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Wind Energy Science Discussion

Date: February 28, 2025 Subject: WES-2024-140 Final Response

Dear Referees,

We would like to express our sincere gratitude for the time and effort you have devoted to reviewing our manuscript and for providing such valuable feedback. Your comments have highlighted important aspects we had not fully addressed in the initial submission, and we believe that incorporating your suggestions has significantly improved the quality and impact of our work.

We have carefully considered each of your comments and have revised the manuscript accordingly.

On behalf of all Authors, sincerely,

Alessandro Fontanella

Response to Referees comments

In this document, **R1** refers to Referee 1, **R2** to Referee 2, and **AC** to our response. We have organized the comments by section and, where possible, grouped similar comments together, providing a single response to avoid redundancy.

After the point-by-point response, we have included a copy of the revised article, where the main changes made to address the Referees' comments are highlighted in blue.

General comments

R1: One of the main considerations overlooked in the current work is potential interaction between periodic turbulence statistics in the wake, and the imposed platform motion. These interactions could lead to harmonics visible in velocity measurements, which are currently shown in terms of the phase of the imposed platform motion.

AC: we introduced a new figure (Fig. 10) that illustrates the evolution of turbulence and periodic wakes structures across the wake through the power spectral density of the velocity.

R2: The results would be much easier to follow and to compare with existing literature (previous lab experiments, field measurements, and models) if they were reported nondimensionally. Dimensional results are specific to the setup and make it hard to compare to other studies.

AC: we agree that reporting results in non-dimensional quantities enhances comparability with other studies. We clarified the dependency on reduced frequency in the figures showing loads and wake quantities. Additionally, we normalized velocities by the free stream wind speed and used diameters for the axes in wake figures.

R2: Relatedly, the authors can more clearly explain and justify the choice of nondimensional parameters that were investigated here. How were the amplitudes and frequencies (Strouhal numbers) selected?

AC: Thank you for your comment. We have added an explanation of how frequency and amplitude values were selected in Sect. 3.2 Platform motion conditions.

R2: The level of agreement with the linear regression in Figure 5 is overstated, which is important, because it affects the degree to which a linear model is justified (it seems not to be justified).

AC: We noted that "The variation in loads for motion cases at a frequency of 2 Hz exceeds the expected linear trend." This behavior was seen in past experiments but not simulations, suggesting that non-idealities in the experimental setup cause the higher-than-linear response at 2 Hz. These non-idealities should be considered when comparing experimental and simulation results but do not significantly impact the trends in wake dynamics.

R2: More problematically, in the conclusions on Line 350, the authors state: "The mean velocity within the wake, regardless of the direction of movement, closely matches that of the bottom-fixed scenario." This statement cannot be made, as it will depend on the nondimensional parameters. For some Strouhal numbers and amplitudes of motion, the wake will match bottom fixed, and for others not. The authors could rephrase as: "The mean velocity within the wake, regardless of the direction of movement, closely matches that of the bottom-fixed scenario for the dynamic regimes we have investigated here" or similar. Otherwise, the authors would need to investigate why their results differ from published studies including Messmer 2024 and others.

AC: We agree with your comment and have rephrased the sentence to clarify that our conclusion is valid for the conditions examined in the experiment.

Abstract

R1: "NETTUNO" — Acronym used before it is defined.

AC: NETTUNO refers to the project titled "Understanding Turbine-Wake Interaction in Floating Wind Farms Through Experiments and Multi-Fidelity Simulations," as noted in the Acknowledgments. For readability, we use the short title in the main text.

1 Introduction

R1: "pre-commercial stage" — Is this still true? There are commercial deployments of floating

offshore turbines already.

AC: It is uncertain whether floating wind has reached a commercial stage or if the current projects are still intended to demonstrate the technology's capabilities. To clarify, we rephrased the sentence to state that there are only a few small-sized projects.

R1: "Very recently, Ozinan et al. (2024) studied the near wake of a 2 MW floating wind turbine and found no evident impact of wave-induced motions on the average velocity of the wake, partially contrasting theoretical speculations." — Does this undercut the motivation for the work? Please clarify how findings in this reference influence the NETTUNO project.

AC: We clarified that full-scale experiments are impacted by uncertainties related to measurements and the lack of information regarding wind conditions and platform movements. Additionally, we explained that controlled wind tunnel experiments are essential for interpreting full-scale measurements and supporting their conclusions.

R1: "10 MW" and "15 MW turbine" — Please specify the model of turbine used as a basis of the scaled experiments.

AC: We specified the wind turbine models.

R2: The literature review is thorough and useful. It may be interesting to discuss the findings of recent articles on the subject of kinematic motion/unsteady inflow affects on power [1] and wakes [2,3], especially relevant to the studies of Messmer in 2024. **AC**: We cited some of the articles suggested by you and the other referee that were useful to comment on our results and support our conclusions.

R2: The motivation for the article on Line 47 is clear and concise. I would suggest also including research questions and hypotheses that you seek to address in this article.AC: We implemented your suggestion by adding the research questions and hypothesis in the introduction.

2.1 Wind turbine

R1: "The design of the blades and tower focused on maximizing stiffness to minimize their aeroelastic response." — Since measurement of structural loads are among the project goals, can the authors provide some more detail on how aeroelastic scaling was approached for the wind turbine model? The authors state that the goal is to maximize blade stiffness, but this is not likely to produce results that scale up to the 10 MW turbine model. More justification for this choice would help readers understand the approach.

AC: In Sect. 2.1, we clarified that the wind turbine scale model is designed to isolate aerodynamic loads and wake response due to platform motion from those caused by blade and tower flexibility, which are beyond this study scope.

R2: Section 2.1: Presumably the previous studies referenced here included validation of the lab model against the DTU 10 MW reference, but that should be made more explicit here (and perhaps included in an Appendix) so that this article is self contained.

AC: An Appendix has been included to present the validation of the aerodynamic design of the wind turbine scale model against its full-scale reference.

R2: Section 2.1: The authors should report the nondimensional numbers that govern the dynamics explicitly for both the wind tunnel tests and also the true DTU 10 MW reference turbine. Specifically: Reynolds number, tip-speed ratio, and Mach number for 'fixed bottom' turbine operation.

AC: The chord-based Reynolds number has been added in section 2.1. It is stated in section 2.1 that the tip-speed ratio of the scaled turbine is identical to that of the full-scale turbine, and section 3.1 specifies that the turbine operates at a TSR of 7.5. Discussing the Mach number seems unnecessary.

R1: "ensuring the rotor was vertical to the wind tunnel floor" — Does this mirror the 10 MW reference turbine design? Typically, the tower is nominally vertical but the rotor is not. Perhaps it would be instructive to state here that aligning the shaft with the mean flow is an intentional simplification, to limit loads to the axial thrust and moment in the baseline case.

AC: We implemented your suggestion at the end of Sect. 2.1.

2.2 Measurements

R1: "six-component loads" — Please be more specific about the loads measured in the introduction. This statement seems to indicate that only aggregate rotor forces and moments were measured. Were the tower bending or twist measured? Blade loads? **AC**: We clarified that the test campaign measured the three forces and three moments at the interface between the tower-top and nacelle. Tower bending and blade loads were not measured because these components are rigid in the wind turbine scale model. Therefore, tower loads can be derived from tower-top loads, and blade loads can be determined from the thrust force measurement.

R1: "The wind velocity in the turbine wake was measured using a hot wire anemometer." — While a hotwire is certainly a good system for pointwise turbulence measurements, using a single probe in isolation makes coherence and spatial correlations impossible. With such a capable traverse system, why not make multiple measurements simultaneously?

AC: The study used a single hot-wire anemometer assuming periodic wake response to platform motion. Velocity measurements were phase-aligned with the platform motion signal for spatial correlation. Accurate turbulence correlation requires simultaneous multipoint measurements, achievable through hot-wire probe arrays or advanced techniques like Particle Image Velocimetry (PIV). This is detailed in Sect. 2.2.

R2: Line 85: It seems limiting to only have a single wire rather than a cross wire, especially if yaw misalignment and sway are investigated, where lateral velocities become non-negligible.

AC: As noted by the reviewer, certain motion conditions may cause oscillatory velocities in unmeasured components. Section 2.2 explains that we measured the velocity in the xi direction (u) because it is predominant in the studied wake region. The time-averaged values of the other two velocity components are expected to be significantly lower than u, so our focus remained on the xi direction. The decision to measure only the u component is consistent with other recent experimental studies on wind turbine wakes.

R1: "The resulting average extended uncertainties, at a 95% confidence level, are 0.17 m/s for flow speed and 0.4% for turbulence intensity." — Thank you for including a quantitative description of the measurement uncertainty. **AC**: Thank you.

R2: Line 83: "Substantial agreement was found between the two measurements." This is a vague statement and the measurements are not shown. The measurements should be shown to confirm this or this statement should be removed

AC: At the start of Section 2.2, we have clarified the purpose of the two Pitot tubes and linked these instruments to the measurements described in the text.

3 Test scenarios

R1: "Turbulence intensity is 1.5%" — These are very low turbulence levels, likely to arise offshore during stable atmospheric conditions. Since the experiment presumably did not apply a temperature gradient across the boundary layer, I think "close to laminar," as the authors state it, is a good description.

AC: we agree with your comment, we left "close to laminar" and we removed the reference to the paper describing turbulence levels in offshore conditions.

R2: Line 107: Relative to the literature review in the introduction, especially Messmer et al., 2024b, the experiments were selected to be performed in laminar inflow. More discussion of this choice would be appropriate, as it will reduce the relevance of the wake measurements to the true system.

AC: Agreed. In Section 3, we explained that laminar inflow ensures a controlled environment, reducing external turbulence. This allows for a precise evaluation of the wake structure and turbulence generated by the turbine and platform motion.

R1: Also, given the confidence interval stated above, does this mean that the actual TI has $a \pm 27\%$ confidence interval (1.5% \pm 0.4% TI)?

AC: We explained in the text that the average turbulence intensity is 1.5% with variations ranging from a minimum of 1.2% to a maximum of 2.1% across the rotor disk.

R1: "reduced frequency:" — It may be helpful to readers to state that the reduced frequency is a normalized frequency similar to the Strouhal number. **AC**: we added a note as you suggested.

R2: Section 2.1: More discussion of the reduced frequencies investigated in this study compared to the full system is needed. Also, this is more commonly called the Strouhal number.

AC: In section 3.1 we added an explanation of how the frequencies of the experiment were selected. In response to R1 comment, we explained that the reduced frequency is similar to the Strouhal number. We prefer to use reduced frequency because this is often used in studies on floating wind turbines.

R1: "Normalized mean velocity and turbulence intensity (I)'' — The inflow average wind speed and TI are surprisingly variable across the rotor area. Does this arise from obstructions in the wind tunnel? Is this level of variability in y and z expected?

R2: Figure 2: There appears to be a consistent structure in the deviations from hub-height wind speed in figure 2(a). Do the authors have explanations for this? i.e. is it a boundary layer from the bottom surface? Interactions with the wall/flow speed up on the outside of the rotor. Boundary layer from the top surface?

AC: we clarified the spatial variability of mean wind speed and turbulence intensity arises from the wind tunnel design and is comparable to the one achieved in other wind tunnel studies on wind turbines.

3.1 Wind conditions and wind turbine settings

R2: Line 118: The experiments are conducted with a constant rotor speed. This is probably a good choice, because it eliminates controller feedback which would make the results more complicated. But I suggest discussing this choice and explaining it more.

AC: We explained that reproducing the actions of a feedback controller in scale model experiments is challenging and introduces uncertainty in the measurement of aerodynamic loads.

3.2 Platform motion conditions

R2: Line 138: Consider adding the angles introduced here to Figure 1. **AC**: done.

R2: Line 148: "With surge, the blades velocity is equal to the platform velocity and the apparent wind experienced by the rotor is: ..." The authors can more carefully define the 'apparent wind' terminology they are using because the velocity at the blades will be affected by induction. Similar comments throughout this section.

AC: We agree with your comment and have revised the text.

4 Data processing

R1: "This process was performed in the frequency domain by computing the complex FFT, keeping the frequency components up to 3 Hz, and then utilizing the inverse FFT to reconstruct the signal in the time domain." — This sort of low-pass filtering can introduce convolution errors that are visible in the filtered time-domain data. Were any other techniques used (e.g., zero-padding, Butter or other optimal filter design) to reduce convolution errors?

AC: The filtering process based on FFT-IFFT was chosen over two other options with optimal filters because the latter were less accurate in isolating the aerodynamic loads response due to platform motion. This is shown below.

Figure 1 presents the spectra of aerodynamic thrust and torque obtained from the tower top load cell under conditions of pitch motion with a frequency of 1Hz and an amplitude of 1.3°. The spectra display peaks corresponding to the rotor frequency (1P) and its multiple harmonics (e.g., 2P, 3P, etc.). These peaks result from a mass imbalance of 10% in one of the blades, aerodynamic imbalance (such as a slight blade-pitch offset), and the tower shadow effect.

The goal of filtering was to remove the loads response due to rotor asymmetry. Three filtering options were compared:

- 4th-order Butterworth filter with a 3 Hz cut-off frequency, compensating for delay.
- Stop-band filters with bandpass of 3-11 Hz and 13-22 Hz, plus a Butterworth filter with a 27 Hz cut-off. Stop-band filters use a minimum order with 60 dB attenuation. Stop-band filters and the low-pass filter compensate for delay. The goal of this filtering was to remove the 2P, 4P harmonics, and frequencies above 6P.
- FFT-IFFT with a 3 Hz cut-off frequency.



Figure 1. Power spectral density (PSD) of aerodynamic thrust and torque in the scenario with platform pitch motion at a frequency of 1 Hz and an amplitude of 1.3° resulting from various filtering procedures. The platform motion frequencies (0.5, 1, and 2 Hz) considered in the experiment are marked by dotted lines.

Figure 2 presents the time series of the aerodynamic thrust and torque after applying filtering and phase-averaging techniques. The thrust force response obtained using Butterworth and FFT-IFFT filtering methods are similar, whereas the response from the Stop-band+low-pass filtering method shows oscillations at the 3P frequency of 12 Hz (i.e., 12 high-frequency cycles within one cycle of platform motion at 1Hz). The torque response filtered with Butterworth and Stop-band+low-pass methods retains some of the response at the 1P frequency of 4 Hz, which is not completely attenuated due to the gradual roll-in of the optimal filters.



Figure 2. Phase-averaged time series of aerodynamic thrust and torque in the scenario with platform pitch motion at a frequency of 1 Hz and an amplitude of 1.3° resulting from various filtering procedures.

Based on these observations, it was decided to discard the filtering based on optimal filters because they cannot eliminate harmonics at the 1P frequency without affecting the response at 2 Hz, which is where platform motion occurs. The FFT-IFFT filtering introduces convolution errors; however: 1) it removes the response at 1P without impacting the response at 2Hz, and 2) the convolution errors are smaller than the oscillations in the signals due to rotor asymmetry, which cannot be fully eliminated with optimal filters.

The same FFT-IFFT filtering process with a cut-off frequency of 3 Hz was also utilized in the OC6-Phase III project (<u>https://doi.org/10.5194/wes-8-465-2023</u>), which examined aerodynamic loads obtained in a wind tunnel experiment using a setup similar to the one used in the present study.

We updated the manuscript by adding a sentence in Sect. 4 to explain the selection of the filtering process.

R1: "12 platform motion cycles" — Is this a large enough sample size to describe spectra and turbulence moments given the non-stationary (periodic) nature of the flow statistics? AC: A sensitivity analysis was conducted to examine the evolution of mean wind speed, turbulence intensity, and amplitude of the velocity spectrum at the frequency of platform motion. This is illustrated below for the scenario involving platform pitch motion with a frequency of 1 Hz and an amplitude of 1.3°. Figure 3 presents the progression of the three evaluated parameters as the number of platform motion cycles increases. Figure 4 depicts the development of the coefficient of determination (R2) of the three parameters in relation to their values obtained from evaluating 12 cycles of platform motion. In this context, R² measures how well a parameter computed on n cycles aligns with the parameter computed over 12 cycles. The results indicate that the three parameters exhibit minimal variation when the number of cycles exceeds 10. Among them, the mean wind speed converges the fastest, followed by turbulence intensity and spectrum amplitude. A sample size of 12 is considered a suitable balance between the accuracy of parameter estimation relevant to this study and the time required to conduct the measurements. A sentence was included at the end of Sect. 4 to summarize this analysis.



Figure 3. Progression of mean wind speed (left), turbulence intensity (middle), and amplitude of the velocity spectrum (right) at the frequency of platform motion as the number of platform motion cycles (nc) increases.



Figure 4. Evolution of the coefficient of determination of the three parameters of Figure 4 for increasing number of platform motion cycles.

5.1 Aerodynamic rotor loads

R1.1: By asking whether the traces in figure 4 represent "nominal curves," I mean to ask whether they are the measured responses of the platform or they represent a modeled or idealized response to the imposed motion.

R2: Figure 4: The experimental curves are remarkably smooth. Has any smoothing / postprocessing been applied to the plot?

AC: We clarified that the curves are derived from measurements, and the smooth time series indicate that the number of motion cycles included in the measurements, as well as the data processing method, effectively isolate the aerodynamic loads response to platform motion.

R2: Section 5.1: Useful to report dimensional loads but it would be much more useful to also report the nondimensional results so it can be more directly compared to the full-scale system, other wind tunnels, and models. Likewise, plotting the results against reduced frequency would be better than dimensional frequency.

AC: we agree that reporting results in non-dimensional quantities enhances comparability with other studies. We clarified the dependency on reduced frequency in the figures showing loads and wake quantities. Additionally, we normalized velocities by the free stream wind speed and used diameters for the axes in wake figures.

R2: Line 217: "The differences in the mean value of Fx are likely due to the zero bladepitch recalibration done during testing." Can the authors elaborate on what this means? **AC**: we added an explanation in Sect. 3.1. The test procedure required setting the zero blade-pitch position to ensure consistent operation of the wind turbine. This process was performed individually for each blade using an inclinometer and it was repeated regularly throughout the test campaign. The resulting zero blade-pitch position achieved an accuracy of $\pm 0.75^{\circ}$.

R1.1: "A linear regression is fitted to the measurements and is included in Figure 5a and Figure 5b. This method of representation demonstrates that the loads change linearly with respect to the platform motion amplitude, as evidenced by the normalized points aligning with the regression line.: --- The loads response indicated by the markers in Figure 5 do not appear to follow a linear trend. For the 2 Hz case, both the thrust and the Torque seem to deviate from a linear trend established by the other cases.

R2: Line 225: "This method of representation demonstrates that the loads change linearly with respect to the platform motion amplitude, as evidenced by the normalized points aligning with the regression line." I'm not sure I agree with this conclusion. It appears to me that the trend is not linear as evidenced by the lack of collapse at the higher frequencies.
R2: Line 227: "Additionally, the loads exhibit a linear increase with frequency." Same comment as above, but now with respect to frequency.

AC: We specified that "the loads change linearly with respect to the platform motion amplitude" is valid for the frequencies of 0.5 Hz and 1 Hz. It was already said that "the variation in loads for motion cases at a frequency of 2 Hz exceeds the expected linear trend".

R2: Line 233: "This deviation, already seen by Bergua et al. (2023), is attributed to fluctuations in rotor speed and the flexible response of the tower, which could affect the wind turbine behavior under these testing conditions." Is there any evidence of this or is this speculation? Specifically, I thought the turbine was designed for rigidity as explained earlier. Are the authors sure there are not other physical mechanisms causing nonlinear response?

AC: We have explained that it is unlikely that the above-linear trend is due to other physical mechanisms. This conclusion was derived from high-fidelity simulations of the wind turbine involved in this experiment. Please inform us if you have an alternative explanation for the nonlinear response.

R2: Line 248: Similar to above, it's not clear to me that the thrust changes are linear.

AC: We changed the sentence saying that "pitch and surge movement leads to periodic thrust changes *driven* by apparent wind speed fluctuations" and avoiding the word "proportional" that can be mistaken for "linearly proportional".

R2: Line 238: Is there a reason the authors selected such a low yaw amplitude? Even fixed bottom turbines exhibit yaw variations of 10 degrees from turbulence.

AC: The referee's comment is unclear. While 10° can be the instantaneous inflow skew angle, dynamic yaw variations of 10° are rare. Additionally, it is challenging to correlate harmonic motion from our experiment to the stochastic motion experienced by floating wind turbines or turbulent inflow skew. Anyway, we have added details on the selection of yaw motion amplitudes in Sect. 3.2.

5.2 Wake with platform pitch and surge motions

R2: Equation 8: Since the authors are using a single probe hot wire, this should be streamwise velocity rather than wind speed. Likewise for the streamwise standard deviation and streamwise TI.

AC: We agree with your comment and have revised the text.

R1.1: "The velocity profile has a double-gaussian shape, and it is mostly unaffected by platform motion" --- Can the authors elaborate why the velocity profile is asymmetric with y? There appears to be a lateral shift in the mean profile of approximately 0.15 m and the peak deficit in the right side of the plot shows complex behavior for all cases.

AC: Thank you for this comment. Previous measurements of the wake of the same turbine also showed asymmetry. This asymmetry is explained by the interaction of the tower with the rotor, as investigated in the study "Wind tunnel investigation on the effect of the turbine tower on wind turbines wake symmetry." The text at the beginning of Sect. 5.2 has been revised to clarify the possible reason for the wake asymmetry.

R1: "Figure 7c examines the velocity at xw = 3D and yi = 0.6 m over one motion cycle" — Wouldn't it be more meaningful to show the velocity measurements and apply the sinusoidal fit to the collective data of all 12 periods of motion? If the velocity trends are truly periodic and the phase relationship is consistent, this should provide a better least-squares fit. Please also indicate fit quality.

R2: Figure 7(c): Are these results phase averaged or instantaneous?

AC: We have clarified that the results shown in Fig. 7c represent phase-averaged data. Additionally, we amended the captions of the remaining figures to emphasize that they display phase-averaged rather than instantaneous data. We also introduced the parameter R^2 in the figures to demonstrate the fit of the sine function to the velocity data points derived from phase averaging. This parameter is now used in the text to provide a more thorough description of the results.

R1: "As shown in the graph, the sine wave at the motion frequency does not fit the velocity data as well as it does in the 0.5 Hz and 1 Hz cases." — How much of the high frequency oscillations in Figure 7c arise from the rotor motion? A rotor speed of 240 rpm corresponds to 4 Hz, which is similar enough to the imposed platform motion that one could expect some interaction.

AC: In response to the referees' comments, we added a figure showing the power spectral density of wind speed at various positions in the turbine wake. The spectra show no flow response at 4 Hz.

R1: "Among the conditions explored in this study, the strongest perturbation of the wake occurs when the reduced frequency of platform motion is 0.6'' — This case is not shown in figure 7. Should it be?

R2: Line 275: "Among the conditions explored in this study, the strongest perturbation of the wake occurs when the reduced frequency of platform motion is 0.6." Is this shown somewhere? If not, please add a result that proves this or remove such statements.

AC: The cases with a frequency of 1Hz have a reduced frequency fr = 1*2.4/4 = 0.6. To facilitate the readers, we include the reduced frequency value in the captions of all plots.

R1: "We believe that coherent flow structures, larger for f = 1 Hz at 3D (Fig. 7), are dissipated as they move downstream by generating small-scale turbulence, which increases the turbulence intensity and accelerates the wake transition." — It is not clear how this conclusion is drawn from the data in Figure 9.

AC: Thank you for your comment. We acknowledge that our comment to the results was not clear. To address this, we have introduced a new figure (Fig. 10) that illustrates the evolution of turbulence across the wake through the power spectral density of the velocity. This provides a clearer explanation of evolution of turbulence and flow structures coherent with platform motion.

R2: Line 282: "The asymmetry might be due to flow in inhomogeneity in the wind tunnel seen in Fig. 2, potentially reducing the correlation between wake flow structures. This hypothesis can be verified through numerical simulations of the experiment, incorporating a high-fidelity model of the wind tunnel inflow." This is a pretty weak statement. Further investigation should be performed with the available measurements.

AC: We agree that our comment was not justified properly. In response to R1 we explained that wake asymmetry can be the consequence of the interaction between the individual wakes of tower and rotor. Moreover, we introduced a new figure (Fig. 10) that illustrates the evolution of turbulence across the wake through the power spectral density of the velocity. With the new figure, we explain that dissipation of coherent flow structures occurs due to interaction with other turbulent structures and because of turbulent mixing. This process can be asymmetric because of the asymmetry of the flow (Fig.2) and the tower wake affecting the rotor wake.

Conclusions

R2: Line 350: "The mean velocity within the wake, regardless of the direction of movement, closely matches that of the bottom-fixed scenario." This statement cannot be made. This will depend on the parameters that govern the movement. One could say: for the parameters investigated, the mean velocity in the wake is similar to the fixed bottom scenario. But the statement as written is not correct.

AC: The conclusions were adjusted to state that, for the wind turbine operating conditions, motion scenarios, and distances from the rotor examined in this experiment, the mean velocity in the wake is comparable to that of the bottom-fixed case.

R2: Line 356: "In offshore environments turbulence intensity depends on the wave height and typically ranges from 3% to 6%." This is a statement which is far too over-simplified. Turbulence intensity in offshore environments does not 'depend' on the wave height only, but may partially depend on the wave height, or better yet, is 'correlated' with the wave height.

AC: The reference to wave height has been removed and the sentences in the conclusions have been rephrased.

References

AC: Among the articles recommended by the reviewers, we have cited the following:

[1] Wei, Nathaniel J., Adnan El Makdah, Jia Cheng Hu, Frieder Kaiser, David E. Rival, and John O. Dabiri. "Wake dynamics of wind turbines in unsteady streamwise flow conditions." Journal of Fluid Mechanics 1000 (2024): A66.

[2] Zhaobin Li, Guodan Dong, and Xiaolei Yang. "Onset of Wake Meandering for a Floating Offshore Wind Turbine under Side-to-Side Motion". In: Journal of Fluid Mechanics 934 (Mar. 2022), A29. ISSN: 0022-1120, 1469-7645. DOI: 10.1017/jfm.2021.1147.

Citing these articles is useful as they provide theoretical context about wake dynamics under alongwind and side-to-side motions, thereby strengthening the manuscript.

Additional changes

We have modified the colors of the lines in the figures after one of the authors noticed that the previous color scheme could be difficult to distinguish when printing the article.

The text has been thoroughly revised to improve readability and clarity.

Wake Development in Floating Wind Turbines: New Insights and Open Dataset from Wind Tunnel Experiments

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Abstract. Floating offshore wind turbines represent a promising advancement in renewable energy, yet they remain in early development stages with limited performance data. As part of the NETTUNO research project, this study investigates how platform motion affects the aerodynamics of a floating wind turbine rotor and connects its load response to the wake evolution. Wind tunnel experiments were conducted using a 1:75 scale wind turbine model subjected to platform motions in various direc-

- 5 tions. Measurements include rotor loads and wake velocities at multiple downstream distances. The results show that surge and pitch motions induce periodic thrust fluctuations, leading to significant variations in near-wake velocity, that have maximum amplitude at a reduced frequency of 0.6. Yaw motion causes oscillations in yaw moment and lateral wake meandering, while combined surge and sway motions result in skewed apparent wind, causing both wake velocity fluctuations and lateral meandering. Increased turbulence intensity near the wake center suggests enhanced mixing and potentially faster wake recovery
- 10 beyond 5 rotor diameters. This new experimental dataset serves as a foundation for validating numerical simulation tools and provides valuable insights for optimizing the design and layout of future large-scale floating wind farms.

1 Introduction

Floating wind power is the technology that can open the doors of deep-water sites to wind energy projects. At present, there is only a limited number of pilot floating wind plants with project sizes ranging from 10 to 50 MW (Barter et al., 2020). However,

- 15 upcoming projects are significantly larger and will include from tens to hundreds of wind turbines, reaching gigawatt scale. On that scale, wind turbines in the farm are going interact through wakes–areas behind rotors with lower energy and higher turbulence compared to free-stream wind (Meyers et al., 2022). To date, research on floating wind power has mostly focused on individual turbines, but upcoming projects raise concerns about wakes and aerodynamic interactions inside the wind farms. Unlike bottom-fixed turbines, floating turbines may experience significant movements that impact rotor aerodynamics and
- 20 wake formation. Moreover, full-scale measurements of floating wind turbine wakes are scarce. Very recently, Özinan et al. (2024) studied the near wake of a 2 MW floating wind turbine and found no evident impact of wave-induced motions on the average velocity of the wake, partially contrasting theoretical expectations. The authors of the study acknowledge that the conclusions are uncertain due to limitations in estimating wind speed, turbulence intensity, wind direction, and the lack

of information about atmospheric stability and floater motions. Controlled wind tunnel experiments are essential to interpret

25 full-scale measurements and support their conclusions.

To date, wind tunnel experiments on the effects of platform motion on the wake have been conducted using model turbines and porous disks. Bayati et al. (2017a) examined the effect of surge motion on the near wake of a 1:75 scale model of the DTU 10-MW wind turbine of Bak et al. (2013) and discovered that the wake has wind speed fluctuations due to motion. Fontanella et al. (2021) expanded the experiment to include more surge motion conditions and described how motion influences the

- 30 formation of tip vortices near the rotor. Fontanella et al. (2022a) carried out an experiment with a 1:100 scale model of the IEA 15 MW turbine of Gaertner et al. (2020), which was subjected to platform movements in different directions. They found that the average velocity and turbulence in the near wake were marginally lower with motion compared to the fixed case. Messmer et al. (2024a) experimentally investigated the wake of a floating wind turbine experiencing harmonic sway and surge movements in a wind tunnel with laminar inflow conditions. They found that sway movements cause wake meandering, while
- 35 surge movements create a pulsing wake, and both types of motion improve wake mixing. Messmer et al. (2024b) studied in the wind tunnel the wake of a scale model wind turbine subjected to surge motion with inflow turbulence intensity up to 3%. They found that as turbulence in the incoming wind increases, the wake recovery is less influenced by platform motion.

Porous disks have been utilized to investigate the wakes of floating wind turbines within atmospheric boundary layer flows. Schliffke et al. (2020) found that at 4.6 rotor diameters downstream, surge motion does not alter wake mean velocity, but it does

40 affect turbulence intensity and turbulent kinetic energy profiles. Schliffke et al. (2024) demonstrated that harmonic platform motion produces distinct frequency signatures in the far wake spectra, whereas broadband motion leaves no easily discernible marks.

The effects of platform motions in different directions on the wake of a floating wind turbine remain uncertain, and the interactions among floating wind turbines through their wakes have yet to be studied. The NETTUNO project (Nettuno, 2023)
aims to address these gaps by combining wind tunnel experiments with multi-fidelity simulations to advance the understanding of wake-turbine interactions in floating wind farms. In its initial phase, the project investigates how large platform movements affect wake development, while the subsequent phase will focus on wake-induced interactions between two floating wind turbines. This article presents a unique experimental approach with two main objectives. First, it examines how platform movements—typical of floating wind turbines and occurring in various directions—impact rotor aerodynamic loads. Second,
it links the rotor response to wake development. Unlike previous studies on floating wind turbine wakes, this work provides a

- publicly available dataset that includes extensive wake measurements at multiple downstream locations under various platform motion conditions. Furthermore, the wind tunnel experiment is conducted on a larger rotor, extensively studied in past international projects, offering robust insights into wake evolution in floating wind turbines. This study addresses two key research questions:
- 55
- How do platform movements in different directions impact rotor aerodynamics and wake dynamics?
 - How do variations in movement frequency and amplitude affect wake recovery and turbulence at various distances from the rotor?

To analyze these questions, we propose the following hypotheses:

- Platform motion-induced fluctuations in thrust force significantly alter wake structure and turbulence.
- 60 The wake response closely follows the periodicity of platform motion.

Insights into wake evolution from the experiment can be utilized to optimize wind farm layouts, design floaters, and enhance control strategies at both the turbine and farm levels. Collected measurement data are accessible to the community and serve as a foundation for validating numerical simulation tools.

The paper is organized as follows. Section 2 presents the setup used for carrying out the wind tunnel experiment, which 65 includes the wind turbine scale model and the measurements taken in the test campaign. Section 3 describes the test scenarios. 65 Section 4 explains the processing that was applied to measurement data collected in the experiment. Key results are discussed 65 in Sect. 5. Section 6 draws the conclusions and proposes suggestions for future work.

2 Wind tunnel experimental setup

The experimental campaign was conducted in the atmospheric-boundary layer test section of Politecnico di Milano wind tunnel,
which is 13.84 m wide by 3.84 m high by 35 m long. The test turbine, shown in Fig. 1, has a rotor diameter (*D*) of 2.38 m and it is mounted on a six-degrees-of-freedom robotic platform that mimics the rigid-body motions of floating foundations. This setup has been used in various experiments about the aerodynamics of floating wind turbines and the data from this turbine, collected in earlier projects (Fontanella et al., 2021), were recently utilized in the International Energy Agency Task 30 to verify the accuracy of the aerodynamic response predicted by various offshore wind modeling tools (Bergua et al., 2023; Cioni et al., 2023).

2.1 Wind turbine

The wind turbine is a 1:75 scaled version of the DTU 10-MW (Bak et al., 2013); its geometry is summarized in Table 1. The design of the rotor blades was aimed at mimicking the load distribution of the DTU 10-MW blades, achieving similar thrust and power coefficients. To meet this goal, the wind speed was adjusted by a scale of 1:3 while ensuring the tip-speed-ratio

- 80 remained the same as that of the full-scale turbine. Due to the reduced wind speed, the wind turbine scale model has lower chord-based Reynolds numbers $(50 \cdot 10^4 \text{ to } 1.5 \cdot 10^5)$ compared to those of the full-scale turbine $(10^6 \text{ to } 10^7)$. The blades utilize the low-Reynolds airfoil SD7032, with modified chord and twist to meet the desired thrust force (Bayati et al., 2017b). The rotor thrust determines the wake velocity deficit and so the chosen scaling approach guarantees that the wake produced by the wind turbine scale model accurately reflects that of a full-size turbine (Wang et al., 2021). The aerodynamic design of the scale
- 85 model rotor was verified prior to this experiment, with the results detailed in Appendix A. The design of the blades and tower focused on maximizing stiffness to minimize their aeroelastic response. This design philosophy aims to distinctly differentiate the wind turbine aerodynamic loads and wake response due to platform motion from the response caused by the flexibility of blades and tower, which is not within the scope of this study.



Figure 1. (a): the wind turbine scale model on top of the robotic platform during testing. The traversing system used to measure the wake is visible in the background. (b): schematic representation of the experimental setup with the coordinate systems: $(x_i - y_i - z_i)$ is the inertial reference frame, $(x_p - y_p - z_p)$ is the platform reference frame, and (x - y - z) is the hub reference frame. x_w is the distance from the wind turbine hub to the section where wake measurements were conducted.

The shaft of the wind turbine scale model had a tilt of 5°, matching the full-scale version of the DTU 10-MW. During wind 90 tunnel tests, the tower was tilted at a negative angle of 5° to align the wind turbine shaft with the mean flow and ensure the rotor was vertical to the wind tunnel floor. This simplification aimed to limit loads to the axial thrust and moment in scenarios without prescribed platform motion.

Table 1. Geometry of the wind turbine used in the experiment.

| Parameter | Unit | Value |
|-------------------------|------|-------|
| Rotor diameter (D) | m | 2.381 |
| Blade length | m | 1.102 |
| Hub diameter | m | 0.178 |
| Hub height | m | 2.190 |
| Rotor overhang | m | 0.139 |
| Shaft tilt angle | 0 | 5 |
| Tower-to-shaft distance | m | 0.064 |
| Tower length | m | 1.400 |
| Tower diameter | m | 0.075 |
| Tower base offset | m | 0.730 |

2.2 Measurements

The test campaign measured the three forces and three moments at the interface between tower-top and nacelle, the undis-

95 turbed wind speed, and the streamwise velocity in the turbine wake. Rotor forces were measured using a six-component force transducer (ATI Mini45 SI-580-20) mounted at the top of the tower. Movements of the robotic platform were recorded with MEL M5L/200 laser sensors and the rotor rotational speed with a magnetic encoder.

The undisturbed wind speed was measured at two locations using Pitot tubes. One tube was positioned 7.5 m ahead of the wind turbine at hub height, while the second was placed laterally at $x_i = 2$ m, $y_i = 5$ m, and $z_i = 3$ m. The Pitot tube positioned upstream of the wind turbine recorded the free-stream wind velocity, while the lateral Pitot tube measured the wind

speed affected by blockage effects.

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The streamwise velocity in the turbine wake was measured using a hot wire anemometer. A DANTEC 55P11 single-sensor probe with a 5 μ m wire diameter was employed. The wire is perpendicular to the probe axis, and it is capable of measuring mean and fluctuating velocities of one-dimensional flow. In the wind tunnel campaign, the velocity in the x_i direction (*u*)

- 105 was measured, consistent with other recent experiments (Messmer et al., 2024a,b). While certain motion conditions in the experiment may induce oscillations in the crosswind components, their time-averaged values are expected to be an order of magnitude lower than u in the studied wake region. The probe was connected to DISA55M systems operating in constant temperature mode, providing a high dynamic response of up to tens of kHz, making it ideal for turbulence measurements. Prior to testing, the probe was calibrated using a reference nozzle over the expected velocity range.
- 110 The use of a single hot-wire anemometer in this study was based on the assumption that the wake response to platform motion is predominantly periodic. Spatial correlation of velocity measurements was reconstructed by phase-aligning the data with the platform motion signal. To describe the spatial correlation of turbulence components that are different from the flow structures associated with the platform periodic motion, simultaneous measurements at multiple points are necessary. This could be achieved using arrays of hot-wire probes or advanced velocimetry techniques such as Particle Image Velocimetry

115 (PIV), which allow for detailed and simultaneous velocity field acquisition.

The measurement uncertainty for the hot wire system was determined using a Monte Carlo simulation, with uncertainties taken into account throughout the entire measurement process, beginning with the probe calibration. Factors contributing to the uncertainty included the accuracy of the data acquisition board and the pressure transducer, which influences both the reference flow velocity during calibration and the upstream wind speed during wind tunnel tests. Additionally, the uncertainty of the thermocouple, which adjusts the hot wire voltage for ambient temperature variations, was considered. The resulting average extended uncertainties, at a 95% confidence level, are 0.17 m s^{-1} for flow speed and 0.4% for turbulence intensity.

The hot wire probe was mounted on an automatic traversing system with two motorized axes, allowing movement in the crosswind $(x_i - y_i - z_i)$ plane. Velocities were measured along lines centered on the rotor hub, both in the horizontal direction (along the y_i -axis) and in the vertical direction (along the z_i -axis). Horizontal measurements were made at 35 evenly spaced points, and vertical measurements at 27 points. The distance between consecutive measurement point is 0.1 m. The traversing system was positioned at three distances downstream of the rotor to study wake evolution: $x_w = [3, 4, 5]D$.

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3 Test scenarios

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The test scenarios were chosen by revisiting previous studies and considering recent findings in floating wind turbine aerodynamics. While some turbine operating parameters and motion conditions were kept consistent with past research for continuity,

- 130 other motion cases were selected to induce various excitations in the turbine wake. All tests were conducted in low-turbulence wind conditions, with relatively uniform velocity and turbulence intensity across the wind tunnel section. This choice was made to isolate the effects of platform motion on rotor aerodynamics and wake dynamics. A low-turbulence inflow provides a controlled and repeatable environment, allowing for a more precise evaluation of the wake structure and the turbulence generated directly by the turbine and platform motion. However, we recognize that offshore environments typically exhibit moderate
- 135 turbulence intensities (e.g., 3–6%), shaped by atmospheric boundary layer dynamics and wave conditions (Türk and Emeis, 2010). As a result, the wake behavior observed in this study may differ from that in real-world conditions. At the same time, large-eddy simulations of Pagamonci et al. (2025) have shown that the experiment low-turbulence inflow is not equivalent to a laminar inflow and still has a significant impact on wake development.

Prior to testing, wind characteristics were measured and are shown in Fig. 2. The mean velocity over the rotor area varies

- 140 by 5% compared to the hub mean velocity and the average turbulence intensity across the rotor disk is 1.5%, varying from 1.2% to 2.1%. These spatial variations of mean velocity and turbulence intensity are consistent with those reported in other wind tunnel facilities conducting experiments on wind turbines (Schottler et al., 2018). The spatial variability in mean wind speed and turbulence intensity originates from the wind tunnel configuration. The facility has 14 fans arranged in two rows to generate airflow, which introduces inflow inhomogeneity within the return duct that houses the atmospheric-boundary layer
- test section. Additionally, turbulence intensity increases near the wind tunnel ceiling due to interactions between the airflow and the tunnel walls. Figure 2c displays the power spectral density of wind speed at the wind turbine hub. The air density during testing was 1.185 kg m^{-3} .

Every test scenario is defined by the free-stream average wind velocity U_0 , the wind turbine rotor speed, the amplitude of platform motion (a), the motion frequency (f), and the oscillation amplitude of the apparent wind speed due to platform displacement (Δu). The motion frequency relates to the free-stream wind through the rotor reduced frequency:

$$f_{\rm r} = \frac{fD}{U_0} \,. \tag{1}$$

The reduced frequency is a normalized frequency comparable to the Strouhal number, which is commonly utilized in studies of unsteady fluid dynamics.

3.1 Wind conditions and wind turbine settings

155 The wind turbine was tested at the free-stream wind speed of 4 m s^{-1} that was also investigated by Fontanella et al. (2021). The wind turbine was operated with a constant rotor speed of 240 rpm, corresponding to the optimal tip-speed ratio of 7.5, and the blade pitch was fixed to 0° (i.e., the wind turbine optimal pitch angle). The rotor speed and blade pitch were set to a constant value to avoid any controller actions that could affect the aerodynamic response. Reproducing the actions of a wind turbine



Figure 2. Wind speed at the wind turbine location measured prior to testing. (a, b) Normalized mean velocity and turbulence intensity (*I*) in the wind tunnel test section; the black circle marks the rotor edge. (c) Wind velocity power spectral density at the wind turbine hub.

feedback controller in scale model experiments is challenging and introduces uncertainty in the measurement of aerodynamic
loads (Fontanella et al., 2023). The blade pitch was fixed individually for each blade using an inclinometer and the procedure was repeated regularly throughout the test campaign. The resulting zero blade-pitch position achieved an accuracy of ±0.75°.

The wind turbine creates about an 8% geometrical blockage effect, calculated from the rotor area and the wind tunnel test section ratio. This causes the wind speed to rise at the rotor section depending on the turbine operating conditions. The wind speed at the rotor, affected by blockage effects, was measured using the lateral Pitot tube and was 4.2 m s^{-1} in all cases.

165 3.2 Platform motion conditions

Platform motions were defined using the inertial reference frame $(x_i - y_i - z_i)$ and the platform reference frame $(x_p - y_p - z_p)$ that are depicted in Figure 1b. The inertial reference frame has the origin O_i on the wind tunnel floor in correspondence of the tower axis when the robotic platform is at rest; z_i points upward, x_i points in the downwind direction and y_i forms a right-hand backhoe with x_i and z_i . The platform reference frame is fixed to the robotic platform and moves with the turbine; it

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has the origin O_p in correspondence of the tower base, on the tower centerline; z_p points along the tower axis, x_p points in the downwind direction when the platform is at rest and y_p forms a right-hand backhoe with x_p and z_p . When the wind turbine is in its rest position, the vertical distance between O_p and O_i is 0.73 m.

The system was studied under harmonic motion of different frequencies and amplitudes. The platform displacement, whether it involves a translation or a rotation, is:

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$$d(t) = a \cdot \sin(2\pi f t)$$
. (2)

The platform velocity is:

$$\dot{d}(t) = a \cdot 2\pi f \cdot \sin\left(2\pi f t\right). \tag{3}$$

The wind turbine was forced to oscillate in different directions resulting in several motion scenarios that are summarized in Table 2. The surge motion corresponds to the translation of $O_{\rm p}$ along $x_{\rm i}$. Pitch is the rotation around the y_i axis, roll is the rotation around the x_i axis, and yaw the rotation around the z_i axis; the rotations are illustrated in Fig. 1. Furthermore, the 180 experiment explored platform movements at different angles relative to the wind direction. The surge-sway motion at an angle γ is the combination of a translation $d(t) \cos \gamma$ in the x; direction and translation $d(t) \sin \gamma$ in the y;. The rotation of the platform at an angle γ with respect to the wind direction is defined as the combination of a roll rotation around the x_i axis, $d(t) \cos \gamma$, and a pitch rotation, $d(t)\sin\gamma$ around the y_i axis. The yaw motion is a rotation of the platform around the z_i axis.

- 185 The amplitudes of surge, pitch, surge-sway, and roll-pitch motions were selected to achieve values of Δu between 0.1 m s⁻¹ and $0.6 \,\mathrm{m\,s^{-1}}$. When combined with a free-stream wind speed of $4 \,\mathrm{m\,s^{-1}}$, these result in apparent wind-to-undisturbed wind speed ratios ($\Delta u/U_0$) ranging from 2.5% to 15%. Comparable values were taken into account in previous wind tunnel studies on the aerodynamic rotor loads of floating wind turbines (Schulz et al., 2024). The yaw motion amplitude was limited to a maximum rotation of 3° by the robotic platform. The motion frequencies were selected to replicate the reduced frequencies 190 investigated in previous experimental studies on floating wind turbine aerodynamics, facilitating direct comparisons with existing results (Schulz et al., 2024; Bergua et al., 2023). Additionally, these frequencies are representative of the behavior of
 - full-scale floating wind turbines (Wise and Bachynski, 2020): reduced frequencies of 0.3 and 0.6 correspond to full-scale frequencies of 0.02 Hz and 0.04 Hz, typical of rigid-body modes in floating turbines; the reduced frequency of 1.2, on the other hand, corresponds to a full-scale frequency of 0.08 Hz, representative of motions at wave frequency.
- 195
- The wind turbine experiences an apparent wind when the system undergoes platform motion. The apparent wind velocity results from combining the incoming wind velocity (U_0) with the rotor structural velocity, as shown in Fig. 3.

With surge and pitch motions the rotor is forced to oscillate in the x_i direction against the incoming wind. With surge, the apparent wind speed is:

$$u_{a}(t) = U_{0} - d(t),$$
 (4)

The oscillation amplitude of the apparent wind speed is $\Delta u = a \cdot 2\pi f$. 200

The pitch motion results in a skewed flow due to the rotor plane tilt angle. However, the pitch amplitude is relatively small, thus the rotor motion is primarily in the x_i direction and can be approximated with the hub motion. The apparent wind speed is:

$$u_{\rm a}(t) = U_0 - \dot{d}_{{\rm hub},x}(t),$$
(5)

with $\dot{d}_{hub,x} = \dot{d}(t) \cdot r_{hub}$, where r_{hub} is the hub distance between O_p and the rotor apex. 205

When surge and sway or pitch and roll movements combine, they generate a flow with considerable skew. The apparent wind has velocity components in the x_i and y_i directions. The component in the x_i direction is obtained using Eq. 5, setting

 $\dot{d}_{\text{hub},x}(t) = \dot{d}(t) \cdot \cos \gamma$ for the surge-sway case and $\dot{d}_{\text{hub},x} = \dot{d}(t) \cdot \cos \gamma \cdot r_{\text{hub}}$ for the roll-pitch case. The velocity component in the y_i direction is:

| 210 | $v_{\rm a}(t) = -\dot{d}_{{\rm hub},y}(t),$ | |
|-----|---|--|
|-----|---|--|

where $\dot{d}_{\mathrm{hub},y} = \dot{d}(t) \cdot \sin \gamma$ for the surge–sway case and $\dot{d}_{\mathrm{hub},y} = \dot{d}(t) \cdot \sin \gamma \cdot r_{\mathrm{hub}}$ for the roll–pitch case.

| 1 | 1 | ſ | ١ | |
|---|---|---|---|--|
| | | Ļ | | |
| | | | | |

| Table 2. Test scenarios and wake measurements taken for each of them. Loads were measured in all cases. "H" stands for horizontal and "V" |
|---|
| for vertical. |

| Direction of | Amplitude (a) | Frequency (f) | Reduced | Δu | Measurement | Measurement |
|-----------------|-------------------|-----------------|-----------------------|------------|-------------|-------------|
| Platform motion | [m] or [°] | [Hz] | frequency (f_r) [-] | [m/s] | section | direction |
| Fixed | _ | _ | _ | _ | 3D, 4D, 5D | H, V |
| Surge | $0.032\mathrm{m}$ | 0.5 | 0.3 | 0.1 | 3D | H, V |
| | $0.064\mathrm{m}$ | 0.5 | 0.3 | 0.2 | 3D | Н |
| | $0.016\mathrm{m}$ | 1.0 | 0.6 | 0.1 | 3D | Н |
| | $0.032\mathrm{m}$ | 1.0 | 0.6 | 0.2 | 3D, 4D, 5D | H, V |
| | $0.048\mathrm{m}$ | 1.0 | 0.6 | 0.3 | 3D | Н |
| | $0.016\mathrm{m}$ | 2.0 | 1.2 | 0.2 | 3D | Н |
| | $0.032\mathrm{m}$ | 2.0 | 1.2 | 0.4 | 3D | Н ,V |
| | $0.048\mathrm{m}$ | 2.0 | 1.2 | 0.6 | 3D | Н |
| Pitch | 1.3° | 0.5 | 0.3 | 0.1 | 3D, 5D | H, V |
| | 2.5° | 0.5 | 0.3 | 0.2 | 3D | H, V |
| | 3.0° | 0.5 | 0.3 | 0.25 | 3D | H, V |
| | 0.6° | 1.0 | 0.6 | 0.1 | 3D | H, V |
| | 1.3° | 1.0 | 0.6 | 0.2 | 3D, 4D, 5D | H, V |
| | 1.9° | 1.0 | 0.6 | 0.3 | 3D | H, V |
| | 0.3° | 2.0 | 1.2 | 0.1 | 3D | H, V |
| | 0.6° | 2.0 | 1.2 | 0.2 | 3D | H, V |
| | 1.3° | 2.0 | 1.2 | 0.4 | 3D, 5D | H, V |
| | 1.9° | 2.0 | 1.2 | 0.6 | 3D | H, V |
| Surge-sway 30° | $0.032\mathrm{m}$ | 0.5 | 0.3 | 0.1 | 3D | Н |
| Surge-sway 15° | $0.032\mathrm{m}$ | 1.0 | 0.6 | 0.19 | 3D | Н |
| Surge-sway 30° | $0.032\mathrm{m}$ | 1.0 | 0.6 | 0.17 | 3D, 5D | Н |
| Surge-sway 45° | $0.032\mathrm{m}$ | 1.0 | 0.6 | 0.14 | 3D | Н |
| Surge-sway 30° | $0.032\mathrm{m}$ | 2.0 | 1.2 | 0.35 | 3D | Н |
| Roll–pitch 15° | 1.3° | 1.0 | 0.6 | 0.20 | 3D | Н |
| Roll-pitch 30° | 1.3° | 1.0 | 0.6 | 0.18 | 3D | Н |
| Roll–pitch 45° | 1.3° | 1.0 | 0.6 | 0.15 | 3D | Н |
| Yaw | 2° | 0.5 | 0.3 | - | 3D | Н |
| | 2° | 1.0 | 0.6 | - | 3D, 5D | Н |
| | 3° | 1.0 | 0.6 | - | 3D | Н |
| | 2° | 2.0 | 1.2 | _ | 3D | Н |

(6)



Figure 3. The motion conditions studied in the experiment are classified into three categories: (a) surge and pitch motions that cause an apparent wind velocity that is aligned to the undisturbed wind; (b) motions that combine surge with sway, or pitch with roll, resulting in an apparent wind velocity with a crosswind component; (c) yaw movement, resulting in an apparent wind velocity gradient across the rotor span.

During yaw motion, because the yaw amplitude is small, the apparent velocity is mainly in the x_i direction. It has a gradient across the rotor width, and its profile function of the radial position is:

$$u_{\mathbf{a}}(t) = U_0 - \dot{d}(t) \cdot p_y, \tag{7}$$

215 where p_y is the distance on the y-axis from the rotor apex.

4 Data processing

The tower-top loads were post-processed to obtain the aerodynamic rotor loads. These tower-top loads include forces from rotor aerodynamics and the rotor-nacelle assembly inertia and weight. Inertia and weight contributions were subtracted from the recorded loads using the post-processing method described by Fontanella et al. (2022a), isolating the aerodynamic forces. The aerodynamic loads are represented in the hub reference frame (x - y - z).

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The aerodynamic rotor loads were analyzed both in the frequency domain and the time domain. The zero-peak amplitude of the aerodynamic loads was computed as the fast Fourier transform (FFT) amplitude at the frequency of platform motion. The phase shift between the aerodynamic rotor load and the platform motion was computed based on the real and imaginary part of the complex FFT at the frequency of interest.

- 225 In the time domain, rotor loads exhibit a response at the blade-passing frequency caused by rotor mass imbalance and at higher harmonics resulting from aerodynamic imbalance. The loads were low-pass filtered with a 3 Hz cut-off frequency to eliminate the effects of imbalance. This process was performed in the frequency domain by computing the complex FFT, keeping the frequency components up to 3 Hz, and then utilizing the inverse FFT to reconstruct the signal in the time domain. Filtering based on FFT and IFFT was selected over other techniques because it eliminates the harmonics at the rotor spinning
- 230 frequency (4 Hz) due to imbalance, without affecting the response at 2 Hz, which is the highest frequency of platform motion considered in the experiment. The filtered loads include approximately 40 periods of platform motion. Subsequently, these loads were binned based on platform motion and phase averaged.

Velocity measurements using a hot wire were taken at a sampling frequency of 10 kHz. To prevent high-frequency electrical noise from skewing the turbulent statistics, the data were digitally filtered with a low-pass filter with a cut-off frequency of 235 100 Hz. The filter cut-off frequency was chosen carefully after analyzing the turbulent scales to identify the smallest and highest-frequency energy-containing scales, ensuring accurate estimation of the turbulent kinetic energy. By calculating the auto-correlation function of the stream-wise velocity at all measurement points, we observed that the smallest and highestfrequency scales appear in the undisturbed flows approaching the turbine and in the free-stream areas beside the rotor wake. In these low-turbulence zones, the autocorrelation consistently shows a correlation time of about 0.12 s, suggesting a correlation scale of around 0.5 m and an integral length scale of around 0.1 m. Filtering at 100 Hz captures the most energetic scales, 240 along with a significant part of the inertial range of the energy spectrum visible in Fig. 2. Additionally, 100-Hz filtering in

- free-stream regions estimated a turbulence intensity of 1.5-2%, aligning with flow measurements conducted before testing. Within the wake of the wind turbine, the energy-containing scales were observed to be larger in size and lower in frequency, with sufficient magnitude to dominate electrical noise. Thus, although the filter was not strictly necessary, it was applied 245 consistently across all regions of the wake.

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The acquisition time for each measurement point corresponded to 12 platform motion cycles. A sensitivity study conducted at the beginning of the test campaign determined that 12 cycles are adequate for achieving convergence of mean wind speed, turbulence intensity, and velocity spectrum amplitude at the frequency of platform motion. This is considered a reasonable compromise between the accuracy of parameter estimation relevant to this study and the time required to conduct the measurements. The velocity time series were binned according to platform motion and phase-averaged, in a similar manner to the

5 Results

loads.

This section presents the findings obtained from a subset of the scenarios introduced in Sect. 3 that was found suitable to showcase the most interesting phenomena seen in the experiment. Section 5.1 reports the response of aerodynamic rotor loads to platform motion in different directions. Section 5.2 explores how platform surge and pitch motions influence the wake,

255 Sect. 5.3 examines the impact of yaw motion, and Sect. 5.4 discusses the effect of crosswind motion from combined surge and sway.

5.1 Aerodynamic rotor loads

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The rotor is perpendicular to the wind tunnel floor and the wind field is uniform in space. When the wind turbine tower base

is fixed, the aerodynamic rotor loads are only the thrust force (F_x) and the torque (M_x) , which were equal to 36.47 N and 2.97 Nm, respectively.

Rotor loads oscillate when the wind turbine experiences platform movement. For surge motion, the rotor axis stays aligned with the wind direction and F_x and M_x are the only aerodynamic loads. Pitch motion and movements resulting from the combination of surge and sway or pitch and roll, result in a skewed flow. In these cases, there are aerodynamic loads in different directions (F_x , F_y , F_z , M_x , M_y , M_z). However, the amplitude of the loads different from the thrust and torque (i.e., F_y , F_z , M_y , M_z) are generally small.

Figure 4 shows the rotor thrust and torque for pitch motion at three different amplitude values, with a frequency of 1 Hz. The loads were obtained from measurements using the data processing method described in Sect. 4, and the resulting phaseaveraged time series are presented over one period of platform motion in the angle domain, using platform-motion phase instead

- of time. The smooth time series indicate that the number of motion cycles included in the measurements and the data processing method effectively isolate the aerodynamic loads response to platform motion. The average values of M_x match the result with fixed-tower base. In cases with $a = 0.6^{\circ}$ and $a = 1.9^{\circ}$, the average F_x is slightly lower than in the fixed case, but it matches the fixed-tower base case when $a = 1.3^{\circ}$. Variations in the mean value of F_x are likely attributable to slight differences in zero blade-pitch recalibration performed during testing. F_x and M_x display a sinusoidal pattern at the same motion frequency, with
- an approximate 90° phase shift relative to the platform displacement. This proves that rotor load fluctuations are mainly due to variations in apparent wind speed caused by platform movement as observed in previous experiments (Fontanella et al., 2021).

To compare loads obtained in the different scenarios involving motions in the surge, sway, roll, and pitch directions, the zeropeak amplitude of thrust and torque, ΔF_x and ΔM_x respectively, were normalized according to the hub motion amplitude in the x_i direction ($d_{hub,x}$) in meters. Figure 5a illustrates the variation in normalized aerodynamic rotor thrust, while Figure 5b 280 presents the variation in normalized aerodynamic rotor torque as the platform motion frequency changes. A linear regression is fitted to the measurements and is included in Figure 5a and Figure 5b. The loads with motion frequencies of 0.5 Hz and 1 Hz change linearly with respect to the platform motion amplitude, as evidenced by the normalized points aligning with the regression line; additionally, the loads exhibit a linear increase with frequency. Thus, at 0.5 Hz and 1 Hz, the rotor aerodynamic 285 loads change linearly with apparent wind variations in the x_i direction (Δu). For example, increasing the amplitude of platform in the x_i direction or the frequency by a factor of 2 would result in changes in rotor aerodynamic loads of the same order. The variation in loads for motion cases at a frequency of 2 Hz exceeds the expected linear trend. For instance, when the surge motion has an amplitude of 0.048 m and a frequency of 2 Hz, corresponding to the maximum Δu investigated in the experiment, the amplitude of thrust force oscillation is 6.99 N and not the 5.6 N predicted by linear regression. The deviation of aerodynamic loads from the linear trend at 2 Hz was observed also by Bergua et al. (2023), but it was not reflected in the high-fidelity 290

simulations of the experiment conducted alongside the measurements analysis. Numerical simulations of the wind turbine scale



Figure 4. Phase-averaged time series of aerodynamic rotor thrust and torque for platform pitch motion at a frequency of 1 Hz, which corresponds to a reduced frequency of 0.6, with three different motion amplitudes (a). "Fixed" refers to the scenario with fixed tower base. Δu is the amplitude of apparent wind changes caused by platform motion; ΔF_x and ΔM_x denote the amplitudes of the thrust and torque oscillations, respectively.

model utilized in this experiment indicate that the amplitude of aerodynamic load oscillations decreases below the linear trend as the frequency increases beyond 2 Hz (Ribeiro et al., 2023). Thus, the higher-than-linear response at a motion frequency of 2 Hz is expected to be due to non-idealities in the experimental setup, primarily fluctuations in rotor speed, with an amplitude up to 0.6 rpm, and a minor flexible response of the tower, where the amplitude of nacelle motion is up to 7% higher than that of the platform base motion. These non-idealities should be taken into account when comparing the experimental load response with the results of numerical simulations. However, they are considered small enough to not impact the observed wake dynamics trends.

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Figure 5c and Figure 5d show the phase shift of the aerodynamic rotor loads with respect to the platform motion. The phase
shift is approximately 90°, indicating that the loads oscillations are in phase with the fluctuations in apparent wind speed. The loads response is consistent with the behavior observed in previous studies investigating platform surge and pitch motions (Bergua et al., 2023).

Yaw motion leads to oscillations in the aerodynamic yaw moment (about the z_p axis). Figure 6 shows the aerodynamic rotor thrust and the aerodynamic yaw moment in one cycle of yaw motion for four yaw motion scenarios. The yaw moment exhibits

an average value slightly different from zero, probably because of a small misalignment of the wind turbine during the tests. The loads are affected by oscillations at the frequency of platform motion. These oscillations are determined by the apparent wind speed created by platform movement u_a , rather than by the misalignment of the rotor relative to the incoming wind, since the yaw angle is relatively small. Rotor thrust oscillations exhibit much smaller amplitudes compared to those observed in surge, sway, roll, and pitch motions. In the 0.5 Hz scenario, the oscillations are minimal and almost indistinguishable



Figure 5. Normalized variation of the aerodynamic rotor thrust (a) and phase shift with respect to platform motion (c). Normalized variation of the aerodynamic rotor torque (b) and phase shift with respect to platform motion (d).

310 from measurement noise, resulting in the irregular pattern shown in the figure. Across all four scenarios, the average thrust is marginally lower than that of the fixed turbine case.

5.2 Wake with platform pitch and surge motions

The analysis of aerodynamic rotor loads indicates that pitch and surge movement leads to periodic thrust changes driven by apparent wind speed fluctuations caused by the motion. These thrust changes perturb the wake of the wind turbine. The effect

315 is assessed by comparing the average velocity and turbulence intensity of the floating case with the bottom-fixed case, and examining velocity fluctuations coherent with platform motion.

Figure 7a shows the average velocity profile of the wake at hub height and $x_w = 3D$, for the wind turbine with a fixed foundation and under four platform pitch motion conditions. The velocity profile has a double-Gaussian shape. In line with previous measurements of the wake of the same wind turbine of Fontanella et al. (2022b), the velocity profile is asymmetric

320 relative to the rotor axis. The wake center is shifted approximately 0.06D toward the negative y_i -axis, and the velocity minima peaks on each side of the wake differ in shape. In the experimental study conducted by Pierella and Sætran (2017), it was observed that the interaction between the turbine tower wake and rotor-induced flow results in wake asymmetry. The tower



Figure 6. Phase-averaged aerodynamic rotor thrust and yaw moment (about the $z_{\rm p}$ axis) time series during yaw motion.

creates a low-velocity zone that is entrained into the wake and rotates with it, persisting up to 3D downstream. The mean velocity profile remains largely unchanged by platform motion, with only minor variations observed in the peak of minimum velocity on the positive side of the y_i -axis.

The turbulence intensity in the wake is computed as:

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$$I(y_{\rm i}) = \frac{\sigma_u(y_{\rm i})}{U(y_{\rm i})},\tag{8}$$

where $\sigma_u(y_i)$ represents the standard deviation of the local streamwise velocity, while $U(y_i)$ denotes its time-average value. Since $U(y_i)$ is nearly the same in both the platform motion and fixed tower base scenarios, any increase in *I* is due to greater variability of the streamwise velocity. The turbulence intensity profiles are reported in Figure 7b. They all show two peaks in correspondence of the rotor edges that are attributed to the wake shear layer. At a platform pitch motion frequency of 1 Hz, the turbulence intensity is increased near the center of the wake, indicating that the mixing layer shifts towards the wake core. This behavior intensifies as the amplitude of motion at a specific frequency increases, causing both apparent wind speed and thrust oscillations to rise.

- Figure 7c shows the streamwise velocity at $x_w = 3D$ and $y_i = 0.6$ m obtained from phase-averaging of velocity time series acquired over 12 platform motion cycles, as described in Sect. 4. The point at $y_i = 0.6$ m is selected because it represents half the rotor radius, where the blade undergoes the greatest local normal force variation (Fontanella et al., 2022b), and it does not align with the wake shear layer in the fixed-turbine scenario. Significant velocity fluctuations occur at this location when the wind turbine undergoes platform movement, thus $y_i = 0.25D$ is used for comparing different motion conditions. The
- 340 data points obtained from phase-averaging were fitted with a sine function with a frequency equal to the motion frequency to emphasize the velocity oscillations that match the platform motion. The determination coefficient R^2 is used to indicate how well the sinusoidal function fits the data points. In the cases examined in Fig. 7c, the fitted sine function captures a significant portion of the velocity variability, indicating the presence of flow structures in the wake that are strongly coherent

with the sinusoidal platform motion. Similar dynamics have been reported by Wei et al. (2024), who demonstrated that periodic

upstream forcing, such as thrust variations caused by platform motion, generates traveling-wave undulations in the wake radius 345 and streamwise velocity.

For the same motion amplitude, the largest velocity oscillations are with f = 1 Hz despite this condition having lower thrust variation amplitude (see Fig. 5) than at f = 2 Hz. When the motion frequency is 2 Hz, oscillations are observed only at the boundaries of the wake (approximately at $y_i \pm 0.45D$) that corresponds to the location of the shear layer. Coherent flow structures are mostly absent at other locations. As shown in the graph, the sine wave at the motion frequency does not fit

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the velocity data as well as it does in the 0.5 Hz and 1 Hz cases. Research on dynamic induction control has shown that the evolution of the wake is influenced by the reduced frequency of periodic fluctuations in thrust force (Munters and Meyers, 2018). We assume that similar effects occur in the wake of a floating wind turbine where fluctuations in thrust force are due to the movement of the platform. Among the conditions explored in this study, the strongest perturbation of the wake occurs 355 when the reduced frequency of platform motion is 0.6. This agrees with the findings of Messmer et al. (2024a) who observed in a wind tunnel experiment that optimal wake recovery occurs with surge motions at a reduced frequency range of 0.5 to 0.8.

1.2 25 0.8 (a) (b) (c) Rotor Rotor 1.1 urbulence intensity [%] c 01 c1 05 1 0.7 0.9 1/0 [-] [-] ⁰n/r 0.8 0.6 0.7 0.6 0.5 0.5 0.4 0 0.4 -1 -0.5 0 0.5 1 -1 -0.5 0 0.5 1 0 90 180 270 360 y;/D [-] y_i/D [-] Motion phase [°] • a = 1.3°, f = 1.0 Hz (f r = 0.6, Δu = 0.2 m/s) ---- a = 1.9°, f = 1.0 Hz (f, = 0.6, Δu = 0.3 m/s) ---- Fixed $a = 1.3^{\circ}, f = 0.5 Hz$ (f = 0.3, $\Delta u = 0.1 m/s$) — $a = 1.3^{\circ}, f = 2.0 Hz$ (f = 1.2, $\Delta u = 0.4 m/s$)

Figure 7. Wake at hub height and $x_w = 3D$ with platform pitch motion. (a) Mean streamwise velocity normalized by the free-stream wind speed (U_0). (b) Turbulence intensity. (c) Phase-averaged time series of streamwise velocity u at $y_i = 0.25D$: the solid lines highlight the first-order sinusoid at the motion frequency that was fitted to the data points obtained from phase-averaging and R^2 is the coefficient of determination assessing the fit quality.

Velocity fluctuations coherent with platform motion vary throughout the wake width. This is examined in Fig. 8 that shows

the phase-averaged velocity in with pitch motion at $x_{\rm w} = 3D$ and at various points along the $y_{\rm i}$ -axis. With motion frequencies of 0.5 Hz and 1 Hz, coherent fluctuations are not observed at the rotor edges, where the shear layer is located, and in the 360 wake center. At $y_i = \pm 0.375D$, the velocity oscillations have similar amplitude and phase shift relative to the platform motion. In other positions within the inner region of the wake, the amplitude of velocity oscillations and their phase in relation to platform motion are not symmetrical with respect to the rotor axis. The development of coherent flow structures is closely associated with their interaction with other turbulent structures and the turbulent mixing process. Variations in the background flow, as shown in Fig. 2, and interaction of the rotor wake with the tower wake may explain the observed asymmetry of velocity fluctuations relative to the rotor axis.

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Figure 8. Phase-averaged time series of streamwise velocity u normalized by free-stream velocity U_0 during pitch motion with $\Delta u = 0.2 \text{ ms}^{-1}$, f = 0.5 Hz and f = 1 Hz at $x_i = 3D$. The solid lines highlight the first-order sinusoid at the motion frequency that was fitted to the data points obtained from phase-averaging and R^2 is the coefficient of determination assessing the fit quality. A satisfactory fit of the sinusoid was not achieved in the positions $y_i = \pm 0.5D$ and $y_i = 0$.

The analysis carried out in Fig. 7 for the wake at $x_w = 3D$ are repeated in Fig. 9 for the wake at $x_w = 5D$. The average velocity exhibits a Gaussian profile that is mostly unchanged by platform motion. The wake velocity in the rotor region (for $|y_i| \le 0.5D$) is known at N_p points in the y_i direction that are symmetrically distributed with respect to the x_i -axis; the average

velocity in this one-dimensional discrete domain is computed as:

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$$U_{\text{avg}} = \frac{\sum_{j=1}^{N_{\text{p}}} |r_j| U_j}{\sum_{j=1}^{N_{\text{p}}} |r_j|},$$
 (9)

where U_i is the time-average velocity at the j-th point which is located at a distance r_i from the x_i -axis. In the rotor region, U_{avg} with motion is approximately 65% of U_0 , slightly exceeding the average of 62.5% of U_0 in the fixed condition.

The turbulence intensity exhibits the two peaks associated with the shear layer, that are situated closer to the wake center than at $x_w = 3D$. In line with findings at $x_w = 3D$, the pitch motion scenario at f = 1 Hz exhibits higher turbulence at the 375 wake center compared to other scenarios. The shear layer occupies a larger part of the wake compared to the fixed case, thus the transition from near to far wake appears to be accelerated and is expected to take place closer to the rotor.

As shown in Fig. 9c, the velocity oscillations that are coherent with platform motion are less pronounced at $x_w = 5D$ than at $x_{\rm w} = 3D$. This is evidenced by R^2 , which indicates that the fitted sine functions capture less of the variability of the velocity.



Figure 9. Wake at hub height and $x_w = 5D$ with platform pitch motion. (a) Mean streamwise velocity normalized by the free-stream wind speed (U_0). (b) Turbulence intensity. (c) Phase-averaged time series of streamwise velocity u at $y_i = 0.25D$: the solid lines highlight the first-order sinusoid at the motion frequency that was fitted to the data points obtained from phase-averaging and R^2 is the coefficient of determination assessing the fit quality. A satisfactory fit of the sinusoid was not achieved with f = 2 Hz.

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The progression of the turbulence components in the wake with increasing distance from the rotor is analyzed in Fig. 10, which presents the power spectral density (PSD) and the phase-averaged streamwise velocity for the scenario involving a pitch motion with an amplitude of 1.3° and a frequency of 1 Hz. At $x_{\rm w} = 3D$ the PSD displays peaks at the platform motion frequency corresponding to velocity oscillations observed in the phase-averaged time series. Flow structures coherent with the periodic platform movement dominate the wake response and a distinct increase and decrease in velocity, matching the periodicity of platform motion, is observable on both sides of the wake. At $x_w = 5D$, the peak in the PSD at the frequency of 385 motion becomes less evident. As the wake begins to dissipate and turbulence develops, there is an increase in spectral energy

from $x_w = 3D$ to $x_w = 5D$ across the wake width and over a broad frequency range. As energy builds up across different frequencies, the direct influence of platform motion on the wake becomes partially obscured by other turbulent structures. This is evidenced by the lower coherency of velocity oscillations with the platform motion at $y_i = 0.25D$, as shown by the sine function fit in Fig. 9, and the reduced smoothness of the phase-averaged wake in Fig. 10d.



Figure 10. Progression of the turbulence components in the wind turbine wake at increasing distances (x_w) from the rotor during pitch motion with $a = 1.3^{\circ}$ and f = 1 Hz. (**a**, **b**): power spectral density (PSD) of streamwise velocity. (**c**, **d**): phase-averaged streamwise velocity. Vertical dashed lines indicate the rotor axis and edges.

Figure 11 compares, at $x_w = 3D$, the wake with surge and pitch motions. Both motion conditions result mainly in thrust oscillations at the frequency of platform motion and are characterized by the same reduced frequency and Δu . Their influence on the wake is comparable in terms of average velocity (at $x_w = 5D$, U_{avg} is 65% of U_0 in both cases) and turbulence intensity. Motion in the pitch direction causes larger velocity fluctuations, attributed to the small skew of the inflow over the rotor that is absent in surge motion.



Figure 11. Wake at hub height and $x_w = 3D$ with platform surge and pitch motions of frequency 1 Hz. The surge and pitch motions are associated with the same amplitude of apparent wind speed $\Delta u = 0.2 \text{ m s}^{-1}$. (a) Mean streamwise velocity normalized by the free-stream wind speed (U_0) . (b) Turbulence intensity. (c) Phase-averaged time series of streamwise velocity at $y_i = 0.25D$: the solid lines highlight the first-order sinusoid at the motion frequency that was fitted to the data points obtained from phase-averaging and R^2 is the coefficient of determination assessing the fit quality.

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Figure 12a-b shows the wake along the z_i -axis at $x_w = 3D$ for both surge and pitch motions at f = 1 Hz with the same Δu . The velocity over one motion cycle is almost identical, and in pitch the wake center moves only slightly in the vertical direction. Figure 12c compares the average velocity along the z_i -axis at $x_w = 3D$ with different pitch motions. Like the horizontal profile of Figure 7a, the vertical velocity profile exhibits a double-Gaussian shape and is largely unaffected by platform movements.



Figure 12. Wake along the z_i -axis and $x_w = 3D$ with platform surge and pitch motions. (a) Phase-averaged streamwise velocity during surge motion with a = 0.032 m, f = 1 Hz (the dashed lines are in correspondence of the rotor axis and the rotor edges). (b) Phase-averaged streamwise velocity during pitch motion with $a = 1.3^{\circ}$, f = 1 Hz (the dashed lines are in correspondence of the rotor axis and the rotor edges). (c) Mean streamwise velocity normalized by the free-stream wind speed (U_0) along the z_i axis with different pitch motions.

5.3 Wake with platform vaw motion

Movement in the yaw direction generates different aerodynamic loads compared to those created by surge or pitch motions. 400 While the thrust variations are minimal, there is an oscillating yaw moment on the rotor. Figure 13 examines the development of the wake at $x_{\rm w} = 3D$ with platform yaw motion. The average velocity is unchanged compared to the fixed scenario. Turbulence intensity increases near the wake center and decreases close to the rotor edges, showing that the shear layer moves toward the wake center, a pattern also noted with surge and pitch motions. However, this is not the effect of periodic fluctuations in thrust 405

force that are almost absent with movement in the yaw direction (see Fig. 6).

Figure 13c shows the velocity over one cycle of yaw motion with $a = 2^{\circ}$, f = 1 Hz. The wake undergoes lateral meandering which is noticeable when tracking the relative maximum velocity at the center and the absolute minimum velocity on the left side of the double-Gaussian (black lines in Fig. 13c). The wake lateral movement matches the platform frequency and shows strong correlation across the wake width. We attribute wake meandering to the periodic yaw moment on the rotor. The

meandering reaches its peak amplitude when f = 1 Hz, is reduced when f = 0.5 Hz, and nearly disappears when f = 2 Hz. 410 Thus, this wake response is also sensitive to the frequency of the oscillations of the rotor loads and has the highest intensity when f_r is equal to 0.6. At $x_w = 5D$, the meandering disappears. With a yaw motion of $a = 2^\circ$, f = 1 Hz, U_{avg} is equal to 65.8% of U_0 at $x_w = 5D$, showing a 5% increase over the fixed-bottom case.



Figure 13. Evolution of the wake at hub height and $x_w = 3D$ with platform yaw motion. (a) Mean streamwise velocity normalized by the free-stream wind speed (U_0). (b) Turbulence intensity. (c) Phase-averaged streamwise velocity during yaw motion with $a = 2^\circ$, f = 1 Hz (the dashed lines are in correspondence of the rotor axis and the rotor edges).

5.4 Wake with platform crosswind motion

415 Whenever movement in the crosswind direction takes place, thus combining both surge and sway, the apparent wind is skewed as depicted in Fig. 3. The thrust exhibits periodic oscillations that are proportional to the rotor forward motion in the x_i direction, similar to those observed in cases of pure surge motion. Additionally, throughout a motion cycle, the wake is released from various positions along the y_i -axis.

Figure 14 shows the wake at x_w = 3D with motions that are the combination of surge and sway. The platform translation
of a =0.032 m and f = 1 Hz is at an angle γ ranging from 0° (i.e., pure surge motion) to 45°. When γ = 45°, the near wake has lateral meandering that is instead absent in the case with γ = 0°. The onset of wake meandering is attribute to the lateral component of the apparent wind v_a and to the periodic release of the wake at different y_i positions, which do not occur in the surge motion case. The wake lateral movement matches the platform frequency and is correlated across the wake width. During one period of platform motion, the minimum velocity on the left side undergoes a lateral displacement of 0.13D, which
is larger than the rotor crosswind motion that has a peak-to-peak amplitude of 0.019D (corresponding to 0.045 m). The large-eddy simulation study of a wind turbine wake under side-to-side motion of Li et al. (2022) shows that this movement can cause an instability in the wake shear layer, which can propagate and amplify downstream the rotor. This can result in wake lateral meandering amplitudes that are up to an order of magnitude larger than the rotor lateral motion, as observed in this experiment.

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With surge and sway motion, the turbulence intensity in the wake center increases more significantly than with surge motion alone. This increase becomes more pronounced as the angle γ grows, as shown in Fig. 14c. The combination of wake meandering and velocity fluctuations appears to further promote the transition from near to far wake.

At $x_w = 5D$, the meandering and the velocity fluctuations disappear. With a motion of a = 0.032 m and f = 1 Hz, with $\gamma = 30^\circ$, the average wake velocity at $x_w = 5D$ is 63.5% of U_0 .



Figure 14. Wake at hub height and $x_i = 3D$ with a platform motion that is the combination of surge and sway. (a) Phase-averaged streamwise velocity in pure surge motion of a = 0.032 m and f = 1 Hz. (b) Phase-averaged streamwise velocity during a translation motion at angle $\gamma = 45^{\circ}$ of a = 0.032 m and f = 1 Hz. (c) Turbulence intensity for translation motions with four values of γ (the surge motion corresponds to $\gamma = 0^{\circ}$).

6 Conclusions

435 This study investigated how platform motion in different directions affects rotor aerodynamics and correlated the rotor response to wake development. Wake measurements, taken with hot-wire probes from 3D to 5D downstream, were compared between scenarios with prescribed platform motions and the bottom-fixed case.

Depending on the direction of platform motion, we distinguished three conditions. With surge and pitch, the rotor moves in the wind direction; this makes both rotor thrust and torque fluctuate at the same frequency of motion. The oscillations in rotor

- 440 loads are driven by the apparent wind created by platform movement. The average values of thrust and torque remain relatively unaffected by the platform motion, indicating that the overall performance of a floating wind turbine is comparable to that of a fixed-bottom turbine. The thrust oscillations translate into fluctuations of the near wake velocity that peak with a motion reduced frequency of 0.6. Yaw motion leads to oscillations of the yaw moment at the rotor and to lateral wake meandering correlated with the platform motion frequency. Combining surge and sway motions results in skewed apparent wind, causing
- 445 lateral wake meandering that adds to the velocity fluctuations caused by oscillations in thrust force. In a floating wind farm, the velocity fluctuations in the wake can lead to fluctuating loads on downstream turbines. Upcoming wind tunnel tests with multiple scale models of floating turbines will investigate this further.

For the wind turbine operating conditions, motion scenarios, and distances from the rotor examined in this experiment, the mean velocity in the wake is comparable to that of the bottom-fixed case. Instead, turbulence intensity increases near the wake center, suggesting a shift of the mixing layer towards the wake core. Increased turbulence is linked to higher wake mixing,

which could mean a faster recovery beyond 5D, the furthest distance examined in the experiment. Future studies should

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examine the wake further downstream to verify whether the increased wake turbulence induced by motion results in quicker wake recovery.

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Greater turbulence intensity in undisturbed wind enhances wake mixing and facilitates the transition to the far wake, helping

- to recover the free-stream speed. In offshore environments, turbulence intensity typically ranges from 3% to 6%, with sheared wind. This experiment used a 1.5% turbulence intensity and a nearly constant vertical velocity profile, thus future testing should examine the effects of more realistic turbulence and wind shear on the evolution of floating wind turbine wakes. The flow within a wind farm is further complicated by the interactions between the wakes of various turbines, each subjected to dynamic platform motion. This condition is recommended for exploration in future experimental campaigns.
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All experimental results obtained in this research aim to provide a reliable benchmark to validate numerical tools and to improve their modeling capabilities. Knowledge gained from experiments, along with better simulation tools can be leveraged to optimize future large-scale floating wind farms.

Data availability. Measurement data of the wind tunnel experiment are accessible at https://doi.org/10.5281/zenodo.13994980.

Appendix A: Assessment of the wind turbine scale model aerodynamic design

- 465 The aerodynamic design of the wind turbine scale model was verified through a dedicated wind tunnel campaign that measured the rotor thrust and torque at various wind speeds within the below-rated region. In this region, the rotor speed was adjusted to achieve the design tip-speed ratio of 7.5, with a fixed blade pitch of 0° . This condition matched the one considered in the rotor design process described by Bayati et al. (2017b). The validation occurred in a uniform wind field with a turbulence intensity of 1.5%.
- 470 Figure A1 presents the rotor thrust and torque measured during characterization as well as those obtained in the fixed towerbase scenario of the current wind tunnel experiment. These results are compared to the DTU 10-MW values at model scale for different wind speeds. The thrust of the scale model aligns with the reference wind turbine values, while the torque is lower across all wind speeds due to the lower efficiency of the SD7032 low-Reynolds airfoil compared to those used in the DTU 10-MW blade.



Figure A1. Steady-state rotor thrust and torque of the wind turbine model compared to the DTU 10 MW values at model scale for different wind speeds.

475 *Author contributions.* All authors prepared and conducted the experiment, and analyzed the measurement data. AF wrote the first draft of the article, while all authors contributed to its review and editing. MB, AB, and VD have procured the funding. MB, SM, AB, and VD have supervised the work.

Competing interests. At least one of the (co-)authors is a member of the editorial board of Wind Energy Science.

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