omega Q}$ is very similar in the experiment and in OpenFAST, and this supports the use of the OpenFAST results in the tuning of the PI TSR-tracking controller; a similar convergence of results is found for $K_{\beta Q}$ in the above-rated region, thus aerodynamic data obtained from the OpenFAST model are suitable also to design the PI pitch controller.

5.3 Verification of the aerodynamic model with platform pitch motion

340 When the wind turbine is subjected to prescribed platform pitch motion the turbine controller dynamically actuates generator torque and collective blade pitch to regulate rotor speed. Platform pitch variations result in oscillations of rotor speed, blade pitch, rotor thrust and torque whose amplitude is proportional to the apparent wind speed created by platform motion. We

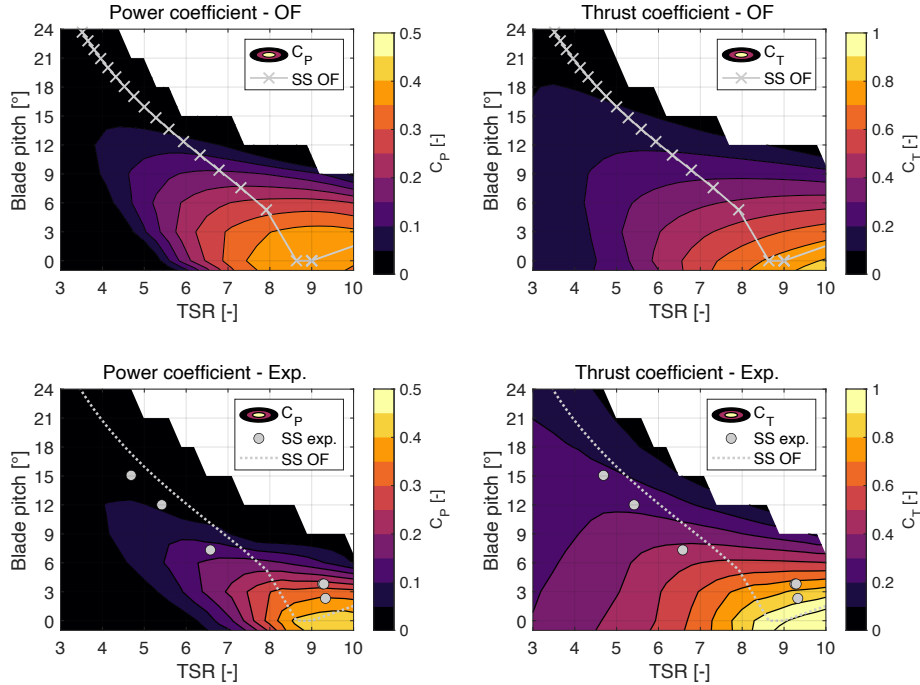


Figure 6. Power coefficient (C_P) and thrust coefficient (C_T) for the wind turbine scale model measured in the wind tunnel (Exp.) and obtained in OpenFAST (OF). “SS OF” and “SS exp.” are the steady-state control trajectories of Fig. 5. Negative values are not shown.

discuss the condition with $A_m = 2.2^\circ$ and $f_m = 1.25$ Hz, which has relatively large ΔU and where it is possible to see the largest differences between experiment and numerical simulations.

345 Calculation of aerodynamic loads in the stand-alone AeroDyn module is verified prescribing the platform pitch, rotor speed, and blade pitch oscillations of the experiment in the numerical model. Time series of rotor speed and blade pitch are filtered to isolate the harmonic at frequency f_m and only this harmonic component is analyzed. The aerodynamic response of AeroDyn is compared to the experiment in Fig. 8 and Fig. 9, which show the zero-peak amplitude and phase shift with respect to motion of the aerodynamic thrust and torque. AeroDyn simulations are repeated with dynamic wake (DW) and static wake (SW) models.

350 In the cases without control, rotor speed and blade pitch are fixed; for the below-rated wind speed, the experimental results with fixed rotor speed/blade pitch are omitted, because when the WTM is controlled with an open-loop rotor speed set point, rotor speed is not fixed but has oscillations of 1.5 rpm, which is comparable to when the ROSCO is used.

Below rated, when rotor speed and blade pitch are fixed, the amplitude and phase shift of aerodynamic loads variations computed with AeroDyn and static or dynamic wake are similar. The result does not change much when rotor speed varies sinusoidally: the thrust phase is slightly lower than in the fixed-rotor speed case, and the torque phase is slightly higher. The estimate of aerodynamic torque of the experiment is affected by uncertainty in the rotor inertia (see Sect. 4.2). Whiskers in Fig. 9 show how the torque amplitude and phase changes due to a variation of rotor inertia of $\pm 20\%$ its mean value

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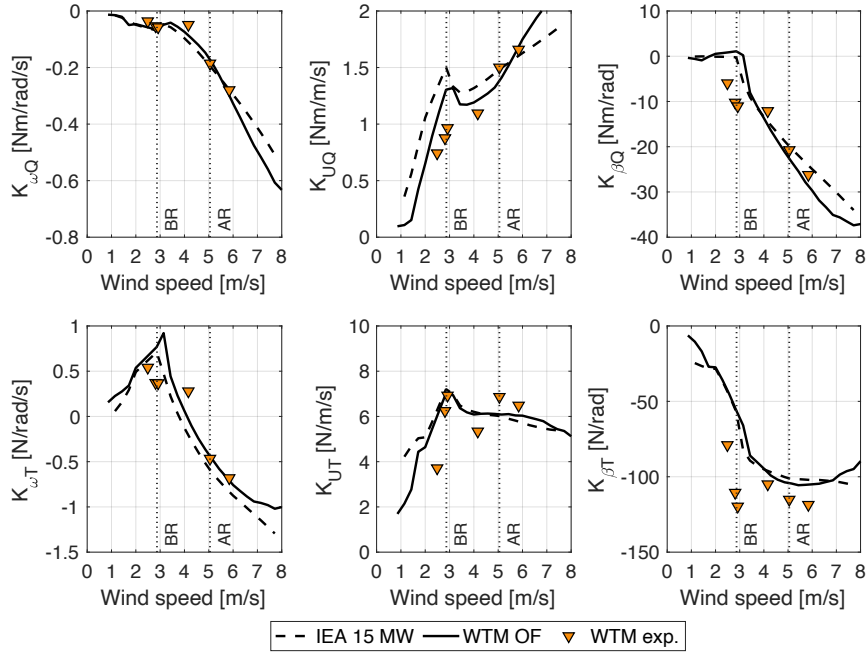


Figure 7. Aerodynamic sensitivities at the steady-state operating points of Fig. 6 obtained from $C_P(\lambda, \beta)$ and $C_T(\lambda, \beta)$ coefficients with the expressions of Appendix A. Experimental results (WTM exp.) are compared to values computed from the OpenFAST model of the turbine scaled model (WTM OF) and to the IEA 15 MW at model scale. Vertical dotted lines identify the below rated (BR) and above rated (AR) operating conditions considered for tests with platform motion.

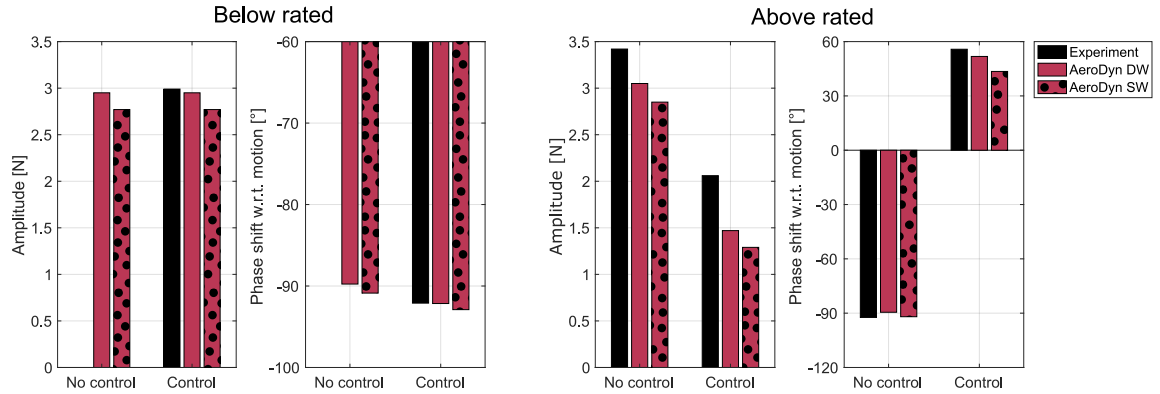


Figure 8. Amplitude and phase shift with respect to platform motion of aerodynamic rotor thrust with $A_m = 2.2^\circ$, $f_m = 2.25$ Hz, with a below rated wind speed of 2.87 m/s and an above rated wind speed of 5.05 m/s, from the experiment and the stand-alone AeroDyn model with dynamic wake (DW) and static wake (SW).

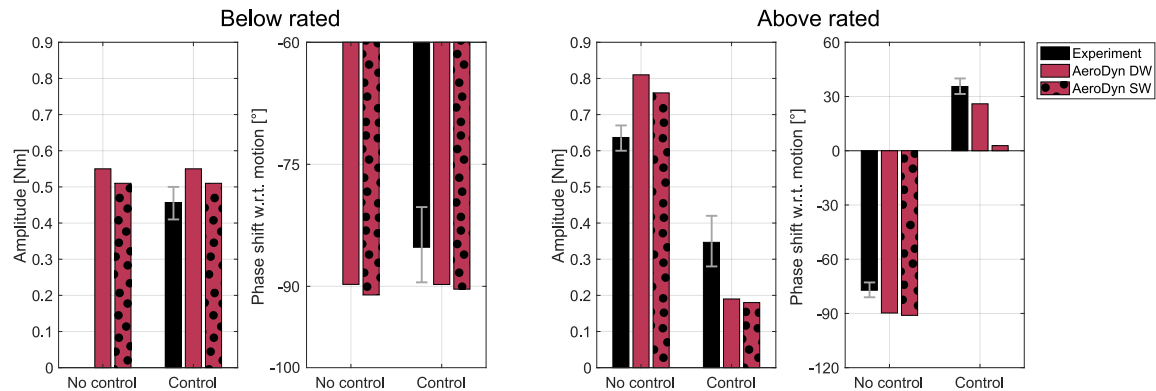


Figure 9. Amplitude and phase shift with respect to platform motion of aerodynamic rotor torque with $A_m = 2.2^\circ$, $f_m = 2.25$ Hz, with a below rated wind speed of 2.87 m/s and an above rated wind speed of 5.05 m/s, from the experiment and the stand-alone AeroDyn model with dynamic wake (DW) and static wake (SW). Whiskers show the experimental aerodynamic torque with a variation of rotor inertia of $\pm 20\%$.

(0.279 kgm²), which is considered a reasonable error for this parameter. Overall, in the below rated condition, the aerodynamic loads are captured by AeroDyn. This can be in part due to the fact that rotor speed oscillations are not large enough to create a challenging unsteady condition (as it is instead done in Bergua et al. (2023)), and indeed the AeroDyn-DW solution is very close to AeroDyn-SW.

Above rated, with fixed rotor speed and blade pitch, the amplitude and phase of aerodynamic loads are again similar for the AeroDyn model with static or dynamic wake. AeroDyn under-predicts the amplitude of thrust measured in the experiment, and over-predicts the amplitude of torque; the phase shift of thrust with respect to motion is approximately -90° and is captured well. The blade pitch actuation alleviates aerodynamic loads, resulting in smaller amplitudes compared to the case without control. The reduction of thrust and torque predicted by AeroDyn is greater than in the experiment, it is similar for the DW and SW models, with the DW model slightly closer to the experiment. The phase shift is captured well by AeroDyn-DW whereas it is under predicted by AeroDyn-SW. In the above rated condition, differences between numerical simulations and the experiment are larger than below rated. This can be in part due to blade pitch actuation, which is known to cause dynamic inflow effects (Snel and Schepers (1995)), and, in the case of torque, can also be due to uncertainties in the estimation of aerodynamic loads from experimental measurements.

5.4 Coupled response with platform pitch motion

The turbine response measured in the experiment is compared to the OpenFAST model of the WTM, which includes the ROSCO. The comparison is done based on the time series of rotor speed, blade pitch, aerodynamic thrust and torque recorded over a number of periods, that are phase-averaged to filter harmonic contributions not due to platform motion.

Figure 10 shows the WTM response with a below-rated wind speed of 2.87 m/s. Blade pitch is saturated and the controller responds with actuation of generator torque. All signals exhibit a first-order sine wave, thus the wind turbine response is driven

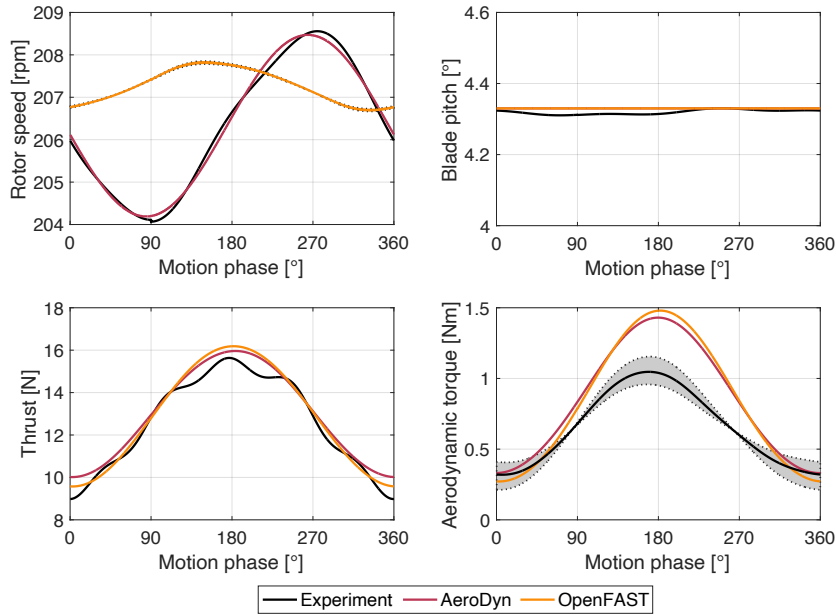


Figure 10. Phase-averaged turbine response with platform pitch motion of $A_m = 2.2^\circ$, $f_m = 2.25$ Hz and a below rated wind of 2.87 m/s, measured in the experiment and computed in AeroDyn (with dynamic wake) with prescribed rotor speed-blade pitch, and in OpenFAST with the Simulink controller. The grey area shows the aerodynamic torque estimate with a variation of rotor inertia of $\pm 20\%$.

by a single frequency corresponding to platform motion. In the OpenFAST-Simulink model, the rotor speed has a peak when the motion phase is 150° , and a peak-to-peak amplitude of 1 rpm. The peak-to-peak amplitude of rotor speed oscillations in the experiment is 4.5 rpm and the maximum is reached when the motion phase is 275° . The thrust force response is similar in the experiment, in OpenFAST, and in the stand-alone AeroDyn simulation. In all cases, the thrust peak is close to 180° , when the hub moves upwind with maximum velocity. Also the torque peak is close to 180° , and the amplitude of oscillations in the experiment is slightly lower than in the OpenFAST model and in AeroDyn (the grey shaded area in the figure shows how the aerodynamic torque estimate changes with a variation of rotor inertia of $\pm 20\%$), but the mean value of torque in the simulations (0.86 Nm) is higher than in the experiment (0.66 Nm), as seen in Fig 5. The variations of thrust and torque of the OpenFAST-Simulink model are close to the experiment and similar to the stand-alone AeroDyn model, despite the different amplitude and phase of rotor speed oscillations. Being aerodynamic torque variations similar in AeroDyn and OpenFAST-Simulink, one plausible reason for the different rotor speed response is the generator dynamics which is not modeled in OpenFAST (in Simulink the torque set point is followed perfectly by the generator, whereas in the physical WTM the generator has a torque control loop with a dynamic response; this dynamic response is hard to model because the torque controller is embedded in the generator drive and its parameters are not disclosed to users).

Figure 11 shows the phase-averaged WTM response with an above-rated wind speed of 5.05 m/s. In this case, generator torque is saturated to its rated value, and the controller responds with actuation of collective blade pitch. Signals generally

exhibit a first-order sine wave, but higher-order effects are seen in the blade pitch and aerodynamic loads of the OpenFAST-Simulink simulation. The peak-to-peak amplitude of rotor speed and blade pitch variations is lower in the experiment (3 rpm

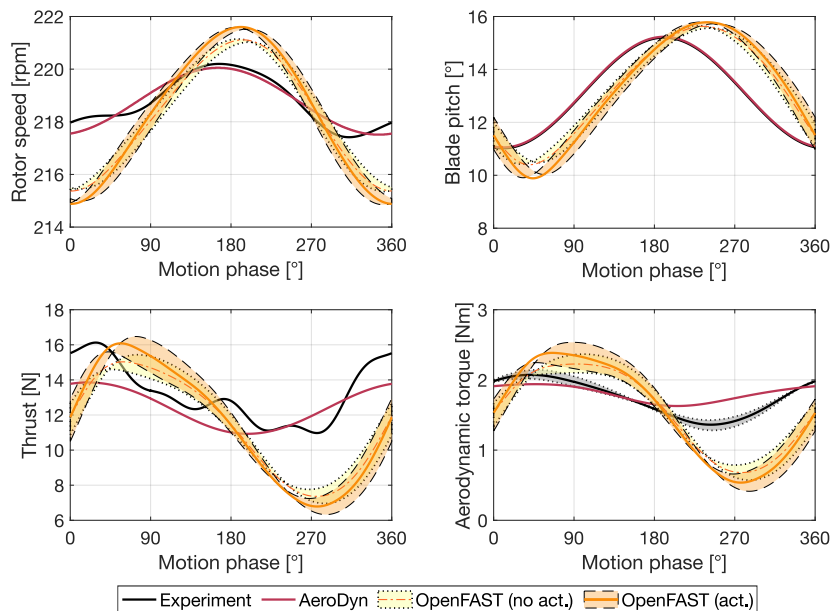


Figure 11. Phase-averaged turbine response with platform pitch motion of $A_m = 2.2^\circ$, $f_m = 2.25$ Hz and an above rated wind of 5.05 m/s, measured in the experiment and computed in AeroDyn (with dynamic wake) with prescribed rotor speed-blade pitch, and in OpenFAST with the Simulink controller. Two OpenFAST-Simulink models are shown, one with and one without the pitch actuator model. The shaded areas show the variation of metrics due to a change in rotor inertia of $\pm 20\%$.

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and 4.1°) than in the OpenFAST-Simulink model (6.7 rpm and 6°); the phase shift between the rotor speed peak and the blade pitch peak is lower in the experiment than in the simulation. When OpenFAST simulations are repeated excluding the pitch actuator model (i.e., $G_{act}(s) = 1$), the amplitude of rotor speed and blade pitch variations is slightly lower than with the actuator model, but still higher than in the experiment. Oscillations of thrust and torque in the experiment have lower amplitude than in OpenFAST-Simulink. The blade pitch excursions in this operating condition are large enough to influence the aerodynamic response of the wind turbine, and modeling the blade pitch control system is critical to capture the wind turbine aerodynamic loading. The pitch actuator may behave differently with wind (i.e., when there is an aerodynamic moment on the blade) than in still air; this difference would not be captured by pitch actuator model and explains the discrepancies in the rotor speed-blade pitch response of OpenFAST compared to the experiment.

405 6 Conclusions

In this article, a theoretical framework is proposed to downscale the reference open-source controller (ROSCO) and use it to control a 1:100 scale model of the IEA 15 MW turbine. The controller preserves the algorithms of its full-scale version, but it

is run in real-time at model scale to respect the time scaling of the experiment. Due to this choice, the controller parameters are downscaled. The controller scaling procedure is model based, and uses information about the aerodynamic response and inertial properties of the model turbine to preserve, at small-scale, the rotor response of the full-scale turbine. The aero-servo-dynamic response of the wind turbine scale model with ROSCO is studied in a wind tunnel experiment with load cases where tower base is fixed and with large prescribed platform pitch motion that mimics a floating foundation.

The experiment is modeled in OpenFAST and results of simulations are compared to those of testing to verify the controller scaling and identify the main sources of uncertainty in the modeling of the wind turbine scale model closed-loop response. The steady-state rotor speed-blade pitch-thrust-torque characteristics of the scaled turbine match those of the IEA 15 MW and are correctly captured by the OpenFAST model. The largest differences are attributed to an offset in blades pitch present in the experiment, to a lower airfoil efficiency for the physical blade compared to its OpenFAST model, and to uncertainty of the drivetrain efficiency required for generator control. The wind turbine controller is scaled making use of a linearized aerodynamic model built based on OpenFAST data. The linearized aerodynamics of the turbine scale model is captured by OpenFAST and is close to the IEA 15 MW.

With platform pitch motion, the turbine response is modeled with different accuracy in the below rated and above rated regions. Below rated, the aerodynamic response is driven by the rotor-apparent wind created by motion more than rotor speed oscillations, which are relatively small. The aerodynamic model of OpenFAST captures the loads measured in the experiment, and the agreement is better for thrust than for torque, whose estimate is affected by the uncertain knowledge of rotor inertia. The rotor-speed response in OpenFAST is different than in the experiment, and this is attributed to the lack of a model for the generator actuator. However, with the control strategy and operating condition we considered here, rotor speed oscillations do not cause significant unsteadiness, thus differences in the rotor speed dynamics are not critical for modeling the aerodynamic loads response due to platform motion. Above rated, rotor speed is regulated with collective blade pitch. The amplitude of aerodynamic loads variations is driven by the combination of apparent wind, and blade pitch oscillations. Differences between the aerodynamic loads computed in OpenFAST and those measured in the experiment are larger than in the below rated condition. In particular, the decrease in loads due to blade pitch variations is overestimated in AeroDyn compared to the experiment. Slightly better agreement is obtained with the use of dynamic BEM compared to a static wake model. Phase of blade pitch variations in OpenFAST is different than in the experiment, and this influences the amplitude and phase of aerodynamic loads in the simulation. The different response obtained in OpenFAST can be due to the scale model blade-pitch actuators behaving differently with wind than in still air (i.e., where the pitch actuator model is obtained), the aerodynamic model of OpenFAST not completely capturing the experiment, or a combination of the two.

In conclusion, this work has provided guidance on how to include wind turbine control functionalities in scale model testing of floating wind turbines. It has also confirmed the aerodynamic load response with platform motion and active control is more difficult to model than when rotor speed and blade pitch are fixed, in particular in the above rated region where large blade pitch excursion occur. It is shown that knowledge of the blade pitch actuator response is important to model the aero-servo-dynamic response of a floating turbine in above rated wind speeds.

In future work, more codes, possibly of higher fidelity than the BEM we considered here, can be used to study the wind tunnel experiment and this will help understand which physics of the wind turbine system are important to consider when modeling the aerodynamic response of a floating wind turbine with active control. The present research examined the case of active turbine control and prescribed motion, of low-frequency and large amplitude, in the pitch direction. Floating wind turbines can experience large wave-frequency motions that should be examined in future experiments. Moreover, only thrust and torque loads have been investigated in the present study, but the other four components of the aerodynamic loading are important for the response of some floater concepts (Bachynski et al. (2015)) and should be addressed in future wind tunnel studies.

Data availability. The OpenFAST model, the MATLAB Simulink version of the reference open-source controller ROSCO, and experimental data can be downloaded at Fontanella et al. (2023b).

Appendix A: Analytical expressions of rotor aerodynamic sensitivities

We provide here the analytical expressions of rotor aerodynamic sensitivities. The rotor speed to rotor torque sensitivity is:

$$K_{\omega Q} = \frac{Q_0}{\omega_{r,0}} \frac{\partial C_Q}{\partial \lambda} \bigg|_0 \frac{\lambda_0}{C_{Q,0}}, \quad (\text{A1})$$

455 the wind speed to rotor torque sensitivity is:

$$K_{UQ} = \frac{Q_0}{U_0} \left(2 - \frac{\partial C_Q}{\partial \lambda} \bigg|_0 \frac{\lambda_0}{C_{Q,0}} \right), \quad (\text{A2})$$

and the collective blade pitch angle to rotor torque sensitivity is:

$$K_{\beta Q} = \frac{1}{2} \rho \pi R^3 U_0^2 \frac{\partial C_Q}{\partial \beta} \bigg|_0, \quad (\text{A3})$$

460 where $\partial C_Q / \partial \lambda$ and $\partial C_Q / \partial \beta$ are the two components of the C_Q gradient. The torque coefficient is computed from the power coefficient as $C_Q = C_P / \lambda$.

The rotor speed to rotor thrust sensitivity is:

$$K_{\omega T} = \frac{T_0}{\omega_0} \frac{\partial C_T}{\partial \lambda} \bigg|_0 \frac{\lambda_0}{C_{T,0}}, \quad (\text{A4})$$

the wind speed to rotor thrust sensitivity is:

$$K_{UT} = \frac{T_0}{U_0} \left(2 - \frac{\partial C_T}{\partial \lambda} \bigg|_0 \frac{\lambda_0}{C_{T,0}} \right), \quad (\text{A5})$$

465 and the collective blade pitch angle to rotor thrust sensitivity is:

$$K_{\beta T} = \frac{1}{2} \rho \pi R^2 U_0^2 \frac{\partial C_T}{\partial \beta} \bigg|_0, \quad (\text{A6})$$

where $\partial C_T / \partial \lambda$ and $\partial C_T / \partial \beta$ are the two components of the C_T gradient.

Appendix B: Frequency response of the turbine controller with input/output scaling

470 We demonstrate that running the turbine controller at full-scale with scaling of input and output signals does not preserve the turbine frequency response.

Let us consider here the response of the wind turbine to a change in wind speed when it function in full load and rotor speed is controlled with a PI collective blade pitch controller. Assuming the wind speed rate of change is low, the variation of rotor speed is:

$$\bar{\omega}_r = - \frac{K_{UQ}}{K_{\omega Q}} \bar{U}. \quad (\text{B1})$$

475 This variation of rotor speed is counteracted by the PI collective pitch controller, which frequency response function for the IEA 15 MW is shown in Fig. B1.

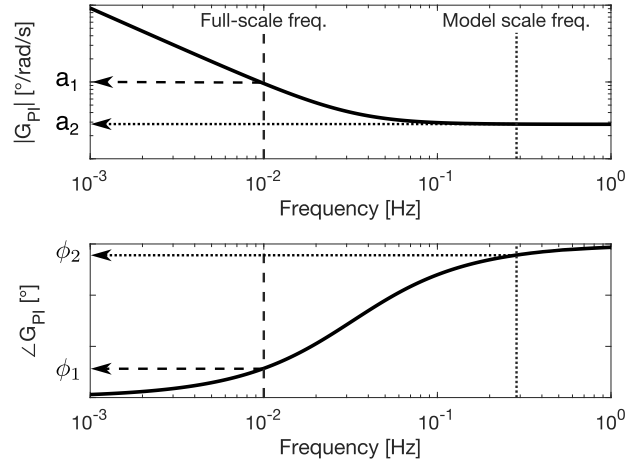


Figure B1. Frequency response function (amplitude and phase) of the IEA 15 MW PI collective blade pitch controller. The frequency response function evaluated at “Full-scale freq.” has amplitude a_1 and phase ϕ_1 . If it is evaluated at “Model scale freq.” it has amplitude a_2 and phase ϕ_2 .

We assume the variation of wind speed for the full-scale turbine is harmonic. In general, this can be due to a wind gust or due to the apparent wind created by motion in case of a floating wind turbine. The wind speed variation is:

$$\bar{U} = u \sin(2\pi f_w t), \quad (\text{B2})$$

480 where f_w is the frequency of the wind speed oscillations. Using Eq. B1, the rotor speed response due to the wind speed is:

$$\bar{\omega}_r = -\frac{K_{UQ}}{K_{\omega Q}}(u \sin(2\pi f_w t)). \quad (\text{B3})$$

The blade pitch controller reaction to this oscillation is:

$$\bar{\beta} = a_1 \left(-\frac{K_{UQ}}{K_{\omega Q}} \right) (u \sin(2\pi f_w t + \phi_1)), \quad (\text{B4})$$

where a_1 and ϕ_1 are the amplitude and phase of the PI pitch controller frequency response function at f_w .

485 Let us consider a scaled version of the full-scale turbine. The wind speed oscillation for the model has scaled amplitude and scaled frequency (e.g., when the turbine is mounted on a scale model of the floating platform):

$$\bar{U} = (u \lambda_v) \sin(2\pi f_w \lambda_f t), \quad (\text{B5})$$

where $\lambda_f = \lambda_v \lambda_L^{-1}$. Assuming the rotor response of the turbine model is ideally scaled:

$$\bar{\omega}_r = \left(-\frac{K_{UQ}}{K_{\omega Q}} \frac{1}{\lambda_L} \right) (u \lambda_v) \sin(2\pi f_w \lambda_f t) \quad (\text{B6})$$

490 When the turbine controller is operated in real-time in full-scale mode, inputs from the turbine model are scaled up to full-scale values before going into the controller. For rotor speed:

$$\bar{\omega}_{r,fs} = \left(\left(-\frac{K_{UQ}}{K_{\omega Q}} \frac{1}{\lambda_L} \right) (u \lambda_v) \frac{1}{\lambda_f} \right) \sin(2\pi f_w \lambda_f t), \quad (\text{B7})$$

where $\lambda_v \lambda_L^{-1} \lambda_f^{-1} = 1$. The blade pitch controller response is:

$$\bar{\beta} = a_2 \left(-\frac{K_{UQ}}{K_{\omega Q}} \right) (u \sin(2\pi f_w \lambda_f t + \phi_2)), \quad (\text{B8})$$

495 where a_2 and ϕ_2 are the amplitude and frequency of the PI pitch controller frequency response function at $f_w \lambda_f$. $\bar{\beta}$ is non dimensional and, with the full-scale controller approach, it is applied to the turbine scale model without any further operation.

Comparing Eq. B4 to Eq. B8 we see they have different amplitude and phase, that are due to the frequency response function of the PI blade pitch controller rather than dimensional scaling.

Appendix C: List of symbols

Table C1 is a list of the symbols that occur most often in the text.

Table C1. List of symbols.

| Symbol | Meaning |
|----------------|--|
| A_m | Amplitude of sinusoidal pitch motion |
| f_m | Frequency of sinusoidal pitch motion |
| h | Drivetrain closed-loop damping ratio |
| J^* | Drivetrain (rotor and generator) inertia |
| $K_{\beta Q}$ | Blade pitch-to-torque sensitivity |
| $K_{\omega Q}$ | Rotor speed-to-torque sensitivity |
| Q_a | Aerodynamic rotor torque |
| Q_g | Generator torque |
| U | Free-stream wind speed |
| β | Collective blade pitch |
| η_g | Drivetrain efficiency |
| λ | Tip-speed ratio |
| λ_L | Length scale factor |
| λ_v | Velocity scale factor |
| τ_g | Drivetrain transmission ratio |
| ω_g | Generator speed |
| ω_r | Rotor speed |
| Ω | Drivetrain closed-loop frequency |

Author contributions. AF and ED devised the methodology to downscale the wind turbine controller and created the simulation model. FN, AF and MB designed the wind tunnel experiment, that was run by AF, ED and FN. ED and AF analyzed the experimental data and run the numerical simulations. MB supervised the research project. All coauthors thoroughly reviewed the article.

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