

### Reply to Associate editor decision: “Publish subject to minor revision”

The authors would like to thank the associate editor and the reviewer for their time and for the useful feedback. The inputs that they provided have contributed to the improvement of the paper.

A list of point-by-point replies to the reviewer’s suggestions is reported in the following. The reviewer’s comments are in black, and our replies in blue.

A revised version of the manuscript is attached to the present reply, with additions highlighted in blue and corrections/deletions marked in red. Please note that, in addition to the highlighted changes, we reorganized the structure as follows. Sections 3.3 and 3.4 have been switched, i.e., the new section order is “3.3 Modeling and control” followed by “3.4 Local WT operating point prediction”; the appendices now appear in the order that they are referenced in the manuscript; the old Figure 3 has been moved to section 4 “Results” and has become Figure 6. Moreover, please ignore that some figure captions are marked in blue, even though they did not change, as we used the “latexdiff” package (<https://ctan.org/pkg/latexdiff>).

The authors

### Suggestions for revision (from anonymous referee #2, see Report #1)

Overall, I think the authors have improved the paper through their revision of it. I still had some difficulty in fully understanding it, as it is quite complex. Indeed, as the authors indicate in their response, the footnote on page 16 is the forward referencing to a figure more than 10 pages later that I had commented on in my original review. There are other forward referencing instances as well. A couple examples, both on Page 10, include:

- Line 249: The reader is pointed to appendix Eq. (A5) for the derivation of the simplified torque rate constraint. When I go to Appendix A and try to understand the simplified torque rate constraint, there is a reference on the 3rd line of Appendix A (on page 33, line 695) to Eq. (18) that is on page 15 of the paper (line 341) which I have not read to yet when I was pointed to appendix Eq. (A5) on page 10!
- Line 259: The reader is pointed to Appendix B for the derivation of the steady-state initialization. Appendix B then refers to Eq. (15) which is on page 14 which I have not gotten to yet when I was first pointed to Appendix B!

So there is some sense of a “wild goose chase” in trying to understand things! As such, I do think my earlier comment is still true: “All this cross-referencing to multiple other places in the paper made it often difficult to understand.”

Thank you for the feedback, and we completely agree with you.

We have deeply reorganized the presentation of the material to avoid back-and-forth referencing. We believe the new version has none left, except in Sect. 3.4.1 when referring to Appendix G. In that case, for clarity we use some results from Sect. 4, which have not yet been presented. To warn the reader, we wrote “Appendix G discusses the case of recovery power limits, with reference to the examples shown later in Sect. 4.” We think this solution should be acceptable. In case not, we can remove the mention of Appendix G from Sect. 3.4.1, as in any case the same appendix is mentioned even later in Sect. 4.

Apart from Appendix G, all others only refer back to the main body of the manuscript when necessary, and when they do so they refer to material that has already been presented, not to material that has not yet been discussed, as it was the case for the previous version.

We also moved all mentions to the examples to Sect. 4 “Results”, this way creating a cleaner description of the methodology in Sects. 2 and 3. Hopefully this will clarify the generality of the

methods that we propose and their wide applicability, beyond the specific examples considered here.

We have also eliminated the footnotes, following the journal style.

We hope that this deep revision of the material has addressed your very valid concerns, leading to a better flow of the discussion and more comprehensible navigation of the rather complex topics that we present.

To the extent possible, I would still recommend that the authors think through further how the paper can be made easier for readers to follow.

We have reworded and simplified several sentences. Hopefully the new version is easier to follow.

Since optimizations must be performed, even if offline, it would be useful for the authors to briefly remark on the computational complexity involved. How long did it take for the optimizations required (and on what computing platform) to create all the look-up tables for the 12-wind-turbine wind farm used in the paper?

We added a comment regarding the numerical cost in Sect. 4.2.3.

In reading the revised manuscript, I do have some more detailed comments and suggestions for improvement:

1. The authors use both “adaption” and “adaptation”. They should be consistent and choose just one format of the word. I personally prefer “adaptation”. :-)  
We replaced all occurrences of “adaption” with “adaptation”.
2. Should Appendices appear in the order that they are referenced in the manuscript? Appendix K (nomenclature) is referenced now in a footnote first. Then, the Appendices are referenced in the order of G, A, B, C, D-F, I, J, H.  
Thank you for spotting this issue. We have now placed the appendices in the order that they are referenced in the manuscript.
3. Page 9, Eq. (6): I may not be understanding properly, but should it be “–” instead of “+” in “ $P_{m,0} - 2 H_v \Omega_{s,max}$ ” rather than “ $P_{m,0} + 2 H_v \Omega_{s,max}$ ”?  
Thank you for your question. The answer is no, since the maximum ROCOF magnitude “ $\Omega_{s,max}$ ” is positive, as stated at the end of the same sentence (line 411). Note that all considered power quantities  $P$  are positive for power generation, as  $P$  is expressed in p.u. of rated power (generation), as stated in line 391.
4. Page 10, Eq. (9): Again, I may not be understanding correctly, but the entries in the constraint vector look to all be negative to me (rather than positive as implied in the equation). Don’t we want to have  $\min \Omega(t)$  to always be greater than  $\Omega_{min}$ ? In which case  $\Omega_{min} - \min \Omega(t)$  would then always be less than or equal to 0 (and not greater than or equal to 0). Similarly for all the other entries.  
Thank you for spotting this typo! We replaced “ $c_i \geq 0$ ” with “ $c_i \leq 0$ ” in Eq. (9), and replaced “ $c_i > 0$ ” with “ $c_i < 0$ ”.
5. Page 14, Line 315: Should be “appendix Eq. (B4)” rather than “Appendix B4”.  
Thank you, we implemented that change.
6. Page 19, Line 453: I’m not sure I understand the sentence: “In contrast, for  $\Omega = 1$ , higher derating significantly reduces the thrust force due to increasing  $\beta$  but constant  $\lambda$ .” To me, it looks like  $\lambda$  is increasing.  
The tip speed ratio  $\lambda$  remains constant when considering a constant rotor speed ( $\Omega = 1$ ) and a constant wind speed. For clarity, the sentence has been rewritten as follows: “In contrast, for initial operation with maximum speed  $\Omega = 1$  at higher or above-rated wind

speeds, higher derating significantly reduces the thrust force due to increasing  $\beta$  but constant  $\lambda$ .”

7. I still think the legends can be moved to the open spaces in Figures 18-19. It seems there is enough white space in the plots to move the legends to more “optimal” locations that don’t cover any of the results actually being shown. Especially in Fig. 19b and 19d, there is plenty of open space in the upper middle of the figures where the legend can be placed without blocking any of the bars. Even in Fig. 18c and 18f, moving the legend a little to the right and to the top of the figures would be better.  
Thank you for the suggestion. However, the legends do not cover or hide any results, since the top ends of the bars are visible in all plots. Therefore we would like to keep the present version of Figs. 18 and 19, which seem to be acceptable to us.
8. Page 30, Line 633: Should it be “Figures 18d-f” rather than “Figures 18c-f”?  
Sorry for the typo! We implemented that change.
9. Appendix B, Eq. (B2): What does “1 = Omega” mean? Should this be “Omega = 1”?  
Yes, apologies, we changed to “Omega = 1” as it should, see appendix Eq. (D2).
10. Appendix K, Line 945: Should the asterisk be a star?  
Yes, thank you, we have replaced the asterisk with a star.
11. The reference “P. Courtier et al.” is in a different format than all the other references.  
Thanks, we changed the format of the reference to comply with the journal style.

# Wind farm inertia forecasting accounting for wake losses, turbine-level control strategies, and operational constraints

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**Abstract.** Future inverter-based resources (IBRs) must provide grid-forming functionalities to compensate for the declining share of conventional synchronous machines (SMs) in the power generation mix. Specifically, decreasing power system inertia poses a significant challenge to grid frequency stability, as system inertia limits the rate of change of frequency (ROCOF). Conventional grid-following control decouples the physical inertia of wind turbines (WTs) from the grid frequency. Novel grid-forming control methods, such as virtual synchronous machine (VSM) control, provide (virtual) inertia to the system ; e.g. by extracting kinetic energy from WTs. Since the grid-forming capability of IBRs depends on volatile operating conditions, future market designs will remunerate inertia provision based on its availability. Thus, estimating grid-forming capabilities of WTs and forecasting inertia of wind farms (WFs) are of interest for both WF and system operators.

In this paper, we propose a method for forecasting WF inertia that accounts for wake effects and WT characteristics. A wake model estimates individual inflow conditions for each WT in the WF based on forecasted site conditions. These inflow conditions ~~are crucial for predicting the~~ enable the prediction of the grid-forming capabilities of each WT. Under varying inflow conditions and derating power setpoints, we simulate the WT inertial responses to a reference frequency event. Taking WT control strategies and operating limits into account, an optimization algorithm computes the maximum feasible inertia provision at the WT and WF level. The proposed approach is demonstrated in a simulation environment, and the results also include a quantification of the uncertainties due to both wind forecasting and wake modeling errors.

## 1 Introduction

### 1.1 Motivation and problem statement

Imbalances between power generation and demand ~~result in~~ give rise to frequency events. ~~Thus~~ To maintain grid frequency within admissible limits, generation or protection units must therefore rapidly compensate for ~~power imbalances to keep the~~ grid frequency within admissible limits ~~such power imbalances~~ (ENTSO-E, 2021). Following an imbalance event, the ~~power system inertia~~ inertia of the power system limits the rate of change of frequency (ROCOF) (ESIG, 2022). Historically, the rotating masses of directly coupled synchronous machines (SMs) provided sufficient inertia to ~~limit the ROCOF~~ keep ROCOF within acceptable bounds. However, ~~with the decreasing the declining~~ share of SMs and the increasing ~~share~~ penetration of

inverter-based resources (IBRs) ~~in the overall generation mix, the power system inertia is decreasing have led to a systematic~~  
25 ~~reduction in system inertia~~ (ENTSO-E, 2021). Additionally, the initial ROCOF ~~also increases due to –~~ defined as the mean  
ROCOF over a time window of a few hundred milliseconds after an event – worsens for increasing power system imbalance  
(ENTSO-E, 2016; Thommessen and Hackl, 2024). <sup>1</sup> During the initial ROCOF, only inertia can limit the drop in frequency,  
before other technologies can activate a (fast) frequency response (AEMC, 2017, p. 13, Fig. 2.4; Thommessen and Hackl, 2024  
, p. 285).

30 Worst-case frequency events are ~~caused by typically associated with~~ faults that split the ~~system into subsystems due to~~  
power system into electrically isolated subsystems, resulting in a sudden loss of ~~electrical~~ import or export power (ENTSO-E,  
2021). ~~Furthermore~~ Moreover, increasing transmission capacities, such as ~~high-voltage direct current~~ high-voltage direct-current  
(HVDC) links, may ~~lead to even higher future~~ further exacerbate worst-case power imbalances during system splits in the  
future (ENTSO-E, 2021). Consequently, IBRs must ~~provide inertia to limit the ROCOF and to avoid~~ increasingly contribute  
35 inertia-like responses to limit ROCOF and prevent blackouts in future power systems ~~(ENTSO-E, 2021)~~.

Wind farms (WFs) can support grid frequency by supplying inertia and fast frequency response through the rotating masses  
of the wind turbines (WTs) and by providing reserves (if available). However, conventional grid-following control decouples  
the “physical” inertia of WTs from the grid frequency and thus can *not* provide inertia to the grid (Bossanyi et al., 2020).  
Advanced grid-following control such as “WindINERTIA” control from General Electric (Clark et al., 2010), or the “inertia  
40 emulation (IE)” control from ENERCON (Godin et al., 2019), can temporarily extract kinetic energy reserves to support grid  
frequency. However, this so-called “synthetic” inertia cannot limit the instantaneous or initial ROCOF subject to a system dis-  
turbance (AEMC, 2017; ENTSO-E, 2021; ESIG, 2022). On the contrary, new grid-forming control methods for IBRs, such as  
virtual synchronous machine (VSM) control, provide (virtual synchronous) inertia that limits the initial ROCOF (ESIG, 2022;  
Bossanyi et al., 2020; Rodriguez-Amenedo et al., 2021; Thommessen and Hackl, 2024; Ghimire et al., 2024). Consequently,  
45 future WFs should integrate grid-forming control to provide inertia and fast frequency response. However, this is not only a  
WT control problem, because what a WT can deliver ultimately depends on the intra-farm wake-dominated flows that develop  
within WFs.

New grid codes and market incentives for grid-forming technologies are paving the way for the stability of future power  
systems (ESIG, 2022). Accordingly, system operators are transitioning towards the procurement of inertia provision ~~by from~~  
50 grid-forming technologies. For instance, due to the high penetration of IBRs in Great Britain, the National Grid Electricity Sys-  
tem Operator already defines technical requirements for grid-forming technologies in the grid code and includes grid-forming  
capability as a market product (ESIG, 2022). Similarly, German system operators plan to establish an inertia market and to  
remunerate inertia provision based on its availability (Bundesnetzagentur, 2024). Accordingly, the new German specifica-  
tions (VDE, 2024a) already define technical requirements for grid-forming control and inertia provision. Ghimire et al. (2024)  
55 present a review of existing functional specifications and testing requirements of grid-forming offshore WFs. Hu et al. (2023)  
design an inertia market to ensure sufficient system inertia and analyze its impact on the power generation mix. Their results

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<sup>1</sup> ~~The initial ROCOF is the mean ROCOF over a time window of a few hundred milliseconds after an event, during which only inertia can limit the drop in~~  
~~frequency, before other technologies can activate a (fast) frequency response (AEMC, 2017, p. 13, Fig. 2.4; Thommessen and Hackl, 2024, p. 285).~~

show that investing in wind resources with virtual inertia ~~facilities~~ capabilities is more cost-competitive than substituting wind resources with thermal generators, not to mention the improved environmental impacts.

System inertia monitoring and forecasting are essential to ensure adequate inertia provision. More precisely, system operators  
60 need to quantify the minimum required system inertia to survive worst-case system splits and need to procure sufficient inertia  
~~provision~~ accordingly. Given the uncertainty and variability associated with renewable energy sources, system operators need  
inertia forecasting to ensure that sufficient inertia is available at any time. Similarly, WF operators need WF inertia forecasting  
to participate in future availability-based inertia markets. In particular, WF inertia forecasting enables reliable and profitable  
inertia provision by taking WF control strategies, WF wind input conditions, and intra-WF effects into account. With the future  
65 development of wind at certain busy sites, WF-to-WF wake effects will also have to be considered.

## 1.2 State-of-the-art

~~The~~ Since 2016, the Electric Reliability Council of Texas has ~~been monitoring and forecasting inertia since 2016, but only  
for SMs based on their operating plans~~ monitored and forecasted system inertia solely based on the operating plans of SMs,  
thereby neglecting potential inertia contributions from IBRs (Matevosyan, 2022). ENTSO-E (2017) and General Electric (GE,  
70 2021) monitor inertia ~~based on~~ by measuring the grid frequency and the power imbalance in a (sub)system. However, this  
requires additional measurement units and appropriate online power stimuli. GE (2021) developed an inertia forecaster based on  
machine learning using grid measurement data. However, this approach is only valid for small-signal analysis, as nonlinearities  
~~such as inverter current saturation,~~ cannot be taken into account during rare events with severe ROCOFs. These approaches  
do not consider the fact that the grid-forming capability of a WF depends on its initial operating point, which varies depending  
75 on wind conditions and chosen derating of WTs (Ghimire et al., 2024; Höhn et al., 2024). Thus, new methods for inertia  
forecasting should take the volatile nature of renewable energy into account.

It appears that the existing research does not adequately address the evaluation of the grid-forming capabilities of WTs and  
the forecasting of WF inertia, despite their key relevance for WF and system operators. Although recent publications (Bossanyi  
et al., 2020; Meseguer Urban et al., 2019; Roscoe et al., 2020; Thommessen and Hackl, 2024; Höhn et al., 2024) propose VSM  
80 control for WTs, they do not offer any insights regarding how to choose the VSM inertia. For instance, Meseguer Urban et al.  
(2019) vary the VSM inertia for only one operating point. When discussing offshore WF inertia provision, Höhn et al. (2024)  
only roughly estimate the grid-forming capability by a linear function, which interpolates between the virtual inertia constants  
at cut-in ~~power~~ and at rated power. Due to a lower WT rotor speed limit, Godin et al. (2019) design inertia provision for  
pre-activation power levels above 25% of rated power, risking saturation of the inertial power response to ROCOFs for lower  
85 power levels. Godin et al. (2019) consider grid-following instead of grid-forming or VSM control. Lee et al. (2016) propose  
a simplified gain scheduling for inertia emulation by grid-following WTs, taking the releasable kinetic energy into account.  
However, Lee et al. (2016) solely consider maximum power point tracking (MPPT) and no derating strategies. Moreover, they  
include power, torque and torque rate limits in ~~the control~~ their control implementation by corresponding limiter blocks, but  
these operating limits are not taken into account for the control gain ~~adaption~~ adaptation or for identifying the inertia emulation

90 capability. Their WF simulation results are based on a simple wake model and include only four ambient wind conditions, which heavily simplifies the actual conditions to which ~~the~~ WFs are exposed.

### 1.3 Proposed solution, contributions, and outline

To the best of our knowledge, a generic approach for evaluating the maximum deliverable inertia from WFs for grid-forming control is still missing. Moreover, the methodology for predicting WF inertia based on operation plans has not yet been discussed, although this is key for the reliable and efficient operation of future power systems. Furthermore, even though the intra-farm turbine-to-turbine interactions have a ~~huge~~ very significant influence on the local inflow at the turbines, they have largely been ignored in the existing studies evaluating inertia provision ~~capability. Thus~~ capabilities. To fill these gaps, this paper proposes a novel and generic approach for WF inertia forecasting. This holistic methodology considers weather prediction models, WF flow effects due to wake interactions among the WTs, control strategies, and operational constraints to predict the maximum deliverable inertia at the WT and WF levels. The contributions of this paper include:

- forecasting WF inertia, considering wake effects and operational constraints, using data-driven and physics-based models,
- formulating a nonlinear optimization problem to maximize the inertia provision capability of individual WTs,
- analyzing WT dynamics and relevant operating limits by simulating the inertial response to a reference frequency event,
- integrating VSM control and modifying WT control for inertia provision and fast frequency response,
- demonstrating the proposed approach for evaluating and forecasting deliverable inertia at the WT and WF levels,
- comparing the proposed approach with simplified ones for estimating WT grid-forming capabilities, and
- evaluating the impact of weather forecast uncertainty, wake effects, control strategies, and WT model inaccuracies on WF inertia forecasting.

The rest of this paper is organized as follows. Section 2 presents the necessary background and fundamentals regarding system inertia, ROCOF, and inertia provision by WTs using the VSM concept. Section 3 presents the proposed approach in detail. This includes all the necessary steps for WF inertia forecasting: (i) ~~forecast of~~ forecast of WF ambient wind conditions ~~forecast in~~ (Sect. 3.1.); (ii) ~~prediction of~~ prediction of local WT operating points ~~prediction in~~ (Sect. 3.2, ~~and~~); (iii) ~~mapping of all operating points, given by local wind inflow conditions and operational setpoints, to the~~ WT modeling and control (Sect. 3.3); and (iv) WF grid-forming capability ~~in Sects~~ (Sect. 3.4–3.3). Section 4 presents a case study ~~for concerning~~ a WF with twelve WTs and discusses the results, including the WT steady states, the WT inertial response to a reference frequency event, and the WF hour-ahead inertia forecasting. Finally, Sect. 5 summarizes the entire work and offers concluding remarks, including outlook for future work.

## 2 Background and fundamentals

The initial ROCOF immediately after a system power imbalance  $\Delta P_s$  between mechanical system power  $P_{m,s}$  and electrical system power  $P_{e,s}$  can be approximated by a one-mass model (ENTSO-E, 2020; Thommessen and Hackl, 2024), written as <sup>1</sup>

$$\dot{\Omega}_s = \frac{\Delta P_s}{2H_s} := \frac{P_{m,s} - P_{e,s}}{2H_s}, \quad (1a)$$

$$\text{where } H_s := \frac{E_{\text{kin},s}}{p_{s,R}} := \frac{\frac{1}{2}\Theta_s\omega_s^2}{\sum p_R}, \quad \Theta_s := \sum \Theta \quad P_{m,s} := \frac{\sum p_m}{p_{s,R}}, \quad \text{and} \quad P_{e,s} := \frac{\sum p_e}{p_{s,R}}. \quad (1b)$$

The system inertia constant  $H_s$  (in s) is the system kinetic energy  $E_{\text{kin},s}$  (in W s) normalized to the rated system power  $p_{s,R}$  (in W), defined as the sum of rated power  $p_R$  of all (V)SMs. Please notice that, here and throughout the paper, we use capital letters to indicate normalized quantities. The system moment of inertia  $\Theta_s$  (in kg m<sup>2</sup>) is defined as the sum of the moment of inertia  $\Theta$  of all (V)SMs. This includes (V)SMs at the generation side but also at the demand or load side, i. e. (V)SMs provide inertia in both generator and motor mode. The system angular velocity is  $\omega_s = 2\pi f_s \approx 2\pi f_{s,R}$  with system frequency  $f_s$  (in Hz) of all synchronously rotating masses and rated system frequency  $f_{s,R}$ . When the admissible-neglecting frequency deviations of up to 0.4% during normal and up to 5% during critical system states (VDE, 2024a) are neglected, it follows that  $\Omega_s := f_s/f_{s,R} \approx 1$ . This assumption leads to equal (normalized) power  $P = \Omega_s M$  and implies that the normalized power  $P$  is equal to the torque  $M$  quantities in Eq. (1a), as  $P = \Omega_s M$ .  $P_{m,s}$  and  $P_{e,s}$  are defined as the sum of mechanical power  $p_m$  and electrical power  $p_e$  of all (V)SMs, respectively, both normalized to  $p_{s,R}$ . More precisely, for a (V)SM,  $p_m$  is the mechanical power of the (virtual) turbine,  $p_e$  is the electrical power of the (virtual) SM, and  $\Theta \neq 0$  is the (virtual) total drivetrain moment of inertia. For grid-following WTs (without VSM),  $p_m$  and  $p_e$  are the mechanical and electrical WT power, respectively, but ; however, the grid-connected moment of inertia is  $\Theta = 0$  due to decoupled because the physical WT inertia is decoupled from the grid. Thus, assuming  $p_m \approx p_e$ , WTs or the WTs and all IBRs that are operating under grid-following control can be neglected in Eq. (1a), i.e. and only (V)SMs contribute to limiting the initial ROCOF  $\dot{\Omega}_s$ .

For a SM, the inertia constant  $H := E_{\text{kin}}/p_R$  is defined as the ratio of the kinetic energy  $E_{\text{kin}}$  and the rated power  $p_R$ . Similarly, for a VSM-controlled WT, the virtual inertia constant  $H_v := E_{\text{kin},v}/p_R$  is defined as the ratio of the VSM kinetic energy  $E_{\text{kin},v}$  and the WT rated power  $p_R$ . Note that  $E_{\text{kin},v}$  differs from the WT physical kinetic energy in general. In particular, a (V)SM always rotates near synchronous speed, whereas the WT speed depends on wind and operating conditions. In contrast to the WT physical inertia constant, the virtual inertia constant  $H_v$  is a tunable control parameter. Finally, aggregating all (V)SMs leads to the system inertia  $H_s$  in Eq. (1a). However, this is only valid for a proper tuning of  $H_v$  because, e. g., emulating a high  $H_v$  may not be feasible due to output power limitations. SMs provide an overload capability of 3 to 5 times, whereas IBRs only allow for an overloading of 1 to 1.5 times, which limits the VSM inertial power response depending on the ROCOF (ESIG, 2022). For a VSM-controlled WT, choosing a high  $H_v$ , e. g.  $H_v > H$  with physical WT inertia constant  $H$ , increases the inertial grid support for low ROCOFs but increases the risk of undesired output power saturation for higher ROCOFs (Höhn et al., 2024). This has to be taken into account when replacing physical inertia by virtual inertia in future power systems.

<sup>1</sup>For the nomenclature used in this work, see Appendix ??.

150 WT curtailment or derating strategies provide power reserve, e. g. for primary frequency or droop control (Kanev and van de  
Hoek, 2017; Bossanyi et al., 2020; Clark et al., 2010). For inertia provision, derating based on the maximum rotation strategy  
(MRS) additionally increases the WT kinetic energy reserve (Ramtharan et al., 2007; Aho et al., 2014; Meseguer Urban et al.,  
2019; Thommessen and Hackl, 2024). Although derating strategies enhance grid frequency support, they also reduce WT  
155 power efficiency. WF and system operators should find a Pareto optimal strategy that considers system stability and efficiency  
to avoid unnecessary curtailment of renewables.

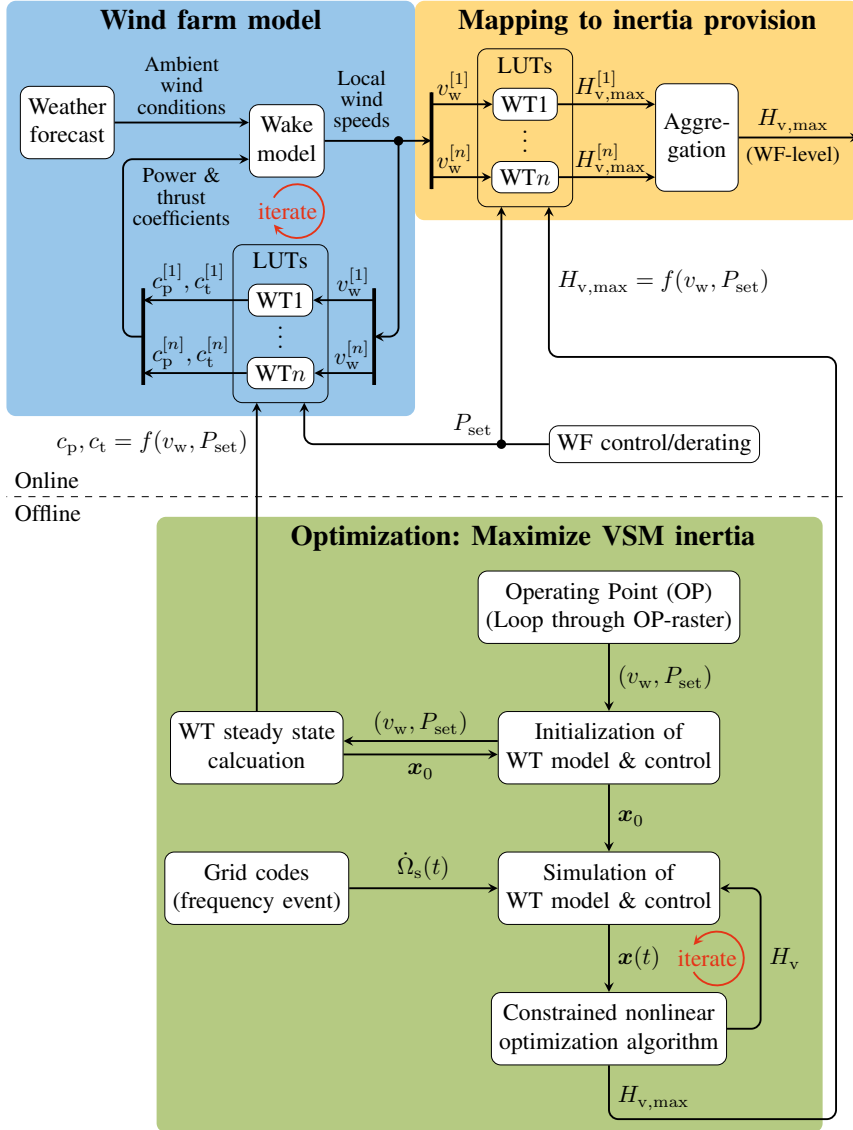
Consequently, WF inertia forecasting is essential for reliable inertia provision through adequate WT derating and precise  
tuning of VSM inertia.

### 3 Methodology

The proposed approach combines online and offline calculations, as depicted in the overview of Fig. 1. First, a data-driven  
160 weather forecast model predicts the site ambient wind conditions. These ambient conditions serve as input to the WF model,  
which incorporates the aerodynamic characteristics of all  $n$  WTs in the WF, given by the power coefficients  $c_p^{[1]}, \dots, c_p^{[n]}$  and the  
thrust coefficients  $c_t^{[1]}, \dots, c_t^{[n]}$ . The WF flow model outputs the local wind speeds  $v_w^{[1]}, \dots, v_w^{[n]}$ , which are fed back to lookup  
tables (LUTs) for the power and thrust coefficients. These LUTs of the form  $c_p, c_t = f(v_w, P_{\text{set}})$  are obtained through offline  
165 calculation of the WT steady states  $x_0$  for all WT operating points defined by wind speed  $v_w$  and power setpoint  $P_{\text{set}}$  (in % of  
available power at the MPP). ~~The wake model iteratively computes local wind speeds at all WTs. Additional-~~

LUTs of the form  $H_{v,\text{max}} = f(v_w, P_{\text{set}})$  are calculated offline in Fig. 1 by solving optimization problems, which maximize  
the VSM inertia constant  $H_v$  for a given WT operating point  $(v_w, P_{\text{set}})$  and a reference frequency event defined by grid codes.  
More precisely, an optimization algorithm iteratively runs simulations of the WT response to a ROCOF  $\dot{\Omega}_s(t)$  with different  
 $H_v$  to find the maximum VSM inertia constant  $H_{v,\text{max}}$  that the WT can provide without violating operating constraints. With  
170 the frequency event starting at  $t = t_0$ , the operating constraints ensure that the WT states  $x(t)$  are within their admissible value  
range for all  $t \geq t_0$ . The LUTs  $H_{v,\text{max}} = f(v_w, P_{\text{set}})$  are evaluated online in Fig. 1 to map the WT operating points to the  
maximum inertia constants  $H_{v,\text{max}}^{[1]}, \dots, H_{v,\text{max}}^{[n]}$  of all  $n$  WTs. Finally, assuming an optimal  $H_v = H_{v,\text{max}}$  tuning for each WT  
in the WF and aggregating  $H_{v,\text{max}}^{[1]}, \dots, H_{v,\text{max}}^{[n]}$ , yields the inertia provision in terms of the maximum inertia constant  $H_{v,\text{max}}$   
at WF level. The proposed approach is generic because it is applicable to different modeling and control formulations of WTs  
175 and WFs.

Although taking wake effects into account for the initial conditions, the proposed approach assumes that the wind conditions  
do not change for the duration  $\Delta t$  of the frequency event. Despite the volatile nature of real wind profiles, such an assumption  
for inertia forecasting at the WF level is reasonable, because of an expected averaging effect of any local fluctuations due to the  
aggregation over several WTs. Moreover, the change in ~~wake behavior~~ turbine waking during the inertial response is typically  
180 ~~probably~~ negligible due to the propagation delay of farm flow effects. For example, ~~for consider~~ a moderate-sized onshore WT  
with a rated wind speed of  $10 \text{ m s}^{-1}$  and a rotor diameter of 130 m. In onshore farms, WTs are usually ~~placed apart at a spaced~~  
between 2D to 5D ~~distance from each other~~, in an optimal layout design subject to spacing constraints (Stanley et al., 2022).



**Figure 1.** Overview of the proposed WF inertia forecasting approach, which predicts the maximum deliverable inertia constant  $H_{v,\max}$  of the WF based on online weather forecasting and offline calculated LUTs.

For the worst-case scenario ~~-, considering of~~ a very short 2D spacing, any change in control action on the upstream WT will take ca. 26 s to reach the downstream WT. This time duration is much greater than the inertial response time or the duration of  
185 a severe ROCOF, which lasts only a few seconds.

### 3.1 Ambient WF wind forecast

Wind conditions are forecasted using fully connected neural networks (FCNNs) based ~~upon-on~~ the methods discussed in Anand et al. (2024). The training targets are the north-aligned component  $v_w^u$  and the east-aligned component  $v_w^v$  of the wind measurements at the site over the forecast horizon. ~~Features from the two~~ Wind speed and direction measurements are obtained  
190 from historical SCADA data; this way, the coarse resolution predictions of numerical weather prediction (NWP) models are  
brought to the specific site where the WF is located. Features from the two NWP models ICON-EU and ARPEGE are used  
as input data (Zängl et al., 2015; Courtier et al., 1991). ~~Furthermore, input data also include lag characteristics, i. e., targets~~  
~~for~~, together with the  $u$  and  $v$  components ~~of-at the present and~~ previous timestamps. The choice of forecast horizon may  
range from several minutes to several hours, depending on the application use case. For example, for a short-term availability  
195 prediction, a forecast horizon of a few minutes to one hour is relevant. However, a forecast horizon of up to 36 hours can be  
necessary for energy market applications.

The probabilistic wind forecast is obtained using a machine learning-based model that utilizes Gaussian mixture distributions formed by superimposing several normal distributions. The resulting probability distribution is given by

$$p(x) = \sum_i^n w_i N(x|\mu_i, \sigma_i), \quad (2)$$

200 where  $w_i$  represents the weight,  $\mu_i$  the mean and  $\sigma_i$  the standard deviation of the  $i$ -th Gaussian normal distribution. Due to the long forecast horizon, an ensemble method consisting of several FCNNs was utilized to predict the parameters  $w_i$ ,  $\mu_i$ , and  $\sigma_i$ , where each network is trained only on a specific segment of the overall forecast horizon. This approach was chosen due to its ability to deliver an improved forecast accuracy for each individual segment, as opposed to using networks designed to forecast over the entire time horizon. By focusing on shorter segments, the model can better capture dynamic variations and  
205 nuances in the data, leading to more precise predictions. The final forecast is obtained by combining the outputs from multiple ensemble networks, each trained on a specific segment of the data. This ensemble method enhances the overall reliability and accuracy of the forecast. In particular, using a configuration with four networks proved to be an effective compromise, striking a balance between maintaining robustness and minimizing the training time required. This formulation allows for sufficient model flexibility while optimizing computational efficiency, making it a practical choice for operational forecasting.

210 To reduce the number of input parameters for the FCNNs, a feature-selection algorithm is applied to each of the FCNNs within the ensemble. This is followed by a hyper-parameter optimization process to determine an appropriate number  $n$  of normal distributions for the mixed distribution, and to fine-tune both the individual FCNN architectures and the training optimizer. The hyper-parameter optimization is automated and utilizes policy gradients with parameter-based exploration (PGPE) (Sehnke et al., 2010). The training process employs the Adam optimizer, using a mean squared error loss function (Kingma and Ba, 2014). The dataset is divided into training (88 %) and validation (12 %) subsets.  
215

### 3.2 Local WT operating point prediction

An ~~engineering wake~~ steady-state engineering flow model is employed to predict the local inflow conditions at each WT within the WF ~~at steady-state~~ (NREL, 2022). The wake flow model takes ambient weather forecasts as inputs and models the ~~wake position and velocity deficit~~ path and flow characteristics of all wakes within the WF, for given turbine characteristics and operational setpoints. ~~This results in local wind condition forecasts at each WT, which is crucial for accurate performance prediction. More precisely, offline~~ Offline computed WT LUTs of the form  $c_p, c_t = f(v_w, P_{\text{set}})$  map the local wind speed  $v_w$  and power setpoint  $P_{\text{set}}$  to the power coefficient  $c_p$  and thrust coefficient  $c_t$ . In general, the flow speed  $v_w$  at ~~the a~~ downstream WT depends on the  $c_t$  of the upstream ~~WTs~~ WTs. Thus, the wake model iteratively computes the local wind speeds at all WTs, as shown in Fig. 1.

### 3.3 WT modeling and control

Figure 2 depicts the overall WT modeling and control used in this work. During normal operation, the MRS-based power setpoint  $\bar{P}_{\text{set}}^*$  defines the electromagnetic torque  $M_e$  ~~in p. u. of rated torque  $m_R$  (at the low speed shaft in N m)~~. During frequency events,  $M_e$  additionally depends on the active power droop and inertia provision through VSM control. Details of these torque controllers follow in Sects. 3.3.2–3.3.5.

~~WT modeling and control for solving the optimization problem in Eq. (14). The simplified control representation for the iterative simulations during optimization is derived based on the VSM control for WTs in Thomassen and Haackl (2024). All saturations or manipulations of the power reference  $P_{\text{ref}}$  for WT protection have been removed and converted into corresponding optimization constraints.~~

~~The WT modeling approach utilized in this work is based on the reference design in Bortolotti et al. (2019), except for the pitch controller that here does not include a tip speed constraint below rated wind speed. Thus, at the rated wind speed  $v_{w,R} = 9.8 \text{ m s}^{-1}$ , the WT operates at its MPP with optimal tip speed ratio  $\lambda^* = 8.5$ . Consequently, the rated tip speed  $\omega_{R,r} = \lambda^* v_{w,R} = 83.3 \text{ m s}^{-1}$  (slightly) exceeds the tip speed limit of  $80 \text{ m s}^{-1}$  assumed in Bortolotti et al. (2019). The resulting A standard pitch control strategy is assumed, with the main objective to limit the WT rotor speed  $\Omega$ , expressed in per unit (p. u.) of rated WT rotor speed  $\omega_R$  is ca. 4.2 larger than the rated value in Bortolotti et al. (2019), but still ca. 5.3 smaller than the maximum assumed rotor speed limit. The higher speed rating increases not only the rated power but also the rated kinetic energy reserve for inertia provision compared with Bortolotti et al. (2019). Neglecting for simplicity any conversion losses from mechanical to electrical power, the WT physical inertia constant at rated speed becomes-~~

$$\underline{H_R = \frac{\frac{1}{2} \Theta \omega_R^2}{p_R} = 3.26 \text{ s.}}$$

~~The WT physical inertia constant  $H$  (which should be better called “inertia variable” due to the variable  $\Omega$ ) is proportional to the WT kinetic energy  $E_{\text{kin}}$ , i. e.~~

$$\underline{H := \frac{E_{\text{kin}}}{p_R} := \frac{\frac{1}{2} \Theta \omega^2}{p_R} = \Omega^2 H_R.}$$

Note that, for (directly grid-connected) SMs (of conventional power plants), it follows that  $H = \Omega_s^2 H_R \approx H_R$  due to  $\Omega_s \approx 1$ .

The blade pitch angle system modeling and control is based on Thommessen and Hackl (2024, Sect. II.C and Sect. III.A). The main objective of the ~~pitch control~~ is to increase the pitch controller, implemented based on Thommessen and Hackl (2024, Sect. III.A), increases the blade pitch angle reference  $\beta_{\text{ref}}$  for above-rated wind speeds to limit the rotor speed to  $\Omega = 1$ . In addition to  $\Omega$ , the wind speed  $v_w$  is also required as input for the pitch control in Fig. 2, in order to adapt the lower pitch angle limit  $\beta_{\text{min}}$  based on the tip speed ratio, i.e.  $\lambda \propto v_w^{-1}$ , i.e.  $\beta_{\text{ref}} \geq \beta_{\text{min}}(\lambda)$ . This is more relevant for derating than for MPPT.

The WT physical inertia constant  $H$  (which should be better called ‘‘inertia variable’’ due to the variable  $\Omega$ ) is proportional to the WT kinetic energy  $E_{\text{kin}}$ , i. e.

$$H := \frac{E_{\text{kin}}}{p_R} := \frac{\frac{1}{2} \Theta \omega^2}{p_R} = \Omega^2 H_R \quad (3)$$

where  $H_R$  denotes the inertia constant at rated speed  $\Omega = 1$ . Note that, for (directly grid-connected) SMs (of conventional power plants), it follows that  $H = \Omega_s^2 H_R \approx H_R$  due to approximately constant SM rotor speed or system frequency  $\Omega_s \approx 1$ .

### 3.3.1 Aeroelastic and mechanical model

The power coefficient  $c_p$  and the thrust coefficient  $c_t$  are modeled as functions of tip speed ratio  $\lambda$  and blade pitch angle  $\beta$  by the corresponding LUTs, see Fig. 6. The fore-aft deflection of the WT tower, excited by the thrust force  $F_t$ , is modeled as a mass-spring-damper oscillator with mass  $m_t$ , damping coefficient  $d_t$  and stiffness coefficient  $k_t$ . The aeroelastic model outputs the WT mechanical torque  $M_m$  (in p. u. of  $m_R$ ) for given inputs  $(v_w, \beta, \Omega)$ , see also Fig. 2, i. e.

$$P_w = \frac{F_w v_w}{p_R} = \frac{1}{2} \rho \pi r^2 v_w^3 / p_R, \quad (4a)$$

$$M_m = \frac{P_m}{\Omega} = \frac{P_w c_p(\lambda, \beta)}{\Omega}, \quad \lambda = \Omega \frac{\omega_R r}{\tilde{v}_w}, \quad \tilde{v}_w = v_w - \dot{s}_t, \quad (4b)$$

$$\ddot{s}_t = \frac{1}{m_t} (F_t - d_t \dot{s}_t - k_t s_t), \quad F_t = \frac{1}{2} \rho \pi r^2 v_w^2 F_w c_t(\lambda, \beta), \quad s_{t,0} = \frac{F_w c_t(\lambda_0, \beta_0) F_t}{k_t}, \quad \dot{s}_{t,0} = 0, \quad (4c)$$

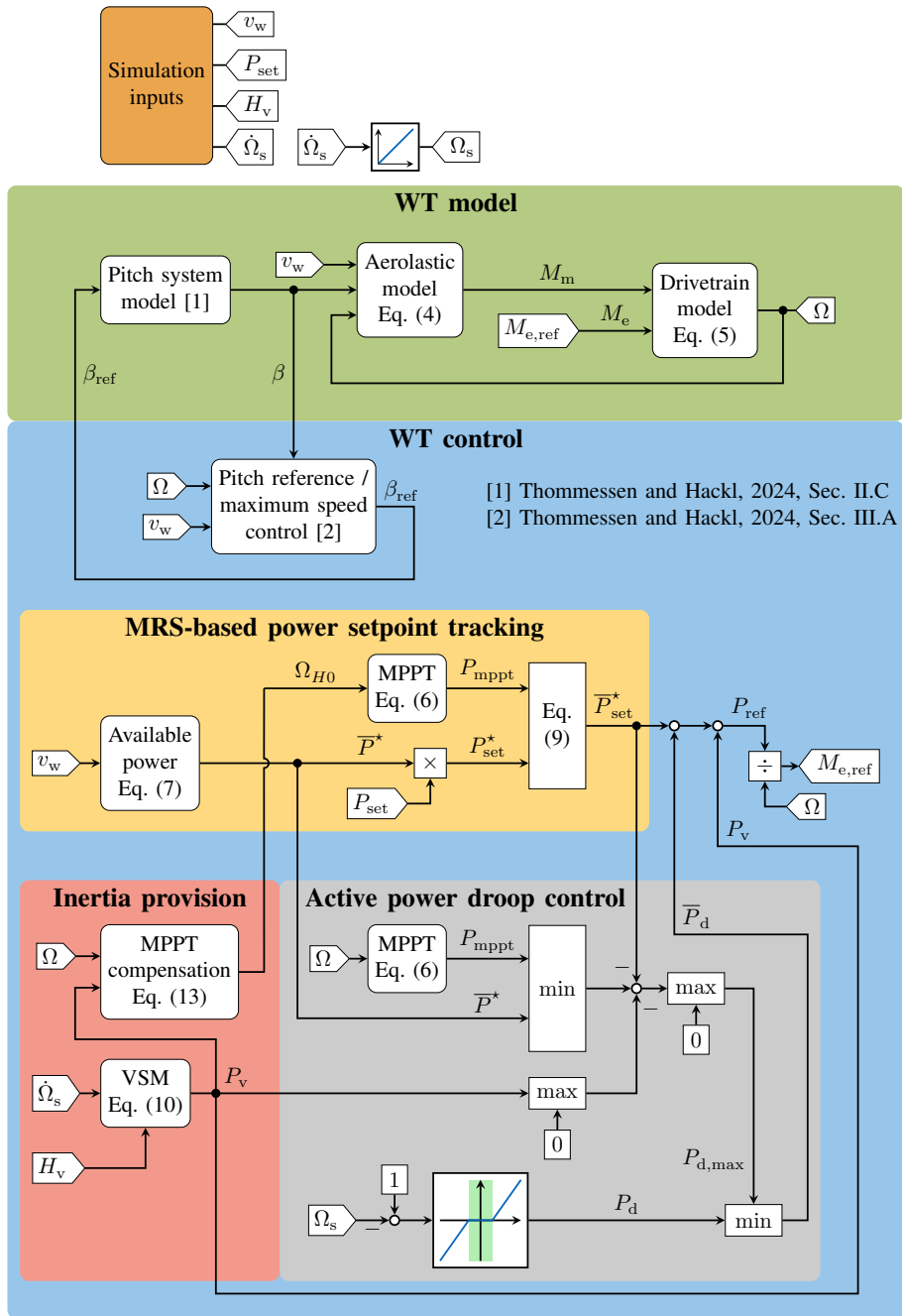
with wind power  $P_w$  (in p. u. of  $p_R$ ), wind-generated force  $F_w$  WT aerodynamic or mechanical power  $P_m$  (in p. u. of  $p_R$ ), air density  $\rho$ , rotor radius  $r$ , relative wind speed  $\tilde{v}_w$ , WT fore-aft tower displacement  $s_t$ , and initial steady-state values  $s_{t,0}, \dot{s}_{t,0}, \lambda_0, \beta_0$ . Indicating with  $H_R$  (in s) the WT total physical drivetrain inertia constant, the WT mechanical dynamics are approximated by a one-mass model, i. e. (Thommessen and Hackl, 2024)

$$\dot{\Omega} = \frac{M_m - M_e}{2H_R}, \quad \Omega(0) = \Omega_0. \quad (5)$$

### 3.3.2 Maximum rotation strategy

The electrical power for MPPT is given by

$$P_{\text{mppt}} := \Omega M_{\text{mppt}}, \quad (6)$$

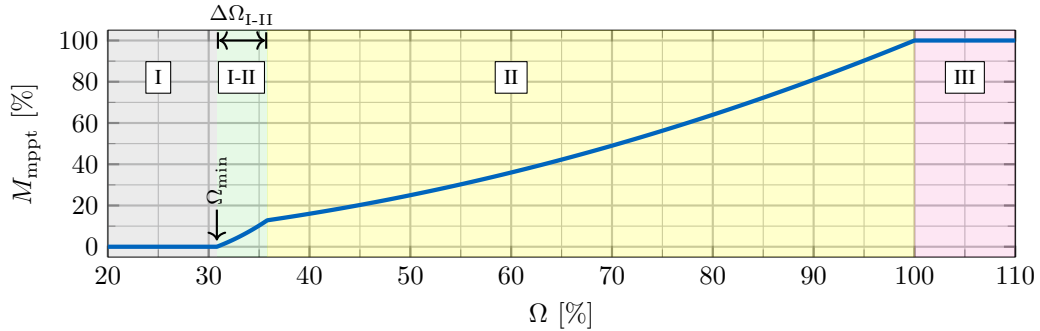


**Figure 2.** WT modeling and control for solving the optimization problem in Eq. (14). The simplified control representation for the iterative simulations during optimization is derived based on the VSM control for WTs in Thommessen and Hackl (2024). All saturations or manipulations of the power reference  $P_{ref}$  for WT protection have been removed and converted into corresponding optimization constraints.

275 where the MPPT torque curve is the function of rotor speed  $M_{\text{mppt}} := M_{\text{mppt}}(\Omega)$  shown in Fig. 3, ~~the electrical power for MPPT is given by~~

$$\underline{P_{\text{mppt}} := \Omega M_{\text{mppt}}.}$$

280 Below rated wind speed in region II,  $M_{\text{mppt}}$  increases proportionally to  $\Omega^2$  for optimal operation at the MPP. Above rated wind speed in region III, the pitch control limits the WT rotor speed to  $\Omega = 1$  such that  $P_{\text{mppt}} = M_{\text{mppt}} = 1$ . Below cut-in wind speed  $v_{w,\text{cut-in}}$  in region I, the WT does not generate power, i. e.  $M_{\text{mppt}} = 0$  for  $\Omega < \Omega_{\text{min}}$ . For a smooth transition to region II, a non-optimal operation is accepted in the small transition region I-II defined by  $\Omega_{\text{min}} \leq \Omega \leq \Omega_{\text{min}} + \Delta\Omega_{\text{I-II}}$ , i. e.  $M_{\text{mppt}}$  is obtained by multiplying the optimal torque at the MPP by a factor that is linearly interpolated between 0 at  $\Omega_{\text{min}}$  and 1 at  $\Omega_{\text{min}} + \Delta\Omega_{\text{I-II}}$ . For a more complete description of the MPPT curve, see Thommessen and Hackl (2024, Sect. III.B.1).



**Figure 3.** MPPT torque as a function of rotor speed  $\Omega$ .

285 The MRS-based derating maximizes the WT kinetic energy reserve for inertia provision. The derating power setpoint  $P_{\text{set}}$  is defined relative to the MPP, i. e.  $P_{\text{set}} = 1$  corresponds to MPPT and  $P_{\text{set}} < 1$  corresponds to derating. Increasing derating (decreasing  $P_{\text{set}}$ ) reduces the electrical power setpoint  $P_{\text{set}}^* := \bar{P}^* P_{\text{set}}$  with available power  $\bar{P}^* \in (0, 1]$ , i. e. the WT accelerates. The As a consequence, the tip speed ratio  $\lambda$  increases such that and the power coefficient  $c_p$  decreases (see Fig. 6). The pitch controller additionally increases  $\beta$  if necessary to limit the rotor speed to  $\Omega = 1$ . In general, the MRS prioritizes increasing WT speed over pitching to provide power reserve.

290 With the measured or estimated rotor-effective wind speed  $v_w$  (Soltani et al., 2013) ~~and the MPPT power coefficient function or LUT  $\bar{c}_p^*(v_w)$  (see Appendix 8) the,~~ the available power in Fig. 2 is defined as

$$\bar{P}^* := P_w(v_w) \bar{c}_p^*(v_w) := \frac{1}{2} \rho \pi r^2 v_w^3 \bar{c}_p^*(v_w) / p_{R,z}, \quad (7)$$

where the MPPT ( $P_{\text{set}} = 1$ ) power coefficient is given by

$$295 \quad \underline{\bar{c}}_p^*(v_w) = \begin{cases} c_p(\bar{\Omega}^*, \bar{\beta}^*, v_w) & \text{if } v_{w,\text{cut-in}} \leq v_w < v_{w,\text{II,min}}, \\ c_p^* := \max c_p & \text{if } v_{w,\text{II,min}} \leq v_w \leq v_{w,\text{R}}, \\ \frac{1}{P_w(v_w)} & \text{if } v_{w,\text{R}} < v_w \leq v_{w,\text{cut-out}}, \end{cases} \quad (8)$$

with cut-in wind speed  $v_{w,\text{cut-in}}$ , minimum region-II wind speed  $v_{w,\text{II}}$ , rated wind speed  $v_{w,\text{R}}$ , cut-out wind speed  $v_{w,\text{cut-out}}$ , and steady-state values  $(\bar{\Omega}^*, \bar{\beta}^*)$ .

Limiting  $P_{\text{set}}^*$  by  $P_{\text{mppt}}$  for rotor speed transients and wind measurement errors, the saturated power setpoint is given by <sup>1</sup>

$$\bar{P}_{\text{set}}^* := \left\{ \begin{array}{ll} \min(P_{\text{set}}^*, P_{\text{mppt}}), & \text{if } P_{\text{set}} < 1 \text{ (derating),} \\ P_{\text{mppt}}, & \text{if } P_{\text{set}} = 1 \text{ (MPPT).} \end{array} \right\} \quad \text{with } P_{\text{set}}^* := \bar{P}^* P_{\text{set}}. \quad (9)$$

300  $\bar{P}_{\text{set}}^*$  is ignored for  $P_{\text{set}} = 1$  (MPPT) in Eq. (9), since (i) no wind measurements are required, and (ii) smaller transient rotor speed overshoots occur due to higher power setpoint adaptation. For example, if  $\Omega > 1$  due to a wind gust, it follows that  $\bar{P}_{\text{set}}^* = P_{\text{mppt}} = \Omega M_{\text{mppt}} > 1$ , whereas  $P_{\text{set}}^* \leq 1$ .

### 3.3.3 VSM control

Grid-forming control is required to limit the initial ROCOF (ESIG, 2022; VDE, 2024a). This paper simplifies the grid-forming VSM control proposed in Thommessen and Hackl (2024), by neglecting fast electromagnetic transients and low-level current control loops. However, the grid synchronization dynamics of grid-forming control define the inertial response and must, therefore, be taken into account.

#### Grid synchronization loop of a freely spinning VSM.

For VSM control, the grid synchronization dynamics are similar to the dynamics of a real (grid-connected) SM, as illustrated in Fig. 4.

The VSM acceleration  $\dot{\Omega}_v$  is proportional to the sum of virtual torques, i. e. the VSM mechanical model is based on a one-mass model with virtual inertia  $H_v$  instead of physical inertia  $H_R$  in Eq. (5). At steady state, the difference between VSM mechanical and electromagnetic torque is zero, i. e.  $M_{m,v} - M_{e,v} = 0$ , and the VSM damping torque is zero, i. e.  $M_{dp,v} = 0$ . Since only the inertial response to ROCOFs or electromagnetic changes are of interest, the VSM mechanical torque is set to zero, i. e.  $M_{m,v} = 0$ , resulting in the freely spinning VSM in Fig. 4. Denormalization of the power system frequency, i. e.  $\omega_s = \Omega_s \omega_{s,R}$  with  $\omega_{s,R} := 2\pi f_{s,R}$ , and subsequent integration of  $\omega_s$  yields the grid or system angle  $\phi_s$ . Similarly, denormalization of the VSM rotor speed, i. e.  $\omega_v = \Omega_v \omega_{s,R}$ , and subsequent integration of  $\omega_v$  yields the VSM rotor angle  $\phi_v$ . The VSM electromagnetic torque  $M_{e,v}$  depends on the (real) load angle  $\delta := \phi_v - \phi_s$  multiplied by the electromagnetic feedback gain  $k_e$ . Due to unknown  $\phi_s$  or  $k_e$ , the VSM controller calculates the torque or power feedback directly based on current and grid voltage measurements. The VSM damping torque  $M_{dp,v}$  is proportional to the VSM slip  $\Omega_v - \Omega_s$ , which emulates the effect of

<sup>1</sup>In Eq. (9),  $\bar{P}_{\text{set}}^*$  is ignored for  $P_{\text{set}} = 1$  (MPPT), since (i) no wind measurements are required, and (ii) smaller transient rotor speed overshoots occur due to higher power setpoint adaptation. For example, if  $\Omega > 1$  due to a wind gust, it follows that  $\bar{P}_{\text{set}}^* = P_{\text{mppt}} = \Omega M_{\text{mppt}} > 1$  whereas  $P_{\text{set}}^* \leq 1$ .

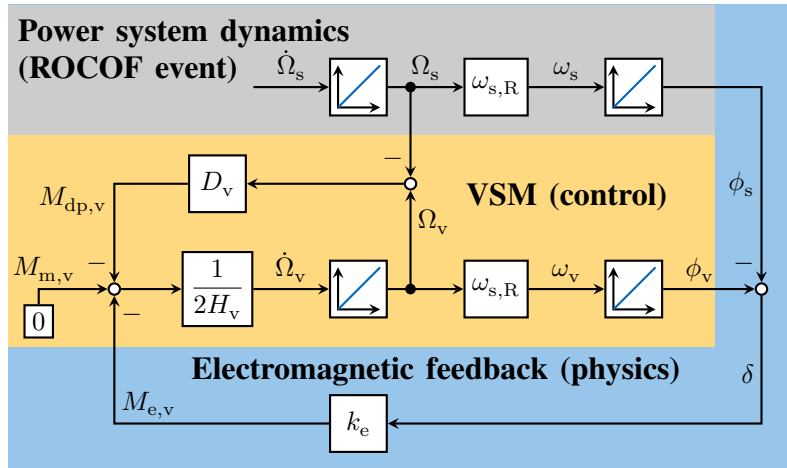


Figure 4. [Grid synchronization loop of a freely spinning VSM.](#)

dampers in SMs. However, unlike real SMs, the VSM enables flexible tuning of the VSM damping  $D_v$ . Grid voltage measurements are required to determine  $\Omega_s$  for VSM damping torque calculation.

The VSM power for inertia provision, added to the power setpoint  $\bar{P}_{set}^*$  in Fig. 2, is defined as

$$P_v = \Omega_v M_{e,v}, \quad (10)$$

325 where  $\Omega_v$  and  $M_{e,v}$  are given by the grid synchronization loop in Fig. 4 with input  $\dot{\Omega}_s$ .

The inertial response in the Laplace domain of the VSM is given by (Thommessen and Hackl, 2024)

$$\frac{M_{e,v}}{\dot{\Omega}_s} = \frac{-k_e \omega_{s,R}}{s^2 + 2\zeta_v \omega_{n,v} s + \omega_{n,v}^2}, \quad \omega_{n,v}^2 := \frac{k_e \omega_{s,R}}{2H_v}, \quad \zeta_v := \frac{D_v}{\sqrt{8H_v k_e \omega_{s,R}}} = 1 \Rightarrow D_v := D_v(H_v) := \sqrt{8H_v k_e \omega_{s,R}} \quad (11)$$

with natural angular velocity  $\omega_{n,v}$  and damping ratio chosen as  $\zeta_v = 1$  to avoid overshooting. With grid synchronized VSM speed  $\Omega_v = \Omega_s \approx 1$  in Eq. (10) and setting  $s = 0$  in Eq. (11), the steady-state VSM power for a constant ROCOF simplifies to

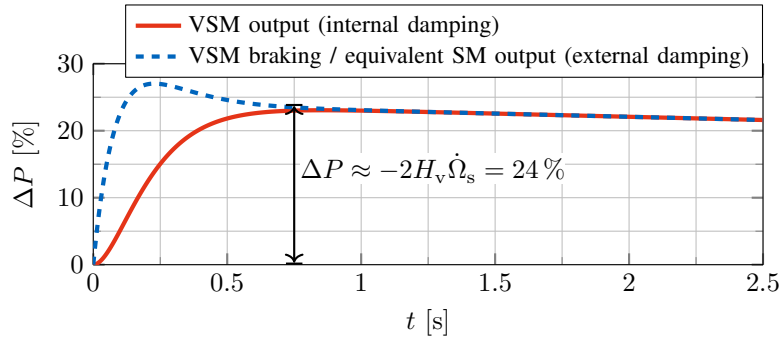
$$330 \quad P_v = \Omega_v M_{e,v} \approx M_{e,v} \approx -2H_v \dot{\Omega}_s. \quad (12)$$

The power system dynamics in Fig. 4 are defined by a reference frequency event. The electromagnetic feedback in Fig. 4 depends on the load angle given by the angle difference between VSM and grid, i. e.  $M_{e,v} = k_e \delta = k_e (\phi_v - \phi_s)$ . For simplicity, this paper assumes a constant electromagnetic feedback gain  $k_e$ . Actually,  $k_e$  depends on the WT operating point and the WT grid connection, i. e.  $k_e$  is a nonlinear function of the load angle  $\delta$ , the grid voltage and the grid impedance (VDE, 2024a;

335 Ghimire et al., 2024). Type 3 WTs use doubly-fed induction machines (DFIMs), where  $k_e$  also depends on the DFIM rotor current or excitation level (Thommessen and Hackl, 2024, Sect. III.E). If not negligible, the dependency of  $k_e$  on  $\delta$  and the excitation level should be taken into account based on the WT operating point. With admissible limits for grid voltage and impedance defined by grid codes (VDE, 2024a),  $k_e$  should be chosen based on the WT grid connection, [see Appendix ?? as explained in Appendix B.](#)

340 This paper assumes internal damping of the VSM (Roscoe et al., 2020; Thommessen and Hackl, 2024), i. e. the damping torque  $M_{dp,v}$  in Fig. 4 is solely virtual and is not converted into real electrical output power. In contrast, for an external damping of a real SM, the damping power is part of the electrical output power (Roscoe et al., 2020). In this regard,  $H_v$  of a VSM and  $H$  of a (real) SM differ, i. e., assuming  $H_v = H$  and equal damping gains, the actually extracted kinetic energy during the inertial response is smaller for a VSM-controlled WT than for a (real) SM, ~~see Fig. 5.~~

345 Figure 5 shows an exemplary inertial power response to a ROCOF with quasi-steady-state amplitude  $\Delta P$  approximated by Eq. (12).



**Figure 5.** Inertial power response to a ROCOF of  $\dot{\Omega}_s = -4\% s^{-1}$  for  $H_v = 3s$  with quasi-steady-state amplitude  $\Delta P$  approximated by Eq. (12). The VSM achieves the desired internal damping of the VSM output power, whereas the VSM braking power overshoots. The VSM output power equals the WT electrical power change. The VSM braking power equals the electrical power of an equivalent real SM with external damping, i. e. the SM braking energy is fully converted into electrical energy according to the law of conservation of energy.

The damping energy corresponds to the area between the two curves in Fig. 5. Strictly speaking, the VSM concept violates the law of conservation of energy since the VSM braking energy is not fully converted into electrical energy. However, high internal damping avoids power overshoots and is beneficial for grid frequency stability (Roscoe et al., 2020; Thommessen and Hackl, 2024). Also, grid codes ~~(VDE, 2024a)~~ require sufficient damping and consider (internal) damping power separately from electrical output power ~~, see VDE (2024a, Kap. 5.1.1.11, Anmerkung 1)~~ (VDE, 2024a, Kap. 5.1.1.11, Anmerkung 1). Finally,  $H_v$  is comparable to  $H$  of a real SM when neglecting the transient damping, i. e. when considering the quasi-steady-state power change  $\Delta P = -2H_v\dot{\Omega}_s$ , as shown in Fig. 5. Thus,  $H_v$  is a suitable measure for the inertial power response and the grid frequency support by inertia provision. <sup>1</sup>

355 ~~Inertial power response to a ROCOF of  $\dot{\Omega}_s = -4\% s^{-1}$  for  $H_v = 3s$  with quasi-steady-state amplitude  $\Delta P$  approximated by Eq. (12). The VSM achieves the desired internal damping of the VSM output power, whereas the VSM braking power~~

<sup>1</sup>The recent draft VDE (2024b) for certification of grid-forming IBRs quantifies inertia provision by the mean power change over a time window starting 0.5s after the ROCOF change and ending at the beginning of the next ROCOF change during the reference frequency event, i. e.  $T_A := \text{mean}|\Delta P(t)/\dot{\Omega}_s| \approx 2H_v$ . Due to a constant initial ROCOF for 1s during the reference frequency event (see  $\dot{\Omega}_s$  in Fig. 10), the considered time window for quantifying inertia provision for the initial ROCOF would be  $0.5s \leq t \leq 1s$ . For simplicity, this paper quantifies inertia provision by the control parameter  $H_v$  (see also Fig. 5), which results in a (slight) overestimation of the inertia provision compared to VDE (2024b).

overshoots. The VSM output power equals the WT electrical power change. The VSM braking power equals the electrical power of an equivalent real SM with external damping, i. e. the SM braking energy is fully converted into electrical energy according to the law of conservation of energy. The recent draft VDE (2024b) for certification of grid-forming IBRs quantifies inertia provision by the mean power change over a time window starting 0.5s after a ROCOF change and ending at the beginning of the next ROCOF change during the reference frequency event, i. e.  $T_A := \text{mean} |\Delta P(t)/\dot{\Omega}_s| \approx 2H_v$ . For example, due to a constant initial ROCOF for 1s during the reference frequency event starting at  $t = 0$ , the first time window for quantifying inertia provision is  $0.5\text{s} < t < 1\text{s}$ . For simplicity, this paper quantifies inertia provision by the control parameter  $H_v$  (see also Fig. 5), which results in a (slight) overestimation of the inertia provision compared to VDE (2024b).

This paper assumes an ideal inertial power response, i. e. the VSM power  $P_v$  is added to the original machine power setpoint  $\bar{P}_{\text{set}}^*$  in Fig. 2. Moreover, the final electromagnetic torque equals the electromagnetic torque reference, i. e.  $M_e = M_{e,\text{ref}}$  in Fig. 2, neglecting low-level current controls with closed-loop time constants that are significantly smaller than the ones of high-level WT or VSM control. Clearly, this is a simplified representation of the actual VSM control, which adjusts the voltage or current phase angle based on the VSM angle  $\phi_v$  to achieve the grid-forming capability (ESIG, 2022; Thommessen and Hackl, 2024). Although the implementation details are beyond the scope of this paper, the simplified representation should take into account the general differences between existing VSM control strategies, as discussed in Appendices ??C – ??E.

### 3.3.4 MPPT compensation

For a negative ROCOF, the WT output power increases for inertia provision, which decelerates the WT, i. e.  $\Omega$  decreases. The MPPT would counteract the deceleration or the desired inertial response by reducing  $P_{\text{mppt}}$  for decreasing when  $\Omega$  according to Eq. (6) decreases. To avoid this, the so-called MPPT compensation manipulates the MPPT input  $\Omega$  (Duckwitz, 2019; Thommessen and Hackl, 2024). This paper simplifies the MPPT compensation proposed in Thommessen and Hackl (2024, Sect. III.B.2). The speed change due to inertia provision is estimated by replacing the numerator of the one-mass model in Eq. (5) by the inertial torque change  $P_v/\Omega$ . Thus, the manipulated MPPT input, equal to the theoretical WT rotor speed for zero inertia provision, is given by

$$\Omega_{H0} := \Omega + \left\{ \begin{array}{ll} \min \left\{ a, \frac{1}{2H_R} \int_{t_0}^t \frac{P_v}{\Omega} dt \right\}, & \text{if } |\dot{\Omega}_v| > \epsilon \text{ (active MPPT compensation),} \\ 0, & \text{if } |\dot{\Omega}_v| \leq \epsilon \text{ (inactive MPPT compensation),} \end{array} \right\}, \quad a := \max \{1 - \Omega, 0\}, \quad (13)$$

with the threshold  $\epsilon$  used for detecting active inertia provision based on the VSM acceleration  $\dot{\Omega}_v \approx \dot{\Omega}_s$ . The integral in Eq. (13) is reset to zero for inactive MPPT compensation. The actual implementation of Eq. (13) includes an additional rate limiter, which ensures  $|\dot{\Omega}_{H0}| \leq \dot{\Omega}_{H0,\text{max}}$  with maximum acceleration  $\dot{\Omega}_{H0,\text{max}}$  for a smooth transition between active and inactive MPPT compensation (Thommessen and Hackl, 2024, Sect. III.B.2).

Assuming active MPPT compensation, a prolonged MPP deviation during a long time period with a small negative ROCOF would lead to excessive WT rotor deceleration. Thus, the threshold  $\epsilon$  in Eq. (13) should not be chosen too small. This also implies less inertia provision for small negative ROCOFs  $|\dot{\Omega}_s| \approx |\dot{\Omega}_v| \leq \epsilon$  than expected by  $H_v$ , due to inactive MPPT compensation the MPPT compensation being inactive. Also, for  $\epsilon < |\dot{\Omega}_s| < \dot{\Omega}_{s,\text{max}}$  with normalized worst-case ROCOF magnitude

$\dot{\Omega}_{s,\max}$ , there may be cases ~~where~~-when the inertia provision is (slightly) smaller than expected by  $H_v$ , if output power saturation is required to protect the rotor speed due to prolonged MPP deviations. However, the proposed approach ensures unsaturated or full inertia provision when reaching the worst-case or reference ROCOF  $|\dot{\Omega}_s| = \dot{\Omega}_{s,\max}$ .

### 3.3.5 Active power droop control

In addition to inertia provision, which supports grid frequency by injecting inertial VSM power  $P_v$  proportional to the ROCOF, active power droop control supports grid frequency by injecting (saturated) droop power  $\bar{P}_d$  proportional to the frequency deviation  $\Delta\Omega_s := 1 - \Omega_s$ . Thus, the final power reference in Fig. 2 is given by  $P_{\text{ref}} = \bar{P}_{\text{set}}^* + P_v + \bar{P}_d$ . For WTs, droop control is inactive during normal operation within a tolerance band of  $|\Delta\Omega_s| \leq 0.4\%$  (VDE, 2024a), i. e. the (unsaturated) droop power is  $P_d = \bar{P}_d = 0$  in Fig. 2. During a critical system state outside of the tolerance band, the WTs have to support grid frequency by a proportional power adaptation when possible. This means that  $\bar{P}_d = P_d > 0$  is only required if wind power reserves are available due to previous derating (VDE, 2024a). For ~~our~~-the present control strategy, this implies that the inertia provision based on additional kinetic energy extraction is prioritized over droop control; see Appendix ~~??~~-F for details.

Ignoring the two max-blocks in Fig. 2, the maximum droop power  $P_{d,\max}$  is given by the total currently available power  $\min(\bar{P}^*, P_{\text{mptt}})$  minus the sum of the power setpoint  $\bar{P}_{\text{set}}^*$  and the VSM power  $P_v$ . The saturation  $\bar{P}_d := \min(P_d, P_{d,\max})$  prevents excessive WT overloading since, without it, the droop power  $P_d$  would add to the inertial power even if the output or reference power  $P_{\text{ref}}$  already exceeds the available one. In other words, the WT control prioritizes inertia provision over droop control. Similarly, for real (grid-connected) SMs, the droop control or speed governor response time is significantly slower than the SM inertial response, i. e. only the SM inertial power limits the initial ROCOF.

The additional saturations by the two max-blocks in Fig. 2 ensure that the droop control power  $P_d$  does not counteract the VSM power  $P_v$  for inertia provision. Without the upper max-block,  $P_d = P_{d,\max} < 0$  could counteract  $P_v > 0$  for a high negative initial ROCOF; without the lower max-block,  $P_d = P_{d,\max} > 0$  could counteract  $P_v < 0$  more than expected for a subsequent positive ROCOF during frequency recovery. The presented control is a simplified version of the actual control with dynamic droop saturation of Thommessen and Hackl (2024, Sect. III.F).

### 3.4 Grid-forming capability of WTs

This section introduces a general approach used for evaluating the grid-forming capability of WTs in terms of maximum inertia provision. First, we present the proposed optimization problem to evaluate the maximum deliverable inertia. Then, we develop two solutions of the ~~optimization~~-problem. The first produces a simplified result derived from the formulations in the existing literature. This is followed by a second complete numerical solution, which utilizes a dynamic model of the WT inertial response within the optimization.

### 3.4.1 Optimization problem for maximum inertia provision

The maximum feasible VSM inertia constant  $H_{v,\max}$  is obtained by solving the optimization problem

$$420 \quad \forall t \geq t_0 : H_{v,\max} := \arg \max_{H_v} \{H_v\}, \text{ s. t. } \left\{ \begin{array}{l} \Omega(t) \geq \Omega_{\min} \\ M_e(t) \leq M_{e,\max} \\ \dot{M}_e(t) \leq \dot{M}_{e,\max} \\ P_e(t) \leq P_{e,\max} \end{array} \right\}. \quad (14)$$

Here, the WT rotor speed  $\Omega$  is expressed in per unit (p.u.) of rated WT rotor speed  $\omega_R$  (at the low speed shaft in ), the WT electromagnetic torque  $M_e$  is in p.u. of rated torque  $m_R$  (at the low speed shaft in ), the WT electromagnetic torque electromagnetic torque rate  $\dot{M}_e := \frac{d}{dt} M_e$  is in  $s^{-1}$ , the WT electrical power  $P_e$  is in p.u. of rated power  $p_R = \omega_R m_R$  (in W), and  $\Omega_{\min}$ ,  $M_{e,\max}$ ,  $\dot{M}_{e,\max}$ ,  $P_{e,\max}$  denote the corresponding admissible operating limits. Note that, although the objective   
 425 function and optimization argument in Eq. (14) are the same, solving Eq. (14) the solution of this problem is not trivial due to the presence of nonlinear optimization constraints. Depending on the grid codes and WT design, Eq. (14) may include additional constraints additional constraints may have to be considered, e. g. for recovery power limits (see Appendix ??) or for stall limits, which are to account for aerodynamic stall limits (here implicitly taken into account here by  $\Omega_{\min}$  in Eq. (14). 14). Appendix G discusses the case of recovery power limits, with reference to the examples shown later in Sect. 4.

### 430 3.4.2 Simplified solution

Lee et al. (2016) evaluate the capability of grid-following WTs to emulate inertia by considering WT rotor speed or available kinetic energy reserve. Here, unlike Lee et al. (2016), we derive we depart from that approach by deriving a simplified solution of Eq. (14) for grid-forming WTs that takes all operating constraints into account, and not only the WT rotor speed limits.

Neglecting any changes of aerodynamic conditions during the inertial response, i. e. assuming constant wind speed  $v_w$ , blade   
 435 pitch angle  $\beta$  values for wind speed, pitch angle and tip speed ratio  $\lambda := \Omega \omega_R r / v_w$  with WT radius  $r$ , it follows that the WT aerodynamic or mechanical power  $P_m$  (in p.u. of  $p_R$ ) mechanical power is constant, i. e.

$$\forall t_0 \leq t \leq t_0 + \Delta t : \left\{ \begin{array}{l} v_w(t) = v_w(t_0) =: v_{w,0} \\ \beta(t) = \beta(t_0) =: \beta_0 \\ \lambda(t) = \lambda(t_0) =: \lambda_0 \end{array} \right\} \Rightarrow P_m(t) = P_m(t_0) =: P_{m,0}. \quad (15)$$

For the simplified solution, we assume that the ROCOF  $\dot{f}_s$  is constant and equal to the worst-case initial ROCOF, until reaching the minimum frequency nadir  $f_{s,\min}$  at time  $t = t_{s,\min}$ , i. e. the considered time duration is given by

$$440 \quad \Delta t := t_{s,\min} - t_0 = \frac{f_{s,\min} - f_{s,R}}{\dot{f}_s}. \quad (16)$$

Assuming that the initial electrical power equals the mechanical power in Eq. (15), i. e.  $P_{e,0} = P_{m,0}$ , and approximating the electrical power change during  $\Delta t$  by an ideal power pulse  $\Delta P_e$  according to the simplified inertial response in Eq. (1a), the

electrical power constraint in Eq. (14) simplifies to

$$P_e := P_{e,0} + \Delta P_e := P_{m,0} + 2H_v \dot{\Omega}_{s,\max} \leq P_{e,\max}, \quad (17)$$

445 where ~~the normalized worst-case ROCOF magnitude is~~  $\dot{\Omega}_{s,\max} := |\dot{f}_{s,0}|/f_{s,R} > 0$ . It follows that additional output power  $\Delta P_e := 2H_v \dot{\Omega}_{s,\max}$  is extracted from the WT kinetic energy reserve ~~, and according to~~

$$\int_{t_0}^{t_{s,\min}} (P_m - P_e) dt = E_{\text{kin}}(t_{s,\min}) - E_{\text{kin}}(t_0),$$

~~from which we get~~  $\Delta P_e \Delta t = H_R (\Omega_0^2 - \Omega(t_{s,\min})^2)$ . Finally, the minimum rotor speed constraint in Eq. (14)

~~simplifies to~~ 
$$\int_{t_0}^{t_{s,\min}} (P_m - P_e) dt = E_{\text{kin}}(t_{s,\min}) - E_{\text{kin}}(t_0),$$

450 ~~from which we get~~ 
$$\Delta P_e \Delta t = H (\Omega_0^2 - \Omega(t_{s,\min})^2),$$

~~and finally~~ 
$$\Omega(t_{s,\min}) = \sqrt{\Omega_0^2 - 2 \frac{H_v}{H} (1 - \Omega_{s,\min})} \geq \underline{\Omega}_{\min}, \quad (18)$$

~~simplifies to~~

$$\Omega(t_{s,\min}) = \sqrt{\Omega_0^2 - 2 \frac{H_v}{H_R} (1 - \Omega_{s,\min})} \geq \underline{\Omega}_{\min}, \quad (19)$$

where  $\Omega_0 := \Omega(t_0)$  ~~and the physical total WT drivetrain inertia constant is~~  $H$ , ~~and~~  $H_R$  ~~includes the total inertia of the WT~~  
 455 ~~drivetrain~~. Note that Eq. (19) depends on the normalized frequency nadir  $\Omega_{s,\min} := f_{s,\min}/f_{s,R}$  and not explicitly on the ROCOF, which justifies the aforementioned assumption of a constant ROCOF in Eq. (16). Based on Eq. (17) and Eq. (19), the torque constraint in Eq. (14) simplifies to

$$\max M_e = \frac{P_e}{\Omega(t_{s,\min})} \leq M_{e,\max}. \quad (20)$$

Finally, with Eqs. (17–20) and the simplified torque rate constraint derived in the appendix Eq. (H5), a nonlinear optimization  
 460 algorithm (MATLAB, 2025b) solves Eq. (14) for given initial values  $\Omega_0$  and  $P_{m,0}$ .

### 3.4.3 Complete numerical solution

The optimization problem expressed by Eq. (14) can be solved in a more general way, where dynamic simulations of the WT inertial response to a worst-case or reference frequency event replace the aforementioned simplified expressions. More precisely, an optimization algorithm iterates the simulations with varying  $H_v$  to find the maximum inertia constant  $H_{v,\max}$   
 465 that does not violate any operating limits (see Fig. 1). Clearly, this approach is generic due to its applicability to different WT models and their controllers. Moreover, this approach allows for more accurate solutions. For instance, derating strategies can

provide additional wind power reserves (Kanev and van de Hoek, 2017; Meseguer Urban et al., 2019; Thommessen and Hackl, 2024), which are only taken into account by the complete numerical solution but not by the simplified one. For the iterative simulations during optimization, we rely on appropriate WT modeling, with steady-state ~~initialization~~ initializations derived in  
 470 Appendix ~~??~~1.

Inertia provision requires power headroom. Accordingly, all saturations or manipulations of the WT power reference  $P_{\text{ref}}$  for protection are not just removed ~~in from~~ the WT control model, but are converted into corresponding inequality constraints, i. e.

$$\forall t \geq t_0 : \mathbf{c} := \begin{pmatrix} c_1 \\ c_2 \\ c_3 \\ c_4 \end{pmatrix} := \begin{pmatrix} \Omega_{\min} - \min \Omega(t), \\ \max M_e(t) - M_{e,\max}, \\ \max \dot{M}_e(t) - \dot{M}_{e,\max}, \\ \max P_e(t) - P_{e,\max} \end{pmatrix}, \quad \text{where } \forall i \in \{1, 2, 3, 4\} : c_i \geq 0. \quad (21)$$

475 Based on Eq. (21), a nonlinear optimization algorithm (MATLAB, 2025b) solves the optimization problem in Eq. (14). The  $i$ -th constraint is considered active if  $c_i = 0$ , or inactive if  ~~$c_i > 0$~~   $c_i < 0$ .

## 4 Results

This section presents results regarding different aspects of the proposed WF inertia forecasting approach. First, Sect. 4.1 discusses the WT steady states for different MRS-based deratings (refer to Sect. 3.3.2). Then, Sect. 4.2 demonstrates the  
 480 simulated WT inertial response to a reference frequency event defined by grid codes, and discusses the mapping of WT operating points to the provision of deliverable inertia. Finally, the overall performance of the proposed WF inertia forecasting is evaluated and compared to the existing approaches in Sect. 4.3.

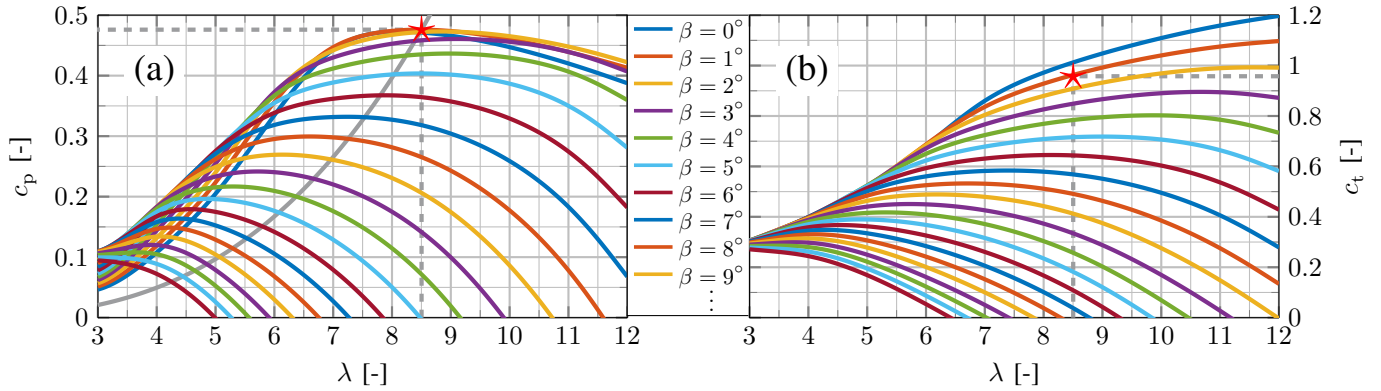
**Power coefficient  $c_p$  (a) and thrust coefficient  $c_t$  (b), as functions of tip speed ratio  $\lambda$  and blade pitch angle  $\beta$ . The MPP is indicated with the symbol  $\star$ . —  $(\lambda^*, \beta^*) = (8.5, 1.1^\circ)$ , —  $c_p^* = c_p(\lambda^*, \beta^*) = 0.48$ , —  $c_t^* = c_t(\lambda^*, \beta^*) = 0.96$ .**

485 **Wind farm layout and wake interactions for ambient wind speed  $v_w = 11 \text{ m s}^{-1}$  and wind direction  $\Gamma_w = 270^\circ$ .**

The WF considered here consists of twelve IEA 3.4 MW reference WTs from Bortolotti et al. (2019) ~~with operating limits~~. The aerodynamic characteristics of the rotor are given in Fig. 6 in terms of the power coefficient  $c_p$  and thrust coefficient  $c_t$ , plotted as functions of tip speed ratio  $\lambda$  and blade pitch angle  $\beta$ .

The operating limits are set as follows:  $\Omega_{\min} = 30.81\%$ ,  $P_{e,\max} = 105\%$ ,  $M_{e,\max} = 106\%$ ,  $\dot{M}_{e,\max} = 150\% \text{ s}^{-1}$ . The  
 490 power limit  $P_{e,\max}$  is chosen based on the inverter design (Höhn et al., 2024), whereas the other limits are chosen based on the aeroelastic design (Bortolotti et al., 2019).

The pitch controller was modified not to include a tip speed constraint below rated wind speed. Thus, at the rated wind speed  $v_{w,R} = 9.8 \text{ m s}^{-1}$ , the WT operates at its MPP with optimal tip speed ratio  $\lambda^* = 8.5$ . Consequently, the rated tip speed  $\omega_{Rr} = \lambda^* v_{w,R} = 83.3 \text{ m s}^{-1}$  (slightly) exceeds the tip speed limit of  $80 \text{ m s}^{-1}$  assumed in Bortolotti et al. (2019). The resulting  
 495 rated WT rotor speed  $\omega_R$  is ca. 4.2% larger than the rated value in Bortolotti et al. (2019), but still ca. 5.3% smaller than the maximum assumed rotor speed limit. The higher speed rating increases not only the rated power but also the rated kinetic



**Figure 6.** Power coefficient  $c_p$ , (a) and thrust coefficient  $c_t$ , (b), as functions of tip speed ratio  $\lambda$  and blade pitch angle  $\beta$ . The MPP is indicated with the symbol  $\star$ .  $(\lambda^*, \beta^*) = (8.5, 1.1^\circ)$ ,  $c_p^* = c_p(\lambda^*, \beta^*) = 0.48$ ,  $c_t^* = c_t(\lambda^*, \beta^*) = 0.96$ .

energy reserve for inertia provision compared with Bortolotti et al. (2019). Neglecting for simplicity any conversion losses from mechanical to electrical power, the WT physical inertia constant at rated speed is

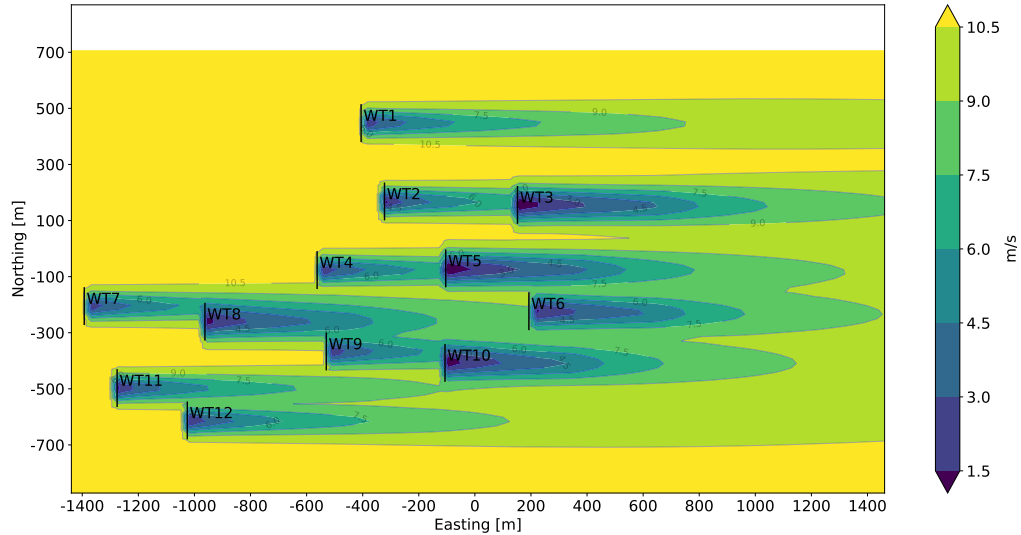
$$H_R = \frac{\frac{1}{2} \Theta \omega_R^2}{p_R} = 3.26 \text{ s.} \quad (22)$$

500 The WTs are arranged in an irregular WF layout on semi-complex terrain characterized by gently rolling hills. Figure 7 shows the WF layout as well as the resulting wake interactions among the WTs in exemplary wind conditions.

Historical data consisting of 2 years of site-specific weather condition measurements are used to train the data-driven weather forecast model. A deterministic model and a probabilistic model predict the weather conditions with a 15-minute resolution for the hour ahead. The deterministic model outputs the expected wind conditions for WF inertia forecasting, whereas, the  
505 probabilistic model additionally considers the wind condition uncertainties, enabling uncertainty quantification of the predicted WF inertia.

#### 4.1 WT steady states

The WT steady states depend on the WT operating point  $(v_w, P_{\text{set}})$ , defined by wind speed  $v_w$  and power setpoint  $P_{\text{set}}$ . Although the WT steady states generally depend on the ~~two-dimensions-pairs~~  $(v_w, P_{\text{set}})$ , they are calculated by solving  
510 one-dimensional optimization (sub)problems. This is obtained through a case analysis of active operating constraints, ~~see Appendix-??~~ which is described in Appendix I. This enables the fast initialization of the WT dynamic model without running time-consuming simulations until reaching steady state. Figure 8 illustrates the WT steady-state conditions as a function of  $v_w$ , with  $P_{\text{set}}$  ranging from the minimum considered value of 90% (dark blue line) in increments of 1% up to the maximum value of 100% (dark red line). Note that  $P_{\text{set}} = 1$  corresponds to MPPT, and  $P_{\text{set}} < 1$  corresponds to MRS-based derating. In Fig. 8  
515 (and in all following figures), all normalized quantities (indicated by [%]) are in per unit of rated values, e. g.  $F_{t,\text{pu}} := F_t/F_{t,R}$

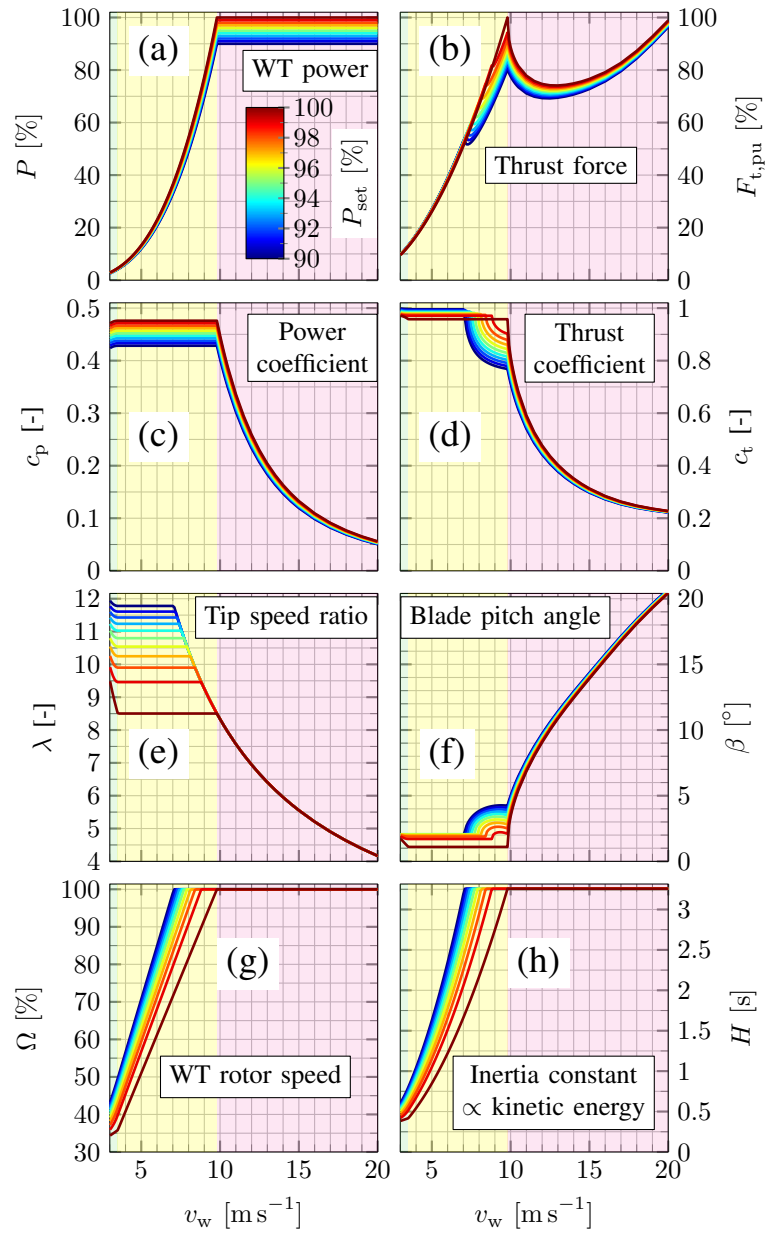


**Figure 7.** Wind farm layout and wake interactions for ambient wind speed  $v_w = 11 \text{ m s}^{-1}$  and wind direction  $\Gamma_w = 270^\circ$ .

with thrust force  $F_t = F_{t,R}$  at the rated WT operating point ( $v_w = v_{w,R} = 9.8 \text{ m s}^{-1}$ ,  $P_{\text{set}} = 1$ ). The only exception is the power setpoint  $P_{\text{set}}$  defined in per unit of available power  $\bar{P}^*$ , see Eq. (9).

In Fig. 8, a higher derating or a lower  $P_{\text{set}}$  increases the WT rotor speed  $\Omega$  at low wind speeds, e. g. at  $v_w = 5 \text{ m s}^{-1}$ , such that the kinetic energy or physical inertia constant  $H$  increases proportional-proportionally to  $\Omega^2$ . For  $P_{\text{set}} = 1$  (MPPT), the blade pitch angle equals its optimal value for below-rated wind speeds in region II, i. e.  $\beta = \beta^* = 1.1^\circ$  (see also Fig. 6); on the other hand, for above-rated wind speeds in region III,  $\beta$  increases to limit the WT rotor speed to  $\Omega = 1$ . In addition to  $\Omega$ , the pitch control requires  $v_w$  as input (see Fig. 2) to adjust the lower pitch angle limit as a function of the tip speed ratio  $\lambda = \Omega \omega_R r / v_w$ , i. e.  $\beta_{\text{ref}} \geq \beta_{\text{min}}(\lambda) := \arg \max c_p(\lambda, \beta)$ . Considering the plots in the third row of Fig. 8, this  $\beta_{\text{min}}$ -adjustment is only relevant for  $\lambda > \lambda^* = 8.5$  due to constant  $\beta_{\text{min}} = \beta^*$  elsewhere. More precisely, for  $P_{\text{set}} = 1$  (MPPT), the  $\beta_{\text{min}}$ -adjustment is only relevant in the small transition region I-II near  $v_{w,\text{cut-in}} = 3.02 \text{ m s}^{-1}$ ; however, for  $P_{\text{set}} < 1$  (derating), the  $\beta_{\text{min}}$ -adjustment is also relevant in region II, as the increased tip speed ratio  $\lambda > \lambda^*$  leads to a higher blade pitch angle  $\beta = \beta_{\text{min}}(\lambda) \geq \beta^*$ .

In Fig. 8, after reaching rated rotor speed  $\Omega = 1$ , the tip speed ratio decreases with increasing wind speed, i. e.  $\lambda \propto v_w^{-1}$ , and the blade pitch angle  $\beta$  increases to limit the rotor speed to  $\Omega = 1$ . Both decreasing  $\lambda$  and increasing  $\beta$  reduce the thrust coefficient  $c_t$  (at least near the optimal operating point, see also Fig. 6). Note that, for MRS-based derating ( $P_{\text{set}} < 1$ ), the rotor speed reaches  $\Omega = 1$  at below-rated wind speeds  $v_w < v_{w,R}$ . For  $\Omega < 1$  and constant  $P_{\text{set}}$ , the thrust coefficient  $c_t$  is constant due to constant  $\lambda$  and  $\beta$ . For  $\Omega < 1$  and varying  $P_{\text{set}}$ , e. g. at  $v_w = 5 \text{ m s}^{-1}$ , higher derating increases  $\lambda$  but only slightly increases  $\beta = \beta_{\text{min}}(\lambda)$  such that  $c_t$  (slightly) increases. This (slightly) increases the thrust force  $F_{t,\text{pu}}$ , although the



**Figure 8.** WT steady-state conditions as functions of wind speed  $v_w$  for MRS-based derating with power setpoint  $P_{\text{set}}$  (see steady state calculation in Appendix [??I](#)).

changes are negligible. In contrast, for initial operation with maximum speed  $\Omega = 1$  at higher or above-rated wind speeds, higher derating significantly reduces the thrust force due to increasing  $\beta$  but constant  $\lambda$ .

535 To sum-up summarize, the MRS-based derating significantly decreases the thrust force for  $\Omega = 1$ , i. e. if no further rotor acceleration is feasible. This is in line with the observed reduction of damage equivalent loads during derating, as reported by Aho et al. (2014). Otherwise, for  $\Omega < 1$ , i. e. especially at low wind speeds or for minor derating in region II, the MRS-based derating accelerates the rotor, but the resulting increase in thrust force is negligible.

## 4.2 WT inertia provision for the reference frequency event

540 This section evaluates WT grid-forming capabilities in terms of maximum inertia provision as a function of WT operating point  $(v_w, P_{set})$ . At first, Sect. 4.2.1 introduces the considered reference frequency event defined by the German grid code and derives a worst-case test scenario for WT inertia provision. Then, Sect. 4.2.2 discusses the resulting dynamic WT simulations for optimized inertia provision, i. e. for  $H_v = H_{v,max}$ , with and without MRS-based derating. Finally, Sect. 4.2.3 discusses the mapping of WT operating points to the maximum feasible inertia constant over a wide operating range.

### 545 4.2.1 Grid codes

Although the grid codes can vary between-among countries and system operators, the core requirements for inertia provision are similar (Ghimire et al., 2024). This paper focuses on the German code for grid-forming control and their-its requirements for inertia provision (VDE, 2024a, b). This grid code defines two reference frequency events with maximum initial ROCOF magnitudes of  $|\dot{f}_s| = 2 \text{ Hz s}^{-1}$ : one for negative inertia provision due to a high positive initial ROCOF, and another for positive  
550 inertia provision due to a high negative initial ROCOF. The latter is considered as the worst-case reference frequency event for WFs, since the output power has to increase for inertia provision, which decelerates the WTs. Emulating this reference frequency event and evaluating the WF power response is required to verify inertia provision.

The considered grid code (VDE, 2024b) defines various tests based on the reference frequency events to verify inertia provision, including operation in (i) fictive or simulated island mode with changing power imbalance  $\Delta P$  due to varying  
555 electrical load, (ii) grid-emulator-connected mode with changing ROCOF  $\dot{\Omega}_s$ , and (iii) real grid-connected mode with changing controller-internal ROCOF, corresponding to the VSM acceleration  $\dot{\Omega}_v$  (see also Fig. 4). In the latter case (iii), the frequency signal defined by the reference frequency events is added as a disturbance to the controller-internal frequency, corresponding to the VSM speed  $\Omega_v$ . It should be noted that some tests consider deactivated droop control. However, the verification principle is always the same, see also Schöll et al. (2024). In all tests, a ROCOF  $\dot{\Omega}$  changes changes  $\dot{\Omega}$  and the inertial power response  
560  $\Delta P$  is measured, or vice versa. The actual inertia provision is quantified by the measured inertia constant  $H_{meas} = |\Delta P / (2\dot{\Omega})|$  at quasi-steady state, see-also-as obtained from Eq. (1a). Note that  $\Delta P$  corresponds to the measured power change  $\Delta P_{meas}$  only if droop control is deactivated. Otherwise, for a correct calculation of  $H_{meas}$ , the droop power change  $\Delta \bar{P}_d$  (depending on the frequency deviation) adds-should be added to the inertial power (depending on the ROCOF) during the frequency event, i. e.  $\Delta P = \Delta P_{meas} - \Delta \bar{P}_d$  for correct  $H_{meas}$ -calculation. Clearly,  $H_{meas}$  must match the expected VSM inertia constant, i. e.  
565  $H_{meas} \approx H_v$ .

For the considered VSM control,  $H_{\text{meas}} \approx H_v$  holds if no power saturation is active (see also Fig. 5). In other words, it is assumed that the actual VSM control implementation would pass all tests (VDE, 2024b) with  $H_{\text{meas}} \approx H_v$  for an arbitrarily chosen  $H_v$  if no power saturation exists. It follows that  $H_{\text{meas}} \approx H_v$  holds if no operating limits are violated in the simulations of the WT model (see Fig. 2). Otherwise, in reality, the protection methods would saturate the output power as the desired inertia provision is not feasible, i. e.  $H_{\text{meas}} < H_v$  due to  $H_v > H_{v,\text{max}}$ . Thus, assuming proper VSM control allows this study to focus on the relevance of interactions with WT control, WT characteristics, and operating limits.

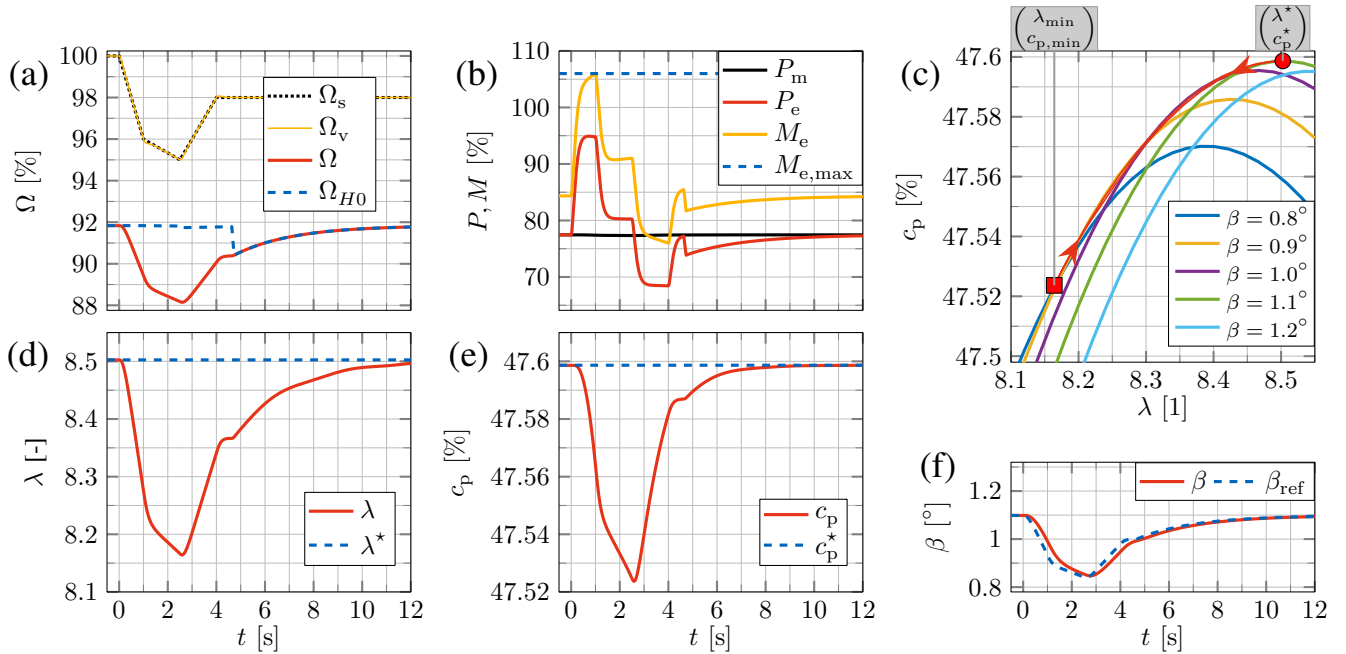
Running and passing all tests defined in the grid code (VDE, 2024b) verifies proper grid-forming control implementation and inertia provision for a given  $H_v$  at ~~some given operating points~~ a given operating point. However, this approach is not suitable for evaluating the maximum feasible inertia provision of WFs over a wide operating range. Thus, this paper considers a single worst-case test scenario to simulate WTs for varying  $H_v$  and varying operating point  $(v_w, P_{\text{set}})$ ; ~~see the corresponding inputs in Fig. 2.~~ The reference frequency event for positive inertia provision with an initial ROCOF of  $\dot{f}_s = -2 \text{ Hz s}^{-1}$  (VDE, 2024b) defines the input  $\dot{\Omega}_s$  in Fig. 2. This can be interpreted as ideal ROCOF emulation at the point of common coupling. As in real grid-connected operation mode, droop control is activated. Finally, a WT survives the worst-case test scenario if no operating limit is violated, i. e. for  $H_v \leq H_{v,\text{max}}$  at a given operating point  $(v_w, P_{\text{set}})$  according to Eq. (14).

#### 580 4.2.2 Optimized time response

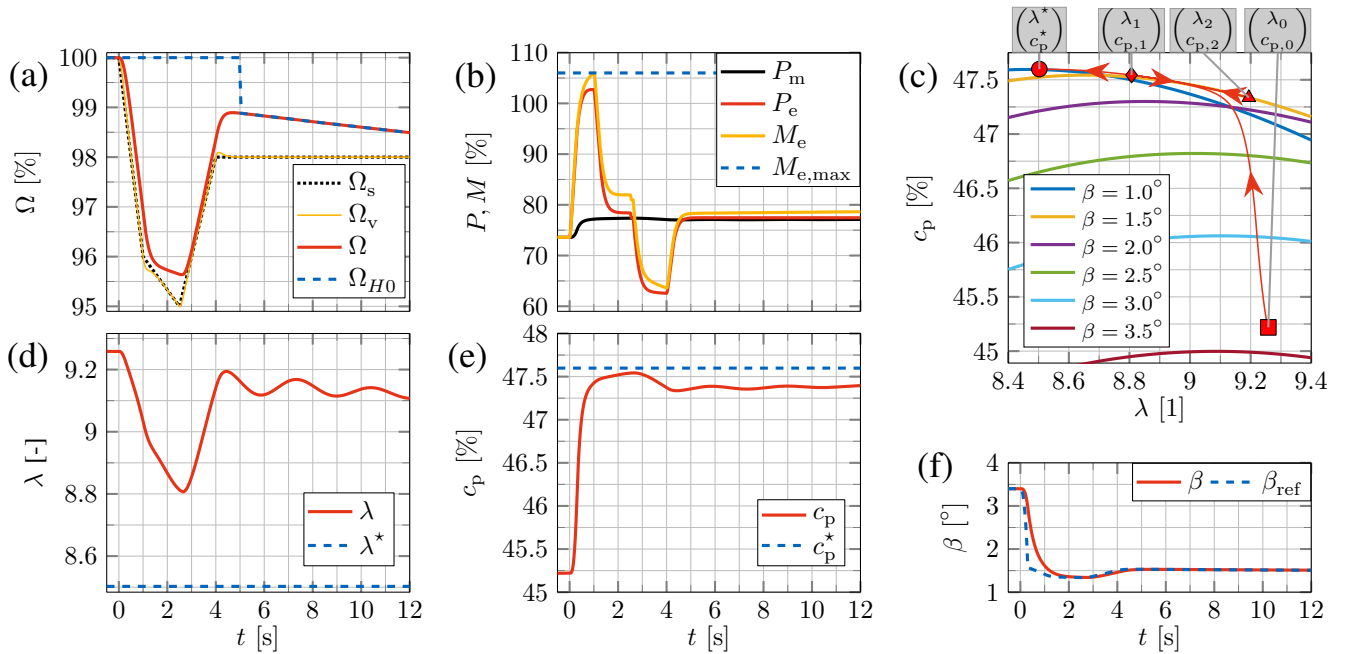
This section considers simulation results of the WT inertial response to the reference frequency event with optimized  $H_v = H_{v,\text{max}}$  at  $v_w = 9 \text{ m s}^{-1} < v_{w,\text{R}}$  for two different power setpoints (i)  $P_{\text{set}} = 100\%$  for MPPT in Fig. 9 and (ii)  $P_{\text{set}} = 95\%$  for MRS-based derating in Fig. 10. The grid frequency is identical in both cases (panel a), i. e.  $\Omega_s$  decreases with the initial ROCOF  $\dot{\Omega}_s = -4\% \text{ s}^{-1}$  for  $t_0 = 0 \text{ s} \leq t < 1 \text{ s}$  and further decreases with  $\dot{\Omega}_s = -2/3\% \text{ s}^{-1}$  afterwards, until reaching the nadir  $\Omega_{s,\text{min}} = 95\%$  at  $t_{s,\text{min}} = 2.5 \text{ s}$ . Then,  $\Omega_s$  increases with  $\dot{\Omega}_s = 2\% \text{ s}^{-1}$ , before staying constant at  $\Omega_s = 98\%$  for  $t \geq 4 \text{ s}$ . In addition, panel (a) shows the VSM speed  $\Omega_v$ , the WT speed  $\Omega$ , and the adjusted MPPT input  $\Omega_{H0}$ . Panel (b) shows the mechanical power  $P_m$ , the electrical power  $P_e$ , the electrical torque  $M_e$  and its limit  $M_{e,\text{max}}$ . Panel (d), (e), and (f) show the tip speed ratio  $\lambda$ , the power coefficient  $c_p$ , and the blade pitch angle  $\beta$ , respectively. Finally, panel (c) shows the resulting aerodynamic trajectory (red line).

In Fig. 9, the power coefficient starts at its maximum value due to MPPT, i. e.  $(\beta, \lambda) = (\beta^*, \lambda^*)$  which yields  $c_p = c_p^*$ . The VSM inertial response increases the electrical power  $P_e$  during the negative ROCOF, whereas the aerodynamic or mechanical power  $P_m$  remains almost constant. Thus, the WT rotor decelerates ~~such that and~~ the tip speed ratio  $\lambda$  decreases. After the ROCOF changes from negative to positive at  $t_{s,\text{min}} = 2.5 \text{ s}$ ,  $P_e$  rapidly decreases below  $P_m$  ~~such that and~~ the rotor accelerates. Accordingly, considering the trajectory  $(c_p, \lambda)$  in Fig. 9c, the WT leaves its MPP  $(\lambda^*, c_p^*)$  during the negative ROCOF, with significantly decreasing  $\lambda$  but almost constant  $c_p$ . Due to the grid synchronization delay of the VSM, the WT reaches its rotor speed nadir or  $(\lambda_{\text{min}}, c_{p,\text{min}})$  at  $t = 2.593 \text{ s}$ , i. e. shortly after the frequency nadir at  $t_{s,\text{min}} = 2.5 \text{ s}$ . Afterwards, with increasing  $\Omega$ , the trajectory converges to  $(\lambda^*, c_p^*)$  for  $t \rightarrow \infty$  ~~again~~.

At  $t \approx 4.65 \text{ s}$  in Fig. 9a, the MPPT compensation resets  $\Omega_{H0}$  to  $\Omega$  such that  $P_e$  in panel (b) rapidly decreases to re-accelerate the WT to its MPP, called WT rotor speed recovery. Clearly, decreasing the rate limit  $\dot{\Omega}_{H0,\text{max}}$  for the transition between



**Figure 9.** WT simulation results for MPPT at ( $v_w = 9 \text{ m s}^{-1}$ ,  $P_{\text{set}} = 100\%$ ) with optimized  $H_{v,\text{max}} = 2.26\text{s}$ .



**Figure 10.** WT simulation results for MRS-based derating at ( $v_w = 9 \text{ m s}^{-1}$ ,  $P_{\text{set}} = 95\%$ ) with optimized  $H_{v,\text{max}} = 3.80\text{s}$ .

600 active and inactive MPPT compensation in Eq. (13) leads to a smoother change of  $P_e$ , which is expected to cause less severe secondary frequency disturbances (Godin et al., 2019). However, this would slow down the WT rotor speed recovery. Finding a reasonable compromise is beyond the scope of this paper, but see Appendix ??-G for further discussion [on this point](#).

In Fig. 10, the initial steady-state power  $P_m(t_0) = P_e(t_0)$  is 5% below its initial MPP value in Fig. 9 due to  $P_{\text{set}} = 95\%$  or  $c_p(t_0) = 95\% \cdot c_p^*$  [with for](#)  $t_0 = 0$ . Note that the lower initial electromagnetic torque  $M_e(t_0)$  leads to (i) higher initial WT speed  $\Omega(t_0) = 1$  in Fig. 10 compared with  $\Omega(t_0) = 91.8\%$  in Fig. 9, and (ii) active pitch control, i. e.  $\beta(t_0) > \beta^*$  in Fig. 10. The trajectory  $(c_p, \lambda)$  in Fig. 10c starts at  $(\lambda_0, c_{p,0})$  with  $\lambda_0 > \lambda^*$  and  $c_{p,0} < c_p^*$ . The VSM inertial response increases the electromagnetic power  $P_e$  during the negative ROCOF [such, so](#) that the WT decelerates and  $\lambda$  decreases. At the same time, the pitch control decreases  $\beta$  due to  $\Omega < 1$ . The trajectory reaches the (local) WT rotor speed nadir or  $(\lambda_1, c_{p,1})$  at  $t \approx 2.62\text{s}$ . Afterwards, during the positive ROCOF,  $P_e$  rapidly decreases such that  $\lambda$  increases, reaching  $(\lambda_2, c_{p,2})$  at  $t \approx 4.44\text{s}$ . Finally, 610 during constant but below-rated frequency  $\Omega_s < 1$  for  $t \geq 4$ , inertia provision is inactive, but the active power droop control increases  $P_e$  to the maximum available MPPT power, which is greater than the initial value  $P_e(t_0)$ . Thus,  $(c_p, \lambda)$  converges to the MPP  $(\lambda^*, c_p^*)$  for  $t \rightarrow \infty$ . Note that the trajectory in Fig. 10 converges to the MPP from the right side of the MPP, whereas, for  $P_{\text{set}} = 100\%$  in Fig. 9, the trajectory converges to the MPP from the left side of the MPP.

After the ROCOF changes from negative to positive at  $t = 2.5\text{s}$  in Fig. 10d, the electrical power  $P_e \approx P_{\text{ref}} = \overline{P}_{\text{set}} + P_v + \overline{P}_d$  615 (panel b) decreases as the VSM inertial power becomes negative, i. e.  $P_v < 0$ . At the same time, the droop power increases as power reserves become available, i. e.  $\overline{P}_d = P_{d,\text{max}} > 0$  [\(see also Fig. 2\)](#). The droop power  $\overline{P}_d > 0$  counteracts the VSM inertial power  $P_v < 0$  during the grid frequency recovery. Thus, both droop control and inertia provision are active at the minimum  $P_e$  at  $t = 4\text{s}$  in Fig. 10b, whereas  $\overline{P}_d = P_{d,\text{max}} = 0$  and  $P_v > 0$  holds for the maximum  $P_e$  at  $t = 1\text{s}$ .

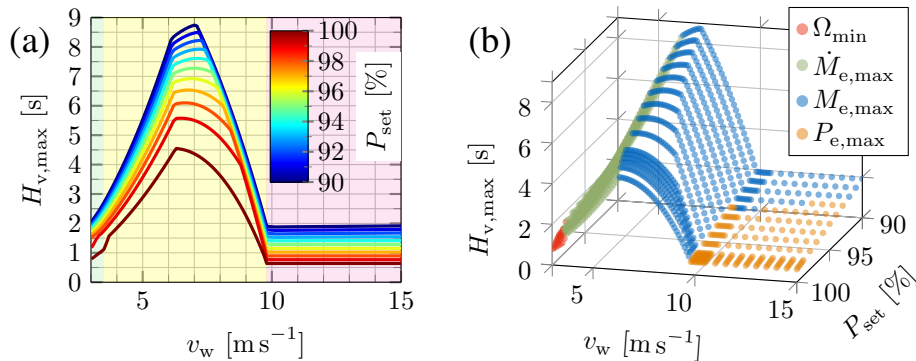
Considering Figs. 9 [and](#) 10 (panel b), the electromagnetic torque  $M_e$  reaches its limit  $M_{e,\text{max}}$  in both cases, i. e. the second 620 constraint of the optimization problem Eq. (14) is active. However, the initial torque  $M_e(t_0)$  for 5% power derating in Fig. 10b is more than 12% lower than  $M_e(t_0)$  for MPPT in Fig. 9b. The torque reduction is higher than the power derating due to an 11% higher initial rotor speed  $\Omega(t_0)$  in Fig. 10a than in Fig. 9a. Clearly, the MRS-based derating maximizes the torque headroom  $M_{e,\text{max}} - M_e(t_0)$  for inertia provision by maximizing the rotor speed.

In general, the VSM inertial response to the initial negative ROCOF increases the electrical power  $P_e$ , which decelerates 625 the rotor in both cases (Figs. 9 [and](#) 10). However, due to a higher initial [rotor speed value](#), the rotor speed does not fall below its MPP-value  $\Omega = 91.84\%$  for MRS-based derating in Fig. 10; [whereas on the other hand](#),  $\Omega$  decreases to 88.15% for MPPT in Fig. 9. Moreover, the WT rotor deceleration leads to constant (or only slightly decreasing) mechanical power  $P_m$  in Fig. 9 but increasing  $P_m$  in Fig. 10. Consequently, the MRS-based derating strategy provides both additional kinetic energy and additional wind energy reserves for inertia provision.

630 To summarize, the optimized values for the VSM inertia constant  $H_v = H_{v,\text{max}}$  are 2.26s and 3.80s for  $P_{\text{set}} = 100\%$  (MPPT) in Fig. 9 and for  $P_{\text{set}} = 95\%$  (MRS-based derating) in Fig. 10, respectively. For these two exemplary operating points, the MRS-based derating of 5% increases the inertia provision capability in terms of  $H_{v,\text{max}}$  by ca.  $3.80/2.26 - 1 \approx 68\%$  compared with MPPT.

### 4.2.3 Mapping of WT operating points

635 The proposed WF inertia forecasting approach uses LUTs to map the operating points of all WTs to the WF inertia (see also Fig. 4), assuming optimal tuning  $H_v = H_{v,\max}$  for each WT in the WF. For the proposed solution of the optimization problem in Eq. (14), Fig. 11 depicts the resulting LUTs of the form  $H_{v,\max} = f(v_w, P_{\text{set}})$  for all operating points  $(v_w, P_{\text{set}})$  within the range  $v_{w,\text{cut-in}} \leq v_w \leq 15 \text{ m s}^{-1}$  and  $90\% \leq P_{\text{set}} \leq 100\%$ . For enhanced visualization and clarity, the panels (a) and (b) show the same results from different perspectives/points of view. In the 2-D plot (panel a), the color coding for the power setpoint  $P_{\text{set}}$  is the same as in Fig. 8; whereas, in the 3-D plot (panel b), the datapoint colors indicate the active optimization constraints in Eq. (14).



**Figure 11.** Maximum inertia provision at different operating points  $(v_w, P_{\text{set}})$ , with active constraints highlighted in panel (b).

The LUTs were computed on a standard desktop computer (FUJITSU Workstation CELSIUS W580, Xeon E-2246G 6C 3.60GHz 12MB, 2 x 16GB DDR4-2666, NVIDIA Quadro P1000 4GB) using MATLAB's Parallel Computing Toolbox, Optimization Toolbox, and Simulink. Four CPU cores were used in parallel via a "parfor" loop (MATLAB, 2025c) to perform solver-based nonlinear optimizations (MATLAB, 2025b), including numerous WT simulations executed in Simulink's Rapid Accelerator Mode (MATLAB, 2025a). The computation time depended strongly on the simulation and solver settings, in particular the optimality and constraint tolerances. Without any attempt at optimizing the software for numerical efficiency, the wall-clock time was typically on the order of four hours for a complete LUT generation. Although execution time is a metric that quickly becomes obsolete and is influenced by many factors, the computational cost of the proposed method is acceptable for realistic applications, especially since the LUT generation is performed offline rather than during operation.

In Fig. 11, the rotor speed limit  $\Omega_{\text{min}}$  is only active for low wind speeds near  $v_{w,\text{cut-in}}$ . For most operating points with  $v_w < 6.3 \text{ m s}^{-1}$ , the torque rate limit  $\dot{M}_{e,\max}$  constraint is active. For  $P_{\text{set}} = 1$ , i. e. even without derating, the maximum inertia constant  $H_{v,\max} = 4.55 \text{ s}$  at  $v_w = 6.3 \text{ m s}^{-1}$  exceeds the rated WT physical inertia constant  $H_R = 3.26 \text{ s}$ . However, for  $6.3 \text{ m s}^{-1} < v_w \leq v_{w,R}$ , the torque limit  $M_{e,\max}$  reduces  $H_{v,\max}$  with increasing wind speed. For above-rated wind speeds  $v_w > v_{w,R}$  and  $P_{\text{set}} = 1$ , the virtual inertia constant  $H_{v,\max}$  is smaller than the WT physical inertia constant

$H = H_R$  due to the electrical power limit  $P_{e,\max}$ . The MRS-based derating increases  $H_{v,\max}$  over the complete wind speed range.<sup>1</sup>

Appendix J reports the  $H_{v,\max}$ -results obtained with the simplified solution derived in Sect. 3.4.2, and compares them with the complete numerical solution. Appendix G reports the results obtained with an additional (optional) constraint for WT rotor speed recovery.

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For each operating point  $(v_w, P_{\text{set}})$  and corresponding optimal  $H_v = H_{v,\max}$  tuning, the rotor speed nadir  $\min \Omega$  and the mechanical power extrema  $\min P_m$  or  $\max P_m$  are evaluated in the inertial response time interval defined in the appendix Eq. (K1). Figure 12a shows the normalized deviation of  $\min \Omega$  to the initial WT speed  $\Omega_0$ , i. e.  $(\min \Omega - \Omega_0)/\Omega_0$ , and Fig. 12b shows the normalized deviation of  $\min \Omega$  to the MPP speed  $\Omega_{\text{mpp}}$ , i. e.  $(\min \Omega - \Omega_{\text{mpp}})/\Omega_{\text{mpp}}$ , where  $\Omega_{\text{mpp}} := \Omega_0$  for  $P_{\text{set}} = 1$ .  
 665 Figures 13a and 13b show the normalized deviation of  $\min P_m$  and  $\max P_m$  to the initial WT mechanical power  $P_{m,0}$ , i. e.  $(\min P_m - P_{m,0})/P_{m,0}$  and  $(\max P_m - P_{m,0})/P_{m,0}$ , respectively.

665

The WT significantly decelerates during the inertial response at low wind speeds, although the rotor speed deviation does not exceed 20% of  $\Omega_0$  in Fig. 12a. Clearly, the lower speed limit  $\Omega_{\text{min}}$  is only relevant for a few operating points at low wind speeds and minor derating, i. e., for all other operating points the available kinetic energy reserve cannot be fully extracted for inertia provision due to active torque or power constraints. Decreasing  $P_{\text{set}}$  increases  $\Omega_0$  such that the rotor speed does not fall  
 670 below its MPP value if the derating is high enough, i. e. for  $\min \Omega - \Omega_{\text{mpp}} > 0$  in Fig. 12b. In this case, no rotor speed recovery is needed after the inertial response.

670

For MPPT, the rotor deceleration decreases the power coefficient  $c_p$  (see also Fig. 6). Thus, the aerodynamic power decreases during the inertial response, i. e.  $\min P_m < P_{m,0}$  for  $P_{\text{set}} = 1$  at below-rated wind speeds in Fig. 13a. However, minor  
 675 derating significantly increases  $\min P_m$ . For  $P_{\text{set}} \leq 95\%$ , the reduction of  $P_m$  is negligible due to  $\Omega_0 \approx \Omega_{\text{mpp}}$ ,  $\min \Omega \approx \Omega_{\text{mpp}}$ , or  $P_m$  even increases due to  $\min \Omega > \Omega_{\text{mpp}}$ , see as shown in Figs. 12a-b and 13a. In summary, the MRS-based derating leads to less severe rotor deceleration due to higher initial rotor speed and thus also higher aerodynamic power during the inertial response.

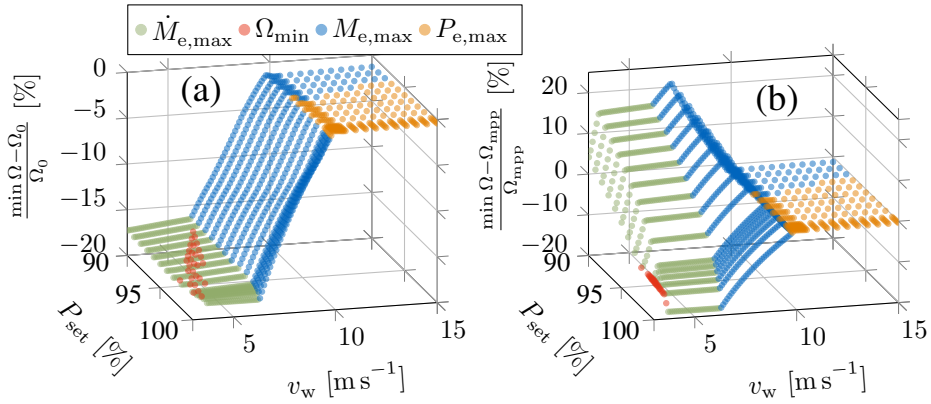
### 4.3 WF inertia monitoring and forecasting

680 Applying the proposed approach to a real WF ambient wind profile over five days, the following figures illustrate:

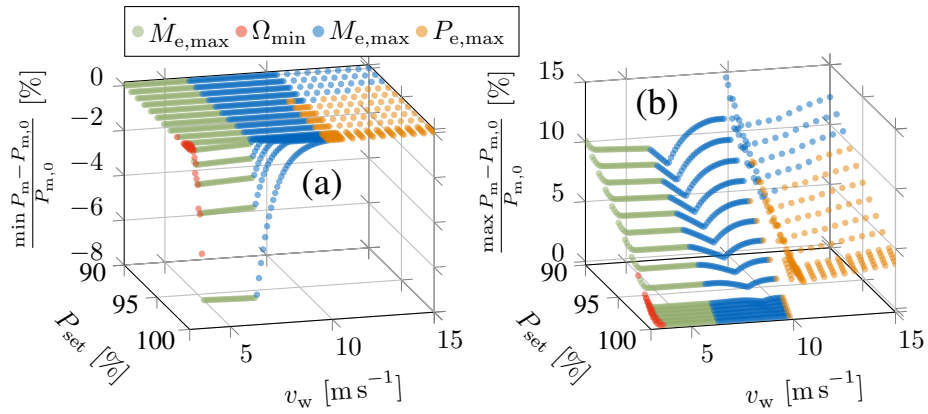
1. actual and forecasted WF ambient wind condition inputs for the wake model (Fig. 14);
2. WF inertia monitoring results, i. e. WF inertia calculations based on actual wind conditions (Figs. 15 and 16);
3. WF inertia forecasting results, i. e. WF inertia calculations based on predicted wind conditions (Fig. 17);
4. uncertainties due to WF ambient wind forecast errors, wake model errors, and WT inertia provision errors (Figs. 17 – 19).

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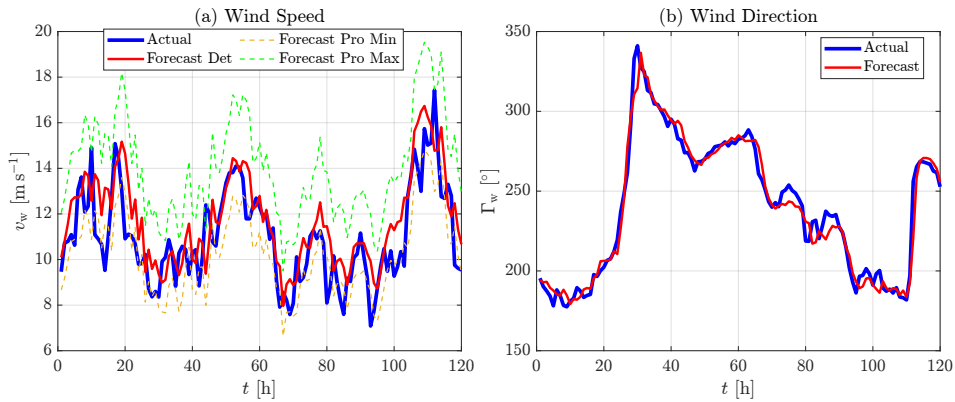
<sup>1</sup> See Appendix ?? for  $H_{v,\max}$ -results with an additional (optional) constraint for WT rotor speed recovery and their discussion. See Appendix ?? for the  $H_{v,\max}$ -results of the simplified solution derived in Sect. 3.4.2 and a comparison with the complete numerical solution.



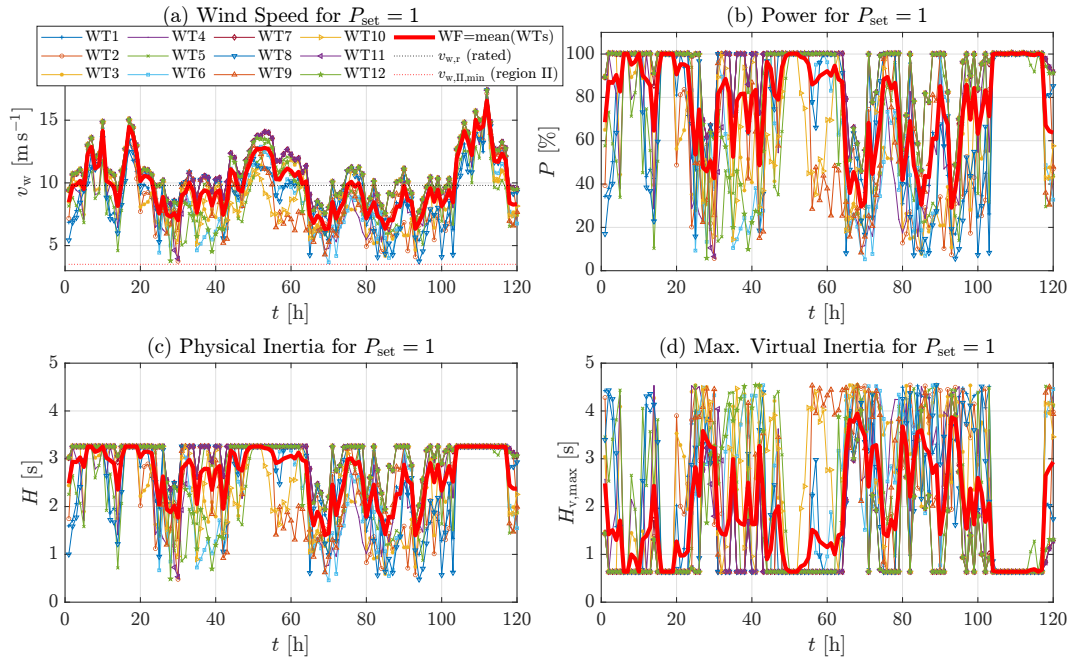
**Figure 12.** WT rotor speed nadir: deviation from the initial operating point (a) and the MPP (b).



**Figure 13.** WT mechanical power extrema: minimum (a) and maximum (b) deviation from the initial operating point.



**Figure 14.** Ambient wind conditions and its hour-ahead forecasts: wind speed (a) and wind direction (b) for actual measurements (Actual) as well as for deterministic (“Det”), probabilistic minimal (“Pro Min”), and probabilistic maximal (“Pro Max”) forecasts.



**Figure 15.** Simulation results for all 12 WTGs operating at  $P_{\text{set}} = 1$  (MPPT) based on the actual WF ambient wind data in Fig. 14.

685 In addition to the actual WF ambient wind speed and direction, Fig. 14a shows the different hour-ahead forecasts. The wind speed  $v_w$  values of the deterministic (“Det”) forecast lie between the values of the probabilistic minimal (“Pro Min”) forecast and the probabilistic maximal (“Pro Max”) forecast. The forecast for the wind direction  $\Gamma_w$  in Fig. 14b is deterministic. All forecast trends match the actual data. The following WF simulation results are based on actual and forecasted WF ambient wind data for inertia monitoring and forecasting, respectively.

690 In Fig. 15a, the maximum (local) wind speed (upper envelope curve) of all twelve WTGs (WT1–WT12) equals the actual WF ambient wind speed in Fig. 14. Clearly, wake effects reduce the wind speed for downstream WTGs, resulting in a lower mean wind speed in Fig. 15. In Fig. 15b, the normalized WT power at steady state is defined as  $P := P_e = P_m$ . Due to the equal power rating of all WTGs, the normalized WF power corresponds to the mean normalized WT power, and the WF (virtual) inertia constant corresponds to the mean WT (virtual) inertia constant. For  $P_{\text{set}} = 1$ , the WF operates at rated WF power  $P = 1$  if the local wind speed at all WTGs reaches  $v_w \geq v_{w,R}$ , e. g. at  $t = 110$  h in Fig. 15b. Accordingly, at  $t = 110$  h in Fig. 15c, all WTGs operate at  $\Omega = 1$  such that the physical inertia constant  $H$  is saturated by its rated value  $H_R$ . In contrast, the maximum virtual inertia constant  $H_{v,\text{max}}$  in Fig. 15d is saturated by its lower limit given by the maximum power constraint –see also (reported earlier in Fig. 11). For lower wind speeds with all WTGs operating in region II, e. g. at  $t = 70$  h in Fig. 15, the lower WT rotor speeds lead to lower  $H$  but higher  $H_{v,\text{max}}$ , as the power constraint becomes inactive. Note that  $H_{v,\text{max}} > H$  is possible  
700 because the WTGs rotate asynchronously, whereas the VSMs synchronize with the grid frequency. Thus, for a given ROCOF,

a VSM-controlled WT with physical inertia constant  $H$  and virtual inertia constant  $H_v > H$  may extract more kinetic energy than a (directly grid-connected) SM with the same  $H$ .

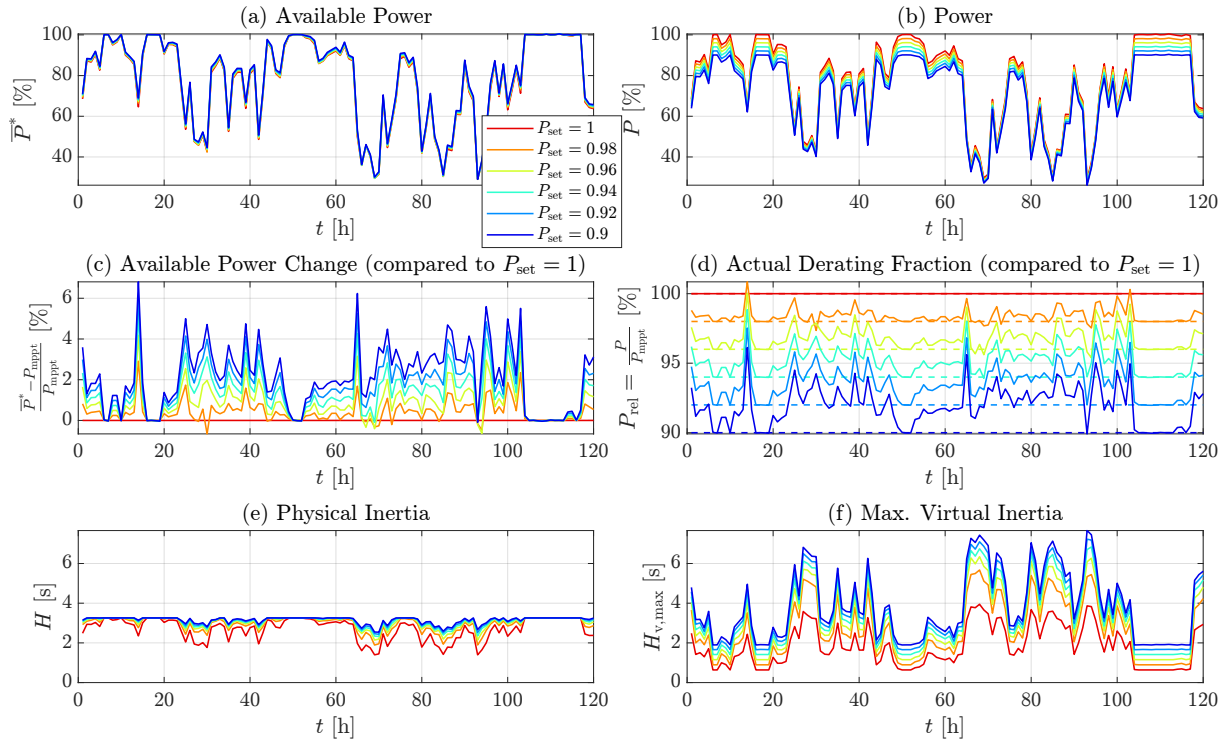
~~Simulation results at WF level for different derating power setpoints  $P_{\text{set}}$  based on the actual WF ambient wind data in Fig. 14.~~

705 ~~In Fig. 16, assuming equal  $P_{\text{set}}$  for all WTs, simulation results are considered~~ Figure 16 reports simulation results for different  $P_{\text{set}}$  at the WF level ~~(i. e. for all turbines)~~. The WF available power  $\bar{P}^*$  increases with lower  $P_{\text{set}}$  (see panel a), i. e.  $\bar{P}^* - P_{\text{mppt}} > 0$  for  $P_{\text{set}} < 1$  holds most of the time (see panel c), where  $P_{\text{mppt}} := P = \bar{P}^*$  for  $P_{\text{set}} = 1$ . This is due to wake effects, i. e. derating increases the local wind speeds at the downstream WTs. In Fig. 16b, the actual WF power  $P$  decreases with lower  $P_{\text{set}}$ , but minor changes are visible for lower wind speeds, since (i) the derating power setpoint is normalized to  
 710 the available (and not to the rated) power, i. e.  $P_{\text{set}} = P/\bar{P}^*$ , and since (ii) higher  $\bar{P}^*$  for lower  $P_{\text{set}}$  partially compensates for derating according to  $P = P_{\text{set}}\bar{P}^*$  ~~see also (cf. Eq. (9)9)~~. Thus, the actual WF derating fraction is defined as  $P_{\text{rel}} := P/P_{\text{mppt}}$  in Fig. 16d.  $P_{\text{rel}}$  is higher than  $P_{\text{set}} = P/\bar{P}^*$  (dashed lines) most of the time, as derating results in higher available power  $\bar{P}^*$ . In Fig. 16e, derating increases  $H$  according to the MRS up to the limit  $H = H_R$ . Besides additional kinetic energy reserve, the MRS-based derating provides wind energy reserve and power headroom for inertia provision. Thus, in Fig. 16f,  $H_{v,\text{max}}$   
 715 increases with lower  $P_{\text{set}}$ , even for  $H = H_R$ .

Figure 17 shows the actual and forecasted  $H_{v,\text{max}}$ -values at WF level for exemplary power setpoints  $P_{\text{set}} = 1$  (panel a) and  $P_{\text{set}} = 96\%$  (panel b), with the different wind input data for the wake model given by Fig. 14. The additional combined minimum (“Comb Min”) forecast in Fig. 17 uses the “Pro Min” and “Pro Max” forecasts. More precisely, the “Comb Min” forecast finds the minimum  $H_{v,\text{max}}$ -values within the forecasted local wind speed range  $v_{w,\text{pro,min}} \leq v_w \leq v_{w,\text{pro,max}}$  for each WT,  
 720 where  $v_{w,\text{pro,min}}$  and  $v_{w,\text{pro,max}}$  are given by wake modeling with “Pro Min” and “Pro Max” forecast input data, respectively. Since  $H_{v,\text{max}}$  is a nonlinear function of  $v_w$  for a given  $P_{\text{set}}$ , a nonlinear optimization algorithm solves  $\min H_{v,\text{max}}(v_w)$  such that  $v_{w,\text{pro,min}} \leq v_w \leq v_{w,\text{pro,max}}$  for each WT. Again, the considered WF values are given by the mean values across all WTs. In Figures 17a and 17b, the “Comb Min” and “Pro Max” forecasts are similar most of the time, but notable deviations occur especially during time intervals with lower wind speeds, which is in line with Fig. 11.

725 Figures 17c and 17d show the “Det” and “Comb Min” inertia forecast errors for different derating. Due to lower  $H_{v,\text{max}}$  variation for lower derating (see panels a, b), the absolute errors tend to be smaller in these cases. During almost all hours, the “Comb Min” forecast error is negative, i. e. the “Comb Min” forecast predicts a lower bound for  $H_{v,\text{max}}$ . This lower bound is especially relevant for (i) WF operators if they have to provide a minimum level of inertia, or for (ii) system operators to analyze worst-case frequency events.

730 Figure 18 compares inertia monitoring errors due to simplified WF modeling for two variants: (i) replacing the numerical by the simplified solution (Sect. 3.4.2) of the optimization problem in Eq. (14) leads to the “Simple Opt” variant, ~~see also Sect. 3.4.2~~; (ii) no wake modeling leads to the “No Wake” variant. Figures 18a–c consider the absolute error (“Abs. Error”)  $\Delta H_{v,\text{max}}$  (in s) and Figures 18d–f consider the relative error (“Rel. Error”)  $\Delta H_{v,\text{max}}/H_{v,\text{max}}$  (in %). Both variants underestimate the WF inertia at all operating points, i. e.  $\Delta H_{v,\text{max}} = -|\Delta H_{v,\text{max}}|$ , such that the negative mean value (“–MEAN”)   
 735 and the mean absolute error (MAE) are equal. The MAEs are 0.37s and 1.22s (panel c) or 11.3% and 38.5% (panel f) for



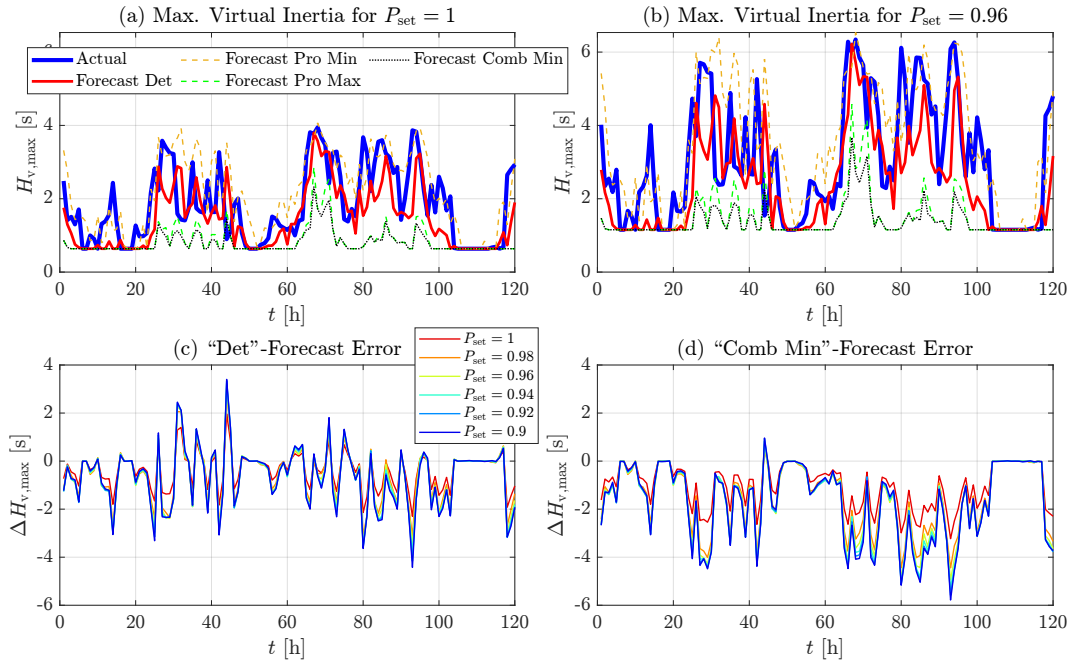
**Figure 16.** Comparison of actual and forecasted Simulation results at WF inertia, with errors (forecasted minus actual values) of level for different derating power setpoints  $P_{set}$  based on the forecast types: deterministic (“Det”), probabilistic minimal (“Pro Min”), probabilistic maximal (“Pro Max”), and combined minimal (“Comb Min”) actual WF ambient wind data in Fig. 14.

the “Simple Opt” and “No Wake” variants, respectively. The error magnitude tends to increase with higher derating for the “Simple Opt” variant (see panels a, d), which is in line with the analysis at the WT level in Appendix ??].

In Figures 19a and 19c, the WF inertia forecast error distributions of the “Simple Opt” and “No Wake” variants are shifted to the left compared with the proposed variant, due to the inertia underestimation in Fig. 18. At the samples with  $\Delta H_{v,max} > 0$  in Fig. 19a, the wind forecast errors overcompensate for the modeling errors  $\Delta H_{v,max} < 0$  in Fig. 18a. For the proposed complete numerical WF inertia forecasting, the MAE is 0.9s or 26.8%, see as shown in Figs. 19b and 19d. For the “Simple Opt” forecasting, the MAE is 1.04s or 30.3%. For the “No Wake” forecasting, the MAE is 1.68s or 46.5%. Thus, the simplified variants “Simple Opt” and “No Wake” increase the WF inertia forecasting MAE (in s) by  $\frac{1.04}{0.9} - 1 \approx 16\%$  and  $\frac{1.68}{0.9} - 1 \approx 87\%$ , respectively.

## 745 5 Conclusions

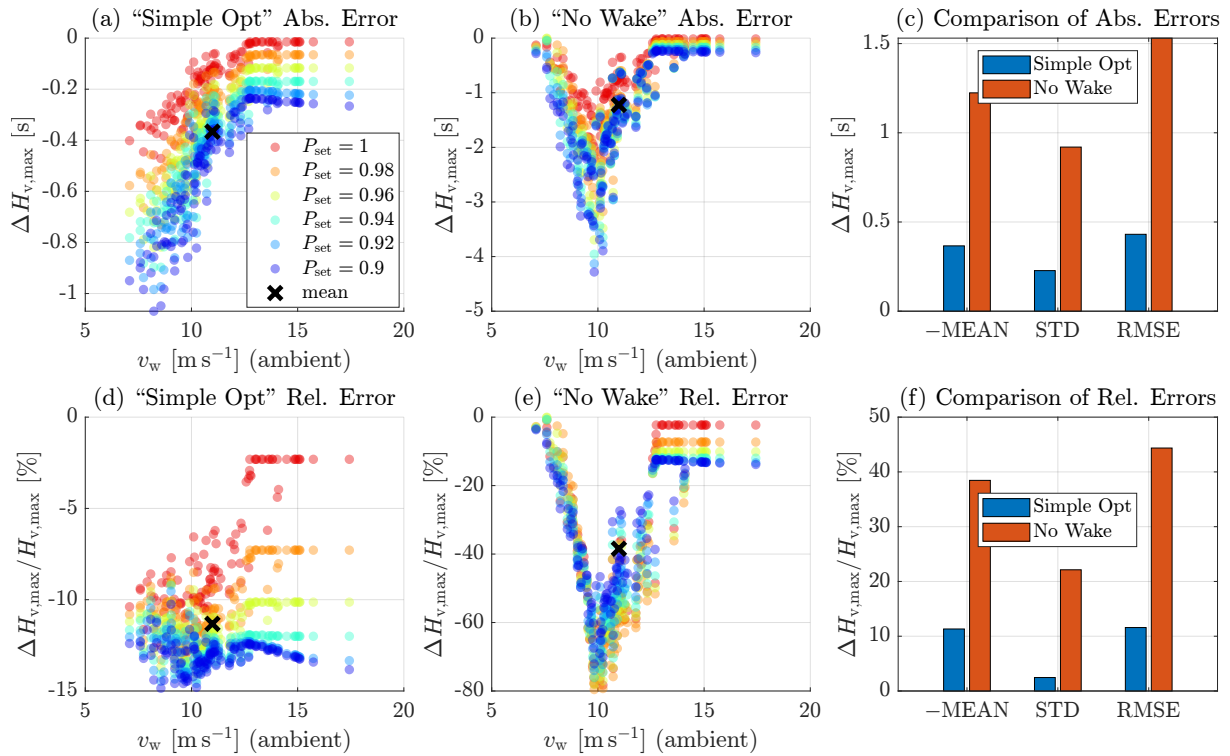
Grid-forming VSM control limits the initial ROCOF by inertia provision and thus enables grid frequency stability in future power systems with a high share of IBRs. VSM inertia can be adapted based on the required grid support. VSM control



**Figure 17.** Comparison of actual and forecasted WF inertia, with errors (forecasted minus actual values) of the forecast types: deterministic (“Det”), probabilistic minimal (“Pro Min”), probabilistic maximal (“Pro Max”), and combined minimal (“Comb Min”).

extracts WT kinetic energy reserve for inertia provision, but physical and virtual WT inertia differ in general. The WT with physical inertia constant  $H$  rotates asynchronously, whereas the VSM with virtual inertia constant  $H_v$  synchronizes with the grid frequency. However, in contrast to the grid-following control without inertia provision, the rotor speed and grid frequency dynamics are not completely decoupled anymore, i. e. the VSM requires the WT energy or power for grid synchronization. Thus, WT operating limits must be taken into account for proper grid synchronization with unsaturated inertia provision. Furthermore, the consideration of intra-farm turbine-to-turbine interactions is of **utmost significance**, as WFs will participate **great significance for a precise quantification of the inertia that can be provided to the grid. The participation** in future inertia markets **-This could be additionally could be** facilitated by short-term prediction of the maximum feasible inertia capability of the WF over varying inflow and operational conditions.

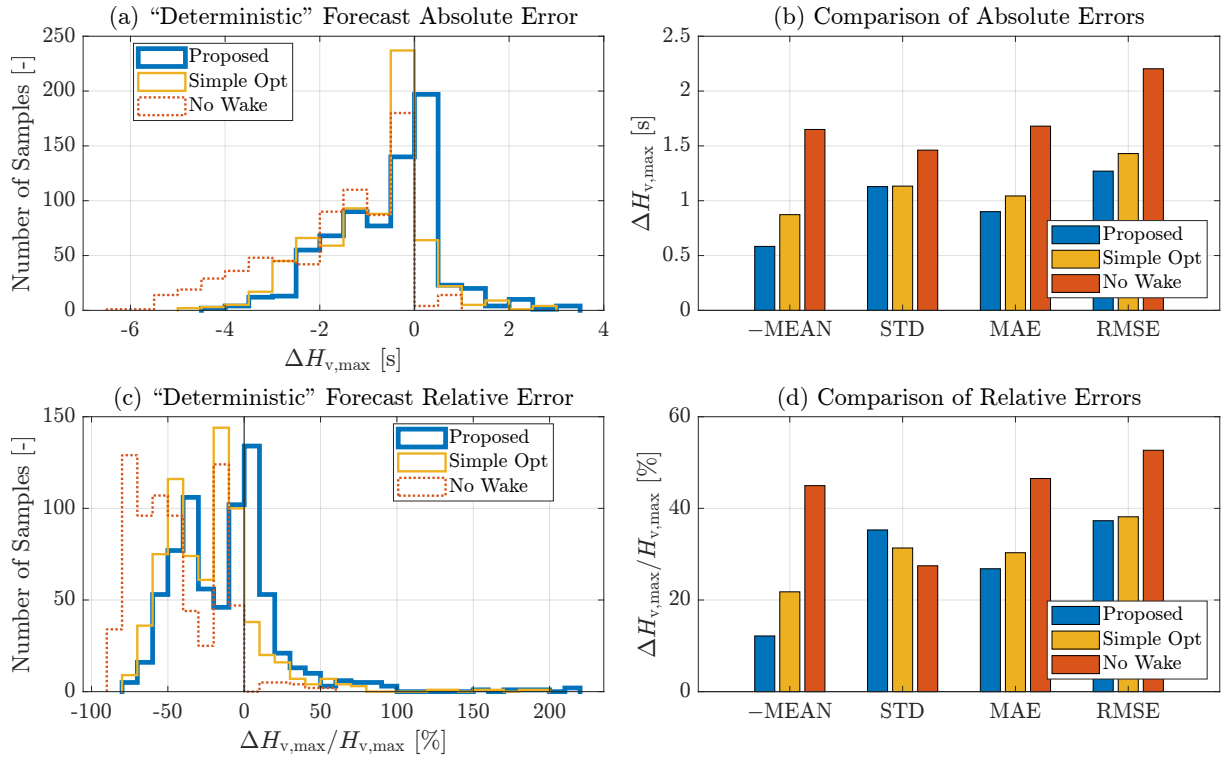
In contrast to existing solutions, the proposed formulation considers the WF grid-forming capability in terms of the maximum feasible inertia constant  $H_{v,max}$  by simulating the inertial response of VSM-controlled WTs. Furthermore, the proposed formulation takes into account the intra-farm turbine-to-turbine interactions, as they have a **huge-large** influence on the local wind condition at the turbines, and have largely been ignored in the existing studies evaluating inertia provision capability. Under varying wind conditions, the derived simplified solution without dynamic WT simulations exhibits a trend similar to that of the complete numerical solution with dynamic WT simulations. However, the simplified solution significantly underestimates the optimal tuning value  $H_v = H_{v,max}$  for maximum inertia provision. In addition, the proposed dynamic WT simulations give



**Figure 18.** Inertia monitoring errors of the simplified optimization (“Simple Opt”) variant and the no wake modeling (“No Wake”) variant. For each considered variant and for each  $P_{\text{set}} \in \{0.9, 0.92, 0.94, 0.96, 0.98, 1\}$ , all 120 WF ambient wind data samples (one data point per hour) of the actual measurements in Fig. 14 are mapped to  $H_{v,\text{max}}$  and to the corresponding errors (variant minus actual/proposed). The bars in panels c and f include all  $6 \cdot 120 = 720$  samples per variant, with negative mean value (“-MEAN”) equal to the mean absolute error (MAE), sample standard deviation (“STD”), and root mean square error (RMSE).

deeper insights into the WT inertial response, including the relevance of operating limits and the interactions among different  
 765 controllers, such as VSM control, MPPT compensation, MRS-based derating, active power droop control, and blade pitch control.

The proposed MRS-based derating increases the WT kinetic energy reserve for inertia provision with negligible impact on the thrust force. However, upon reaching the maximum rotor speed, further derating increases the pitch angle, significantly reducing thrust force and wake effects. Regarding Considering the aerodynamic trajectory of the power coefficient  $c_p(t)$  as a  
 770 function of the tip ratio  $\lambda(t)$  during the inertial response to the worst-case ROCOF, the MRS-based derating provides kinetic energy and power headroom for inertia provision. This is achieved by initial operation with  $(\lambda(t_0) > \lambda^*, c_p(t_0) < c_p^*)$  at the right side of the MPP  $(\lambda^*, c_p^*)$ , i. e. the power coefficient  $c_p(t)$  increases during WT deceleration. If the initial derating and thus the initial rotor speed  $\Omega_0$  is high enough, the rotor speed  $\Omega(t)$  remains above its MPP value during the inertial response. In this case, no rotor speed recovery is required afterwards. The simulation results show that the proposed MRS-based derating



**Figure 19.** Inertia forecasting errors of the proposed (“Proposed”) and simplified variants (“Simple Opt”, “No Wake”). For each considered variant and for each  $P_{\text{set}} \in \{0.9, 0.92, 0.94, 0.96, 0.98, 1\}$ , all 120 WF ambient wind data samples (one data point per hour) of the deterministic (“Det”) forecast in Fig. 14 are mapped to  $H_{v,\max}$ . Thus, the shown distributions include  $6 \cdot 120 = 720$  samples per variant, with negative mean value (“-MEAN”), sample standard deviation (“STD”), mean absolute error (MAE), and root mean square error (RMSE).

775 increases the WF grid-forming capability in terms of  $H_{v,\max}$  over the complete wind speed range. For instance, considering a below-rated wind speed of  $v_w = 9 \text{ m s}^{-1}$  in Sect. 4.2.2, a derating of 5% significantly increases  $H_{v,\max}$  by ca. 68%.

WT curtailment or derating increases the energy or power reserve for inertia provision, but wasting renewable energy should be avoided. Therefore, quantifying and forecasting inertia is essential for maximizing efficiency while ensuring grid stability. This paper demonstrated a generic approach for quantifying and forecasting the inertia provision of a WF based on weather  
780 forecasting, wake modeling, and mapping local WT operating points to grid-forming capabilities. The actual WF power loss is smaller than expected by local WT derating setpoints, since derating reduces wake effects and, thus, increases the available power. The  $H_{v,\max}$  forecast error depends on the uncertainty of the WF ambient wind forecast. Taking this into account, the proposed lower bound prediction for  $H_{v,\max}$  combines probabilistic minimal and maximal wind forecasts. Even for this conservative estimation,  $H_{v,\max}$  varies over time, and the WF can provide more inertia than at rated power during many  
785 hours. Besides the proposed wind forecasting, the WF inertia forecasting includes (i) the proposed wake modeling and (ii) the proposed The proposed optimization of the WT inertial response to avoid high avoids large errors due to (over)simplification.

Future work should validate the simulated WT inertial response based on ~~high-fidelity modeling, e. g. including multi-mass modeling or based on higher fidelity model or using~~ real measurements. ~~However, the modeling of the proposed generic approach can be readily adjusted, if necessary.~~ Utilizing the presented simulation results for WF control is straightforward when operating all WTs with the same power setpoint  $P_{\text{set}}$  for derating. However, future work may also consider an optimal WF control that distributes individual  $P_{\text{set}}$ -values to each WT, e. g. to maximize ~~the~~ WF power while providing the desired inertia. Based on the new grid codes and inertia market requirements, the future mechanical and electrical designs of WTs and WFs should consider enhanced capabilities for grid-forming control, including inertia provision.

## Appendix A: Nomenclature

795	DFIM	Doubly-fed induction machine
	FCNN	Fully connected neural networks
	HVDC	High voltage DC current
	IBR	Inverter-based resources
	LUT	Lookup tables
800	MAE	Mean absolute error
	MPP	Maximum power point
	MPPT	Maximum power point tracking
	MRS	Maximum rotation strategy
	NWP	Numerical weather prediction
805	OP	Operating point
	PGPE	Policy gradients with parameter-based exploration
	RMSE	Root mean square error
	ROCOF	Rate of change of frequency
	SCR	Short circuit ratio
810	SM	Synchronous machines
	STD	Standard deviation
	VSM	Virtual synchronous machine
	WF	Wind farm
	WT	Wind turbine
815		
	$c_p$	Power coefficient
	$c_t$	Thrust coefficient
	C	Capacitance

820	$d_t$	Damping coefficient of tower spring-mass-damper system
	$D$	Diameter
	$D$	Damping factor
	$\Delta P$	Power imbalance
	$\Delta t$	Time duration
825	$E_{\text{kin}}$	Kinetic energy
	$f$	Frequency
	$F_t$	Aerodynamic thrust force
	$F_w$	Force of the incoming wind
	$H$	Inertia constant
830	$k_e$	Electromagnetic feedback gain
	$k_t$	Stiffness coefficient of tower spring-mass-damper system
	$m$	Torque
	$M$	Non-dimensional torque
	$m_t$	Mass of tower spring-mass-damper system
835	$n$	Total number of turbines
	$\bar{c}_p$	Power coefficient corresponding to the available power
	$\bar{P}$	Non-dimensional available power
	$p$	Power
	$P$	Non-dimensional power
840	$P_w$	Power of the incoming wind
	$r$	Radius of the rotor
	$s$	Laplace-domain variable
	$s_t$	Tower-top fore-aft displacement
	$t$	Time variable
845	$u$	Voltage
	$v_w^u$	North-aligned component of the wind measurement
	$v_w^v$	East-aligned component of the wind measurement
	$v_w$	Wind speed
	$\tilde{v}_w$	Relative wind speed
850	$w$	Weight variable
	$x$	State variable
	$\beta$	Blade pitch angle

855	$\delta$	Load angle
	$\epsilon$	Threshold
	$\lambda$	Tip-speed ratio
	$\mu$	Mean
	$\omega$	Angular frequency
860	$\omega_n$	Natural angular frequency of a second order system
	$\Omega$	Non-dimensional angular frequency
	$\phi$	Electrical angle
	$\rho$	Air density
	$\sigma$	Standard deviation
865	$\Theta$	Moment of inertia
	$\zeta$	Damping ratio of a second order system
	$\square_{\text{comb}}$	Value that combines min. and max. probabilistic forecasts
870	$\square_d$	Droop
	$\square_{\text{det}}$	Deterministic forecast value
	$\square_{\text{dp}}$	Damping
	$\square_e$	Electrical
	$\square_m$	Mechanical
875	$\square_{\text{max}}$	Maximum
	$\square_{\text{meas}}$	Measured
	$\square_{\text{min}}$	Minimum
	$\square_{\text{mpp}}$	Value at the maximum power point
	$\square_{\text{mppt}}$	Value for maximum power point tracking
880	$\square_R$	Rated
	$\square_{\text{ref}}$	Reference
	$\square_{\text{pu}}$	Per unit value
	$\square_{\text{rec}}$	Value for an additional WT rotor speed recovery constraint
	$\square_s$	System / Grid
885	$\square_{\text{set}}$	Setpoint
	$\square_{\text{pro}}$	Probabilistic forecast value
	$\square_v$	Virtual / VSM
	$\square_0$	Initial / steady-state value

$\square^{[i]}$	Value for $i$ -th turbine
$\square^*$	Optimal value
$\dot{\square}$	Time derivative $d\square/dt$

## 895 Appendix B: Grid impedance and electromagnetic feedback for VSM control

The grid impedance comprises several physical impedances, such as transformer or line impedances, but also include virtual impedances emulated by inverter control (Taul et al., 2020; VDE, 2024a). Besides, for type 3 WTs, which use DFIMs, the physical grid impedance includes the DFIM stator impedance (Thommessen and Hackl, 2024). For type 4 WTs, which use full-scale back-to-back inverters, the physical grid impedance includes LC(L)filter impedances connected to the grid-side inverter (Taul et al., 2020). A high grid impedance (corresponding to a weak grid connection) leads to a lower electromagnetic feedback gain  $k_e$  (Thommessen and Hackl, 2024, Sect. III.E) and a lower natural angular velocity  $\omega_{n,v}$  in Eq. (11), resulting in a lower inertial response time. In contrast, a higher grid voltage increases  $k_e$  and  $\omega_{n,v}$  (VDE, 2024a).

From the perspective of system operators or grid stability, minimum  $k_e$  or  $\omega_{n,v}$  values characterize the worst case in terms of a weak grid connection and a slow inertial response. For instance, the weak grid connection of (offshore) WFs is characterized by a typical short circuit ratio  $SCR < 2$  (Ghimire et al., 2024). Thus, Ghimire et al. (2024) recommend an operating point sweep between maximum and minimum SCR for testing grid-forming capabilities and (load angle) stability analysis.

From the perspective of WF operators or WT protection, the worst case is characterized by maximum  $k_e$  or  $\omega_{n,v}$  values due to fast inertial response with high torque rates. Accordingly, the considered worst-case analysis of the WT grid-forming capability should assume maximum  $k_e$  or  $\omega_{n,v}$  values. Grid codes requirements (VDE, 2024a) implicitly define an admissible  $k_e$  or  $\omega_{n,v}$  value range by specifying lower and upper limits for the grid impedance and voltage.

## Appendix C: VSM torque versus power synchronization

Replacing the torque with power quantities in Fig. 4 results in a power instead of a torque synchronization loop (Roscoe et al., 2020). However, the normalized VSM power and torque are approximately equal (see Eq. 12). For type 4 WTs with full-scale back-to-back inverters, the VSM-controlled grid-side inverter (approximately) outputs the VSM power (see Appendix ??D), which is in line with the proposed simplified representation. However, for type 3 WTs with DFIMs, the VSM control is not implemented at the grid-side but at the machine-side inverter connected to the DFIM rotor (see Appendix ??E). Considering VSM torque synchronization, the DFIM emulates the VSM torque rather than the VSM power (see Appendix ??E). In this case, the WT inertial *power* response (quasi-steady-state value) depends on the DFIM or WT rotor speed  $\Omega$  according to  $P_e = \Omega M_e$ . Consequently, for type 3 WTs with VSM torque synchronization, it would be necessary to distinguish between (i) the VSM parameter  $H_v$ , representing the inertial *torque* response  $\Delta M_e = M_{e,v} \approx P_v \approx -2H_v \dot{\Omega}_s$  (see also Eq. 12), and (ii) a

power-equivalent inertia constant  $H_{\text{eq}}$  for inertia provision, representing the inertial *power* response, e. g.  $\Delta P_e \approx \Delta M_e \Omega =: -2H_{\text{eq}} \dot{\Omega}_s$ , with  $\Omega \neq \Omega_s \approx 1$  in general. In contrast, for type 4 WTs,  $H_v$  defines the inertial *power* response of both WTs and VSMs, i. e.  $\Delta P_e = P_v \approx -2H_v \dot{\Omega}_s$ , and no additional definition of  $H_{\text{eq}}$  is needed for quantifying inertia provision.

For simplicity, this paper considers type 4 WTs, for which torque and power synchronization loops result in similar inertial responses. For type 3 WTs with *torque* synchronization, the proposed generic approach is also applicable by taking  $H_{\text{eq}} \neq H_v$  into account. In this case, the VSM representation in Fig. 2 must simply be adjusted by redefining the VSM power in Eq. (10) as  $P_v := \Omega M_{e,v}$ . For type 3 WTs with *power* synchronization, it must be simply taken into account that the VSM synchronization speed  $\omega_{n,v}$  depends on the DFIM speed (see Appendix ??E).

#### Appendix D: VSM control for type 4 WTs and assumptions

For type 4 WTs based on full-scale back-to-back inverters, the grid-side and machine-side inverters are connected via a DC-link. The grid-side inverter tracks the WT power reference based on VSM control, whereas the machine-side inverter controls the DC-link voltage  $u_{\text{dc}}$  (Roscoe et al., 2020; Meseguer Urban et al., 2019; Nguyen et al., 2022). Strictly speaking, the power of the machine-side and grid-side inverter are *not* equal during  $u_{\text{dc}}$  transients due to the DC-link buffer energy  $E_{\text{dc}} = \frac{1}{2} C_{\text{dc}} u_{\text{dc}}^2$  with DC-link capacity  $C_{\text{dc}}$ . However,  $E_{\text{dc}}$  is usually negligible in comparison with the WT kinetic energy reserve, i. e.  $E_{\text{dc}} \ll E_{\text{kin}}$ , see also Hackl et al. (2018). Thus,  $E_{\text{dc}}$  is usually *not* relevant for inertia provision. Moreover, a small  $E_{\text{dc}}$  requires a fast-reacting or aggressively tuned DC-link voltage control to keep  $u_{\text{dc}}$  within its small admissible voltage range (Thommessen and Hackl, 2024). Thus, this paper neglects the power difference or delay between machine-side and grid-side inverters.

#### Appendix E: VSM control for type 3 WTs and assumptions

For type 3 WT based on DFIMs, the directly grid-connected DFIM stator generates most of the power. The DFIM rotor is connected to the grid via back-to-back inverters, with the power flow depending on the DFIM slip  $\Omega - \Omega_s$  (Dirscherl and Hackl, 2016). With the VSM torque defined as the DFIM torque, i. e.  $M_{e,v} := M_e$ , the DFIM stator power approximately equals the VSM power, i. e.  $P_{\text{stator}} = \Omega_s M_e \approx P_v$  (Rodriguez-Amenedo et al., 2021; Thommessen and Hackl, 2024). However, the total electrical DFIM power includes the DFIM rotor power as well, i. e.  $P_e = P_{\text{stator}} + P_{\text{rotor}} = \Omega_s M_e + (\Omega - \Omega_s) M_e = \Omega M_e \neq P_v$  (Dirscherl and Hackl, 2016). Alternatively, Shah and Gevorgian (2020) propose grid-forming control for DFIMs based on a power instead of a torque synchronization loop with a power feedback  $P_v := P_e$  instead of a torque feedback  $M_{e,v} := M_e$ . In this case, the DFIM electrical power feedback gain  $k_e$  varies with the WT rotor speed  $\Omega$  according to  $P_e = \Omega M_e$ , and thus the VSM inertial response time or  $\omega_{n,v}$  in Eq. (11) also varies.

#### Appendix F: Prioritization of inertia provision over droop control

Both the inertial response and droop control are essential for grid frequency stability. Firstly, the inertial response is always needed to limit the magnitude of ROCOF regardless of its sign, i.e., even during frequency recovery. Otherwise, if the ROCOF

is too high, grid-connected resources trip, e.g., due to loss of synchronism or load angle instability. Secondly, active power droop control (or primary frequency control) is needed to limit the steady-state frequency deviation. Although WTs can reduce their power to support over-frequency events, WTs in MPPT mode cannot permanently increase their power for prolonged support of under-frequency events. Thus, droop control ~~only~~ has to increase the WT output power only if quasi-steady power reserves are available due to previous derating (VDE, 2024a). A temporary droop control activation, i.e. a temporary power offset proportional to the frequency deviation based on kinetic energy extraction, is often proposed for a fast frequency response of grid-following WTs (Godin et al., 2019; Lee et al., 2016). Although this would temporarily increase frequency support, the slower frequency response would hide the urgency of power adaptation for grid support from other grid-connected resources: ~~More precisely, the slower frequency change or lower frequency deviation would delay or decrease the power response from other resources, thereby delaying or decreasing their response.~~ This may also lead to a secondary frequency event after the temporary support. Thus, such a temporary droop control activation is not considered in the German grid code for grid-forming technologies (VDE, 2024a) ~~or~~, nor in this work.

In summary, the inertial response limits the ROCOF, which determines the time available to react, e.g. for protection devices to shed loads, or for other resources to ramp up power production. Thus, the inertial response ~~is of~~ has the highest priority and uses all power reserves, including kinetic energy. With increasing frequency deviation, droop control (or primary frequency control) limits steady-state frequency deviation. For under-frequency events, droop control is only feasible for WTs if wind power reserves are available due to previous derating. WTs should *not* use kinetic energy reserves for temporary droop control activation in order to properly communicate the droop power demand to other resources via the grid frequency.

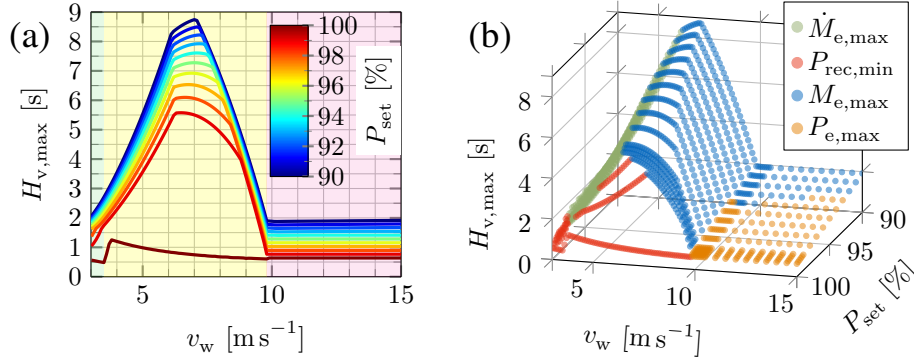
## Appendix G: WT rotor speed recovery and requirements

Assuming initial operation at the maximum power point (MPP), the WT deceleration during the inertial response decreases the aerodynamic power (~~see~~: For the example shown in Sect. 4, this is illustrated by the plot shown in Fig. 9). Thus, during the so-called recovery phase after the frequency event, the output power has to decrease below the initial power to maintain speed. In fact, the WT power is reduced even further to (slowly) re-accelerate the rotor, which affects the grid frequency recovery and may lead to a secondary frequency drop (Bao et al., 2016; Duckwitz, 2019; Höhn et al., 2024). Thus, a former draft of the German grid code (VDE, 2024a, Version 0.1) explicitly prohibited reducing the output power for speed recovery. ~~More precisely, after the ROCOF returns to zero at  $t = 4$  s in Fig. 9~~ With reference to the example considered in this work, reducing the output power below its initial value to re-accelerate the rotor would not be allowed ~~after the ROCOF returns to zero at  $t = 4$  s in Fig. 9~~.

However, this constraint is probably more restrictive than necessary, ~~as discussed next~~.

In fact, the frequency event defined in the grid code (VDE, 2024b) may not represent a realistic situation, but can be regarded as a worst-case test to verify the inertia provision capability of a grid-connected unit. Godin et al. (2019) consider a frequency event with less severe ROCOFs, which requires a lower power increase  $\Delta P$  over a longer period  $\Delta t$ , before reaching the frequency nadir. However, the kinetic energy extraction  $\Delta P \Delta t$  may be similar to the one used in our work. Although Godin

et al. (2019) assume a slower frequency recovery afterwards, the rotor speed recovery is similar, since speed recovery takes  
 985 much longer than frequency recovery (see Fig. 9). Considering the power reduction due to MPP deviations during the recovery  
 phase of a real WF, Godin et al. (2019) observe that similar reductions can be expected simply from changes in wind conditions.  
 Moreover, the inertial response superimposes all variations in WT operating points, which, in reality, results in a smoothing  
 effect on the grid frequency during the recovery phase (see also Thommessen and Hackl, 2024). Consequently, **the requirements  
 for the recovery power requirements for power recovery** should not be too restrictive. As per the recent version of the German  
 990 grid code (VDE, 2024a), no recovery power constraints are considered in this paper by default, although they can be easily  
 included if necessary, as demonstrated next.



**Figure G1.** Proposed solution with additional recovery power constraint: maximum virtual inertia constant for different operating points  $(v_w, P_{set})$  and with active optimization constraints highlighted in panel (b).

The optimization results in Fig. G1 include an additional constraint for the recovery phase after the WT inertial response. More precisely, the power  $P_{mppt}$  available for rotor speed recovery after the frequency event ends at  $t_{end} := 4.1 \text{ s}$  is not allowed to fall more than 1% below the initial power, i. e.  $P_{rec,min} := P_{m,0} - 1\% \leq P_{mppt}(\Omega(t_{end}))$ . This  $P_{rec,min}$  constraint  
 995 significantly limits the inertia provision at below-rated wind speeds for low derating, since most of the WT kinetic energy reserve cannot be used. In this case, the WT rotor speed limit  $\Omega_{min}$  is irrelevant, as the  $P_{rec,min}$  constraint is active instead.

## Appendix H: Simplified torque rate constraint

Limiting the torque rate  $\dot{M}_e$  in Eq. (14) contradicts the simplifying assumption of an ideal power pulse in Eq. (17) with infinite torque rate. However, instead of neglecting the actual limit  $\dot{M}_{e,max}$ , a simplified torque rate constraint can be derived as follows.  
 1000 Transforming Eq. (11) in the time domain, the VSM torque response to a ROCOF step change from zero to  $\dot{\Omega}_s$  is given by

$$M_{e,v}(t) = \dot{\Omega}_s \frac{4H_v + D_v t}{2} e^{-\frac{D_v}{4H_v} t} - 2H_v \dot{\Omega}_s. \quad (\text{H1})$$

By differentiating with respect to time one gets

$$\dot{M}_{e,v}(t) = -\dot{\Omega}_s \frac{D_v^2 t}{8H_v} e^{-\frac{D_v}{4H_v} t}, \quad (\text{H2})$$

and

$$1005 \quad \ddot{M}_{e,v}(t) = -\dot{\Omega}_s \frac{D_v^2}{8H_v} \left(1 - \frac{D_v}{4H_v} t\right) e^{-\frac{D_v}{4H_v} t}. \quad (\text{H3})$$

Zeroing Eq. (H3) and inserting the solution into Eq. (H2) yields the maximum VSM torque rate

$$\max \dot{M}_{e,v}(t) = \dot{M}_{e,v} \left( t = \frac{4H_v}{D_v} \right) = -\frac{1}{2e} \dot{\Omega}_s D_v. \quad (\text{H4})$$

Neglecting the WT deceleration until reaching the maximum torque rate, i. e.  $\Omega(t) \approx \Omega_0$ , and scaling Eq. (H4) by  $1/\Omega_0$  due to asynchronous WT rotation with respect to VSM speed or grid frequency, the simplified WT torque rate constraint is given by

$$1010 \quad \max \dot{M}_e = \frac{\max \dot{M}_{e,v}}{\Omega_0} = \frac{\dot{\Omega}_{s,\max}}{2e\Omega_0} D_v(H_v) \leq \dot{M}_{e,\max}. \quad (\text{H5})$$

### Appendix I: Steady state calculation

Solving the optimization problem in Eq. (14) for a given WT operating point defined by  $(v_w, P_{\text{set}})$  requires initializing the simulation model to a steady state. However, the steady-state conditions are unknown in general. Instead of running time-consuming simulations until reaching steady state, an alternative is proposed to speed up the optimization. At the equilibrium  
1015 point, mechanical and electrical power are equal, i. e. the initial steady states  $(\Omega_0, \beta_0)$  solve the minimization problem

$$(\Omega_0, \beta_0) := \arg \min |e(\Omega, \beta)|, \quad (\text{I1})$$

where  $e(\Omega, \beta) := P_m - P_e = P_w(v_w)c_p(\Omega, \beta, v_w) - \Omega M_e(\Omega, v_w, P_{\text{set}}) = 0$ .

The following analysis evaluates  $(v_w, P_{\text{set}})$  and identifies active constraints in the different WT operating regions to convert Eq. (I1) into simpler one-dimensional subproblems.

1020 Firstly, for MPPT with  $P_{\text{set}} = 1$  and  $M_e = M_{\text{mppt}}$ , Eq. (I1) can be rewritten as (Thommessen and Hackl, 2024)

$$(\bar{\Omega}^*, \bar{\beta}^*) := \arg \min |\bar{e}^*(\Omega, \beta)| \quad \text{s. t.} \quad \begin{cases} \Omega_{\min} \leq \Omega \leq 1, \\ \beta_{\min} \leq \beta \leq \beta_{\max}, \\ \beta_{\min} = \beta & \text{if } v_w \leq v_{w,R}, \\ \Omega = 1 & \text{if } v_w > v_{w,R}, \end{cases} \quad (\text{I2})$$

where  $\bar{e}^*(\Omega, \beta) := P_m - \Omega M_e = P_w c_p - \Omega M_{\text{mppt}}$ , for a given wind speed  $v_w$ . ~~With minimum region wind speed  $v_{w,II,\min}$ , one-dimensional optimizations solve~~ One-dimensional optimizations are solved for the following problems

$$\left\{ \begin{array}{l} v_{w,\text{cut-in}} \leq v_w < v_{w,II,\min} \Rightarrow \begin{pmatrix} \bar{\Omega}^* \\ \bar{\beta}^* \end{pmatrix} = \begin{pmatrix} \arg \min |\bar{e}^*| \\ \beta_{\min} = \beta_{\min}(\lambda) \end{pmatrix} \\ v_{w,II,\min} \leq v_w \leq v_{w,R} \Rightarrow \begin{pmatrix} \bar{\Omega}^* \\ \bar{\beta}^* \end{pmatrix} = \begin{pmatrix} \lambda^* \frac{v_w}{\omega_R r} \\ \beta_{\min} = \beta^* \end{pmatrix} \\ v_{w,R} < v_w \leq v_{w,\text{cut-out}} \Rightarrow \begin{pmatrix} \bar{\Omega}^* \\ \bar{\beta}^* \end{pmatrix} = \begin{pmatrix} 1 \\ \arg \min |\bar{e}^*| \end{pmatrix} \end{array} \right. \quad (\text{I3})$$

1025 with either  $\bar{\Omega}^*$  or  $\bar{\beta}^*$  known in advance (except for the transition region I-II in the first case, where  $\bar{\beta}^* = \beta_{\min}(\lambda)$  is calculated based on  $\lambda = \bar{\Omega}^* \omega_R r / v_w$ ). For the available WT power  $\bar{P}^*$  in Eq. (7), the MPPT power coefficient is given by

$$\underline{c_p^*}(v_w) = \begin{cases} c_p(\bar{\Omega}^*, \bar{\beta}^*, v_w) & \text{if } v_{w,\text{cut-in}} \leq v_w < v_{w,\text{II},\text{min}}, \\ c_p^* := \max c_p & \text{if } v_{w,\text{II},\text{min}} \leq v_w \leq v_{w,\text{R}}, \\ \frac{1}{P_w(v_w)} & \text{if } v_{w,\text{R}} < v_w \leq v_{w,\text{cut-out}}, \end{cases} \quad (14)$$

Eq. (8) with  $(\bar{\Omega}^*, \bar{\beta}^*)$  given by Eq. (13).

Secondly, for derating with  $P_{\text{set}} < 1$  and  $P_e = \bar{P}^* P_{\text{set}}$ , Eq. (11) can be rewritten as

$$1030 \quad (\Omega_0, \beta_0) := \arg \min |\tilde{e}(\Omega, \beta)| \quad (15)$$

where  $\tilde{e}(\Omega, \beta) := P_m - P_e = P_w c_p - \bar{P}^* P_{\text{set}}$ , which is solved in two substeps. In the first substep, ignoring the speed limit by assuming inactive pitch control, leads to the theoretical steady-state rotor speed

$$\tilde{\Omega}_0 := \arg \min |\tilde{e}(\Omega, \beta = \beta_{\min})| \quad (16)$$

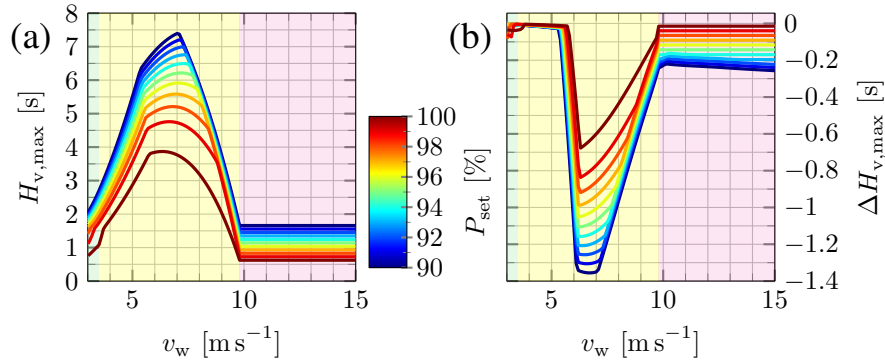
with  $\beta_{\min} = \beta_{\min}(\lambda)$ . The second substep takes the speed limit  $\Omega = 1$  into account, i. e.

$$1035 \quad \Rightarrow \begin{cases} \tilde{\Omega}_0 \leq 1 \Rightarrow \begin{pmatrix} \Omega_0 \\ \beta_0 \end{pmatrix} = \begin{pmatrix} \tilde{\Omega}_0 \\ \beta_{\min} \end{pmatrix} \\ \tilde{\Omega}_0 > 1 \Rightarrow \begin{pmatrix} \Omega_0 \\ \beta_0 \end{pmatrix} = \begin{pmatrix} 1 \\ \arg \min |\tilde{e}| \end{pmatrix}. \end{cases} \quad (17)$$

This way, the ~~conversion of the~~ original two-dimensional optimization problem ~~in given by~~ Eq. (11) ~~into has been converted~~ into the simpler one-dimensional subproblems ~~as in~~ Eq. (13), (16), (17) ~~has been achieved~~.

## Appendix J: Results of the simplified solution for maximum inertia provision

The results of the simplified solution derived in Sect. 3.4.2 are shown in the left plot of Fig. J1. In comparison with the results  
 1040 of the complete numerical solution in Fig. 11, the simplified solution underestimates the WT inertia provision capability. Considering the right plot of Fig. J1, significant simplification errors can be observed, e. g.  $|\Delta H_{v,\text{max}}| > 1 \text{ s}$  at  $v_w = 7 \text{ m s}^{-1}$  for  $P_{\text{set}} \leq 96\%$ . For  $P_{\text{set}} = 1$ , one reason for the underestimation is that the simplified solution assumes a constant electrical power in Eq. (17), whereas in reality, the VSM inertial power  $P_v$  decreases with  $\Omega_v \approx \Omega_s$  in Eq. (10). The underestimation significantly increases with higher derating, since the simplified solution only considers kinetic energy reserve and does not  
 1045 account for wind power reserve or changing aerodynamics. More precisely, the MRS-based derating leads to initial operation at the right side of the MPP. During the inertial response, the decreasing  $\lambda$  in combination with decreasing  $\beta$  increases the aerodynamic or mechanical power, which counteracts the WT rotor deceleration, as also shown in Fig. 10.



**Figure J1.** Simplified solution: maximum virtual inertia constant for different operating points (a) and simplification error or deviation from the complete numerical solution (b).

### Appendix K: Further definitions and parameters

The inertial response time interval for evaluating extreme WT operating points in Figs. 12 ~~and~~ 13 is defined as

$$1050 \quad t_0 \leq t \leq \min\{t_1, t_2\}, \quad t_1 := \max\{t_{s,\min}, \arg \min \Omega(t)\}, \quad t_2 := \left\{ \begin{array}{l} t_1, \text{ if } \forall t \geq t_0 : \bar{P}_d(t) \leq 0, \\ \min t \text{ s. t. } \bar{P}_d(t) > 0 \wedge t \geq t_{s,\min}, \text{ otherwise,} \end{array} \right\} \quad (\text{K1})$$

where  $t_1$  takes into account that the WT speed nadir may occur (shortly) after the grid frequency nadir at  $t_{s,\min}$ , and  $t_2$  takes into account that the droop control may cause further uncritical WT deceleration for  $t \geq t_{s,\min}$ .

*Data availability.* The content and data of figures 3-4, 6-18, G1, and H1 can be retrieved in Python pickle format at <https://doi.org/10.5281/zenodo.15176373> (review note: this is a preliminary repository, the final DOI will be generated if the paper is accepted).

1055 *Author contributions.* AT, AA and CMH developed the formulation of WF inertia forecasting. AT and AA carried out the research, with CLB developing the concept of including wake effects. AT and CMH evaluated the grid-forming capability of WTs, whereas AT implemented WT modeling and control with inputs from AA and CMH. AA developed the ambient wind forecaster based on the formulation proposed by CLB, developed and implemented the operational constraints with CLB, and implemented the wind farm model with inputs from AT. AT prepared the manuscript, with contributions from AA, CMH and CLB particularly in the sections about forecasting and farm modelling. AT  
1060 generated and interpreted the inertia forecasting results based on the actual and forecasted data that AA provided. CLB and CMH supervised the overall research. All authors provided important input to this research work through discussions, feedback, and improved this manuscript.

*Competing interests.* The authors declare that they have no conflict of interest, except for CLB who is the Editor-in-Chief of the Wind Energy Science journal.

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